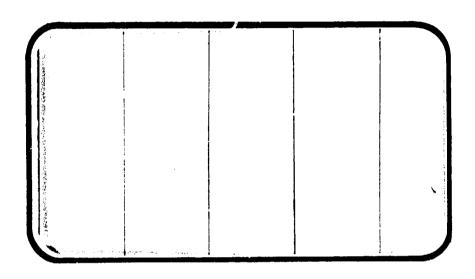


NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

NASA CR-13 +118



N75-30239

(NASA-CP-134118) AFRCDYNAMIC RESULTS OF A SEPARATION EFFECTS TEST CONDUCTED IN THE AFDC 40 BY 40 INCH TUNNEL A FACILITY ON THE FOCKWELL INTERNATIONAL LAUNCH CONFIGURATION 3 (MODEL 32-OTS) INTEGRATED VEHICLE (IA13),

Unclas G3/18 33794

SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER
HOUSTON, TEXAS

DATA MANagement services

SPACE DIVISION CHRYSLER
CORPORATION

DMS-DR 2062 NASA CR- 134,118

AERODYNAMIC RESULTS OF A SEPARATION EFFECTS
TEST CONDUCTED IN THE AEDC 40 x 40 INCH TUNNEL A
FACILITY ON THE ROCKWELL INTERNATIONAL LAUNCH
CONFIGURATION 3 (MODEL 32-OTS) INTEGRATED VEHICLE
(IA13)

Vol. II of III

J. H. Campbell, II Rockwell International Space Division

Prepared under NASA Contract Number NAS9-13247

bу

Data Management Services Chrysler Corporation Space Division New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center National Aeronautics and Space Administration Houston, Texas

TUNNEL TEST SPECIFICS:

Test Number: AEDC VA-323 NASA Series Number: IA13 Test Dates: July 5-17, 1973

FACILITY COORDINATOR:

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

AERODYNAMIC RESULTS OF A SEPARATION EFFECTS

TEST CONDUCTED IN THE AEDC 40 x 40 INCH TUNNEL A

FACILITY ON THE ROCKWELL INTERNATIONAL LAUNCH

CONFIGURATION 3 (MODEL 32-OTS) INTEGRATED VEHICLE

(IA13)

bу

J. H. Campbell, II Rockwell International Space Division

ABSTRACT

Experimental aerodynamic investigations were conducted from July 5 through July 17, 1973, in the AEDC/VKF tunnel A facility, on a 0.01 scale model (Model 32-OTS) of the Rockwell International launch configuration 3 Integrated Vehicle (excluding the left-hand booster). The AEDC captive trajectory system was utilized in conjunction with the tunnel primary sector to obtain "grid-type" data for tank (ET) abort from the Orbiter (SSV), and for nominal separation of one Booster (SRB) from the Orbiter-Tank combination. Booster separation was investigated with and without separation motors plume simulation. The plumes were generated by eight M₁ = 2.15 nozzles using a 1500 psia cold air supply.

Free stream data were obtained for all models (orbiter, tank, orbiter-tank, and right-hand booster) to provide baselines for evaluation of proximity effects.

The entire investigation was conducted at a nominal free stream Mach number of 4.5 and at Reynolds number per foot ranging from approximately 7.0×10^5 to 6.6×10^6 .

This report is published in three volumes. Volume I contains data figures through page 894. Volume II contains the remaining data figures and the appendix (tabulated source data) is found in Volume III.

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	⋖	13-15
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Schedule of Coefficients Plotted:

- A) CN, CA, CIM, CBL, CY, CYN VS APLHA
- B) CN, CLM, CY, CYN VS ALPHA

- C) CN, CA, CLM, CBL, CY, CYN vs Z
- D) CN, CLM, CY, CYN vs Z

NOMENCLATURE

SYMBOL	PLOT SYMBOL	DEFINITION
a		speed of sound; ft/sec
v		velocity; ft/sec
M	MACH	Mach number, V/a
ď∞	Q(PSF)	dynamic pressure, $\rho V^2/2$; psf
ρ	RHO	mass density; slugs/ft3
$A_{\mathbf{b}}$		base area; ft ²
c _{Ab}	CAB	base force coefficient, $-A_b(P_b - P_{\infty})/q_{\infty}S_w$; $\frac{base \ force}{q_{\infty}S_w}$
$^{\mathrm{C}_{\mathbf{A_f}}}$	CAF	forebody axial force coefficient; $c_A - c_{A_b}$
C _A	CA	axial force coefficient excluding adjustments for base and sting/balance cavity pressure measurements; $\frac{axial\ force}{q_{\varpi}S_{\pmb{w}}}$
c_N	CN	normal force coefficient; $\frac{\text{normal force}}{q_{\omega}S_{W}}$
CL	CBL	rolling moment coefficient; $\frac{\text{rolling moment}}{q_{\infty}S_{w}\mathbf{ref}}$
C _m	CLM	pitching moment coefficient; $\frac{\text{pitching moment}}{q_{\infty}S_{W}\ell_{ref}}$
c_{Y}	CY	side force coefficient; $\frac{\text{side force}}{q_{\infty}S_{W}}$
c_n	CYN	yawing moment coefficient; yawing moment $q_{\infty}S_{\mathbf{w}}^{\ell}$ ref

NOMENCLATURE (Continued)

SYMBOL	PLOT SYMBOL	DEFINITION
s_w	SREF	reference wing area; ft ²
CPS		captive trajectory system
P_{b1}/P_{∞}	PB1/P	number one base pressure ratio
P_{b2}/P_{∞}	PB2/P	number two base pressure ratio
P_c/P_{∞}	PC/P	sting/balance cavity pressure ratio
P _{b1}	PB1	number one base pressure; psia
P _{b2}	PB2	number two base pressure; psia
Pc	PC	SRB separation nozzle chamber pressure; psia
P_{∞}	P	freestream static pressure, psia
$P_{t_{\boldsymbol{\infty}}}$	PTOTAL	freestream total pressure, psia
c _{pb1}	CPB1	number one orbiter base pressure coefficient, $(P_{\mbox{\footnotesize{b}}\mbox{\footnotesize{]}}$ - $P_{\mbox{\footnotesize{\varpi}}})/q_{\mbox{\footnotesize{\varpi}}}$
$c_{P_{1,2}}$	CPB2	number two orbiter base pressure coefficient, $(P_{b2} - P_{\infty})/q_{\infty}$
$c_{P_{C}}$	CPC	sting balance cavity pressure coefficient, $(P_{_{\rm C}} - P_{_{\rm D}})/q_{_{\rm D}}$
R/L	RN/L	freestream unit Reynolds number x 10^{-6} ; per foot
^L ref	LREF BREF	reference dimension used to reduce the measured balance moment data to coefficient form, orbiter model body length, inches
δa	AILRON	aileron deflection; degrees
δ _e	ELEVTR	elevon deflection; degrees

NOMENCLATURE (Continued)

SYMBOL	PLOT SYMBOL	DEFINITION
δ _r	RUDDER	rudder deflection; degrees
Δα	DALPHA	parameter name for nominal angle of attack difference, $(\Delta \alpha = \alpha_{SRB} - \alpha_{OT})$ or $(\Delta \alpha = \alpha_{ET} - \alpha_{O})$; legrees
Δβ	DBETA	parameter name for nominal angle of sideslip difference, $(\Delta\beta = \beta_{SRB} - \beta_{O/T})$ or $(\Delta\beta = \beta_{ET} - \beta_{O})$; degrees
^Q O/T	ALPHA	orbiter tank angle of attack. All source data are presented as a function of $\alpha_{O/T}$, except for isolated tests and those where the tank is separating from the orbiter; degrees $\alpha_{SRE} = \alpha_{O/T} + \Delta \alpha$
α ₀	ALPHA	orbiter angle of attack. Source data are presented as a function of α_0 for isolated tests and those where the tank is separating from the orbiter; degrees $\alpha_{ET} = \alpha_0 + \Delta \alpha$
$\alpha_{ m ET}$	ALPHA	external tank angle of attack. Source data are presented as a function of $\alpha_{\hbox{\footnotesize{ET}}},$ only, for tests of the isolated external tank, degrees
^α SRB	АІРНА	solid rocket booster angle of attack, source data are presented as a function of $\alpha_{\mbox{SRB}}$, only, for tests of the isolated SRB, degrees
^β 0/T	BETA	orbiter tank angle of sideslip. All source data are presented as a function of $\beta_{O/T}$ except for isolated tests and those where the tank is separrating from the orbiter; degrees $\beta_{SRB} = \beta_{O/T} + \Delta\beta$
β _O	BETA	orbiter angle of sideslip. Source data are presented as a function of β_0 for isolated tests and those where the tank is separating from the orbiter; degrees $\beta_{ET} = \beta_0 + \Delta\beta$

KMI

NOMENCLATURE (Concluded)

SYMBOL	PLOT SYMBOL	DEFINITION
β _{ET} .	BETA	external tank angle of sideslip. Source data are presented as a function of $\beta_{ET},\ \text{only, for}$ tests of the isolated external tank, degrees
β _{SRB}	BETA	solid rocket booster angle of sideslip. Source data are presented as a function of β_{SRB} , only, for tests of the isolated SRB, degrees
δ_{SB}	SPDBRK	speed brake flare angle, degrees
c.g.		center of gravity
Tto '	TTOTL	freestream total temperature; °R
T _∞	T	freestream static temperature; °R
ΔΧ	x	separation distance measured along the longitudinal X axis; inches. See figures 2d and 2e
	XMRP	X station of the moment reference center; inches. See figures 2b and 2c
ΔΥ	Y	separation distance measured along the lateral, Y axis; inches. See figures 2d and 2e.
	YMRP	Y station of the moment reference center; inches. See figures 2b and 2c.
	ZMRP	Z station of the moment reference center; inches. See figures 2b and 2c.
ΔΖ	Z	separation distance measured along the vertical Z axis; inches. See figures 2d and 2e.

INTRODUCTION

The NASA Space Shuttle Integrated Vehicle (SSV) is comprised of four components: an orbiter (0), an external fuel tank (ET) for the orbiter, and two solid rocket boosters (SRB's). During the ascent phase of the SSV the ET and SRB's are separated from the orbiter, and return to earth. In order to assure clean separation of the ET and the SRB's from the orbiter it is necessary to know the aerodynamic forces acting on these components, and the orbiter, during the separation procedure. This test was conducted to determine the interacting aerodynamic effects for two situations: 1) nominal abort of the SRB from the orbiter/external tank combination and 2) emergency abort of the external tank from the orbiter.

Data were obtained at a freestream Mach number of 4.5 with Reynolds number varying from 7.0×10^5 to 6.6×10^6 per foot. The SRB has small rocket motors which are utilized to assure positive separation. During these tests motor operation was simulated by using cold air jets and data were recorded with and without these motor plume simulations.

The isolated vehicle components were tested over an angle of attack and sideslip range as shown below:

Vehicle Component	α Range	β Values
Orbiter	-10° to 30°	0°, 5°, 10°
External Tank	-40° to 10°	0°, 5°
Orbiter/External Tank	-10° to 30°	0°, 5°
Solid Rocket Booster	-30° to 15°	-10°, 0°, 5°, 10°,
		20°, 30°

INTRODUCTION (Continued)

In order to examine the interacting flow field effects during the two abort situations, the separating component was held at attitudes relative to the parent configuration and moved to various field positions to record aerodynamic data. The relative attitudes tested were:

Vehicle Component	Δα Range	Δβ Values	
External Tank	-30° to 5°	0°, ±5°	
Solid Rocket Booster	-20° to 5°	0°, ±5°, -10 , -2	0°
where for external tank $\Delta\alpha$ = ($\alpha_{\rm ET} - \alpha_{\rm O}$), $\Delta \beta = ($	$\beta_{\rm ET}$ - $\beta_{ m O}$) and for sol	id
rocket booster $\Delta \alpha = (\alpha_{SRB} - \alpha_0)$	$_{\rm O/ET}$), $\Delta \beta = (\beta_{\rm SRB})$	- $\beta_{O/ET}$).	

The vertical (Z), longitudinal (X) and lateral (Y) separation distance langes were:

Vehicle Component	X-Inches(f.s.)	Y-Inches(f.s.)	Z-lnches(f.s.)
External Tank	0 to 1200	-200 to 200	0 to 1300
Solid Rocket Booster	0 to 1600	0 to 800	0 to 100
where for the external tan	k, X, Y, and Z	= 0 when the ET	nose is in the
mated position with respec	t to the orbite	r. (See figure	2d.) For the
SRB, X, Y, and $Z = 0$ when	the SRB nose is	in the mated po	sition with
respect to the Orbiter/Ext	ernal Tank. (S	ee figure 2e.)	

During SRB separation testing (see figure 2e.) the external tank was rigidly attached to the orbiter, which was supported inverted on the tunnel primary sector. The booster was supported on the CTS via a flow-through balance and sting.

During external tank separation testing (see figure 2d.), the

INTRODUCTION (Concluded)

orbiter was supported inverted on the primary sector and the ET was supported on the CTS.

No base or cavity pressure taps were built into the models. To obtain these pressures (two base, one cavity per model), hardline tubing was routed to the vicinity of the model bases and bent into areas where pressures were desired.

As an aid in clarifying the interdependence of the test configuration with the tabulated source data in the Appendix, the following array is a necessary adjunct to the Run Summary Schedule of Table II.

DATASET	BALANCE	TEST CONFIGURATION	APPENDIX PAGE NUMBER
RTJ001 through	Orbiter		1
RTJ099		Tools Samuel B. O	89
RTJT01 through	Tank	Tank Separating From Orbiter	90
RTJT99	•	,	178
RTJ100 through	Orbiter Tank)	179
RTJ283		ann a	361
RTJ300 through	SRB	SRB Separating From Orbiter Tank	362
RTJ483		•	543
RTJ500 (RTJ501)	Tank		543
`			544
RTJ502 through RTJ507	Orbiter		545
!		Isolated	550
RTJ508 RTJ509	Orbiter Tank		551 552
RTJ510 through	SRB	J	553
RTJ536		11	565

CONFIGURATIONS INVESTIGATED

The 0.01-scale 32-OTS models consists of an orbiter, tank, and one booster (with metric high-pressure separation thrusters, fore and aft). See figures 2b and 2c.

The configuration of each vehicle component tested is given below.

Pertinent dimensional data for these model components may be found in

Table III.

Orbiter 09 = $(B_{19}C_7F_5M_4)$ $(W_{107}E_{23})$ (V_7R_5) where:

Component	<u>Definition</u>
B ₁₉	Fuselage per Rockwell line VL70-000139B. (Model drawing SS-A00062)
c ₇	Canopy per Rockwell Lines VL70-000139B. (Model drawing SS-A00062)
E ₂₃	Elevons per Rockwell lines VL70-000139B. (Model drawing SS-A00109)
F ₅	Body flap per Rockwell lines VL70-000139B. (Model drawing SS-A00062)
M ₄	Orbital Maneuvering System per Rockwell lines VL70-000139B. (Model drawing SS-A00062)
R ₅	Rudder per Rockwell lines VL70-000139B and VL70-000095. (Model drawing SS-A00062)
v ₇	Vertical tail per Rockwell lines VL70-000139B and VL70-000095. (Model drawing SS-A00062)
W ₁₀₇	Wing per Rockwell lines VL70-000130B. (Model drawing SS-A00109)

The external tank (T_{10}) is not broken down into subassemblies and was constructed to Rockwell lines VL72-000088 and VL78-000041 (Model Drawing SS-A-00108).

CONFIGURATIONS INVESTIGATED (Concluded)

The solid rocket booster (S_8) is not broken down into subassemblies and was constructed to Rockwell lines VL72-000088 and VL77-000036 (Model Drawing SS-A-00113).

TEST FACILITY DESCRIPTION

The AEDC von Karman Facility (VKF) Tunnel A is a continuous, closed-circuit, variable density wind tunnel with an automatically driven flexible-plate-type nozzle and a 40- by 40-in. test section (see Figure 3). The tunnel can be operated at Mach numbers from 1.5 to 6 at maximum stagnation pressures from 29 to 200 psia, respectively, and stagnation temperatures up to 750°R (M = 6). Minimum operating pressures range from about one-tenth to one-twentieth of the maximum at each Mach number. A description of the tunnel and airflow calibration information may be found in the Test Facilities Handbook*.

^{*}Test Facilities Handbook (Ninth Edition). "von Karman Gas Dynamics Facility, Vol. 3", Arnold Engineering Development Center, July 1971.

DATA REDUCTION

Six-component aerodynamic force and moment data were recorded for the Orbiter, External Tank, and Orbiter/ET. Four-component force and moment data were recorded for the SRB. Thrust loads generated by the SRB separation motors were treated as tares and were subtracted from balance recorded loads before computing coefficients. For Orbiter freestream and ET abort testing, the Orbiter was mounted on the AEDC 0.85-inch 4.00-Y-36-037 balance. For ET freestream and ET abort, the ET was mounted on the AEDC 0.85-inch 4.00-H-36-049 balance. For SRB separation, the Orbiter/ET was mounted on the 4.00-Y-36-049 balance. For SRB freestream and separation, the SRB was mounted on the Rockwell 0.625-inch 4.00-H-34-065 flow-through balance.

All force and moment data were reduced to coefficient form in the body axis system. Base and balance cavity pressures were recorded. However, no pressure adjustments were made to the force and moment data. All coefficients are based on the following reference dimensions.

 S_{REF} = orbiter wing reference area = 0.269 ft²

 $k_{REF} = b_{REF} = \text{orbiter body length} = 12.903 inches$

Moments are about the following reference c.g. locations:

Orbiter (Isolated and with ET separating):

XMRP = 8.516 inches aft of orbiter nose (see figure 2c)

ZMRP = 4.0 (fuselage reference line 4.0)

External Tank (Isolated and while separating from orbiter):

XMRP = 10.94 inches aft of ET nose (see figure 2c

ZMRP = 0.0 (tank centerline)

Orbiter/ET (Isolated and with SRB separating)

XMRP = 7.43 inches aft of ET nose

(see figure 2b)

ZMRP = 0.47 inches above ET centerline

SRB (Isolated and while separating from Orbiter/ET)

XMRP = 9.67 inches aft of SRB nose (see figure 2b)

ZMRP = 0.0 (SRB centerline)

TEST: IA13 (VA 323)

DATE : POST-TEST

TEST CONDITIONS

MACH NUMBER	TOTAL PRESSURE (Pounds/Sq. Inch)	DYNAMIC PRESSURE (pounds/sq.,inch)	STAGNATION TEMPERATURE (degrees Fahrenheit)
4.53	110	5.27	145
4.52	80	3.85	
•	58	2.80	
4.51	28.9	1.41	
4.48	11-6	0.58	
ļ	10.8	0.54	•
···			

BALANCE UTILIZED: 4.00-Y-36-037; 4.00-Y-36-049; 4.00-Y-34-065

CAPACITY:

ACCURACY:

COEFFICIENT TOLERANCE:

1F	±150	±200	±135	±0.2	±0.3	±0.25	 CCURACY
F	± 75	\$100	±100	\$0.15	10.3	10.2	
۱F	±25	150	_	\$0.15	±0.3		
M	1300	+380	±360	\$0.5	手110	\$1.0	
M	120	180		\$0.2	±0.4		
М	±150	\$ 190	£275	10.5	\$1.0	10.75	† '

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- 1) THE 037 MOUNTED IN ORBITER FOR "ORBITER ISOLATED" AND "TANK SEMRATION"
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- 3). THE -OLS MOUNTED IN BOOSTER FOR "BAISTER ISOLATED "AND " BOOSTER SEPARATION."

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O DATA - ORBITER DATA TAIA - EXTERNAL TANK DATA

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Ŋ	CORMAC.	EXTERMEL TANK SPARATION DATA	T.A.				T/	TABLE	II.	(Con	(Continued)	d)								ſ
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1			2	SCHD.			ļ	ă	RAME	rers/	PARAMETERS/VALUES	ړ			_ \o\	1	MACH NUMBERS	MBERS	 	
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	120				-20	0	-	E							/	104	105	100	107	NUM
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A, A	3,0%,	K, B, DK, D/3 - DEGGES; X, Y - INCHES (FALL SCALE)	<u> </u>	IN	wes(Rut	SCALE	; ; (; DEL, DER		- DEGRECS	Recs ,	pr-	PT-BIA	RF- DEGRESS	766G	25.			
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(Continued)	
TABLE II.	

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4) 200, 300, 400, 600, 800, 1000 T) 600, 800, 1000

0) 300, 400, 600, 800, 1000, 1300

SCHEDULES

I) 400, 600, 800, 1000

c) 200, 300, 400, 600, 800, 1000, 6) 800, 1000,

K, B, DK, DB - DEARTES , X, Y - INWES (AULKALE), DEL, DER - DEARTES ; PT-7514, RF - DEARTES (RUDDER FLACE)

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«, p, D«. DB - DEGREES ; K.Y - INUMES (PULSCALE) ; DEL, DER - DEGREES ; DT- PSIA ; RF - DEGREES (RUPDER PLATE)

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	ŭ	TOMAL	EXTEDNAL TANK SEPARATION	DATA			Ξ.	TABLE J) '11	(Continued)	nnea	_							1
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W, B, DK, DB - DELMEES; X, Y - INCHES (FULLSCALE); DEL, DEP - DESMEST; PT-PSIA; RF- DELMEES
(MUDBER PLARE)

	BookTER (SRB) SEMEATION	SEE PE	RA	707		[4]	TABLE	11.	(Cor	(Continued)	d)							1
TEST: 14	1413	\vdash		DATA	ľ	/RUN	NOM	BER	COLL	ATION	SET/RUN NUMBER COLLATION SUMMARY		DATE		POST_TEST	27.		1
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DATA SET IDENTIFIER	CONFIGURATION	8	SCHO.	ă	80	>	K	MKH	PT	MKH PT PCH	RF		RUNS	0	300	900	900	
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107						0	V	4.51	29	750				307	308	309	310	
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104						400	4							315	3/6	317	318	
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901		_				0	4			1500				323	324	325	328	ST F
107						28	A							330	329	328	377	RUN
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109		-				008	800	-	-					335	336	33.7	338	BER
7011						0	V	4.52	28		_			339	340	178	342	s
						200	4							346	345	344	343	
112		-	_			400	A		-					347	348	349	350	
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1	A) = 0.	8	0	200	300	COEF	μς,	SOO.	1000	שנהבלו	1000 DIKINES(VAILS/ALE)			יסי	DVAR (1)	IDVAR	(3)	>02
K SCHEDIII ES			}				t I		1									

ARTTIOO THROUGH RITZ83 - ORBITER/1944 DATA RITJ300 THROUGH RIT 483 - SRB DATA

AUMBER COLLATIO PARAMETERS/VALU WAKE PT WAKE PT WAKE PT WAKE PT PARAMETERS/VALU PT PARAM	SCHD. SCHD. SCHD. O O O YOO A H453 110 O O O YOO A H453 110 O O O O YOO A H453 110 O O O O O O O O SOO SOO SOO SOO SOO S	DATA SET/RUN NUMBER COLLATIO BOX DA X	DATE: POST-TEST	NO. X MACH NUMBERS	RUNS 0 300 600 900	369 370 371 372	373 374 375 376	377 378 379 380	384 383 382 381	385 382 387 388	389 390 391 392	st i	PUN !	395 398 =	396 399	90h	10h 10h 80h 411h	Sot tot 804 214	904 204 804 014	413 414 415 416	oct 614 814 LIH		425 427 429 432	57 61 67 75 76	2 1 1 1 2 1 1 1 1 2 1 1	10VAR (1)	D) = 200, 300, 400, 600, 800, 1000
SET/RUN NUMBER COLL PARAMETERS 0 400 A	DATA SET/ 8 Dx D/8 X 0 0 0 4 0 0 0 8 1 2 2 1 4 4 1 5 6 2 5 6 2 7 7 2 8 8 2 7 7 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 2 8 8 3 1 6 600, 800 5 6 600, 800 6 600,	CONFIGURATION SCHD. CONFIGURATION SCHD. TO/SE (PLUMES ON) O O O O O O O O O O O O O O O O O O	ATION SUMMARY	VALUES	PCH	1500	-	1															 				
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(Continued)	
TABLE II.	
BOOTTER (SRB) SEMENTON	

VRUN NUMBER COLLATION SUMMARY DATE : POST-AFEST	MODERN ORDINERALDA			No. Del										-	(٢
7 E MACH PT PCH RF RUNS O 300 US SO G G G G G G G G G G G G G G G G G G	TEST: IA13	(3			DA	⋖	T/RUI	YON 7	ABER C	מרוֹ,	TION	SUMA	IARY	DATE	12021	7E57			
7 E MALT PT PCH RE NOUS O 300 600 900 900 000 0 000 0 000 0 000 0 000 0 000 0 0				SCHD.	L			"	ARAMET	TERSA	ALUE	s		NO	×	AACH NL	JMBERS		
200 96 C*** \$\frac{1}{4} \frac{1}{4} \frac	CONFIGURATION			6	-	84	<u>></u>	1	<u>\</u>	1/401/	74	PC#	RF	RUNS		300	-	006	
200 G G H H H H H H H H H H H H H H H H H	RTJ 194 00 Tro/58 0		0	0	_	┝	200	96	1		110	1	0		9Zþ	82h		<i>(</i> 2)	
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800 800 4 445 447 448	134			┝──			400	g							#1	244	443	444	
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200 A 450 - - 454 400 A 461 451 454 454 200 A 462 461 460 457 457 60 H 460 461 460 459 470 471 471 471 471 471 471 471 471 471 471 470 <t< td=""><td>141</td><td></td><td></td><td></td><td>0</td><td>5</td><td>0</td><td>A</td><td></td><td></td><td></td><td></td><td></td><td></td><td>449</td><td>ı</td><td>١</td><td>254</td><td>: E</td></t<>	141				0	5	0	A							449	ı	١	254	: E
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6. 80. (000)	A) - A co. 444 (B	A) - A co. 444 (B	744 / 175	•		1	Ū	DEFFI	CENTS			7	/00/	25.	10.8	AR (1)	IC A R	(2	ū
	SCHEDULES D)= 200, 30, 400, 600, 800, 100, 800		DO, 400, 60	7 Z		1,600	200	00/12	7770			9/100	150.20	2,300,40	6	07-00	2		

6) = 600, 800, 1000 L) = 300, 400, 600, 800, 1000

H)= 400,600,800,000

BOOKTER!	KIE Z	Booster (SRB) Separation St: 1413	<u>ğ</u> _		DAT	TABL	TABLE II.		nt fu	(Continued)	ATION	TABLE II. (Continued) DATA SET/RUN NUMBER COLLATION SUMMARY	ARY	
			ן			7	אין		DER	רטרו	101 K	SOM W	1	
DATA SET	SET	MOLEVENION	SC	SCHD.				Ġ.	ARAM	ETERS,	PARAMETERS/VALUES	S		
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RTJ 154	24	09 Tio/58	0	0	0	-20	0	H		4.53	011	_	0	
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													-	
1	158	(PLUMES ON)			5	0	0	B		4.51	4.51 28.9	1500		
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1) = 200, 200, 400, 600, 800, 1000 F) = 150, 200, 300, 400, 600, 800, 1000 H) = 400, 600, 800, 1000

4) =0,50, 100, 120, 200, 300, 400, 600, 800, 1000 INCHES (ALLSCARE) B) = 0,50,100, 120, 200, 300, 400, 600

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(d) = 100, 150, 200, 300, 400, 600, 800, 1000

g = 50, 100, 10, 200, 300, 400, 600, 800, 100

G = 600, 800, 1000

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	ATE:	OL OL	RUNS																				3	280.	000)	009
	<u>a</u>	h																				55]	B = 0, 50, 100, 150,	H) = 400,600,800,1000	K) = 200, 300, 400, 600
3	> -		2	Á	-													Н						50, 10	600	300,
	MAR		اللا	0																	1	49	1	= 0,	400	380
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ued)	10C	TERS/	MAKH PT	£5 #																	•			THUMES (PAUSONE)		
(Continued)	BER (RAME	/	,																		37		ENTS ING	l I	
3	TA SET/RUN NUMBER COLLATION SUMMARY	Ā	Z	8	8	8	В	8	J	8	K,8*	7	8,5	*7'8	7	C	C	CA.	, A.	٨	<i>H</i>			COEFFICENTS	900	
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FRA			Ž O																			19	2	100/	57'04	% 60°C
BOOSTER (SRB) SEPARATION	1		CONFIGURATION	ين	TEE																	_	>	05/0:	J= 50,140,10, 200, 300, 400, 600, 800, 100	J)= 400,600 L)= 300,400,600
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	2	×	0	109	603	609	6/3	620	129	625	629	632	633	149	549	646	647	[pt]	299	199	663	61	뇤		162	9.	
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(Continued)	NUMBER	PARAMETERS/VALUES							***			***	12.0						D, E	فر		37		COEFFICENTS			
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Booster (SRB) SERVENTON	TEST: IA13	DATA SET	IDENTIFIER	RTJ 209	270	117	212	213	214	215	216	217	218	219	220	121	m	223	TAT	225	226		1	Į	SCHE		
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228	(PUMES OFF)			-20	0	ξ						·		737	738	743	744	
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230					400	<u> </u>								141	742	747	748	
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TABLE III. MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY (B19)			· · · · · · · · · · · · · · · · · · ·
GENERAL DESCRIPTION: Configura	ition 3 ligh	tweight Orbiter fo	ıselage
Model Scale = 0.010 Model	Drawing No.	SS-A00062	
Maria - C. Olo Model	DIANLE 1.0.		
DRAWING NUMBER: VI.70	<u>-0001393</u>		
DIMENSIONS:	•	FULL-SCALE	MODEL SCALE
Length ~ in.		1290.3	12,903
Max. Width \sim in. (9 $X_0 = 15$	328.3 in.)	267.6	2.676
Max. Depth \sim in. (@ X = 148	80.52 in.)	244.5	2.445
* Fineness Ratio		4.846	4.846
Area \sim st ²			
Max. Cross-Sectional ((0 X ₀ = 1480	.52 <u>in.)</u> 386.67	0.03867
Planform	•		
Wetted		****	
Base			

^{*} Finaness Ratio is the fuselage length divided by the equivalent diameter for the maximum cross-sectional area.

MODEL COMPONENT: Canopy (C7)		·
GENERAL DESCRIPTION: Configuration 3 light	ntweight Orbiter	Canopy
Model Scale = 0.010 Model Drawing No.	SS-A00062	
DRAWING NUMBER VL70-000139B		
DIMENSION:	FULL SCALE	MODEL SCALE
Length ($X_0 = 433$ IN. to $X_0 = 670$ IN.) Max Width		2.370
Max Depth	·	
Fineness Ratio		
Area		
Max Cross-Sectional		
Planform		
Wetted		
Base		

MODEL COMPONENT: Elevon (E23)		
GENERAL DESCRIPTION: Configuration 3 light Data for (1) of (2)		levons
Model Scale = 0.010 Model Drawing No.	SS-A00109	
DRAWING NUMBER: V1.70-000139B		
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area ~ ft. ²	205.52	0.02055
Span (equivalent)~in.	353.34	3.5334
Inb'd equivalent chord ~ in.	114.78	1.1478
Outb'd equivalent chord~ in.	55.00	0.550
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.208	0,208
At Outb'd equiv. chord	0.400	0.400
Sweep Back Angles, deg		
Leading Edge	0.00	0.00
Tailing Edge	-10.24	-10,24
Hingeline .	0.00	0.00
Area Moment (Normal to hinge line)~ft ³	1548.07	0.00155

MODEL COMPONENT: Bo	dy Flap (F ₅)	
GENERAL DESCRIPTION: Ca	nfiguration 3 lightweight orbiter body flap	<u> </u>
Model Scale = 0.010	Model Drawing No. SS-A00062	
DRAWING NUMBER	VL70-000139P	
DIMENSION:	FULL SCALE MODEL S	CALE
Length ~in.	<u>84.70</u> 0.8470	
Max Width~in.	267.6 2.6760	
Max Depth		
Fineness Ratio		
Area~ft.2		
Max Cross-Sectional		
Planform	142.5195 0.0142	5195
Wetted		
Base	38.0958 0.00380	2958

TABLE III. MODEL DIMENSIONAL DATA (CONTINUED)

MODEL COMPONENT:	OMS Pod (M,)
	Configuration 3 lightweight Orbiter Orbital Manuevering System Pod.
Model Scale = 0.010	Model Drawing No. SS-A00062
DRAWING NUMBER	V170-000139B
DIMENSION:	FULL SCALE MODEL SCALE
Length-in.	346.0 3.460
Max Width~in.	108.0 1.080
Max Depth~in.	113.0 1.13
Fineness Ratio	
Area	
Max Cross-Section	al ·
Planform	
Wetted	
Base	

MODEL COMPONENT: Rudder (R ₅)		·
GENERAL DESCRIPTION: Configuration 3 lighty	weight Orbiter :	rudder
Model Scale = 0.010 Model Drawing No.	SS-A00062	
DRAWING NUMBER: VI70-0001393 VI70-000095		
DIMENSIONS:	FULL-SCALE	MODEL SCALE
Area~ft. ²	106.38	0.010638
Span (equivalent)~in.	201,0	2.010
Inb'd equivalent chord~in.	91.585	0.91585
Outb'd equivalent chord~in.	50.833	0.50833
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	0.400	0.400
At Outb'd equiv. chord	0.400	0.400
Sweep Back Angles, deg.		
Leading Edge	34.83	34.83
Tailing Edge	26.25	26.25
Hingeline	34.83	34.83
Area Moment (Normal to hinge line)~ft3	526.13	0.00053

MODEL COMPONENT: VERTICAL TAIL (V7)		
GENERAL DESCRIPTION: Configuration 3 lightweig	ht orbiter vertical	tail,
double wedge airfoil sect	ions with rounded 1	eading edge.
Model Scale = 0.010 Model Drawing No	. SS-A00062	
DRAWING NUMBER: VL70-000139 B VL70-000095.		
DIMENSIONS:	FULL-SCALE	MODEL SCALE
TOTAL DATA		
Area (Theo)~ft ² Planform Span (Theo)~in. Aspect Ratio Rate of Taper Taper Ratio Sweep Back Angles~deg. Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) WP	413.25 315.72 1.675 0.507 .404 45,000 26.249 41.130	0.04132 3.1572 1.675 0.507 .404 45.000 26.249 41.130
Tip (Theo) WP MAC Fus. Sta. of .25 MAC W. P. of .25 MAC B. L. of .25 MAC Airfoil Section Leading Wedge Angle ~ deg. Trailing Wedge Angle ~ in. Void Area ~ ft ² Blanketed Area ~ ft ²	268.50 108.47 199.81 1463.50 635.522 0.00 10.000 14.920 2.00 13.17 12.67	2.6850 1.0847 1.9981 14.6350 6.35522 0.00 10.000 14.920 0.020 0.0013 0.0012

MODEL COMPONENT: WING (W107) GENERAL DESCRIPTION: Configuration 3 lightweight	Corbiter wing	
Model Scale = 0.010	Model Dra	wing No. SS-A0010
	DWG. NO. VI	L70-000139B
DIMENSIONS:	FULL-SCALE	MODEL SCALE
TOTAL DATA Area (Theo.)~ft ² Planform	2690,.00	0.2/20
Span (Theo) ∼ in. Aspect Ratio Rate of Taper	936.68 2.265 1.177	0.2590 9.3658 2.255 1.177
Taper Ratio Dihedral Angle, deg. (@ T.E. of Elevon) Incidence Angle, deg. Aerodynamic Twist, dag.	0,200 3,500 0,500	0,200 3,500 0,500
Sweep Back Angles, deg. Leading Edge Trailing Edge	+3,000 -45,000 -10,2 <i>l</i> ,	<u>43,000</u> <u>45,000</u> -10,24
0.25 Element Line Chords:~in. Root (Theo) B.P.O.O. Tip, (Theo) B.P.	35,209 . 689,24 137,85	35,209 6,8924 1,3785
MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC	274.81 136.89 299.20 182.13	1.763 1.7491 11.3689 2.9920 1.8213
EXPOSED DATA Area (Theo) ~ ft ² Span, (Theo) ~ in. (From B.P. 108 in.) Aspect Ratio	1752.29 720.68	0.17523 7.2058
Taper Ratio Chords: ~ in. Root (6 B.P. 108)	2.058 0.2451 562.40	2,058 0,2451
Tip MAC	137.85 393.03	5.621,0 1.3785 3.9303
Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC Airfoil Section (Rockwell Mod NASA)	1185,31 300,20 251,76	11.8531 3.0020 2.5176
Root $\frac{T}{C} = 0$ Y _o 199 to NAXA 0010	0.10	0.10
Tip $\frac{T}{c}$ = Data for (1) of (2) Sides	0.32	0.12
Leading Edge Cuff Planform Area ~ ft ² Leading Edge Intersects Fus M. L. • Sta Leading Edge Intersects Wing • Sta		<u>^.00183</u> 5.C2 10.834

MODEL COMPONENT: External Tank (T)	ი)	Participation of the second of
GENERAL DESCRIPTION: Configuration 3 E	xternal Oxygen Hy	drogen Tank
Model Scale = 0.010 Model Drawin VL72-000088		
DRAWING NUMBER V178-000041		
DIMENSION:	FULL SCALE	MODEL SCALE
Length~in.	1865	18.65
Mox Diameter~in.	324	3,24
Max Depth		
Fineness Ratio (Length/Max. Dia.) Area~ft ²	5.75617	5.75617
Max Cross-Sectional	572.555	0.05726
Planform		
Wetted		
Base		

MODEL COMPONENT:	Booster (Sg)		
GENERAL DESCRIPTION:	Configuration 3 Boo	oster Solid Rock	cet Motor
	Data for 1 of 2 Box	osters	
Model Scale 0.010	Model Drawing	No. SSA-00113	
DRAWING NUMBER	VL72-000088 VL77-000036		
DIMENSION:		FULL SCALE	MODEL SCALE
Length (Includes	Nozzle)~in.	1741.0	17.410
Mox Width (Tank Dia.)~in.		142.0	1.420
Mox Dia. (Max. Nozzle Shroud Dia.)~in		205.0	2.050
* Fineness Ratio		8.49268	8.49263
Areo~£t ²			
Max Cross-Secti	onal (Nozzle Shroud)	229,21	0.02292
Planform			
Wetted			
Base			

^{*} Length divided by nozzle shroud diameter.

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axes have been displaced from the center For clarity, origins of wind and stability of gravity 2.

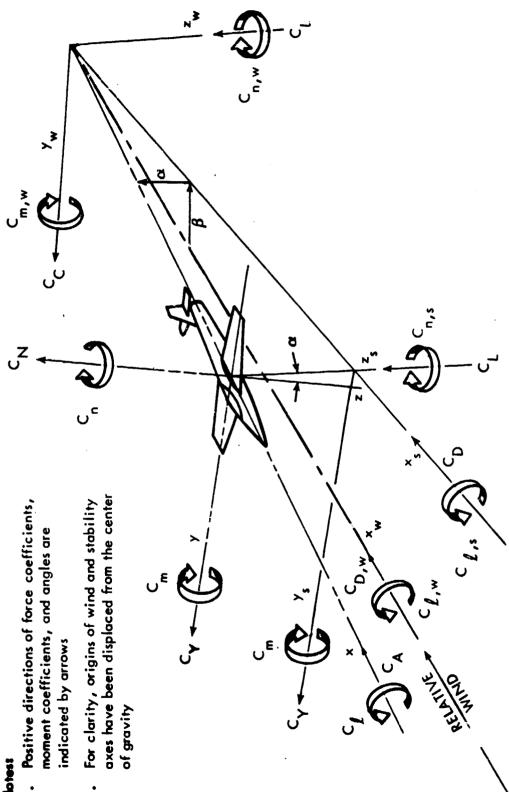
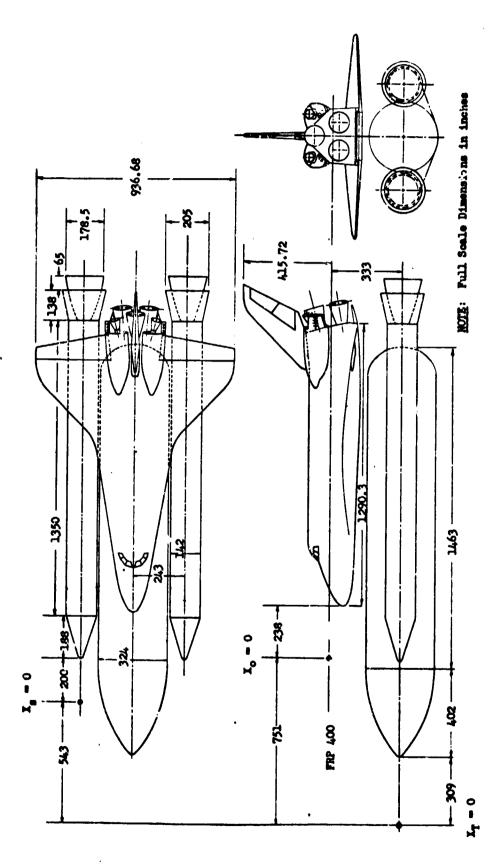
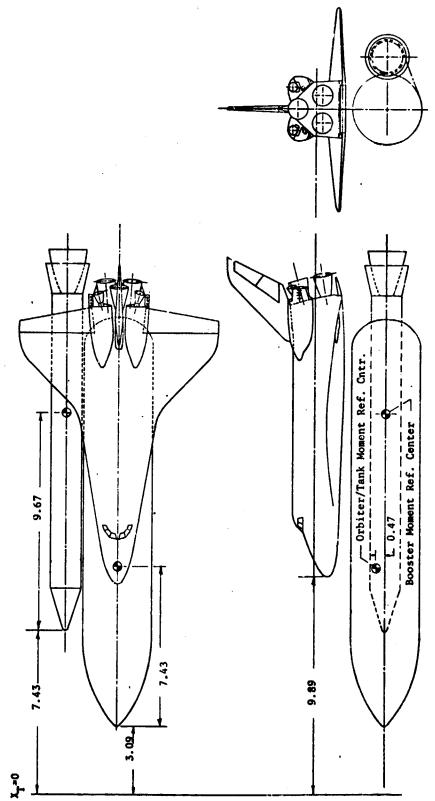


Figure 1. Axis Systems



a. Integrated Vehicle Configuration 3 (Mated)
Figure 2. Model Sketches

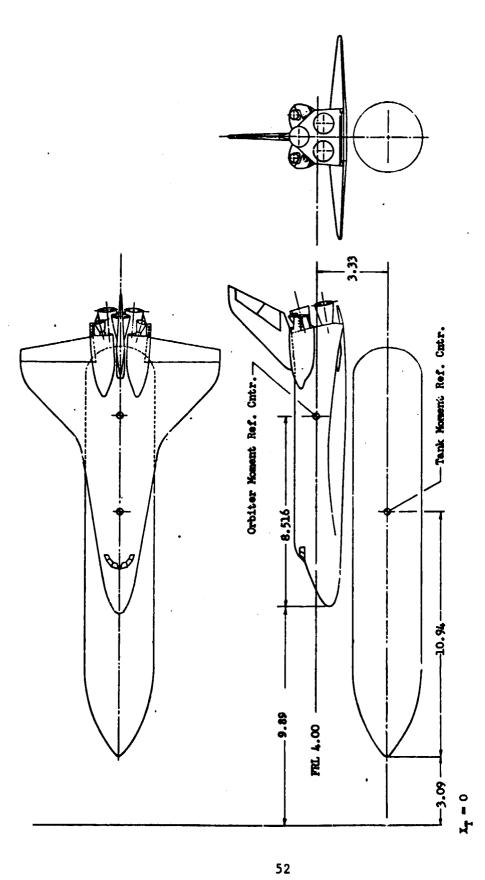


Total children and a second and a second

HOTES:

- Only righthand booster used for Booster Separation testing. Orbiter rigidly attached to tank.
 Mated position shown.
 All dimensions in inches.

Model 32-OTS Integrated Vehicle Configuration 3 Figure 2. - Continued Ъ.



NOTES:

- Models for Tank separation from Orbiter testing. Mated position whom.
 All dimensions a inches.

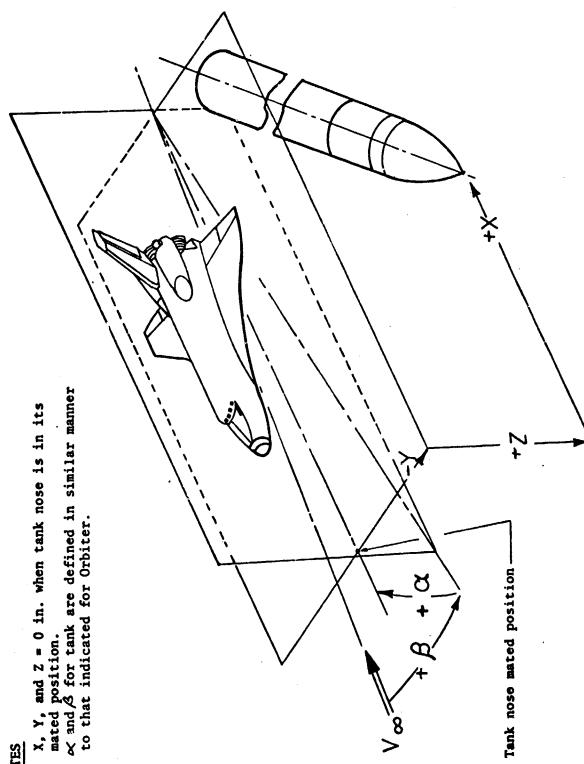
Model 32-OTS Orbiter and Tank Configuration 3 Figure 2. - Continued

NOTES

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and Z = 0 in. when tank nose is in its

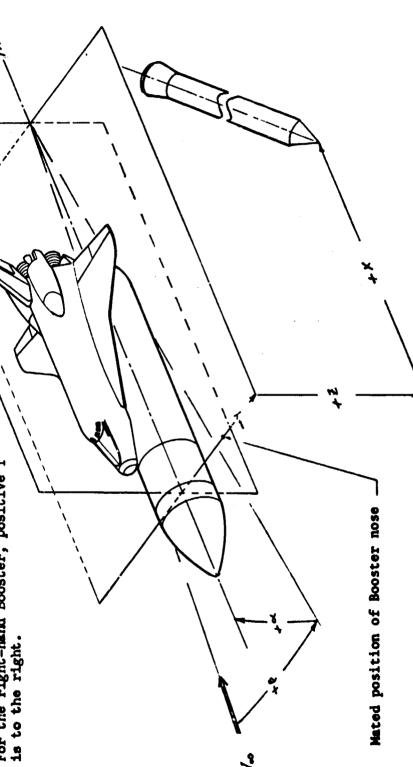
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Tank Separation from Orbiter Figure 2. - Continued

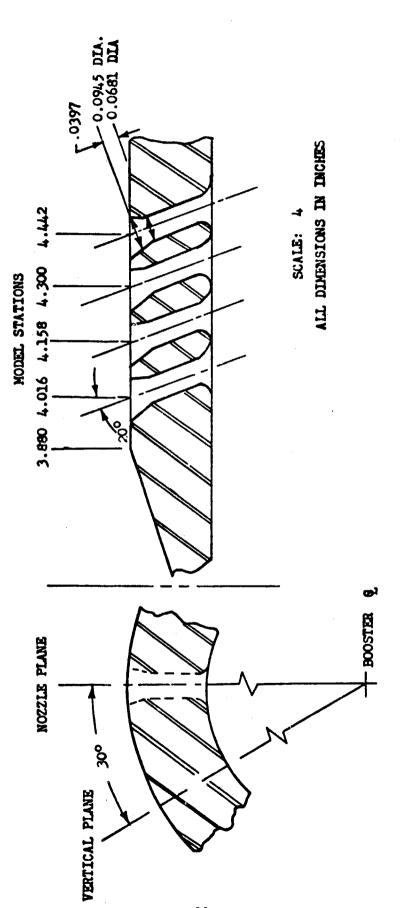
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- 1. X, Y, and Z = 0 when Booster nose is in its mated position.
- α and β for Booster are defined in similar manner to that indicated for the Tank. તં
- For IAl3, it is the right-hand Booster that is tested (not the left-hand, as indicated here). For the right-hand Booster, positive I ь.



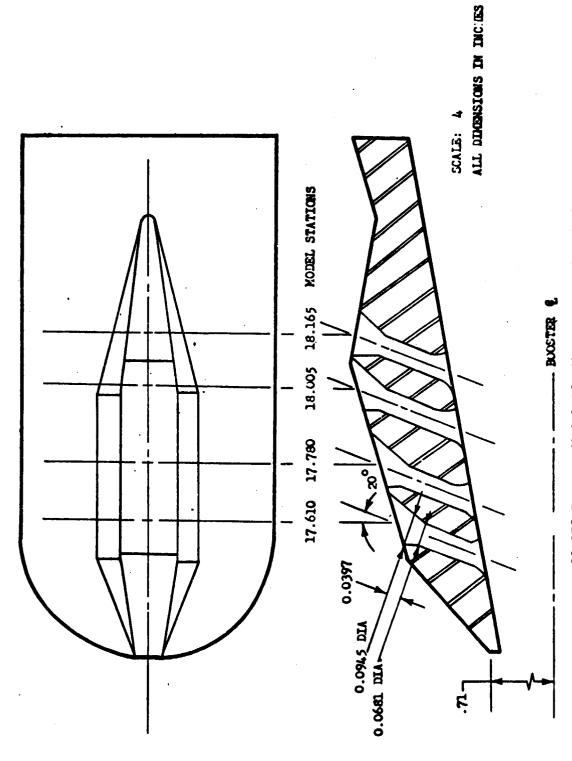
e. Booster Separation from Orbiter/Tank

Figure 2. - Continued

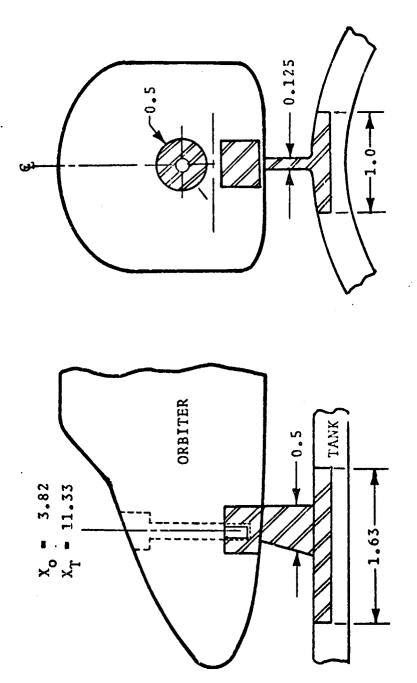


. 32-OTS Booster Model Forward Thruster Nozzle Block Figure 2. - Continued

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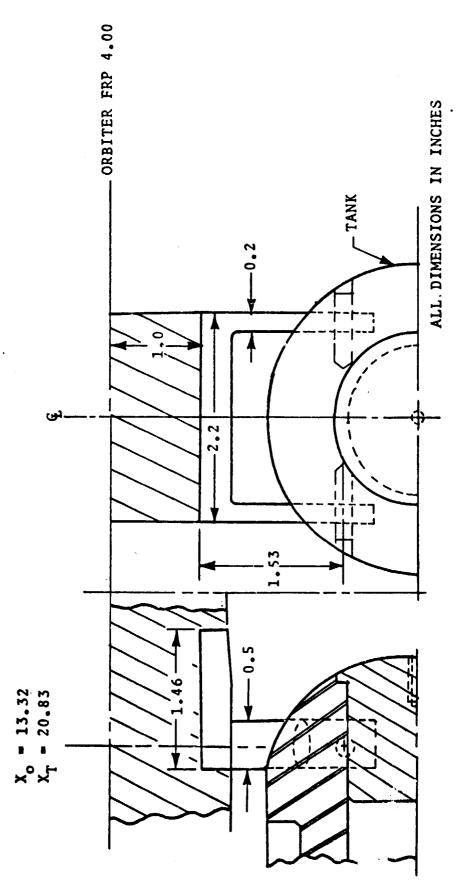


8. 32-OTS Booster Model Aft Thruster Nozzle Block Figure 2. - Continued



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h. 32-OTS Orbiter/Tank Forward Attachment Figure 2. - Continued



. 32-OTS Orbiter/Tank Aft Attachment Figure 2. - Continued

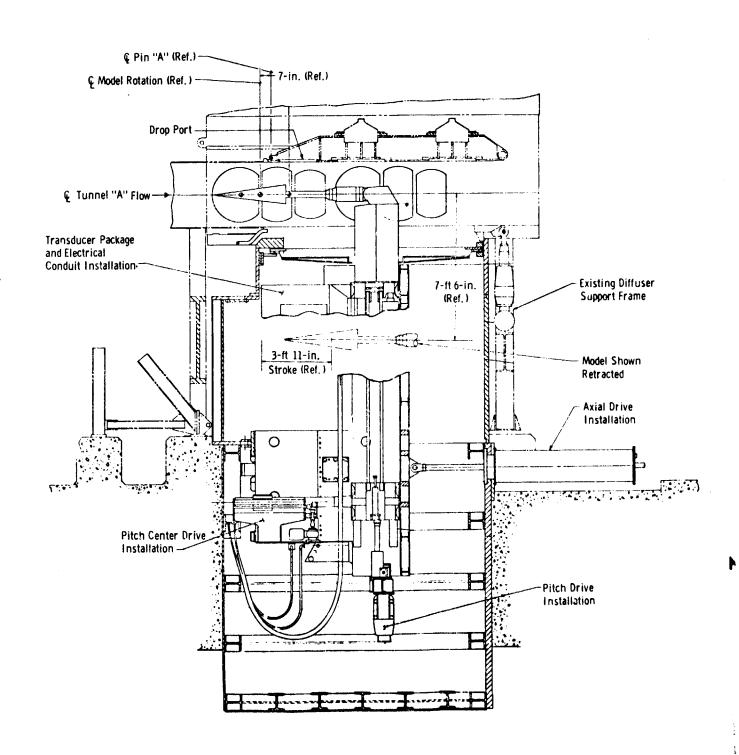


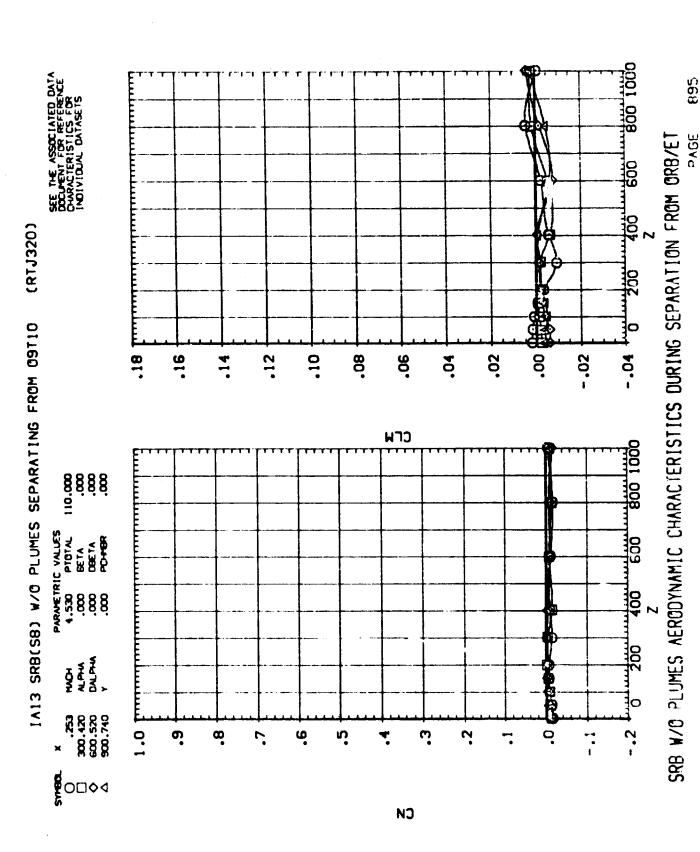
FIGURE 3. AEDC von Karman Gas Dynamic Facility Tunnel A

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I	Data Figures (pages 1 through 894)
II	Data Figures (pages 895 through 1498)
III	Tabulated Source Data

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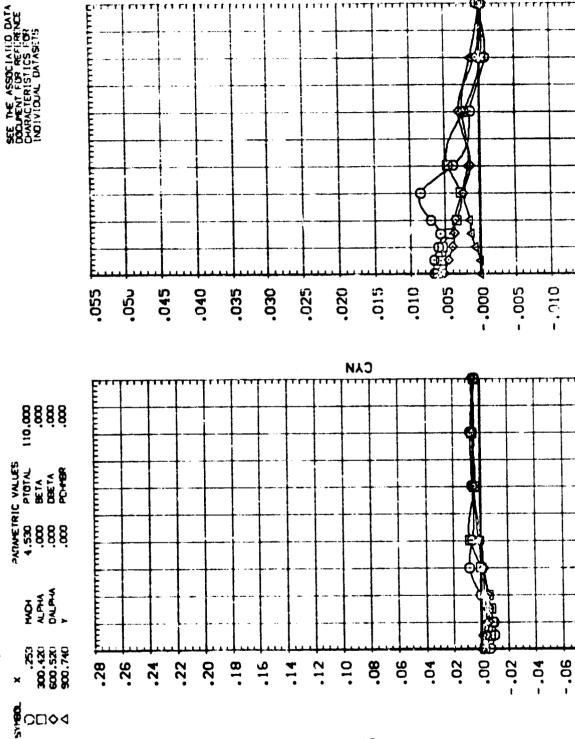
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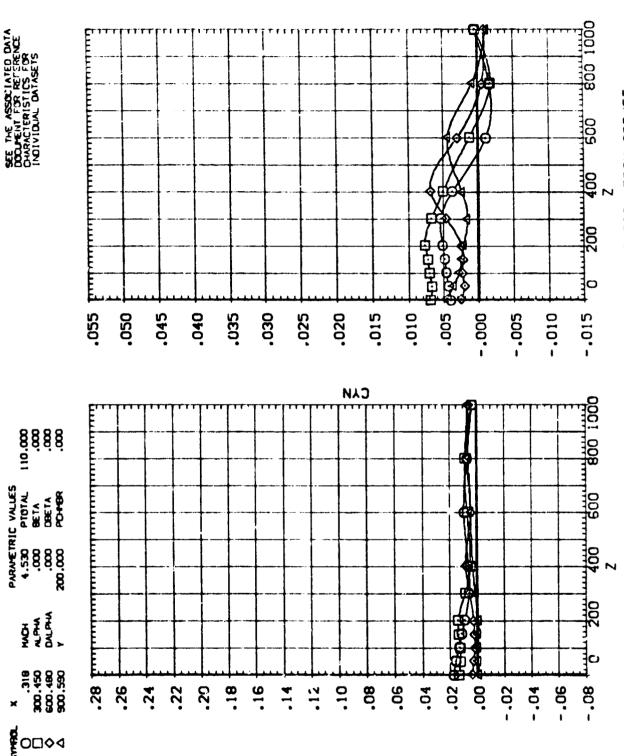
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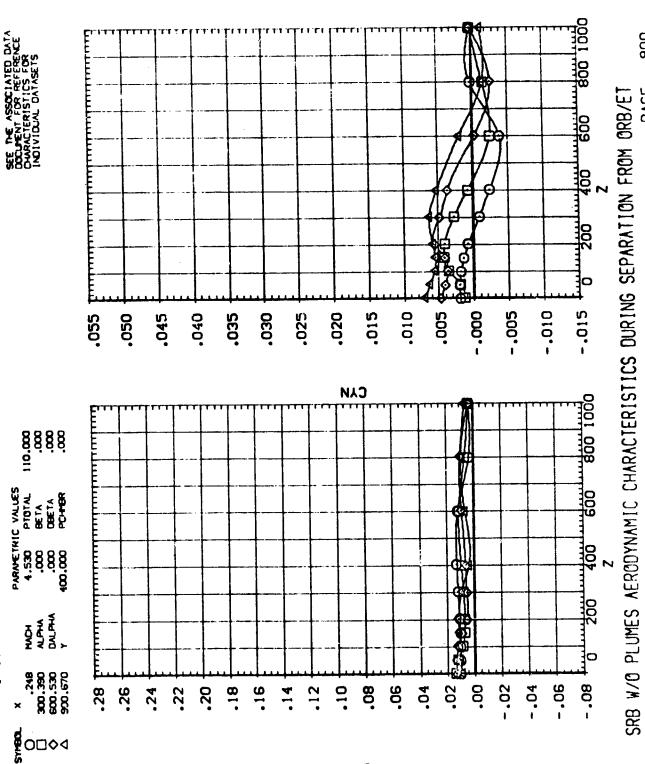
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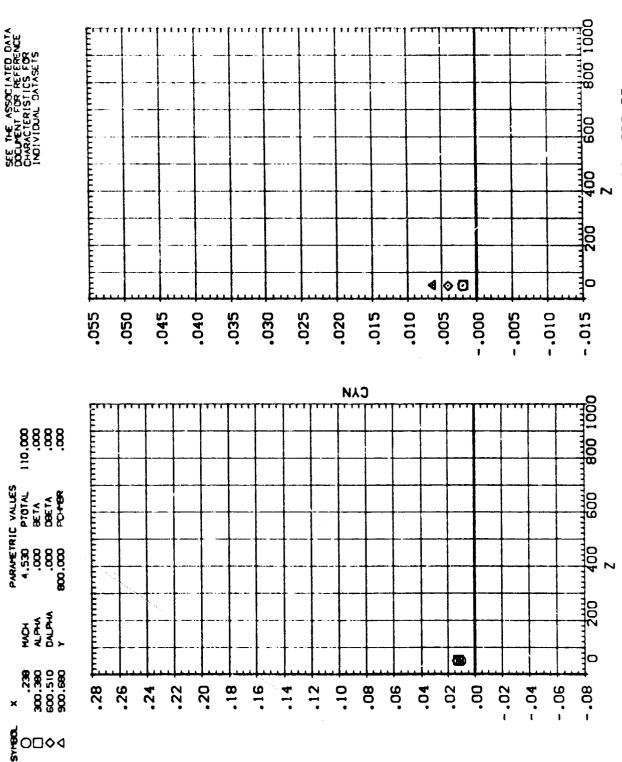
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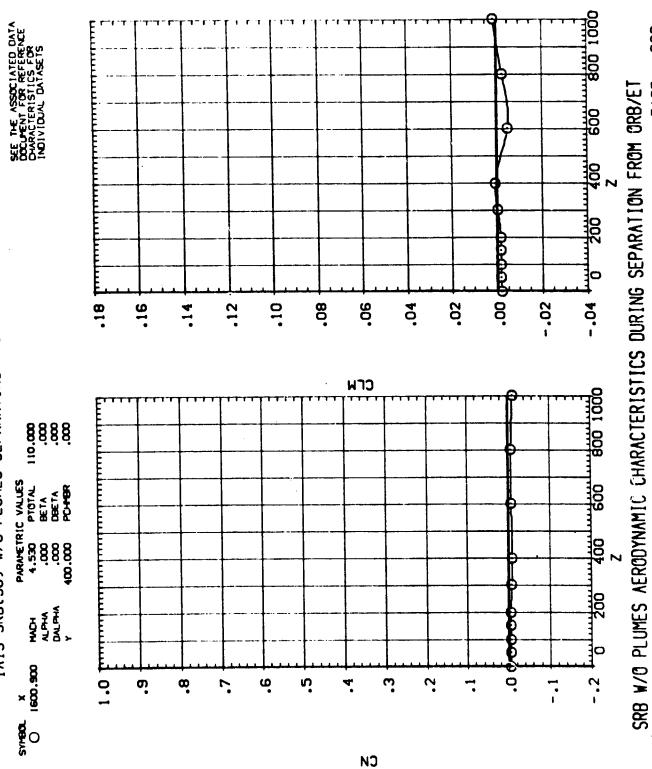
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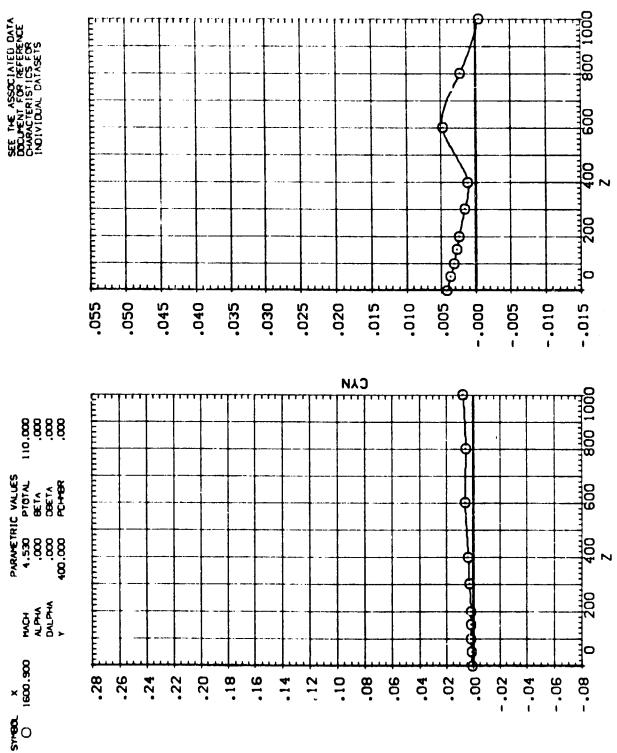


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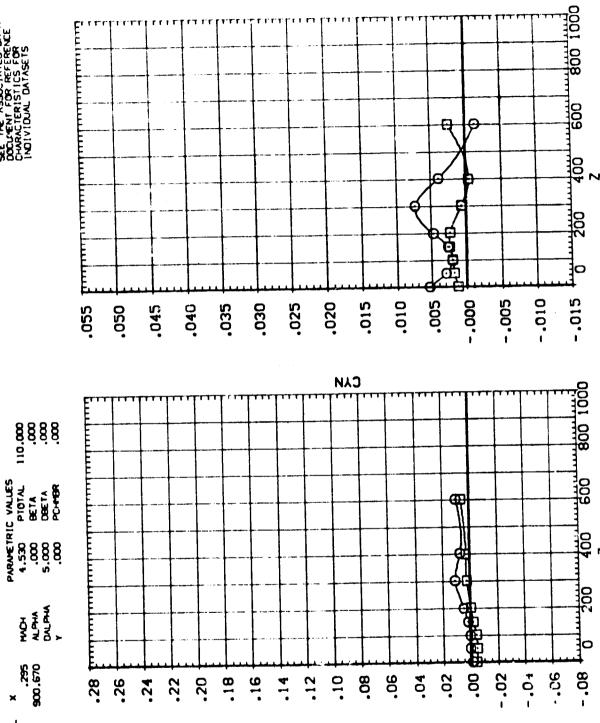


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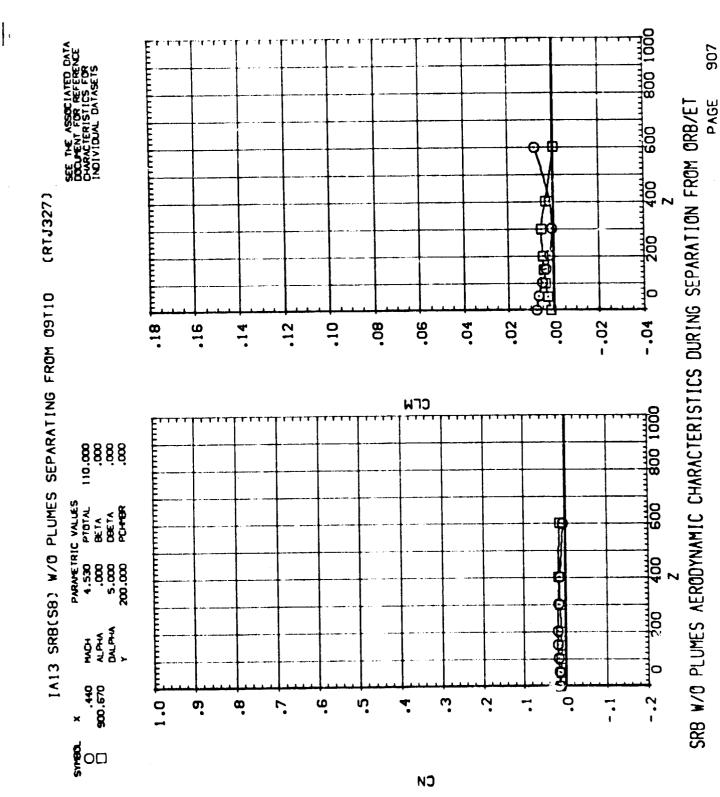
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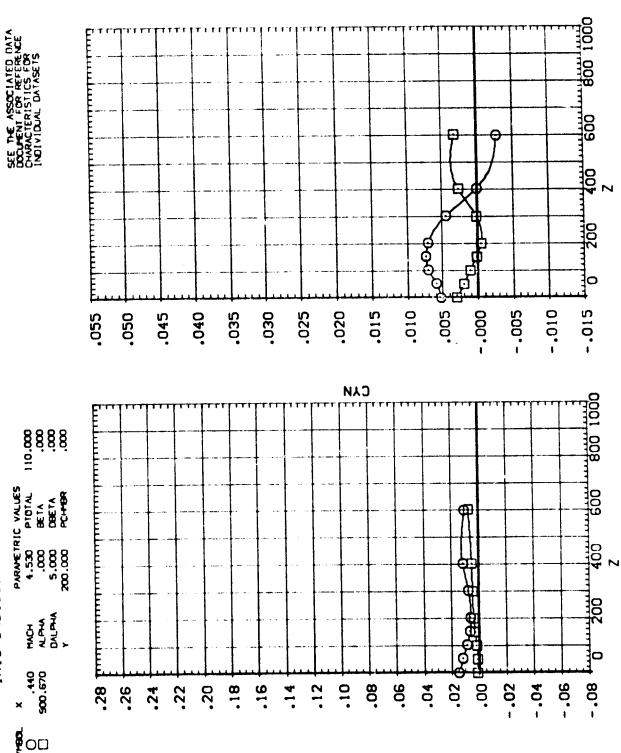


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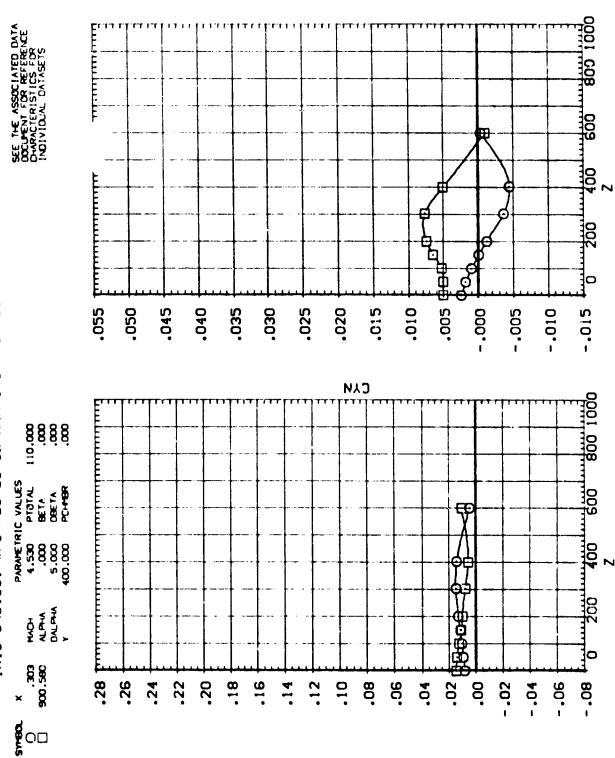
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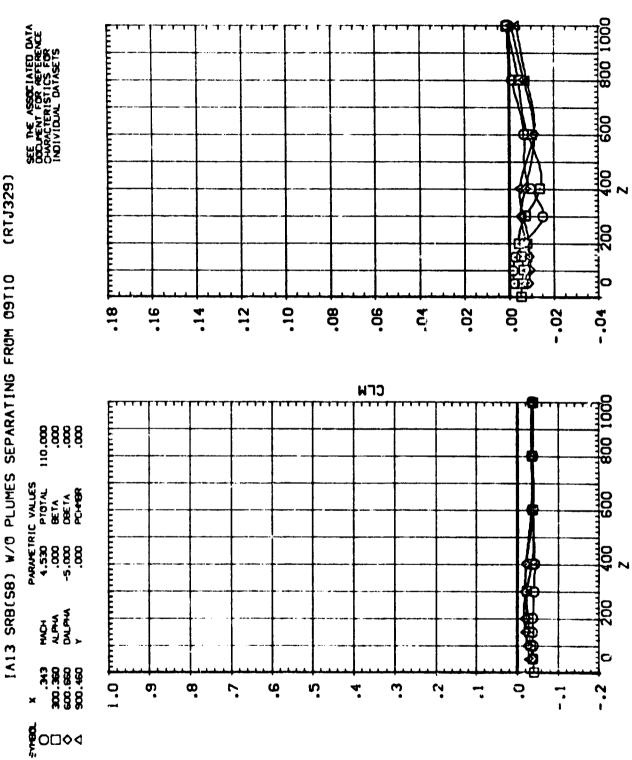
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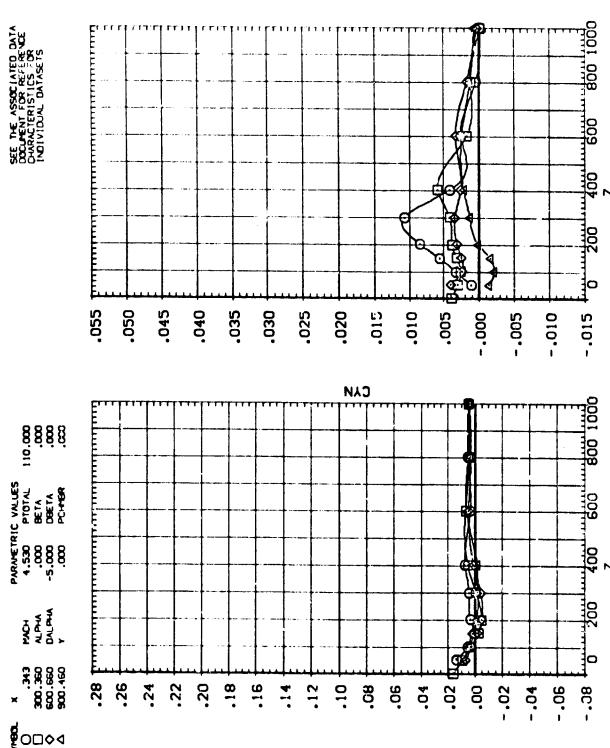
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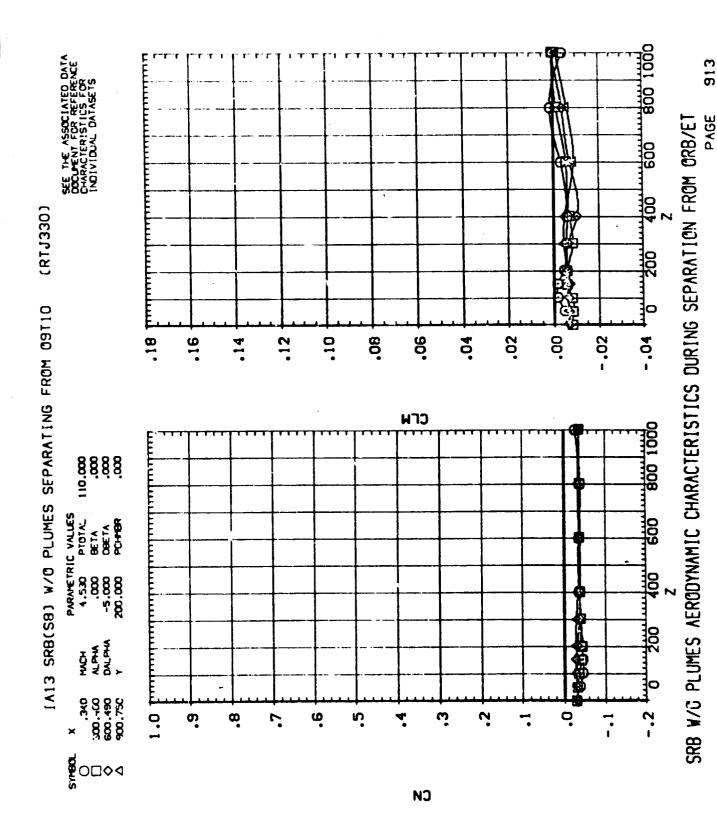
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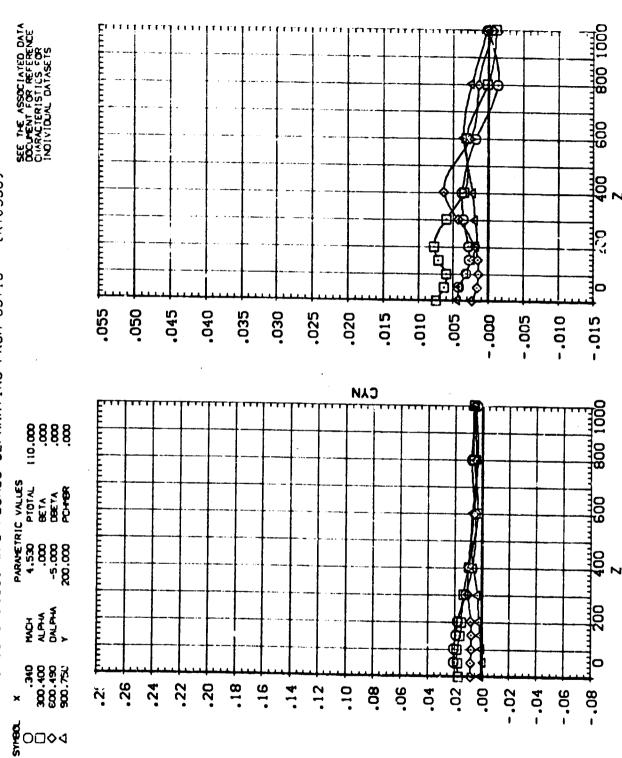


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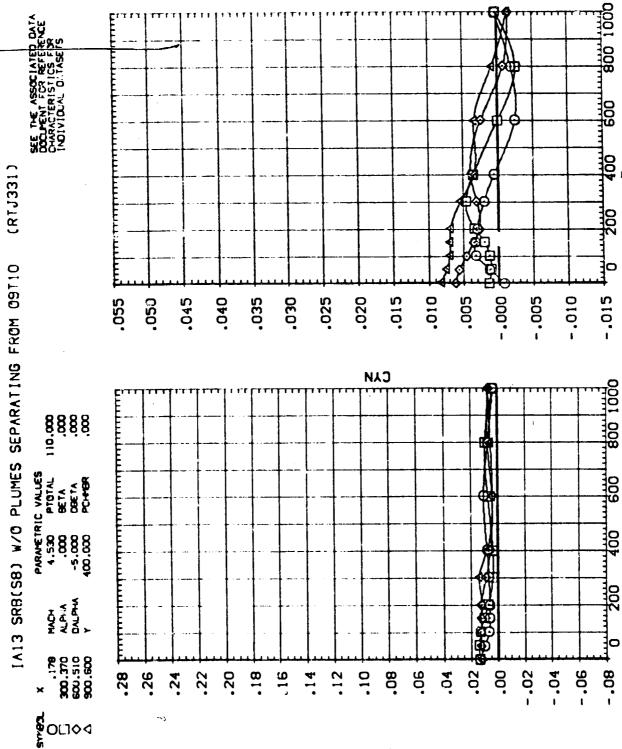
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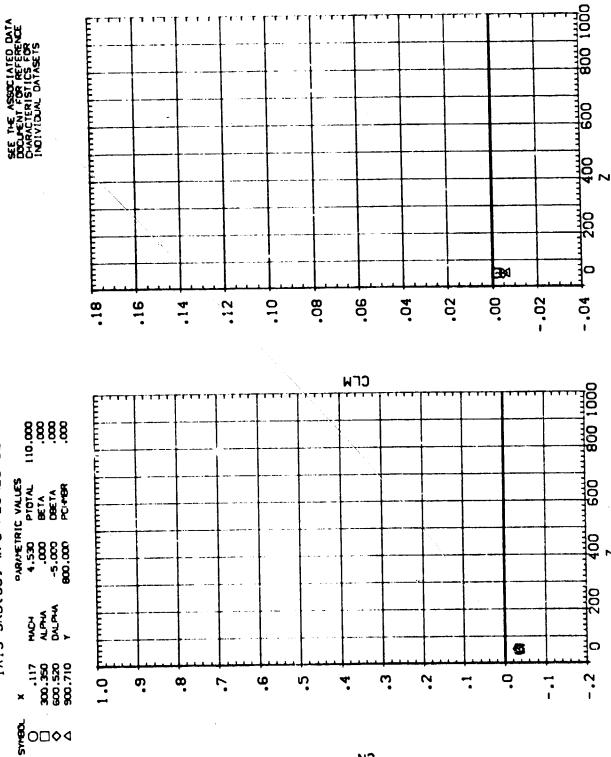


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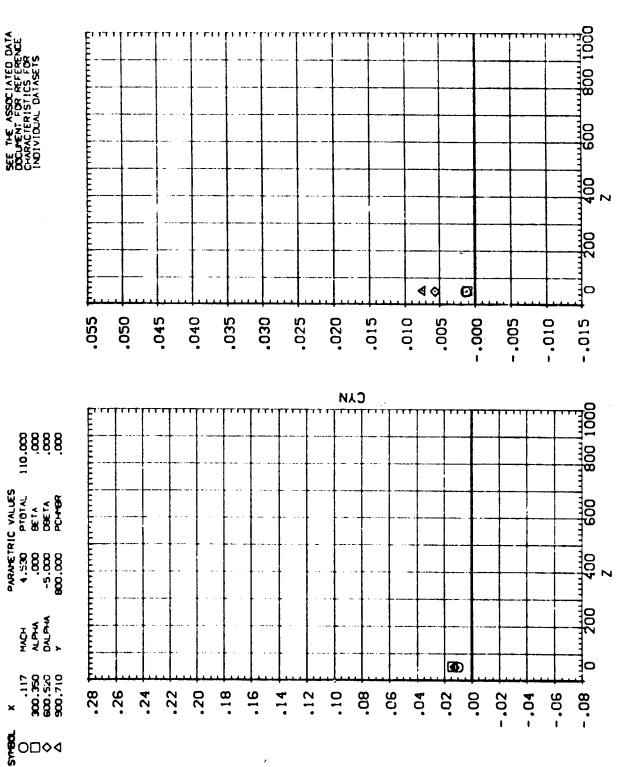
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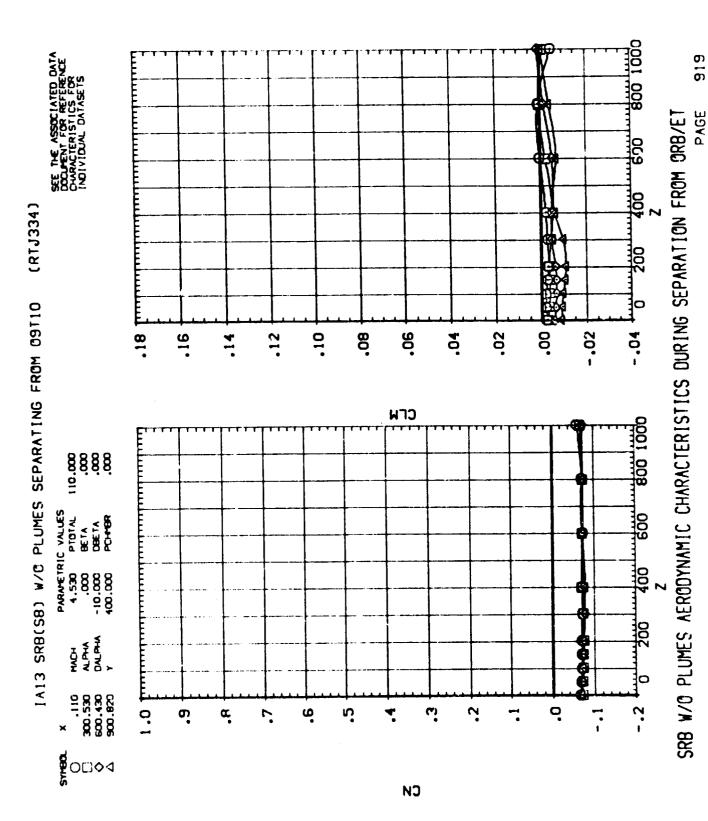
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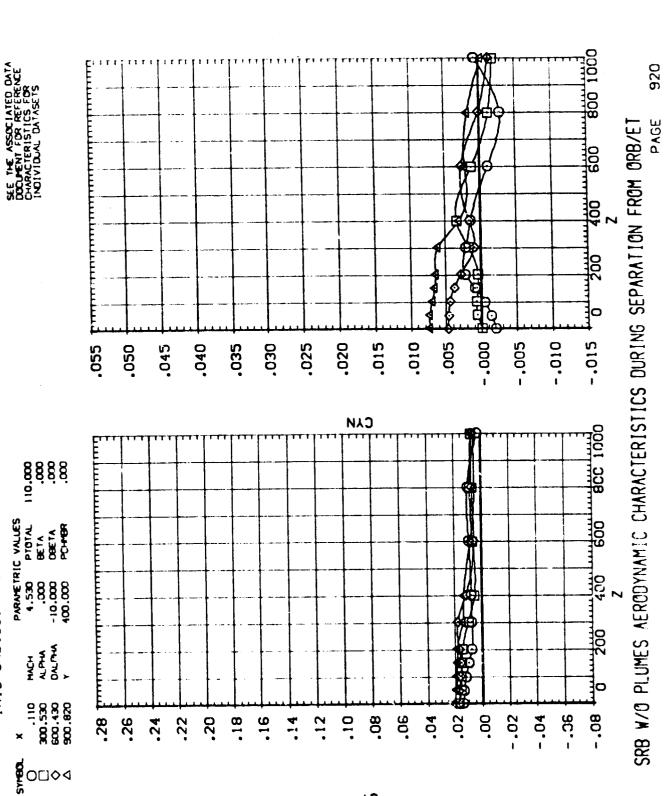
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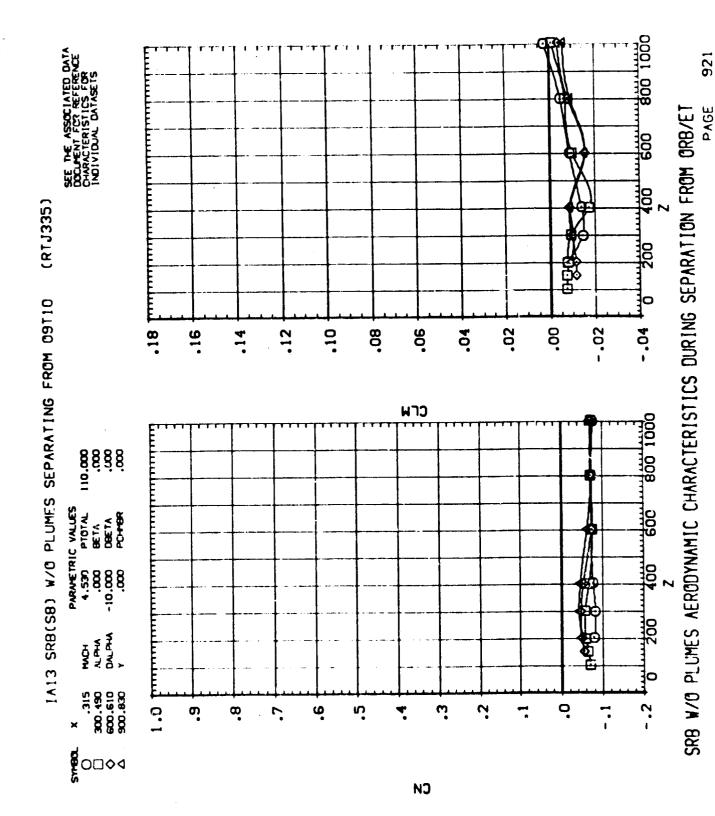




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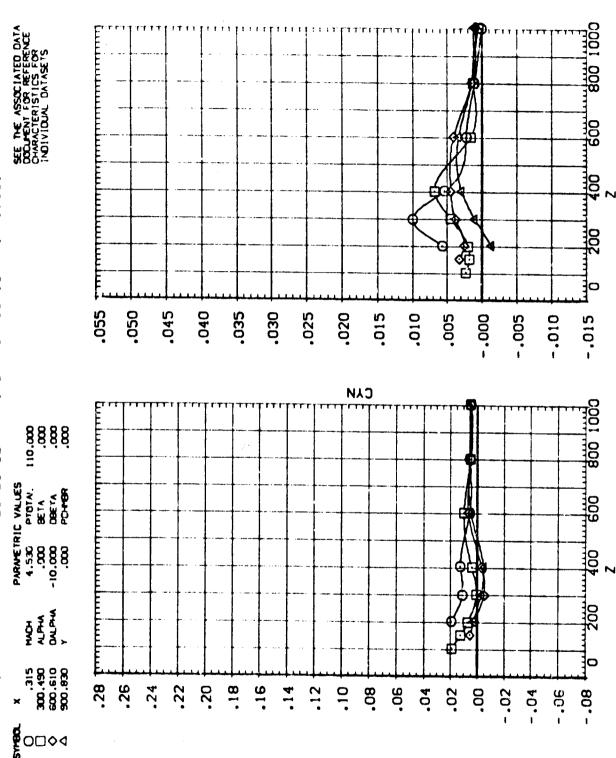


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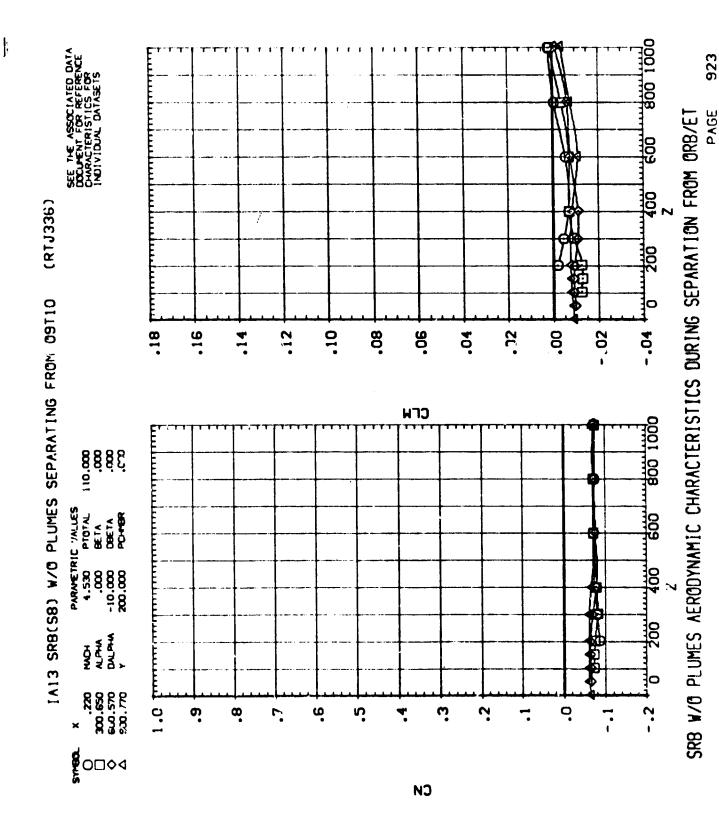
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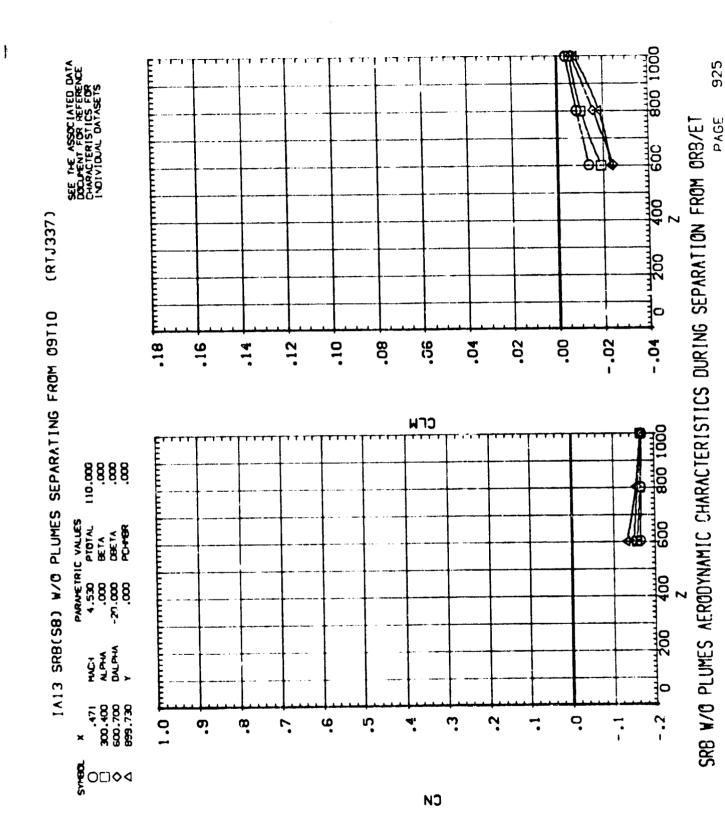
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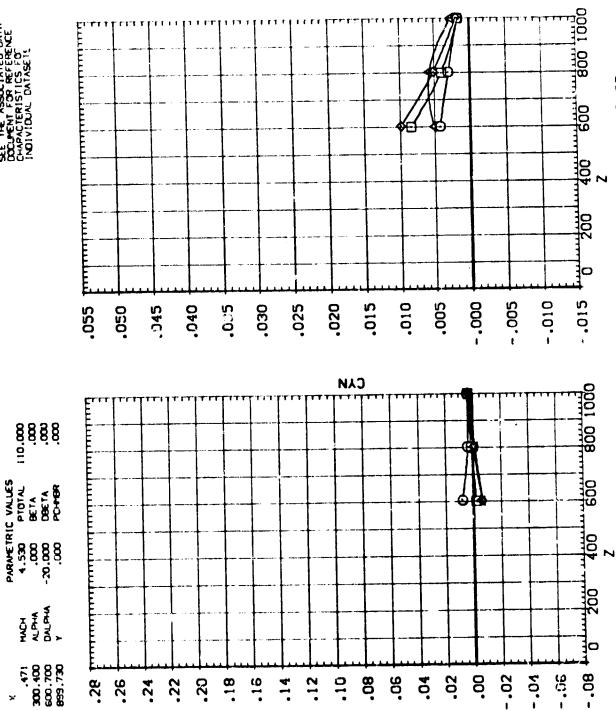


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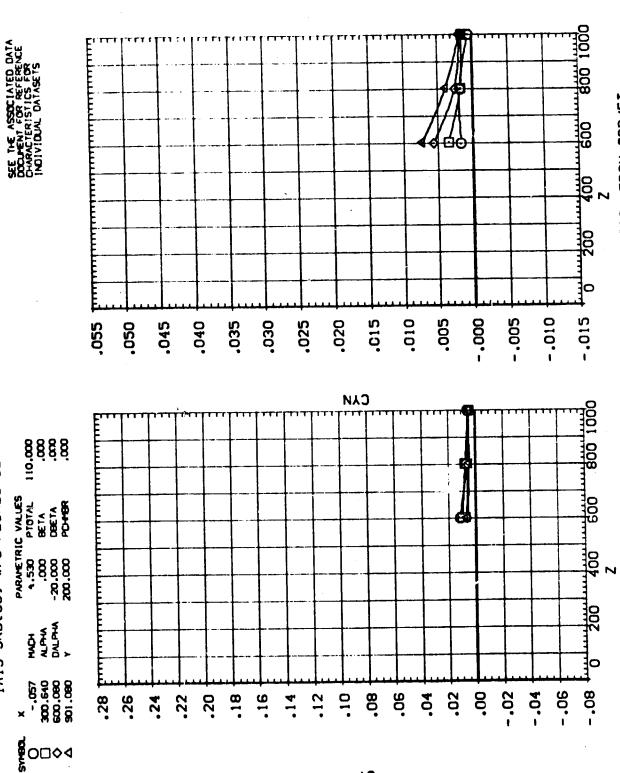
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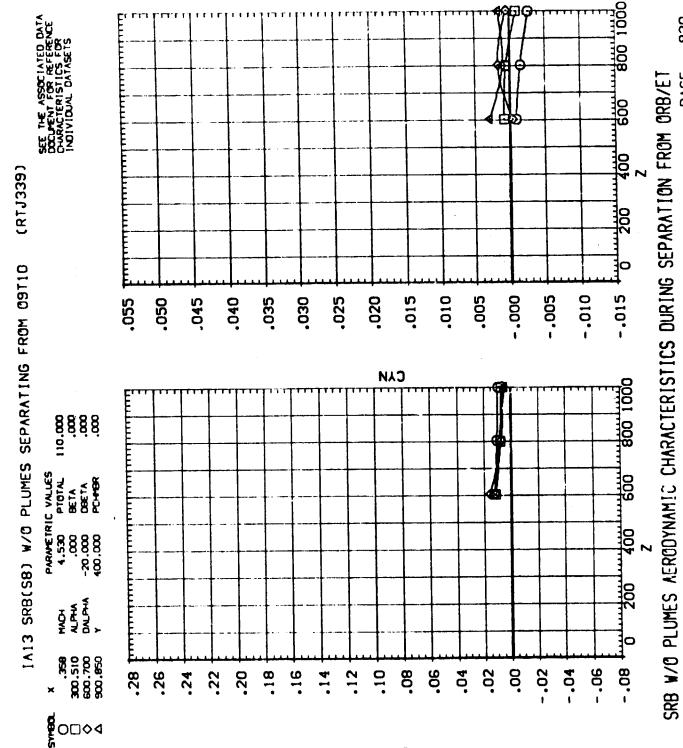
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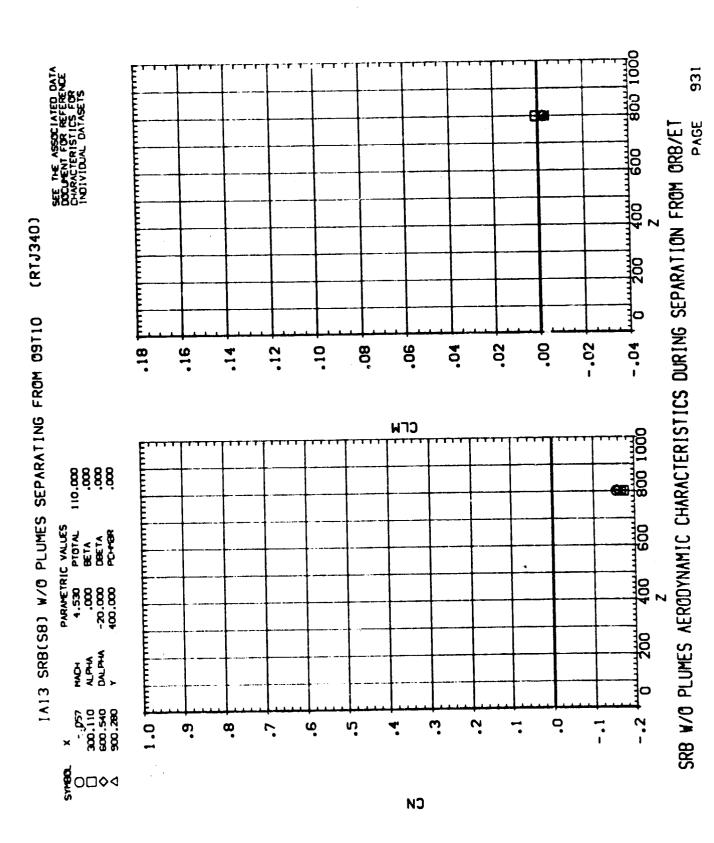
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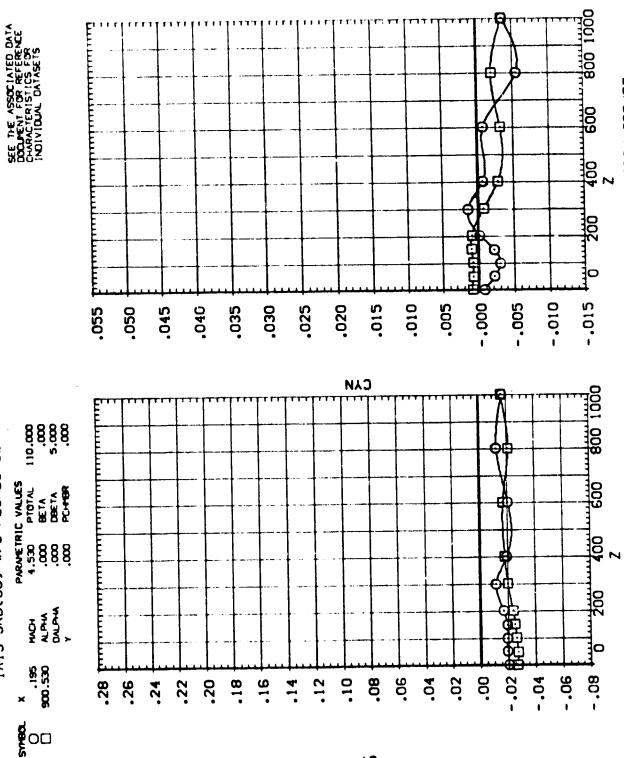
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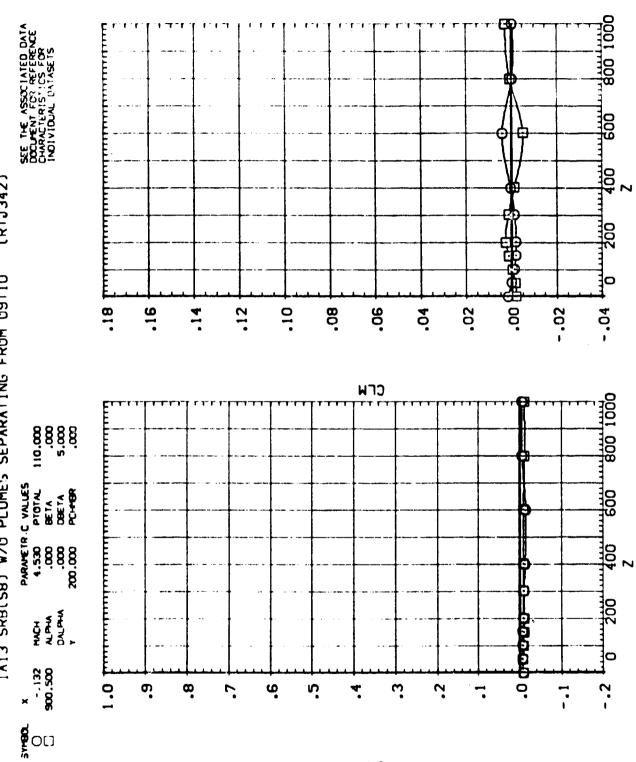


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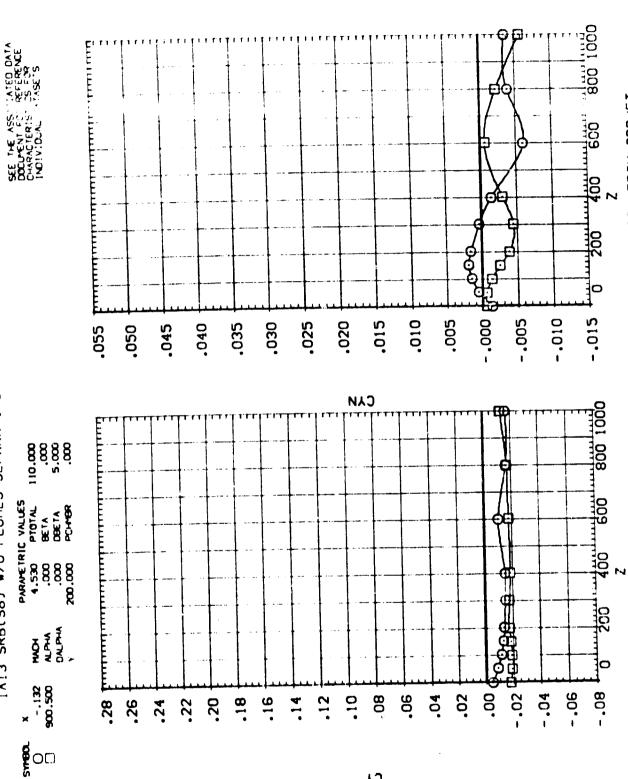


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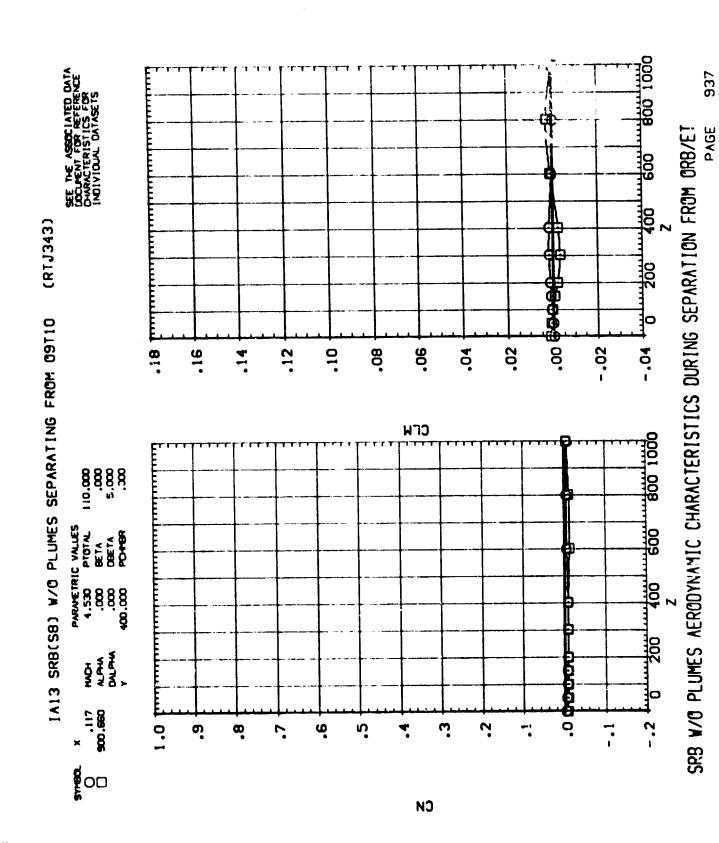
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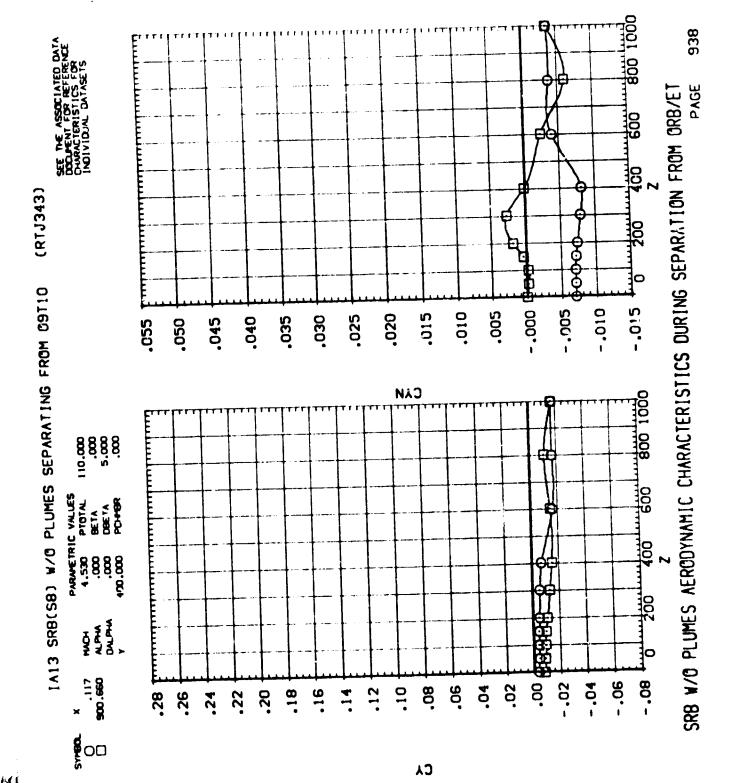
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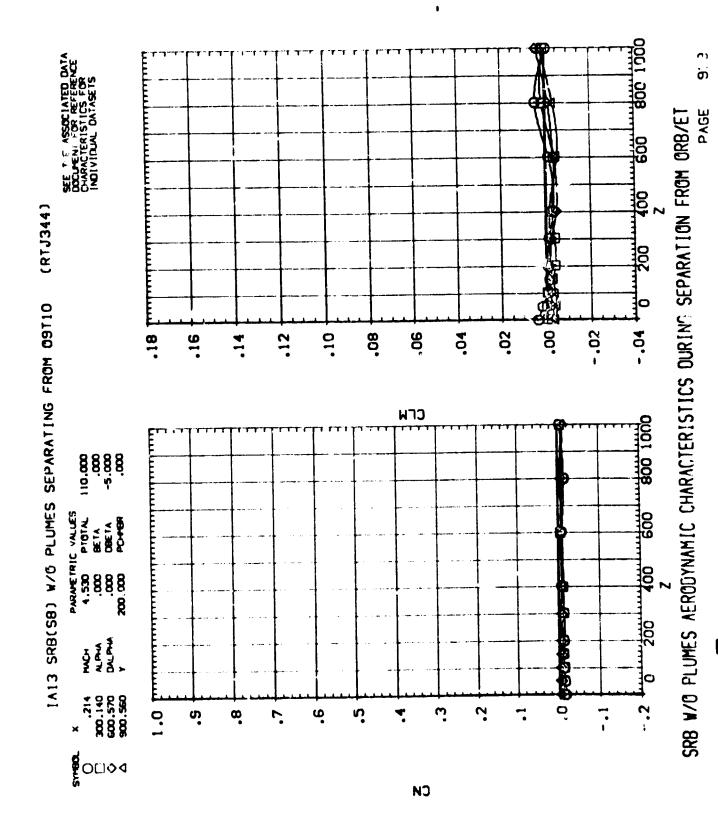
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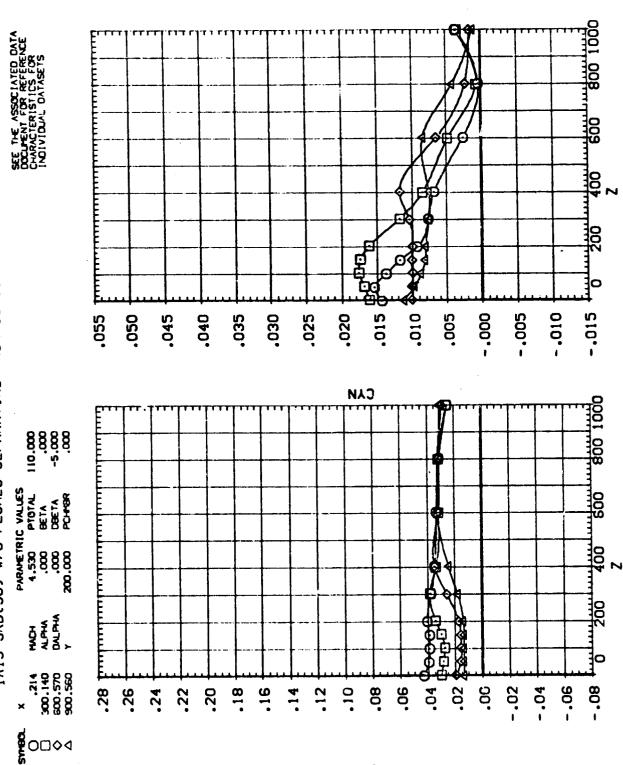




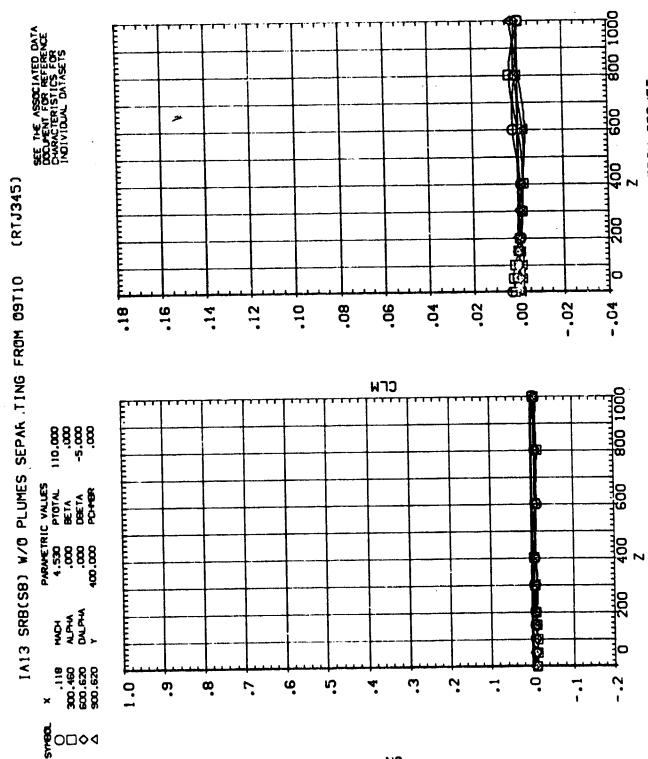
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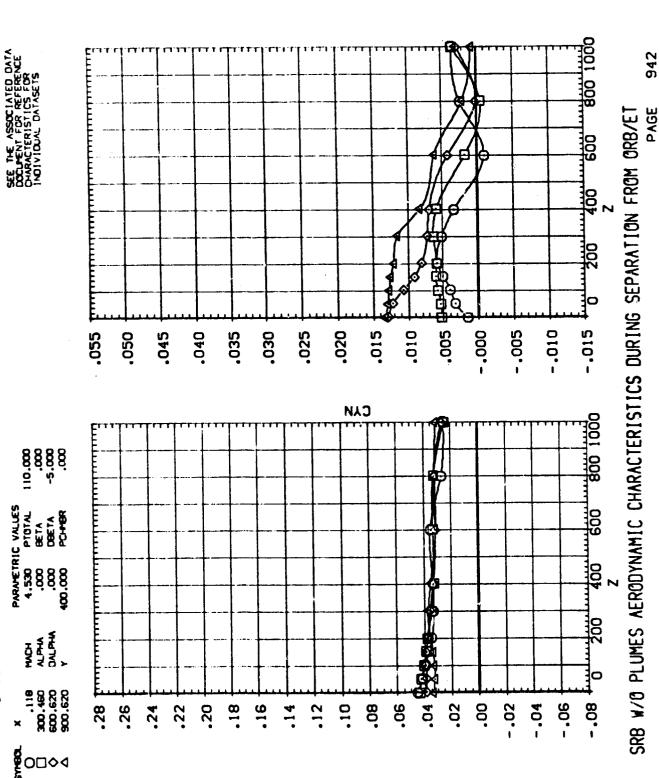


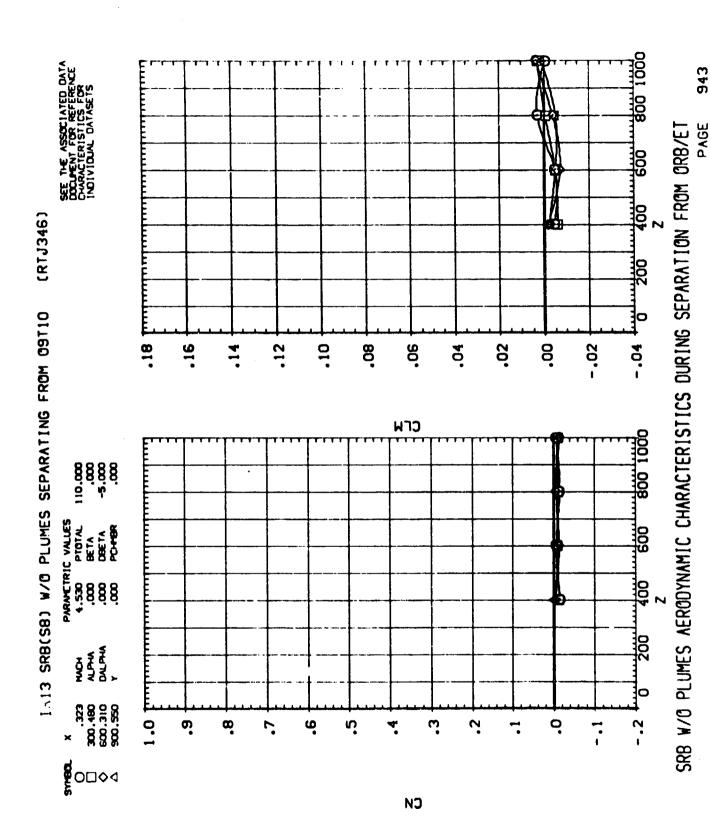
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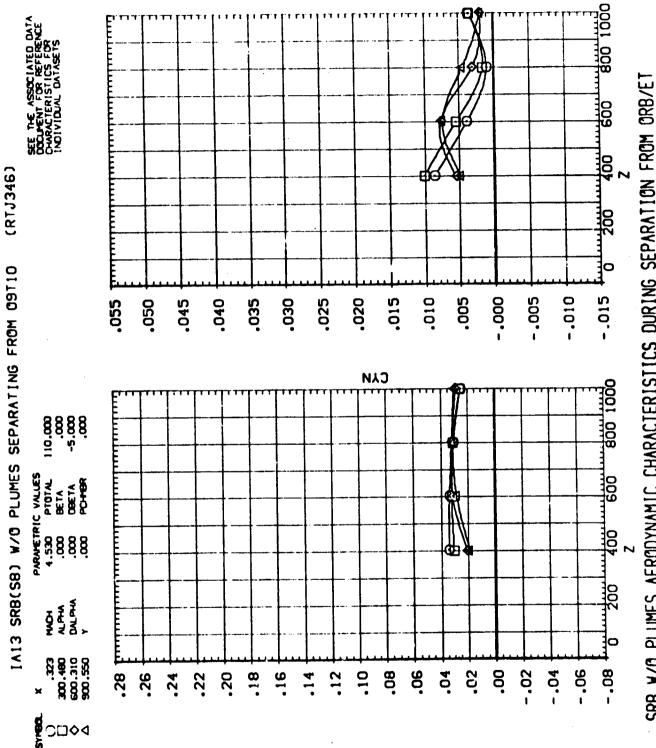
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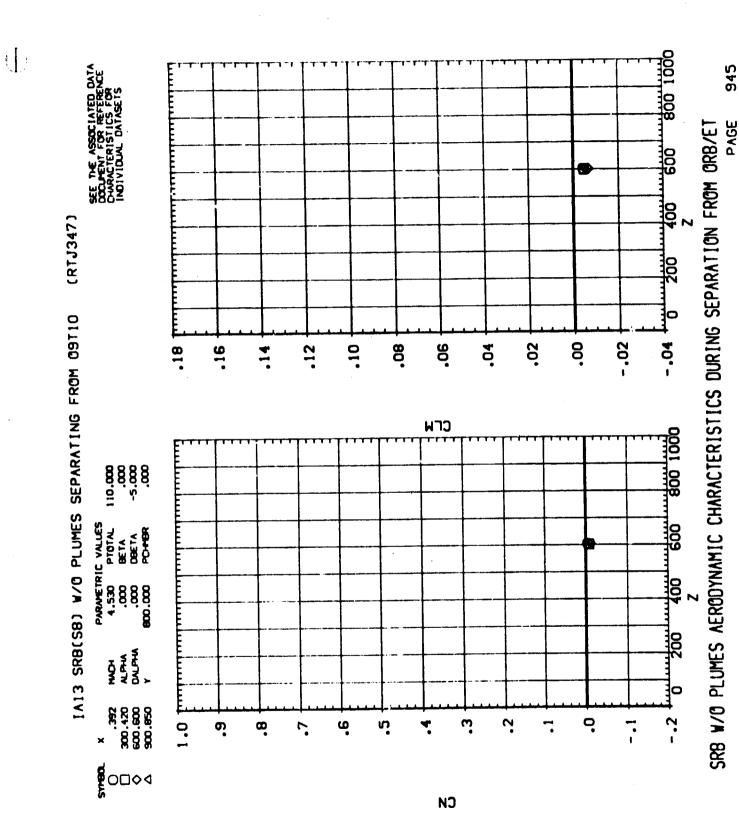


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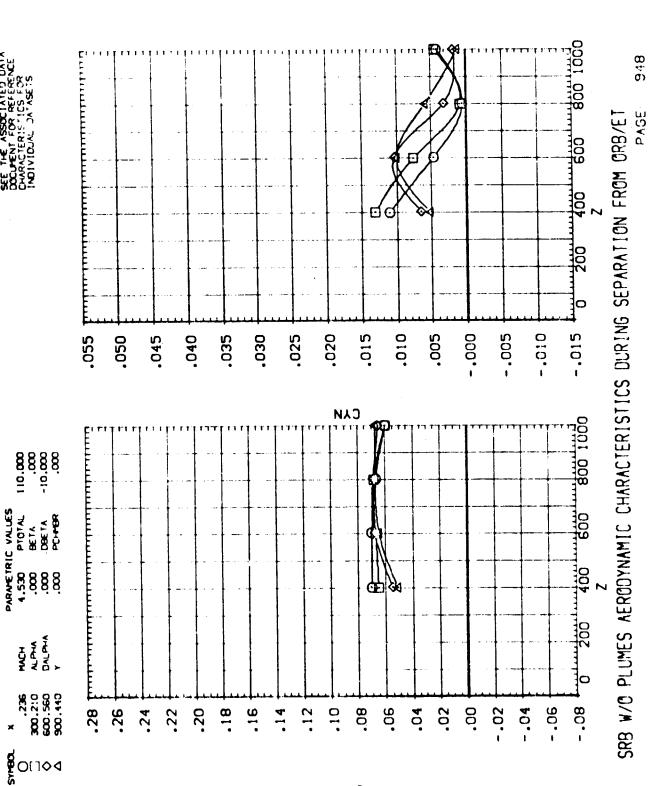
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ348)



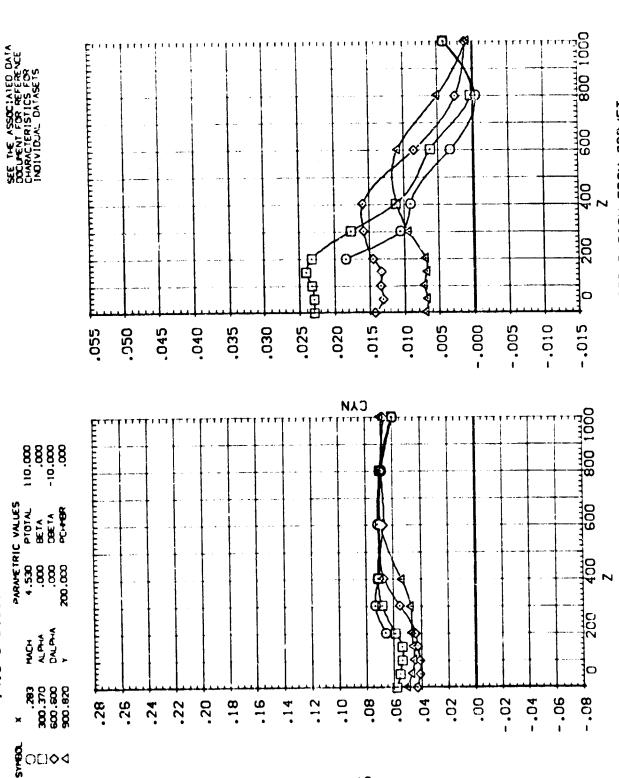
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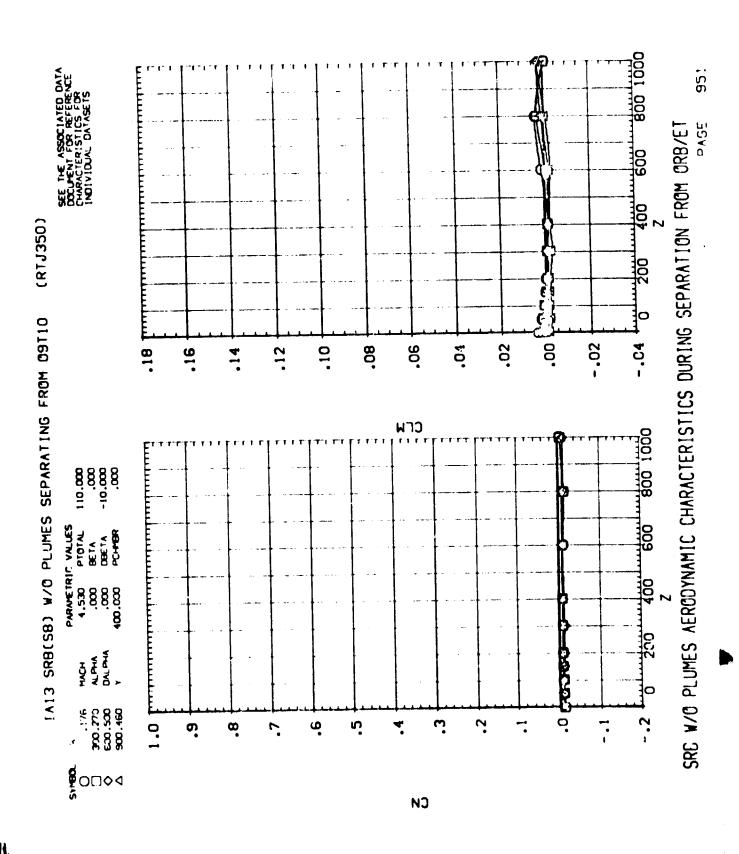
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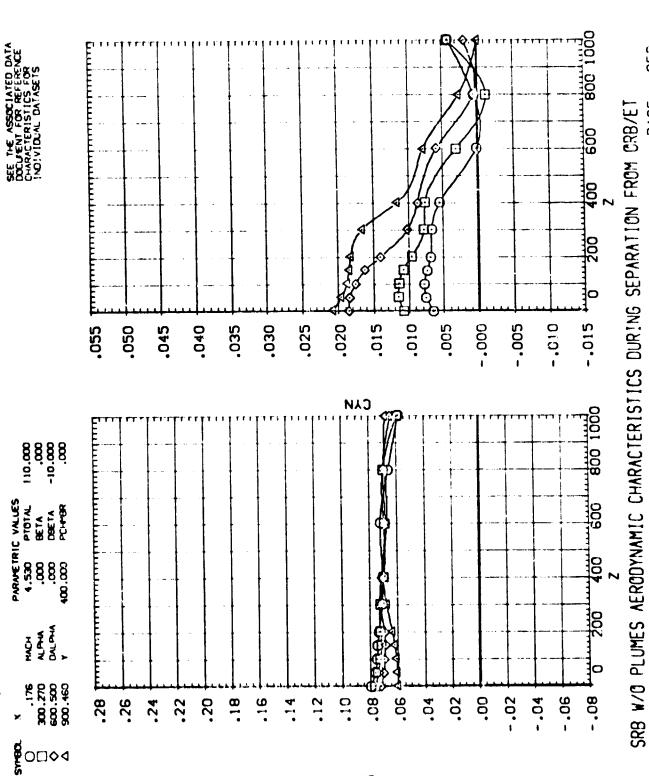
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ349)

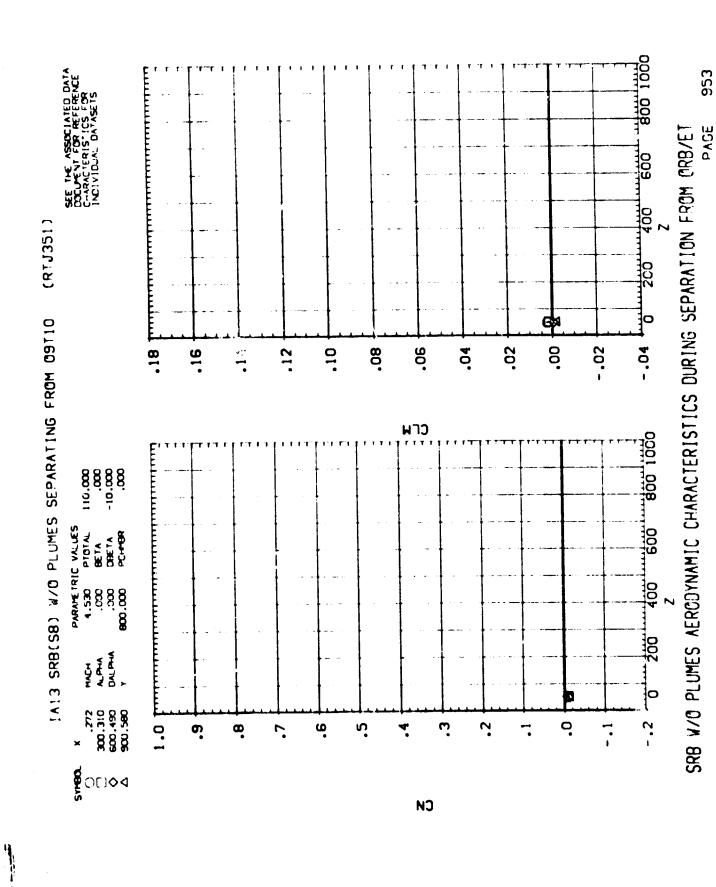


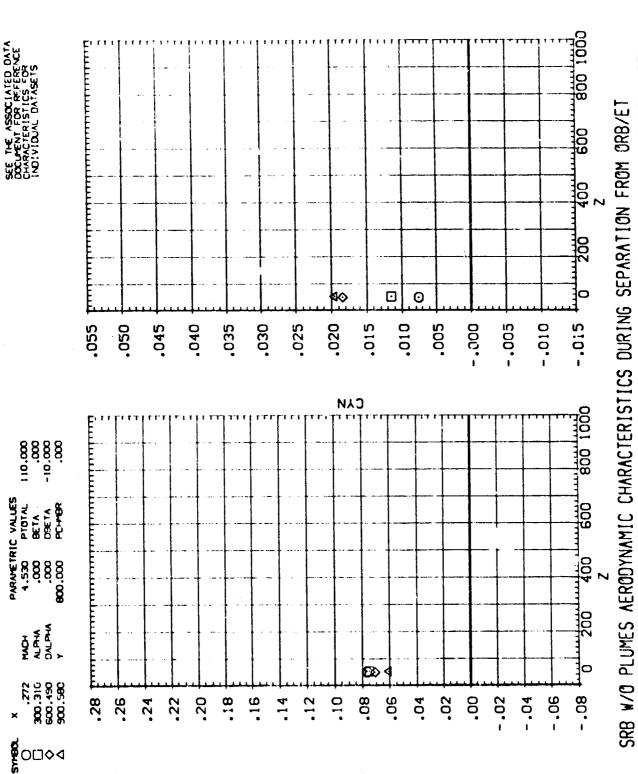
PAGE SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



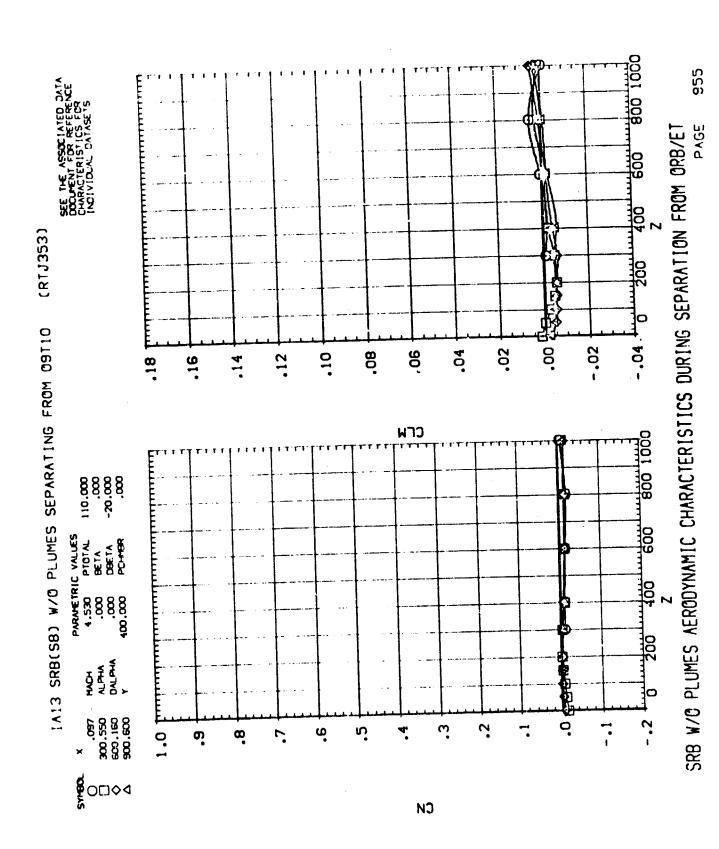


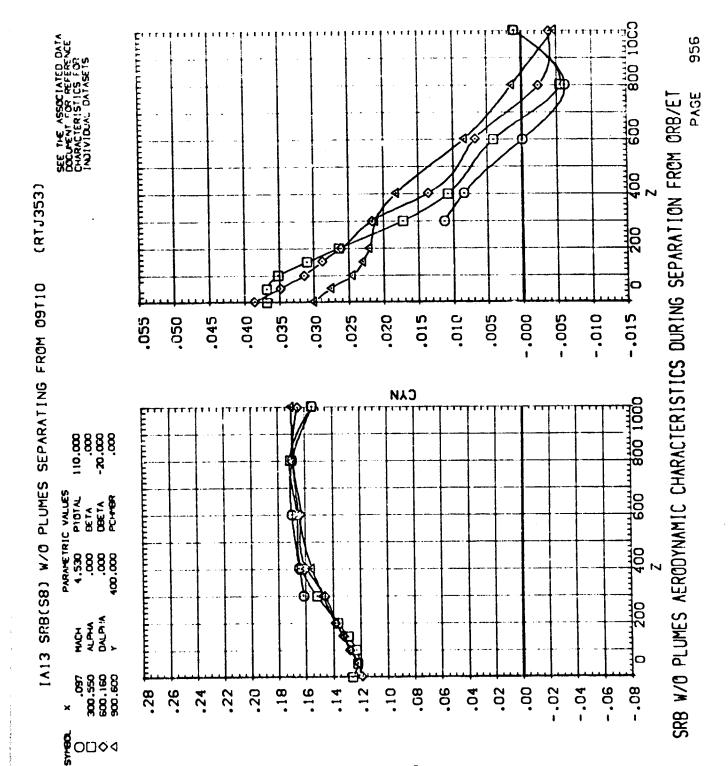
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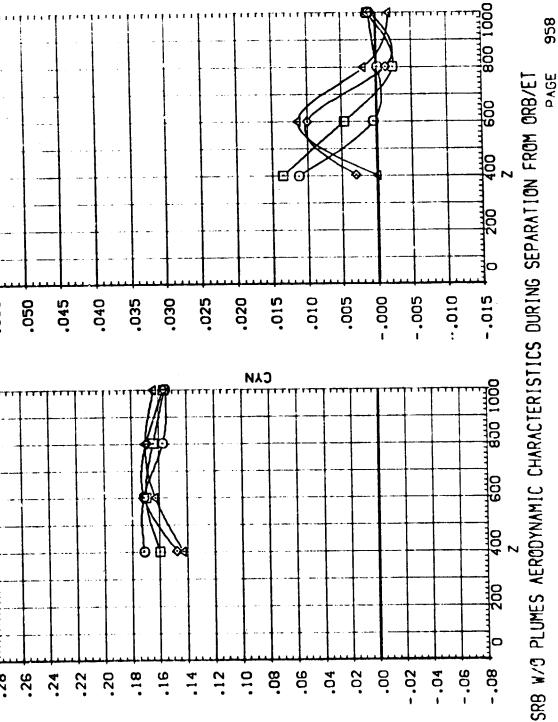
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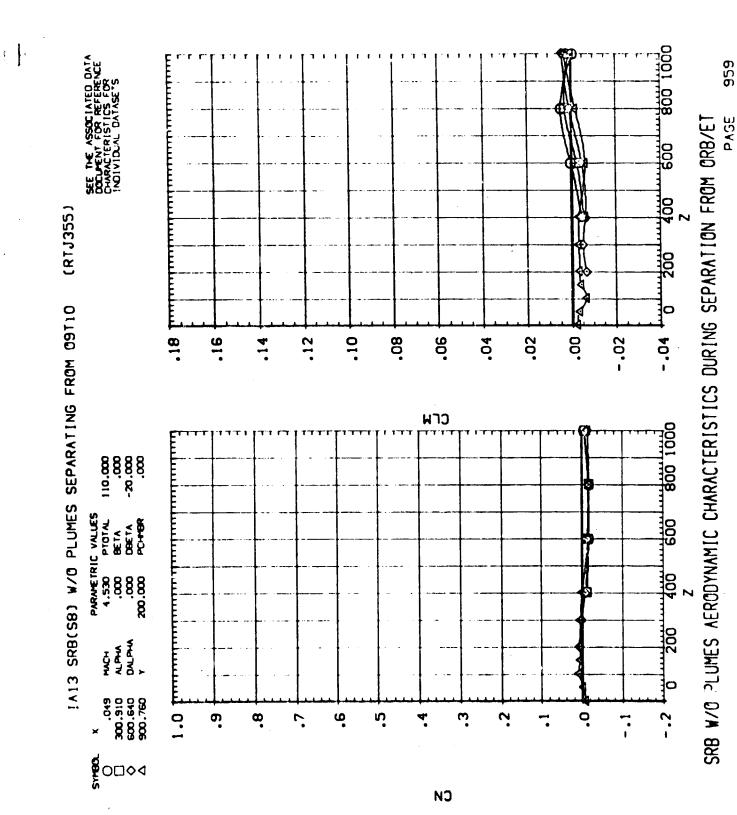
SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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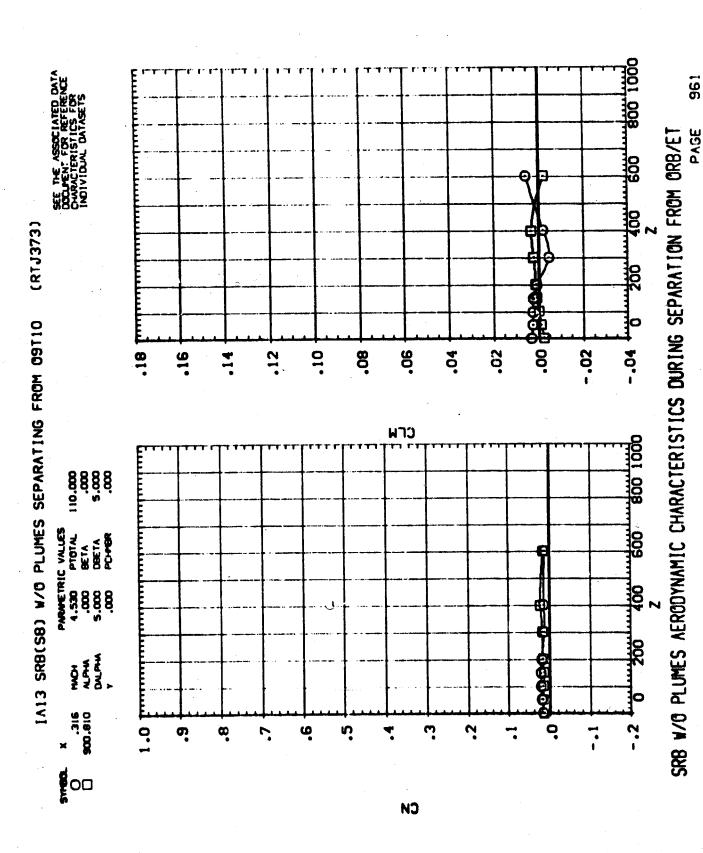
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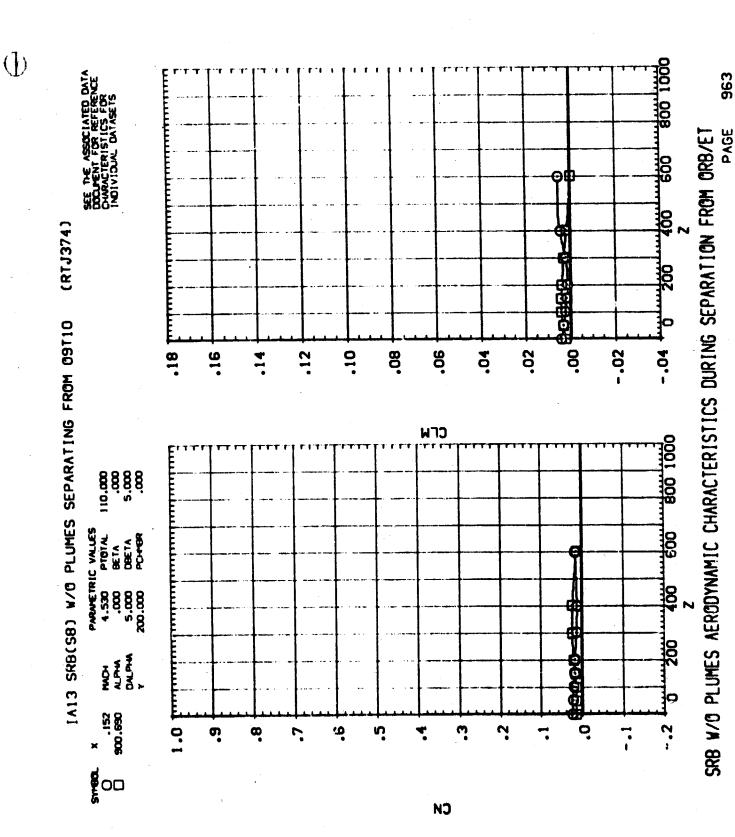


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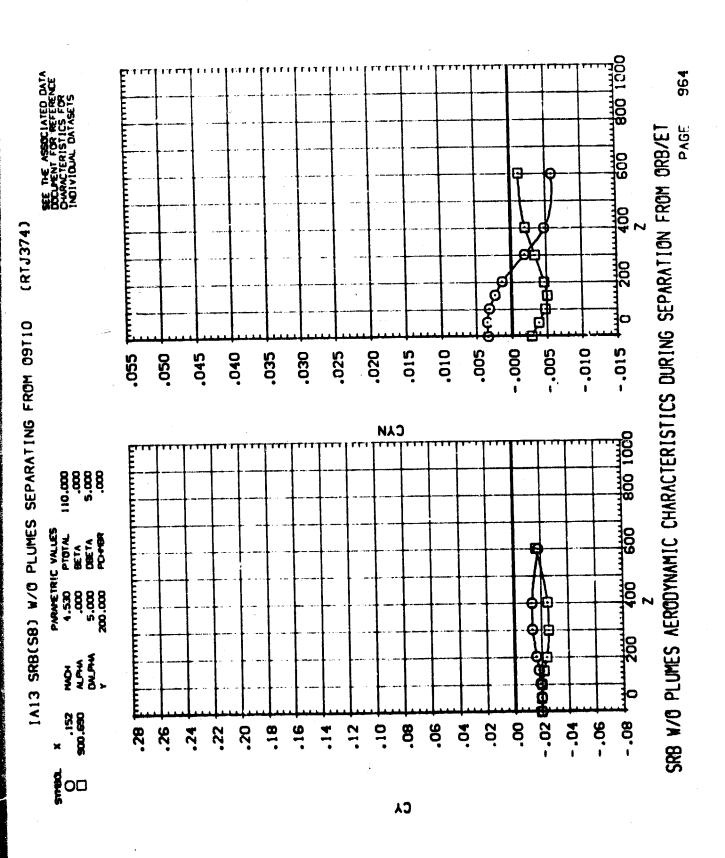
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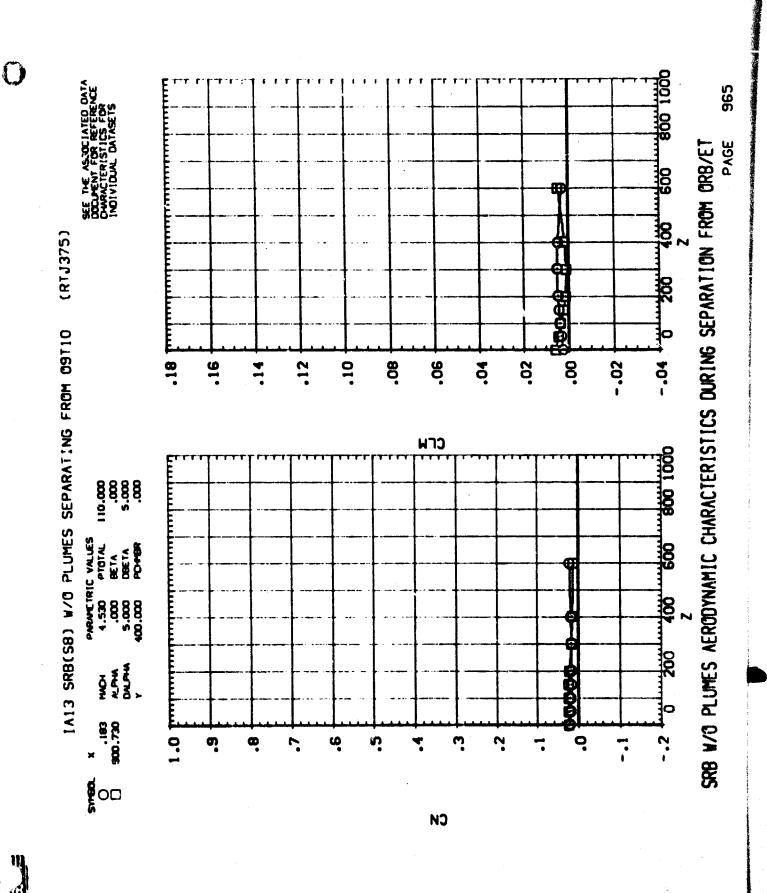
SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

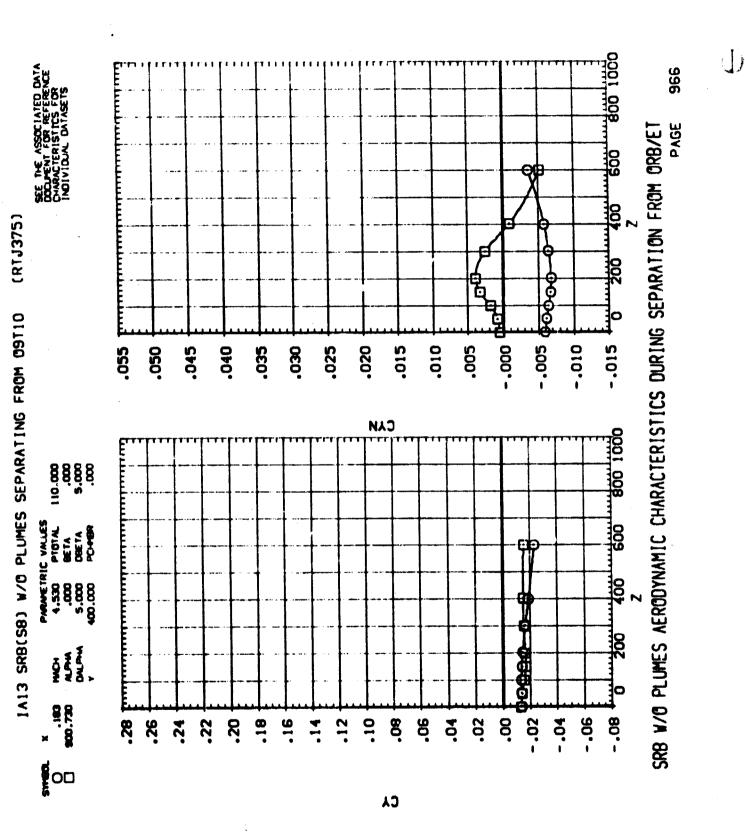
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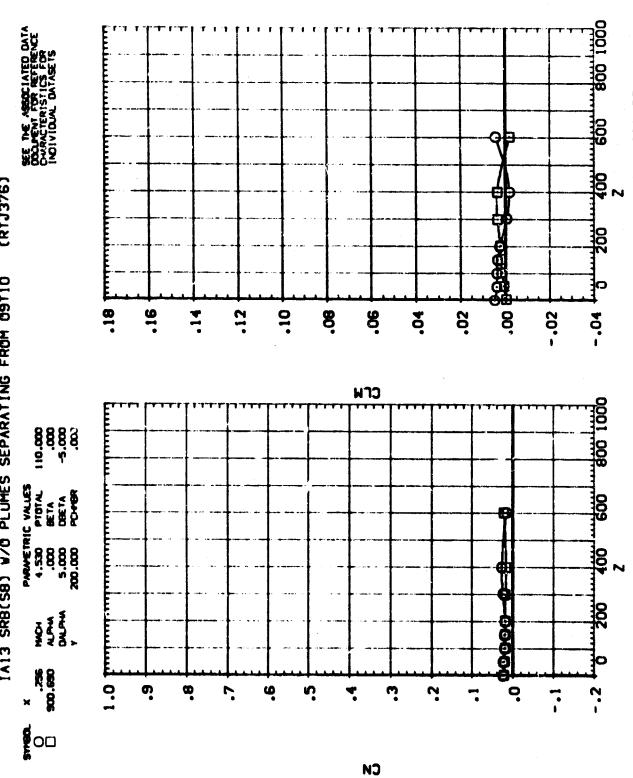






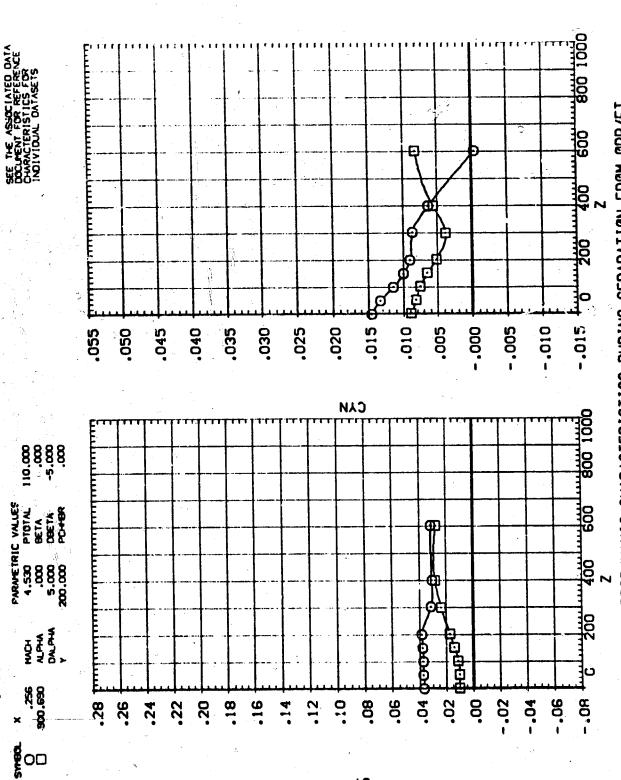
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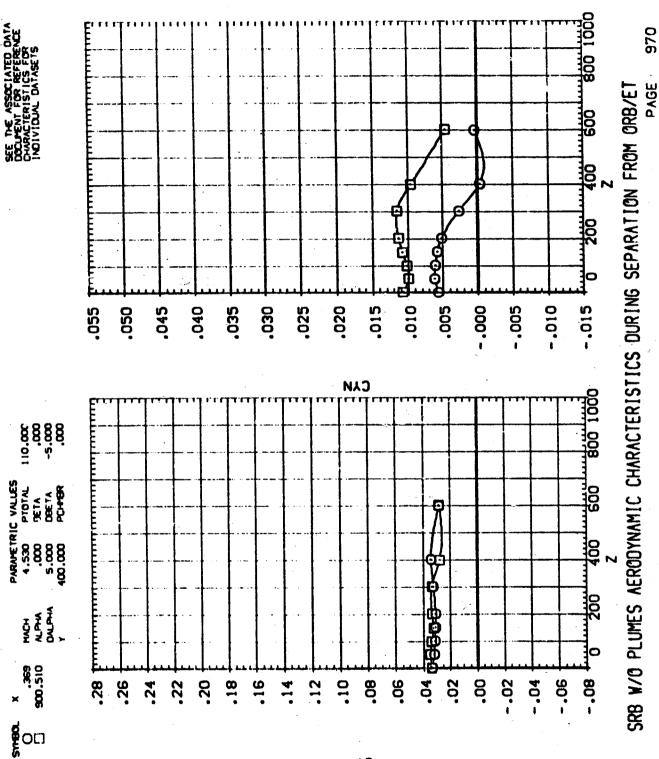


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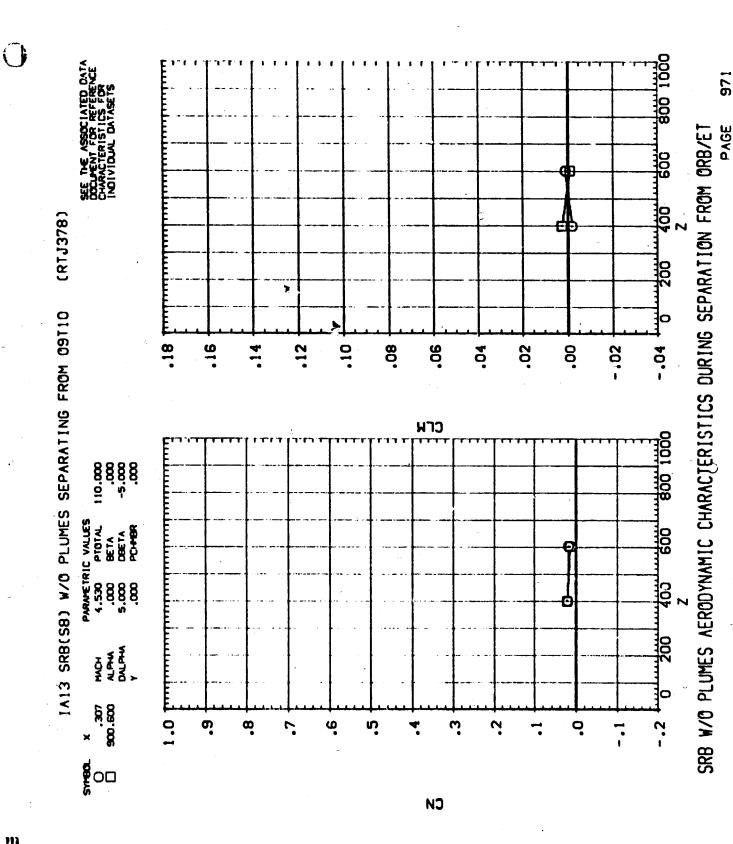
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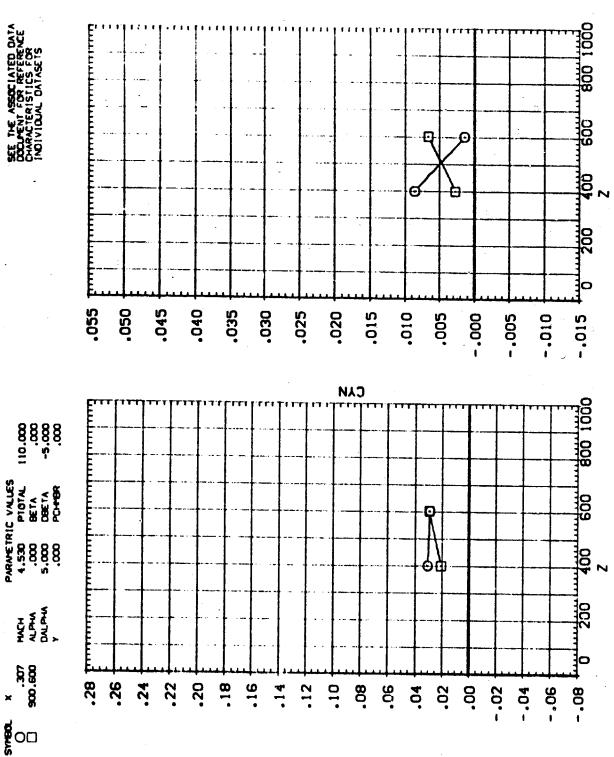


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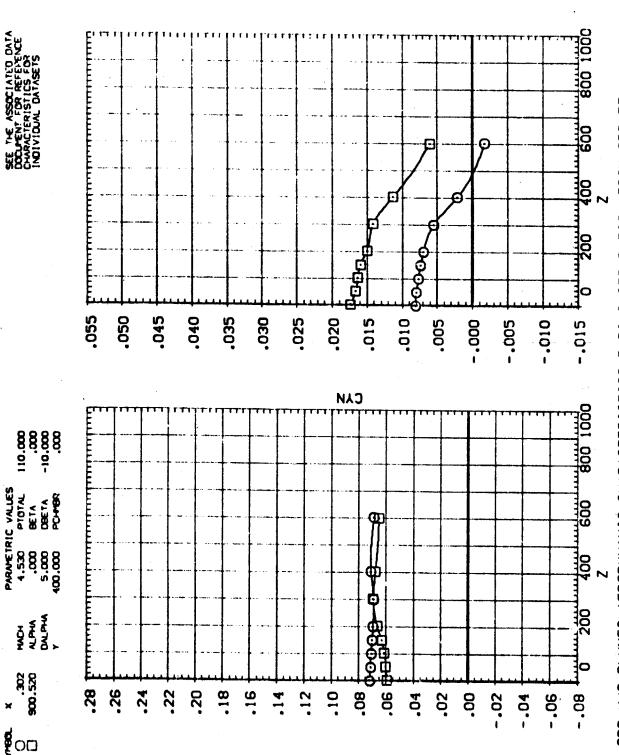
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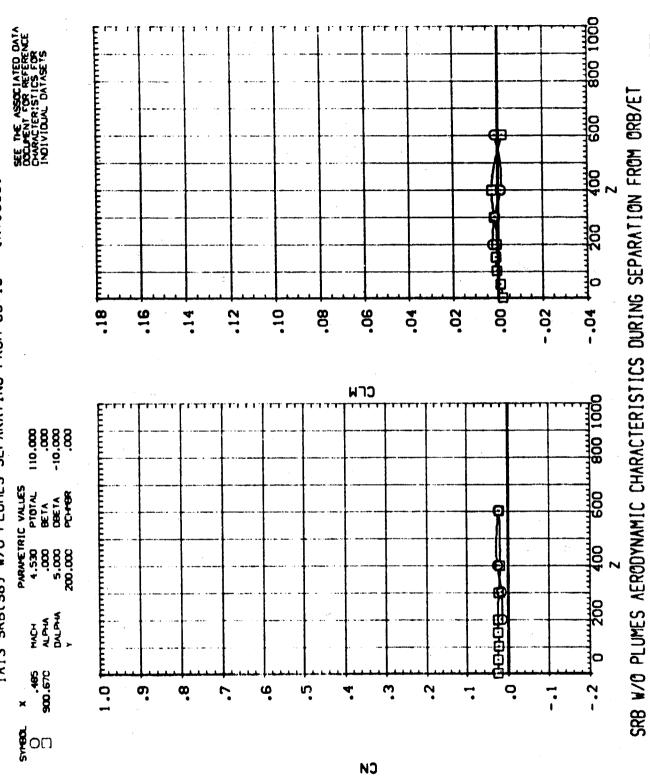
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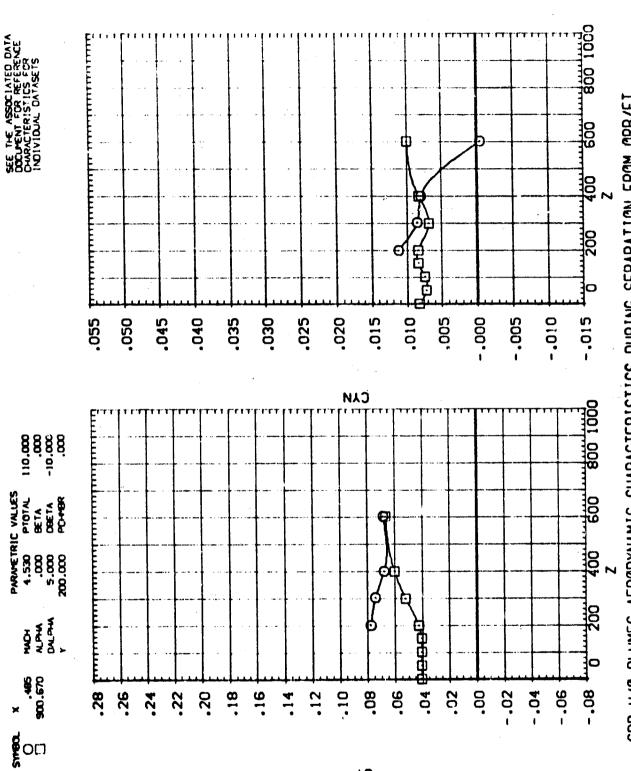
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(RTJ380) IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110



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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTEPISTICS FOR INDIVIDUAL DATASETS 909 **2**2 (RTJ381) 200 IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09710 -.04 18 .02 8 -.02 9: 90. 90. .04 .16 .14 .12 CLM 800 1000 110.000 1.00.01-10.000.01-1.00.000. PARAFETRIC VALUES
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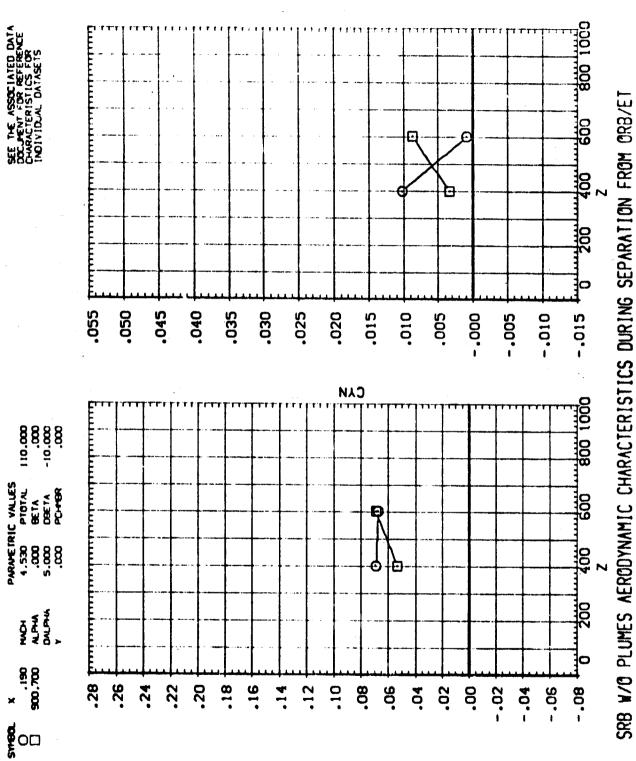
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ381)

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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 600 400 Z (RTJ382) 200 IA13 SRB(SB) W/O PLUMES SEPARATING FROM 05:10 P 8 -.04 .16 .12 .10 8 8 90. .02 .14 9. -.02 כרש 800 1000 20.00 00.05 00.00 00.00 PARAMETRIC VALUES 4.530 PTGTAL .000 BETA 5.000 DBETA 200.000 PC-498 909 Đ 6 8 7 8 A DE PAR фофоч Ю 85.37 37.37 0. Ģ ທຸ -.2 o. œ 1. 'n ? o -: NOU NOU CM

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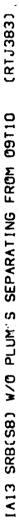
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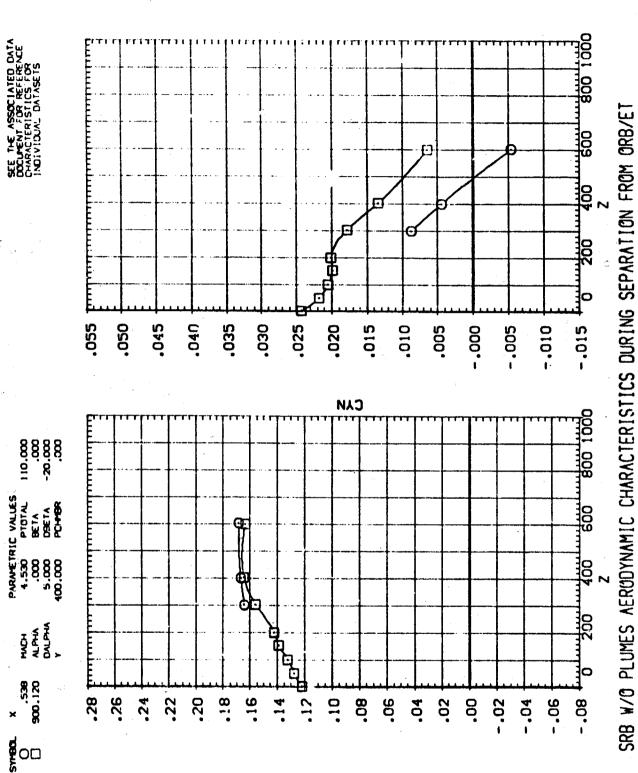
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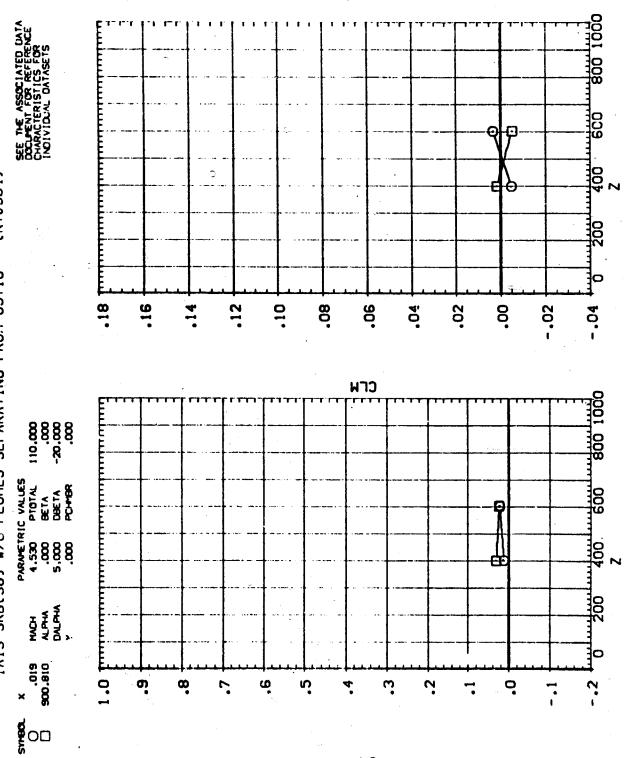
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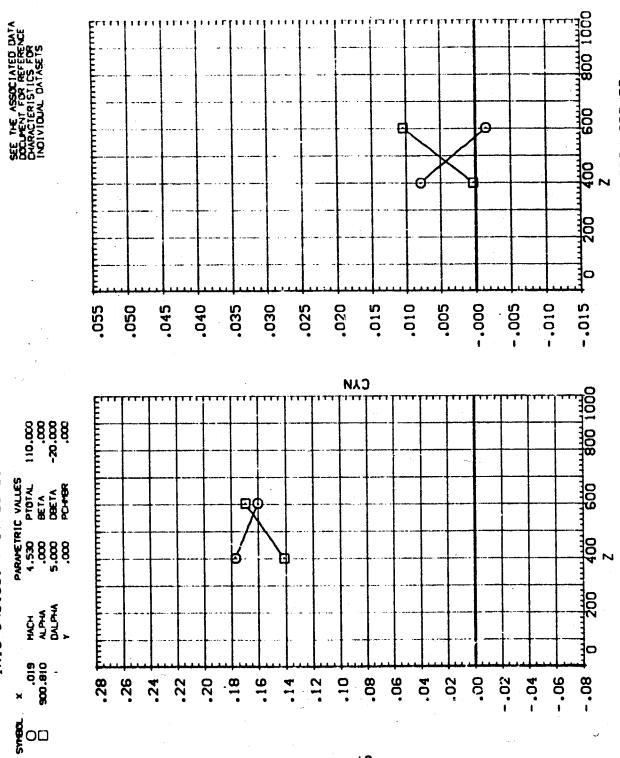


PAGE SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ384)



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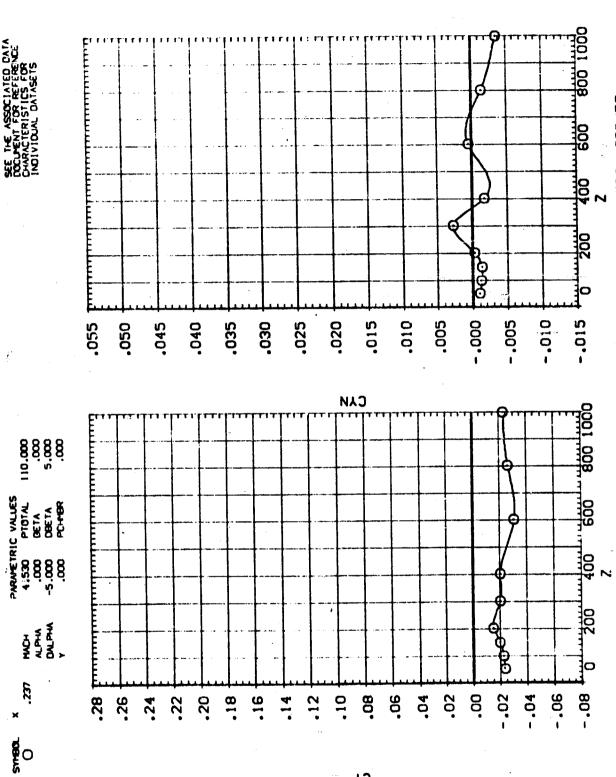
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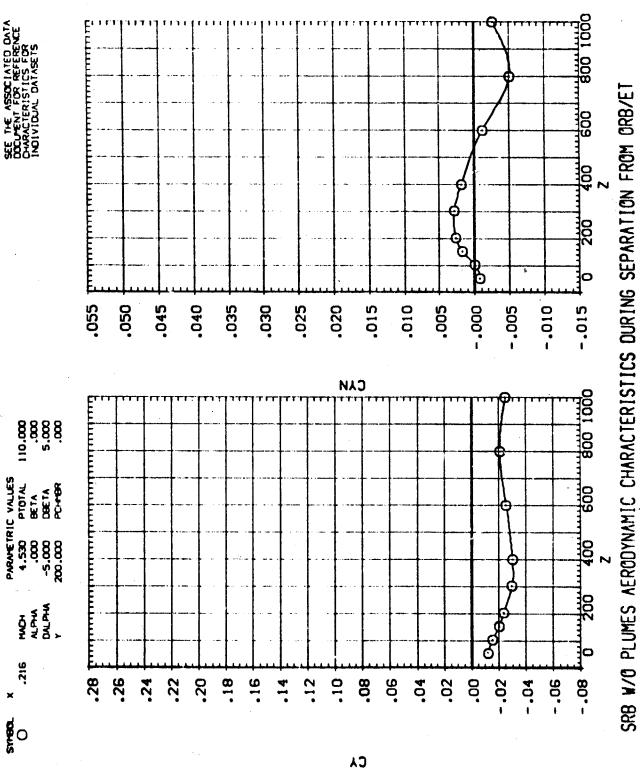
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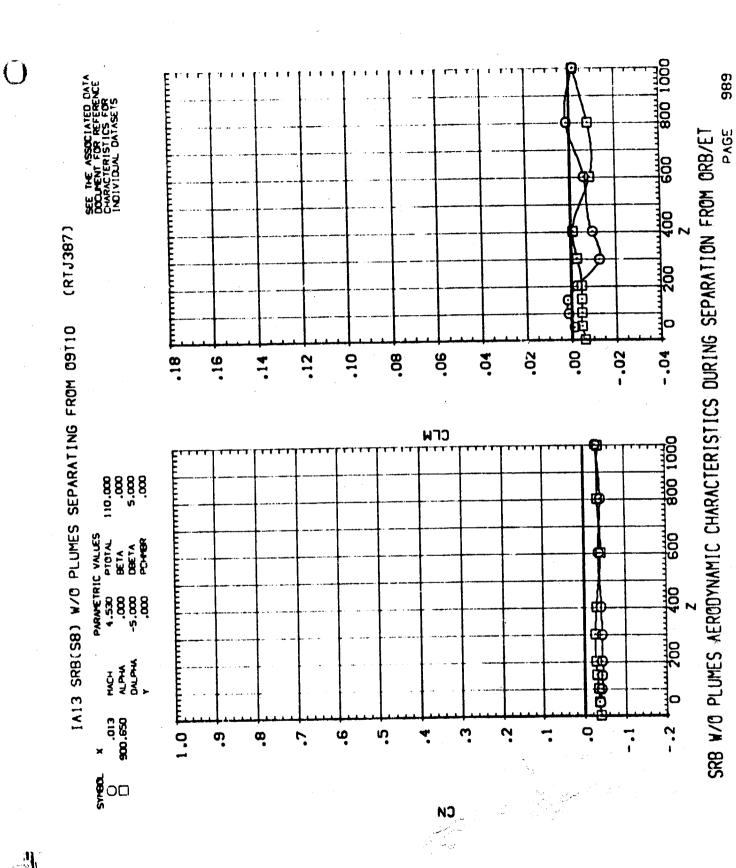
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(RTJ386) SEPARATING FROM 09110 IA13 SRB(SB) W/O PLUMES

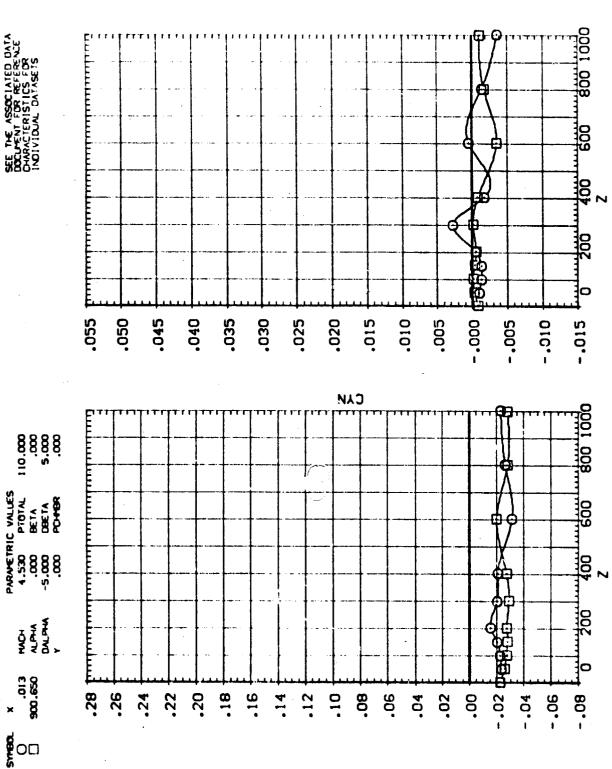


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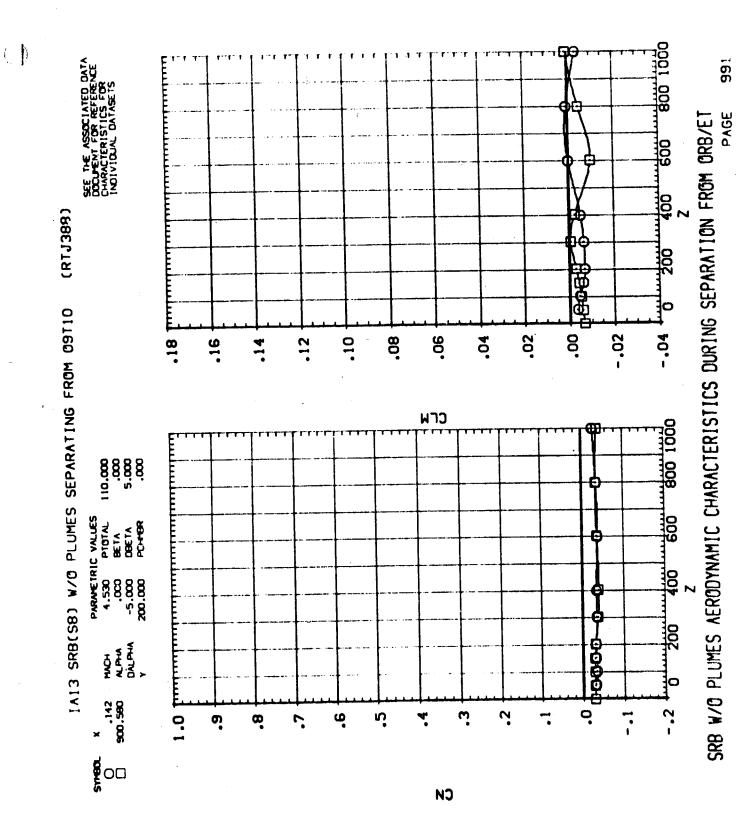


IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ387)

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SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



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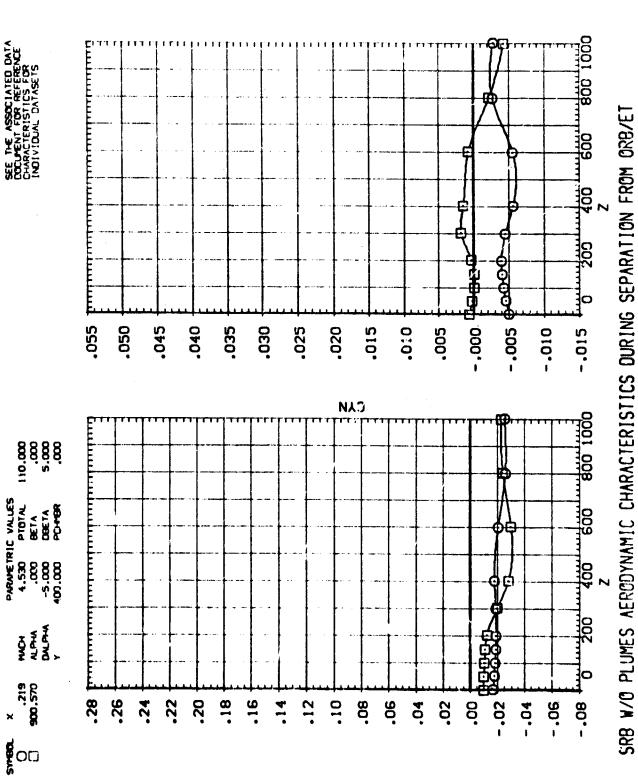
IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09710

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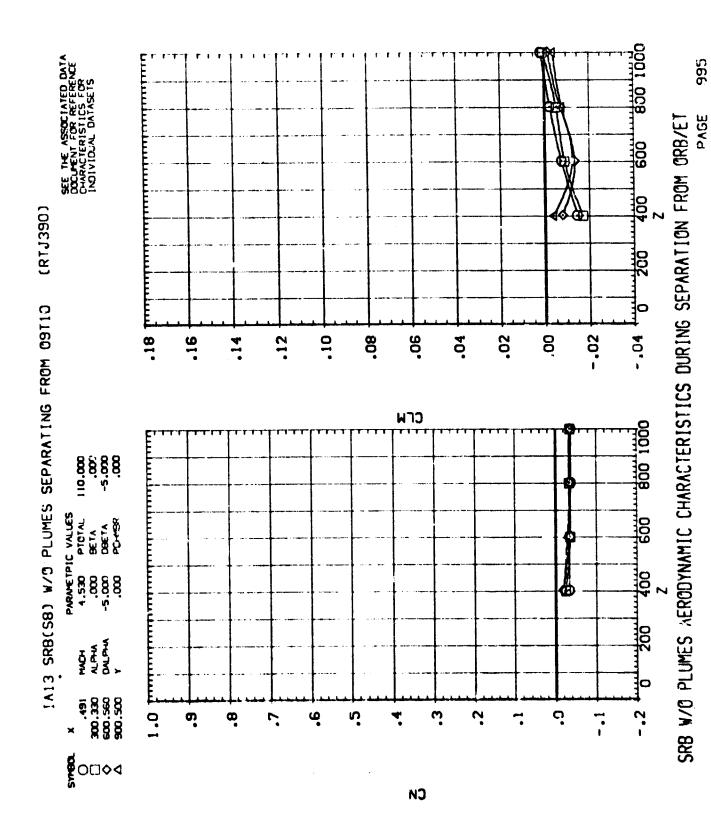
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ389)

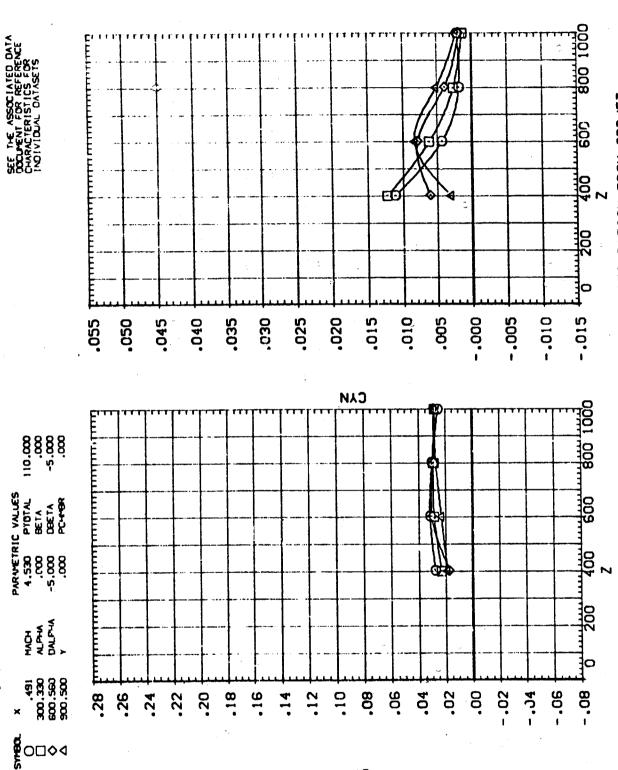


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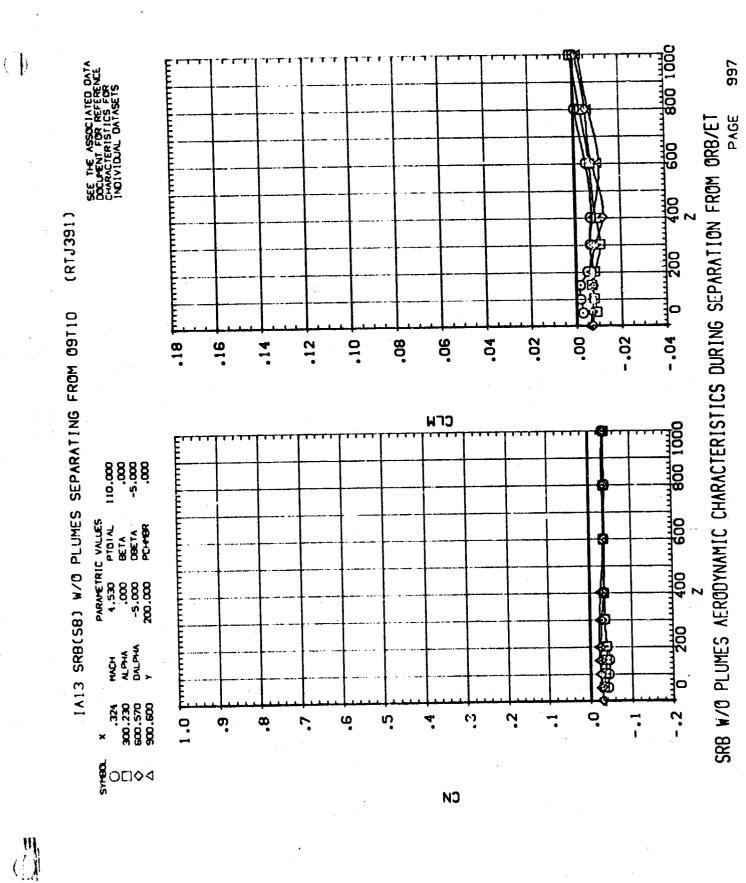


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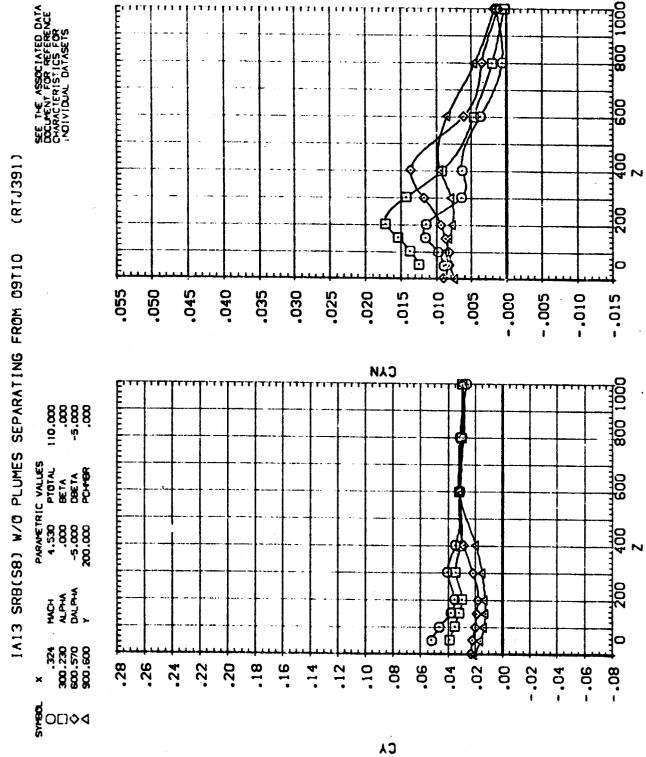
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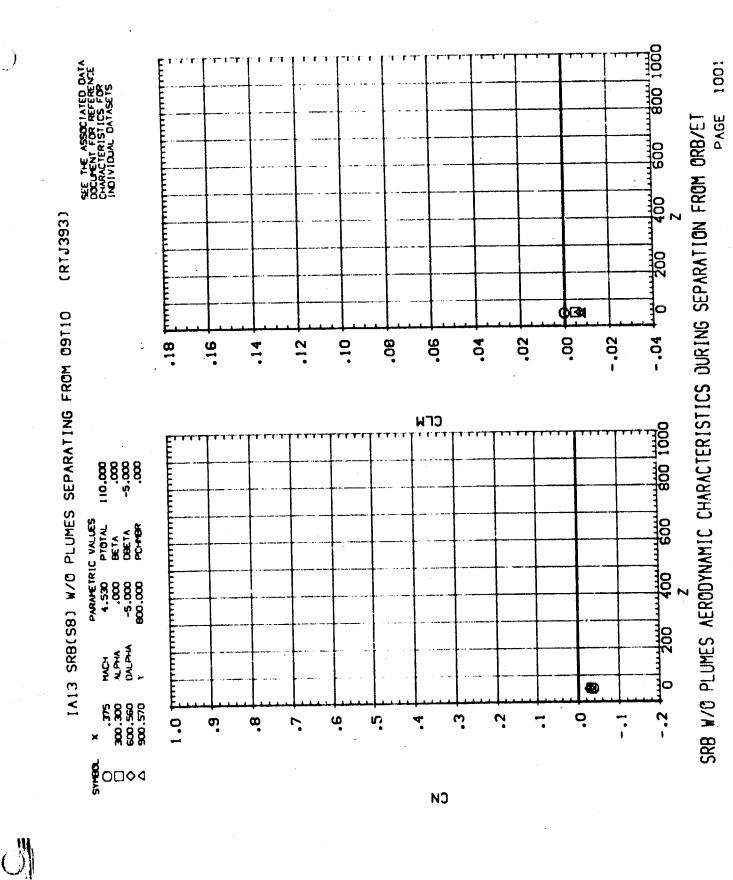
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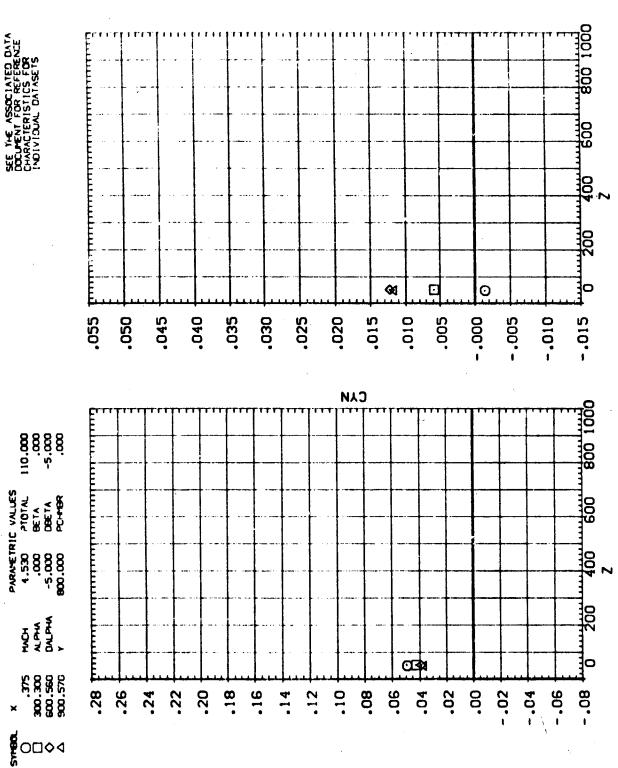
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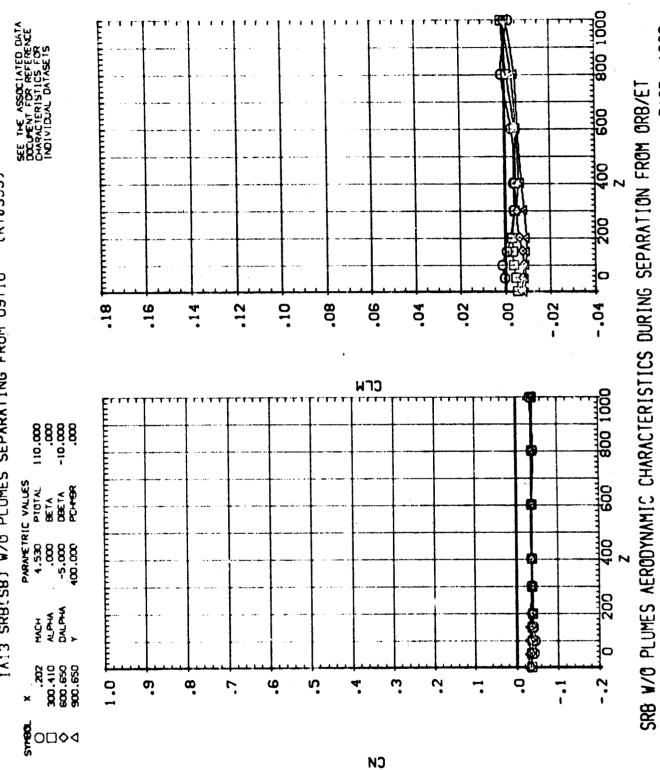


PAGE SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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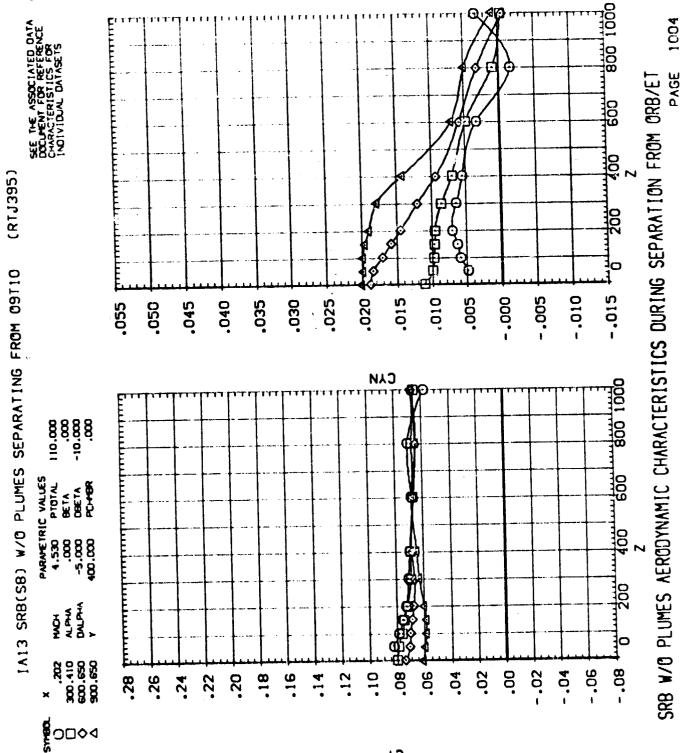
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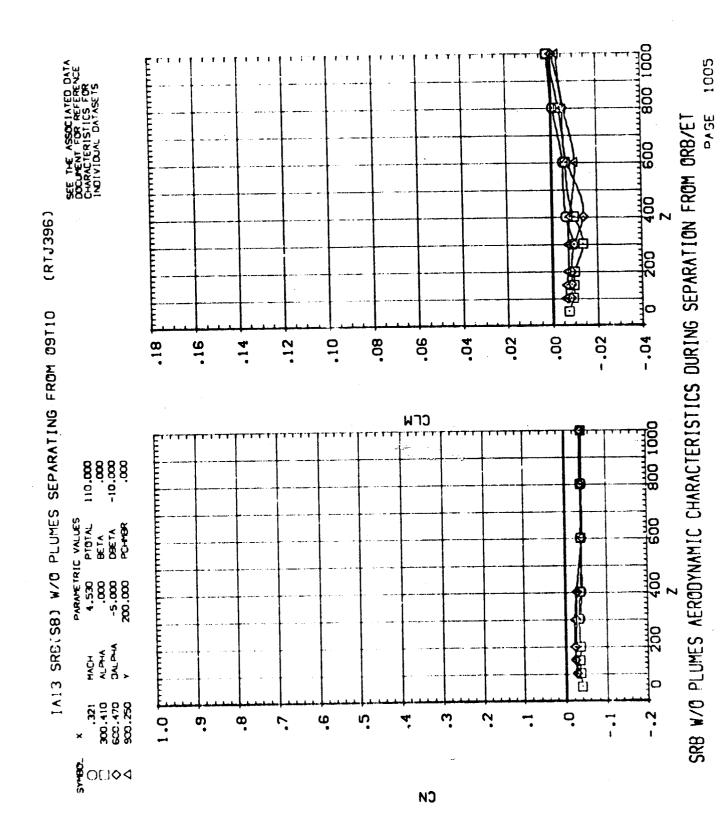


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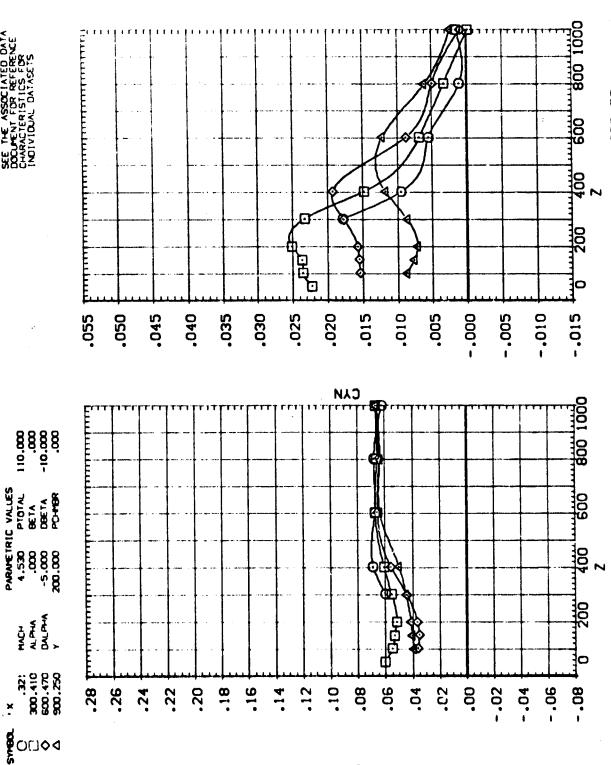
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ396)



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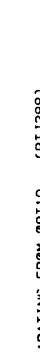
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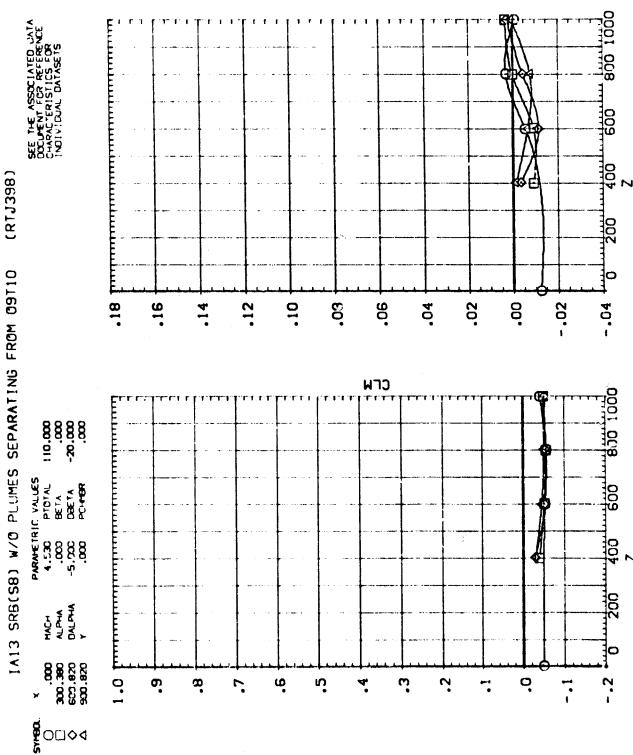
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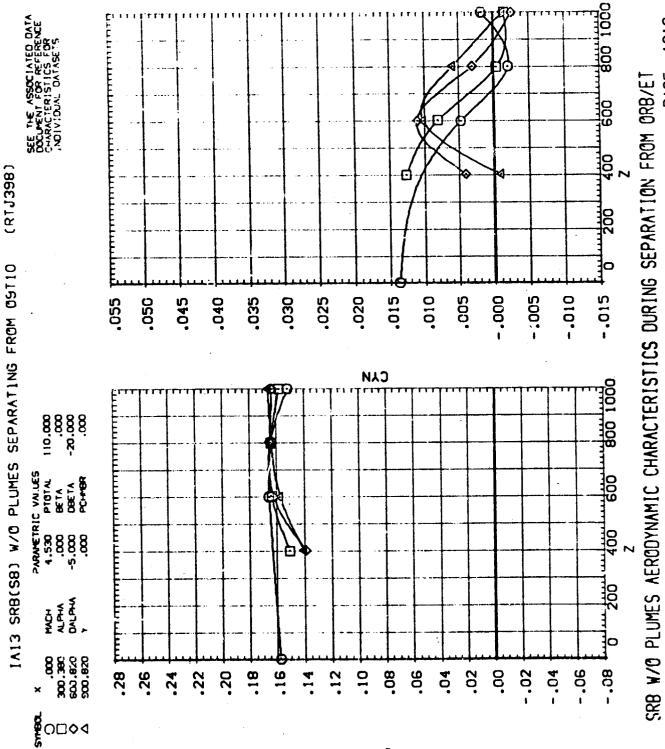
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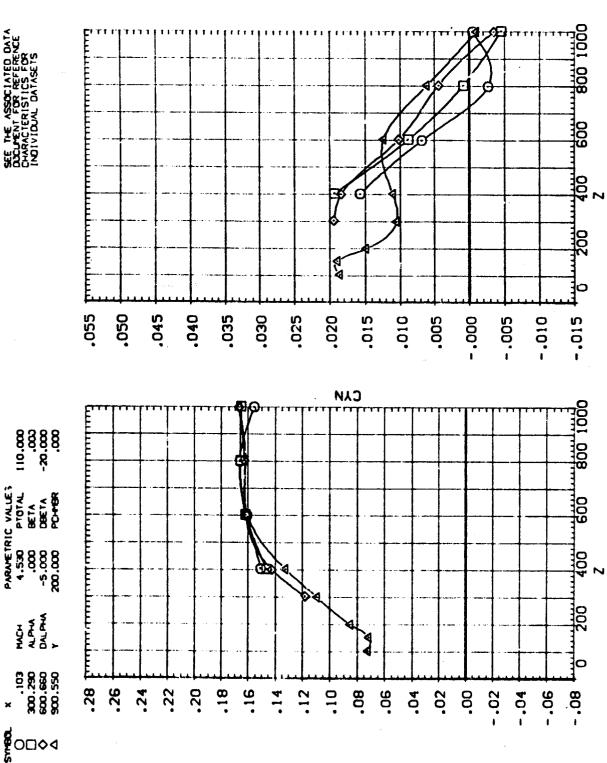
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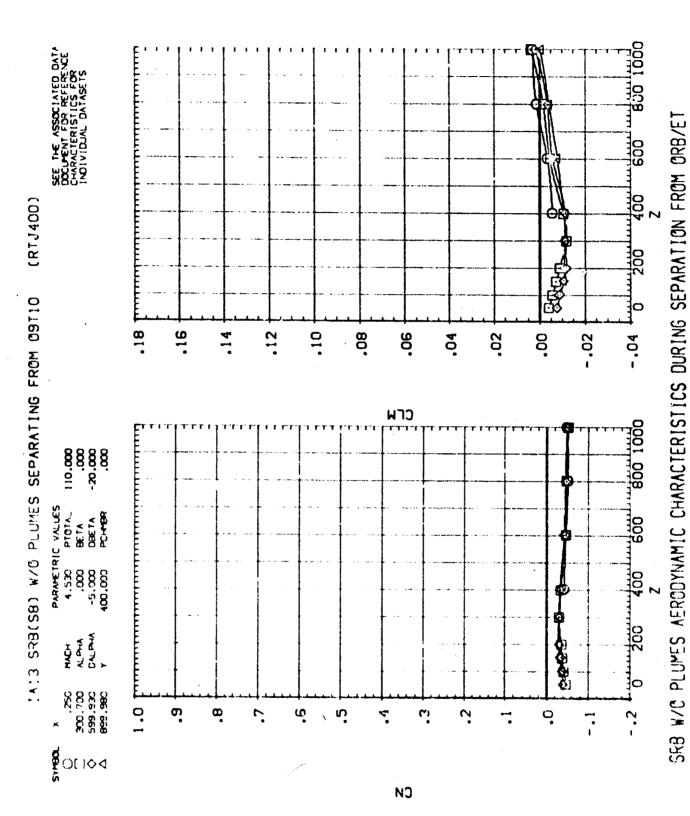
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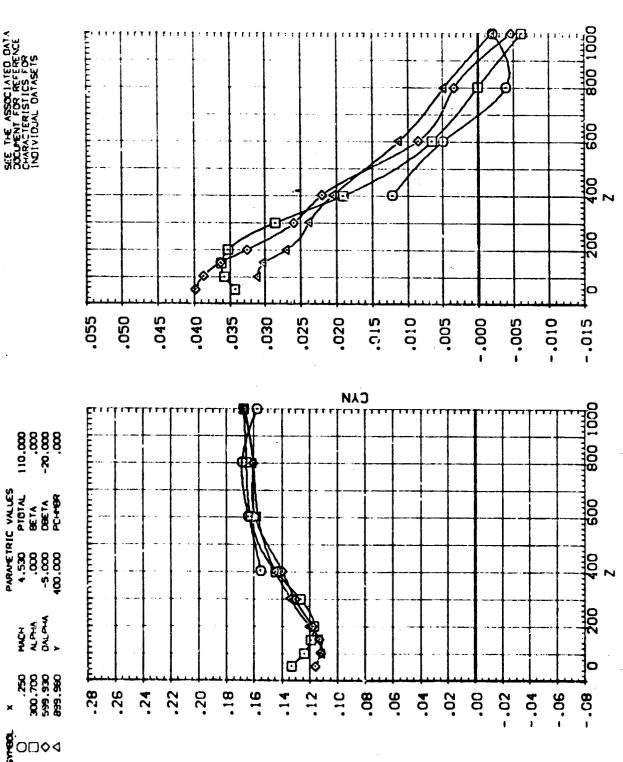
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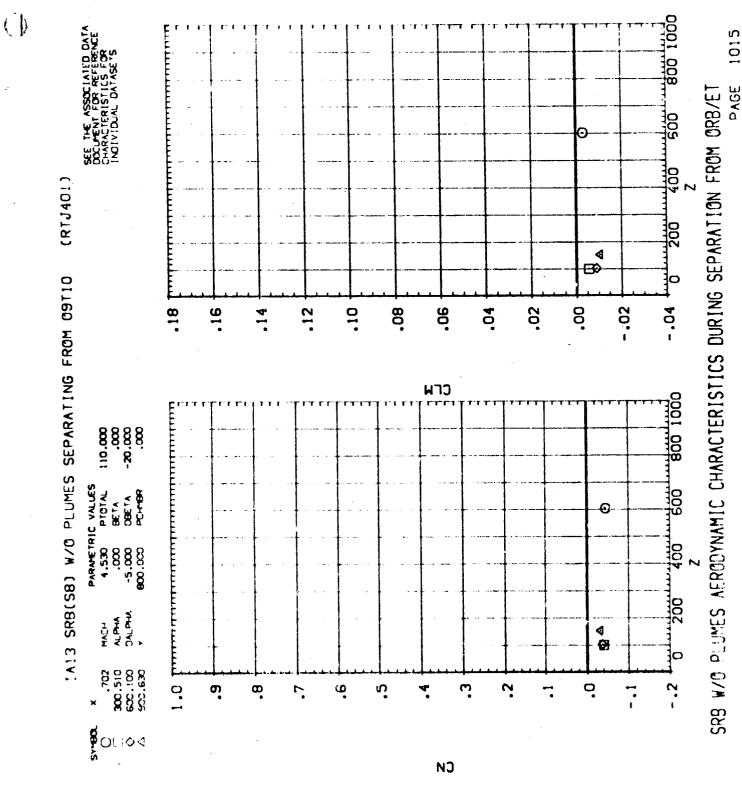
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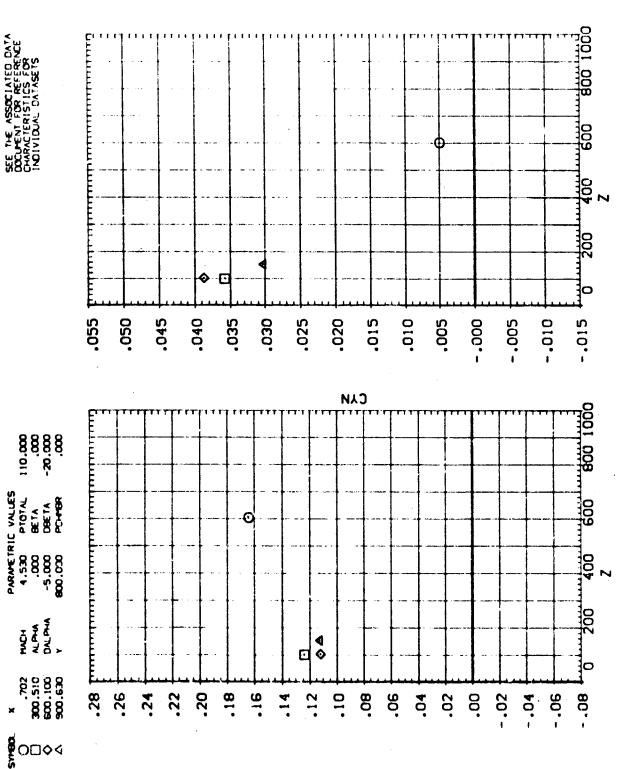


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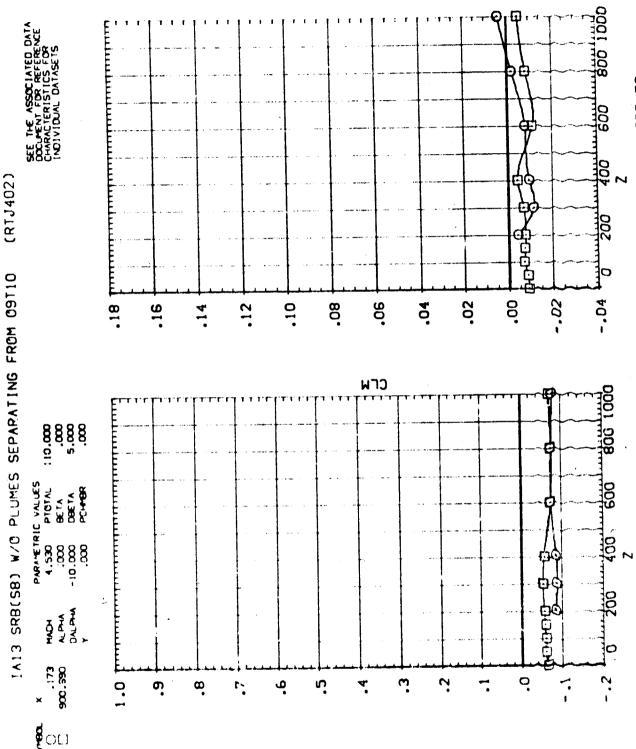
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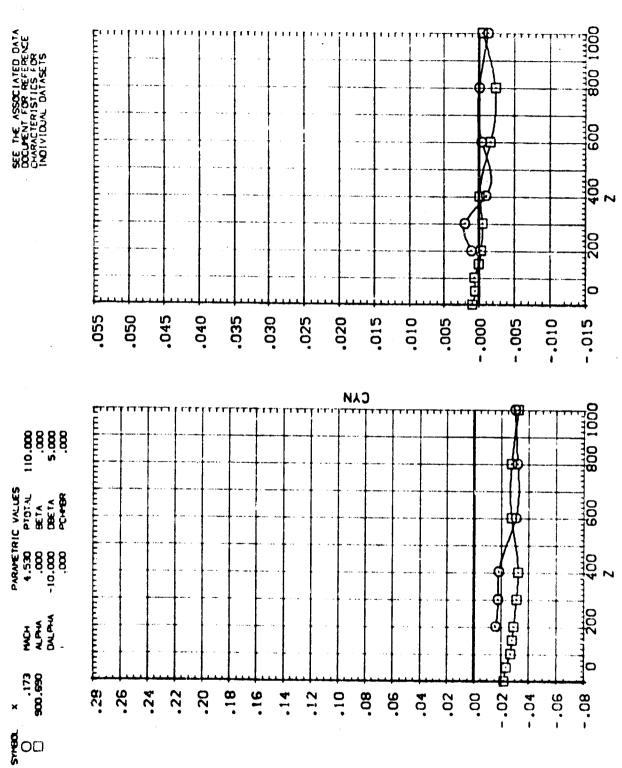
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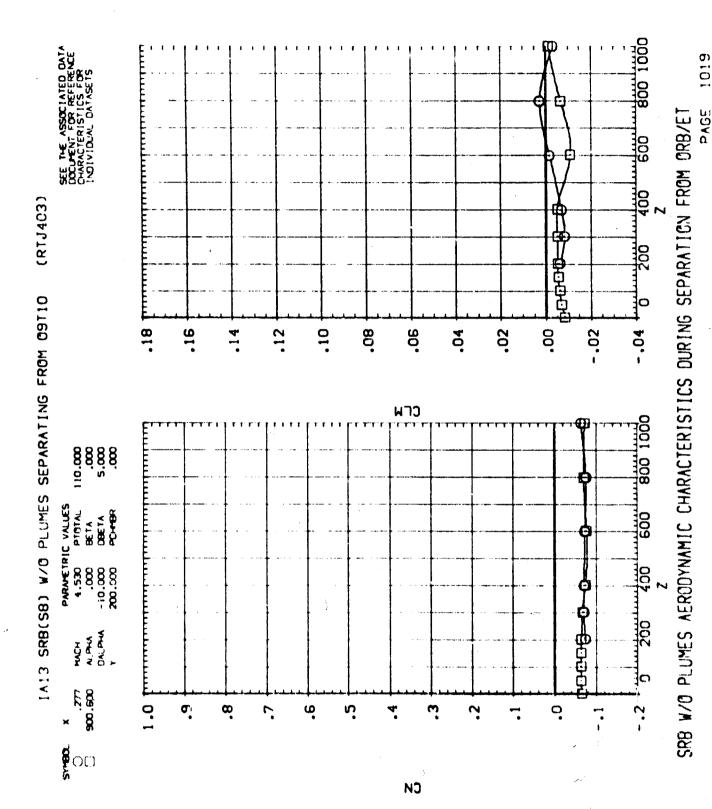
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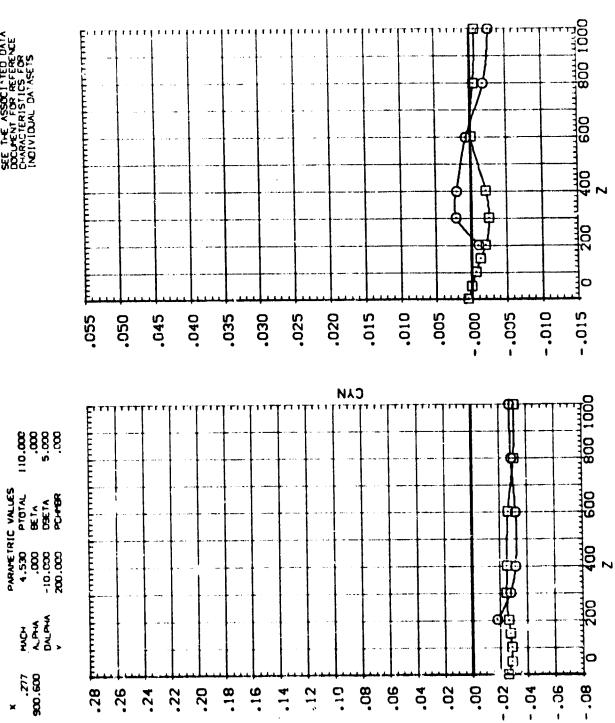




IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110

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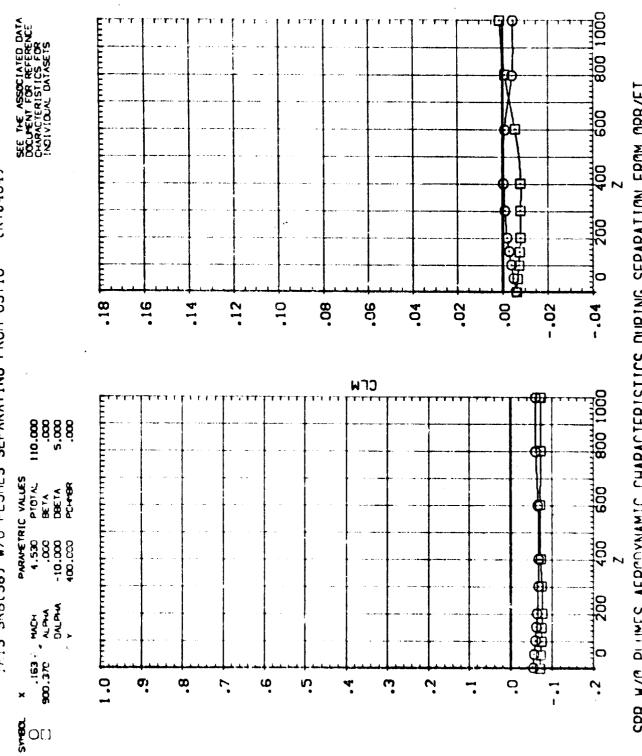
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1020 PAGE SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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(RTJ404) IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110



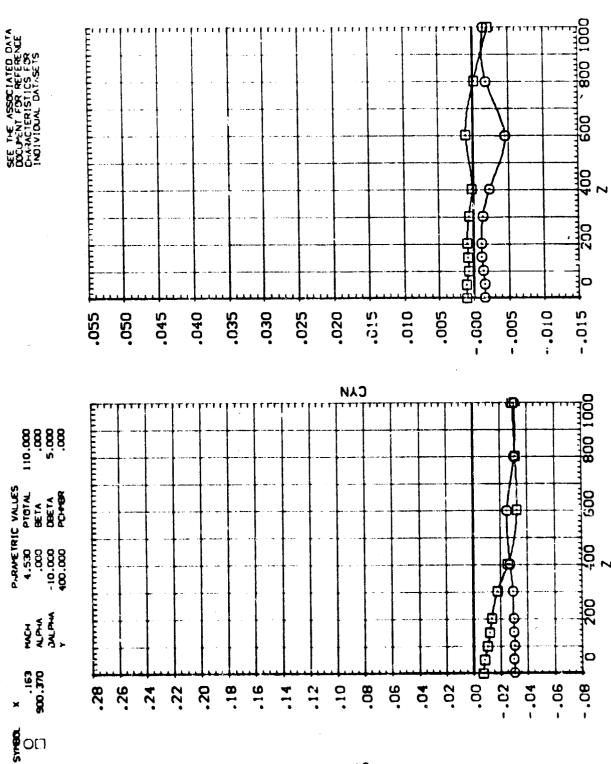
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(RTJ404) SEPARATING FROM 09110 IA13 SRB(SB) W/0 PLUMES

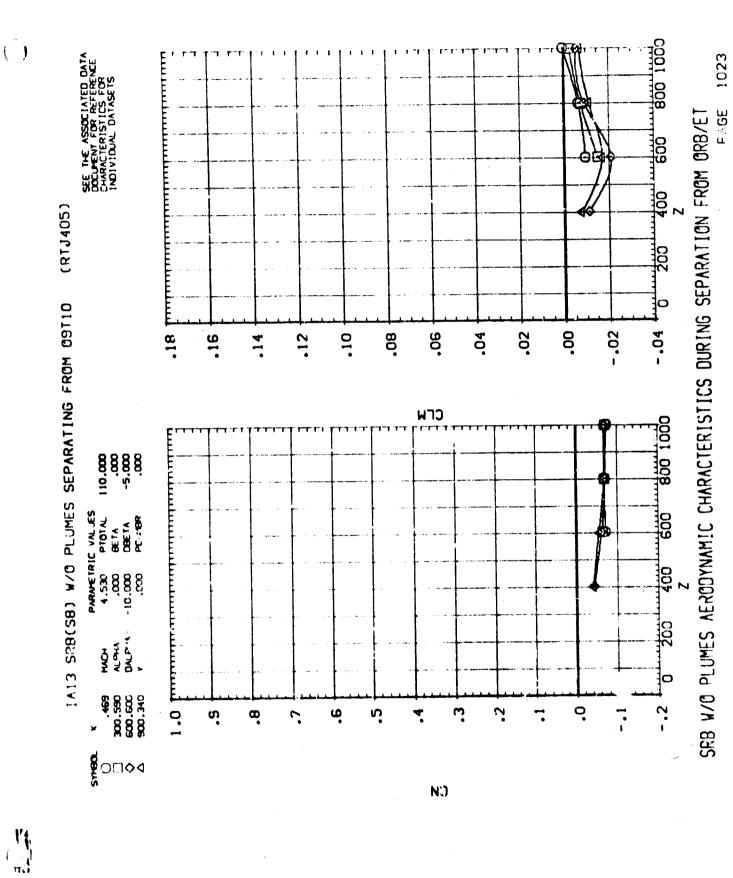


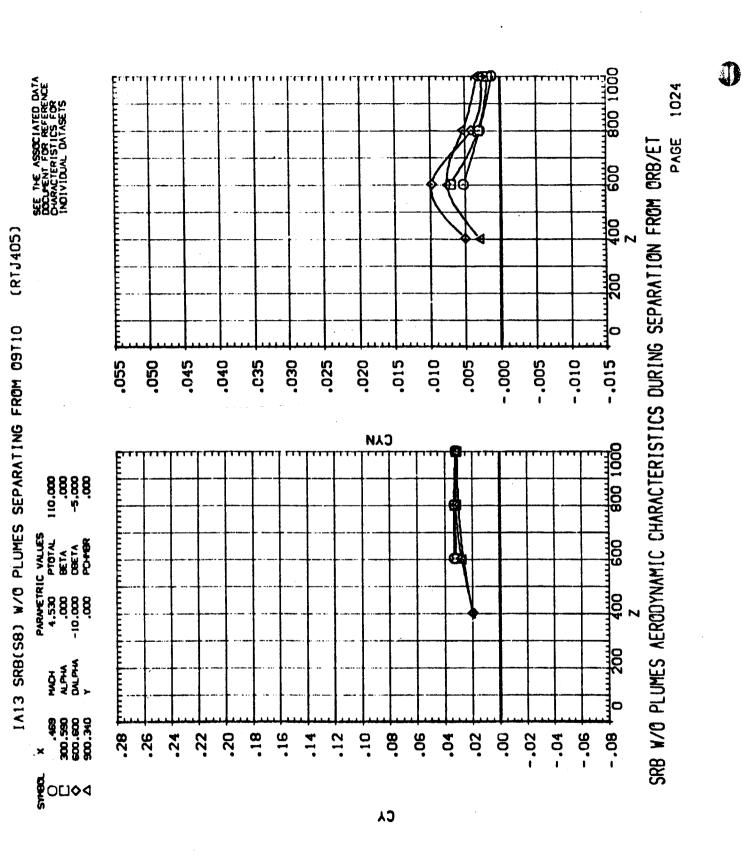
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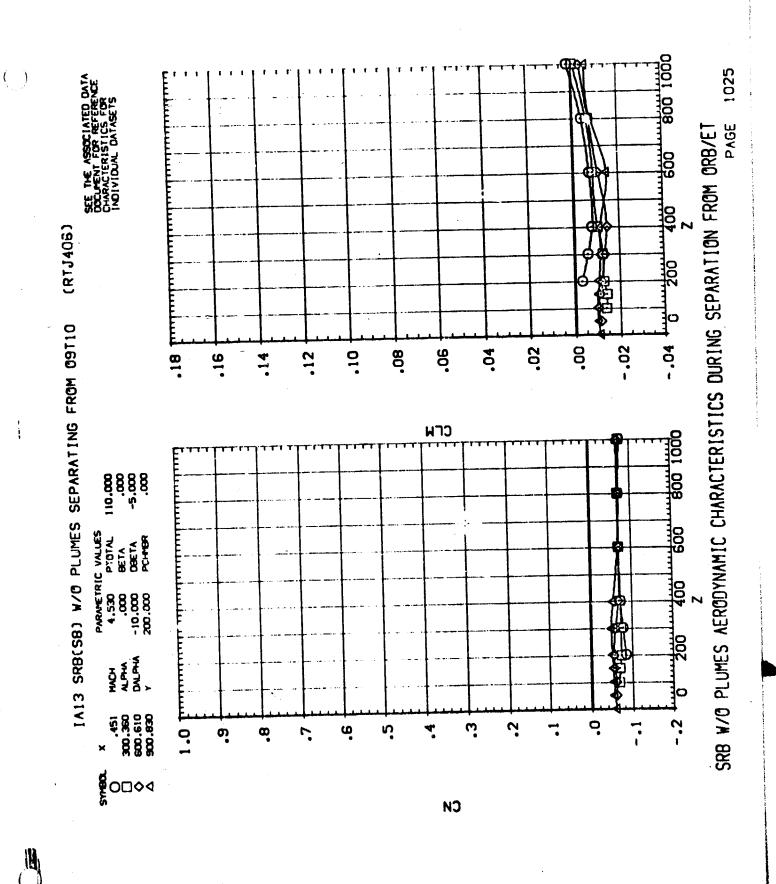
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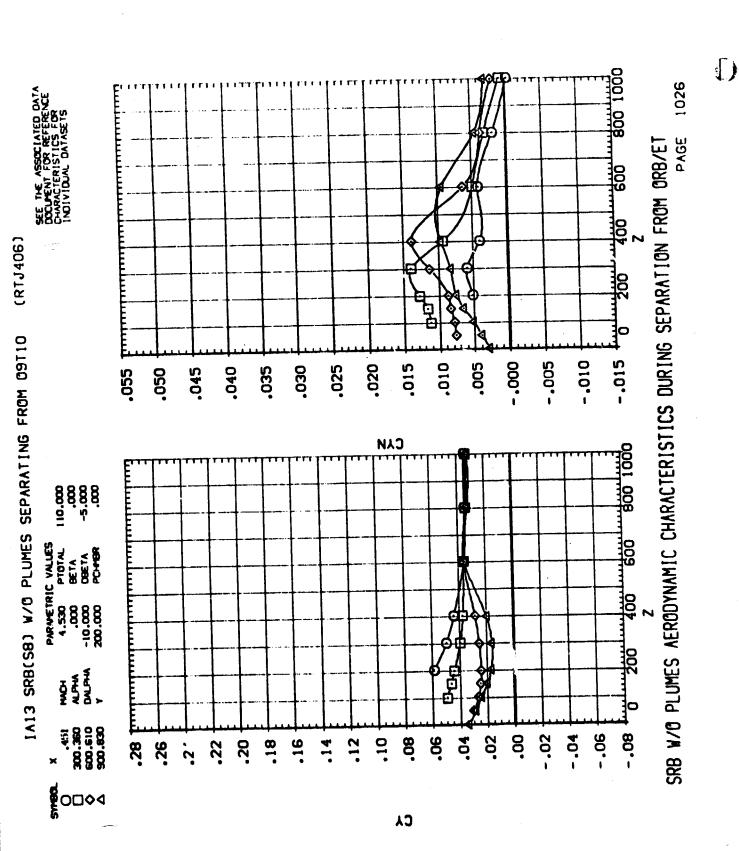
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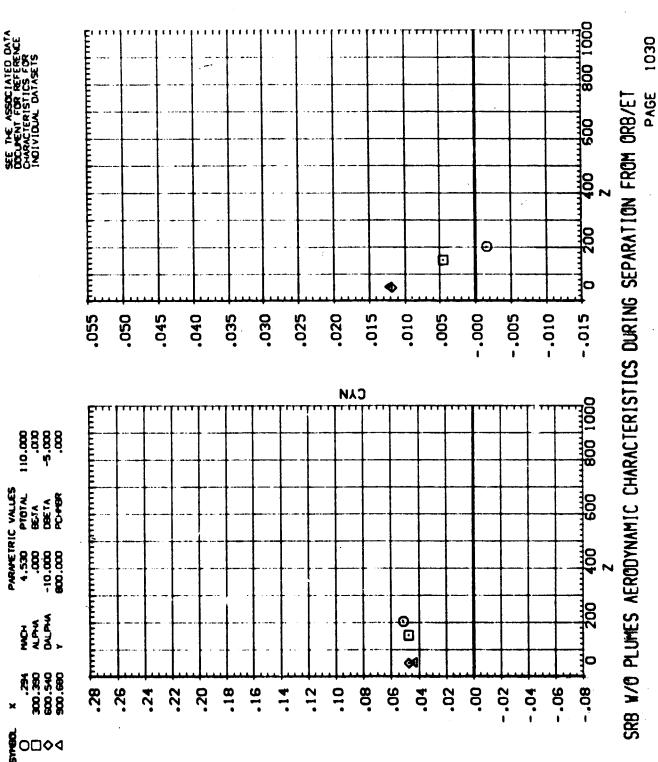
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IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 (RTJ408)



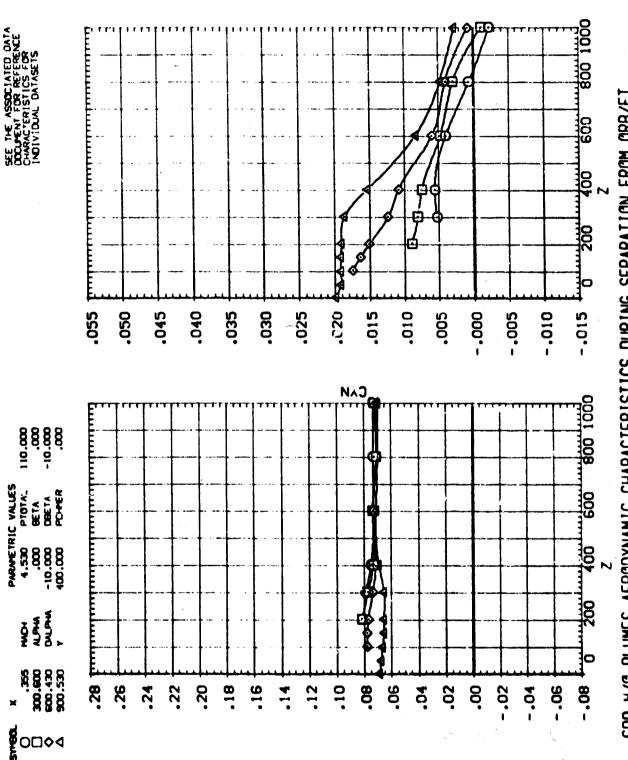
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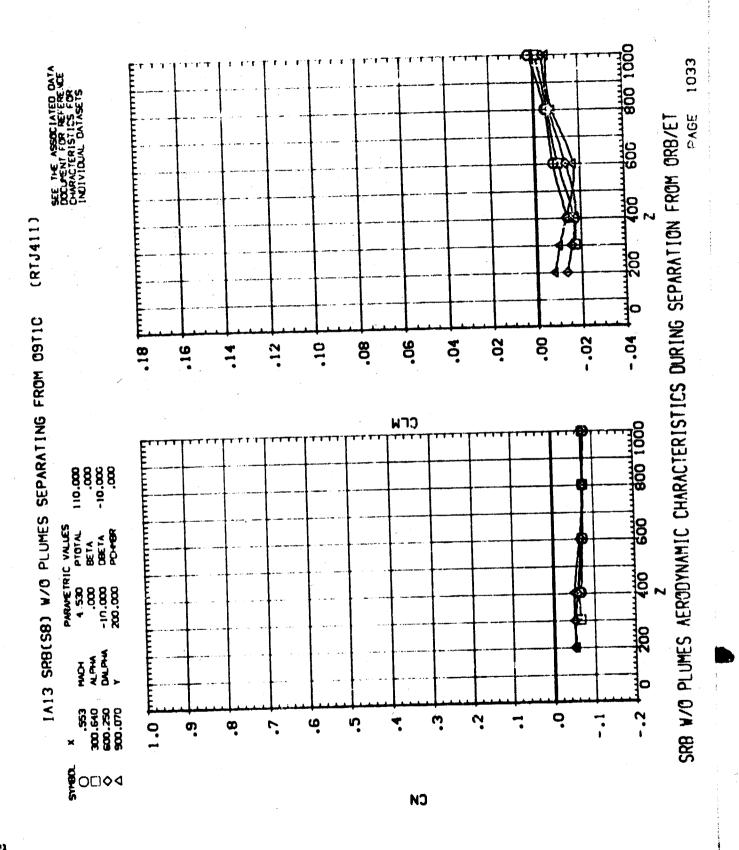




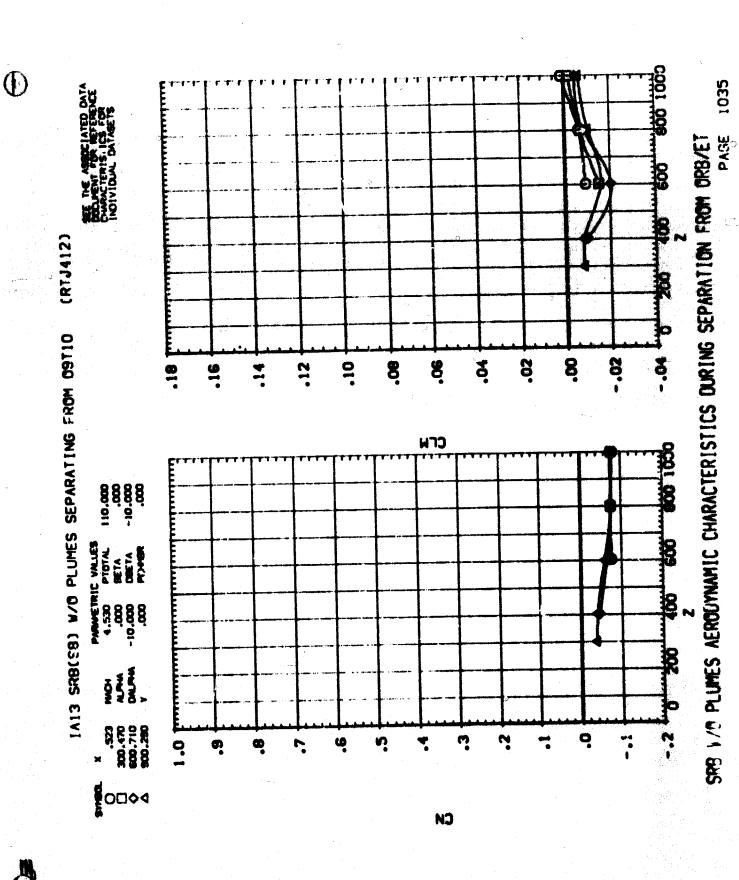
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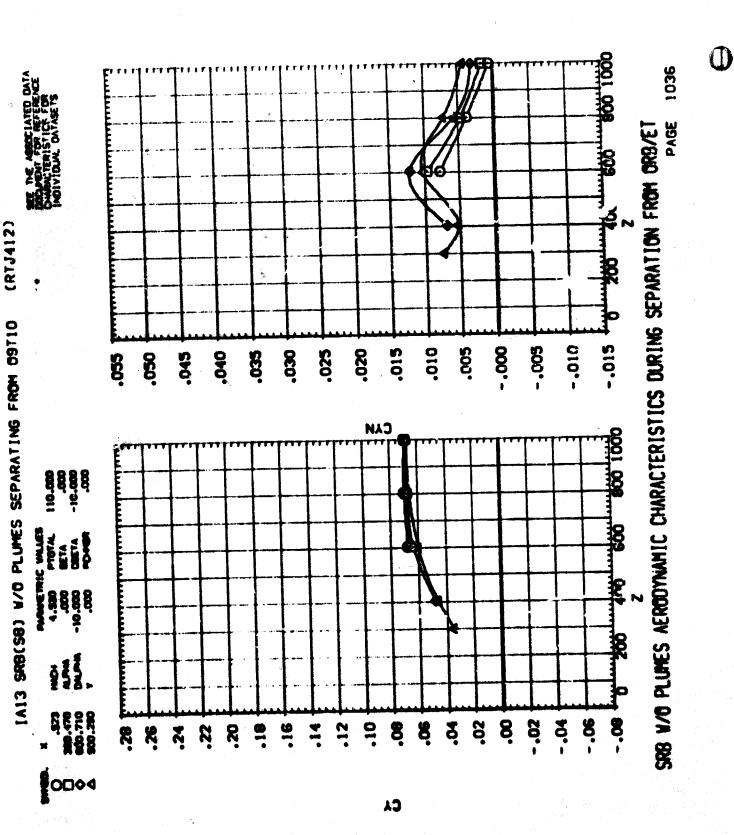
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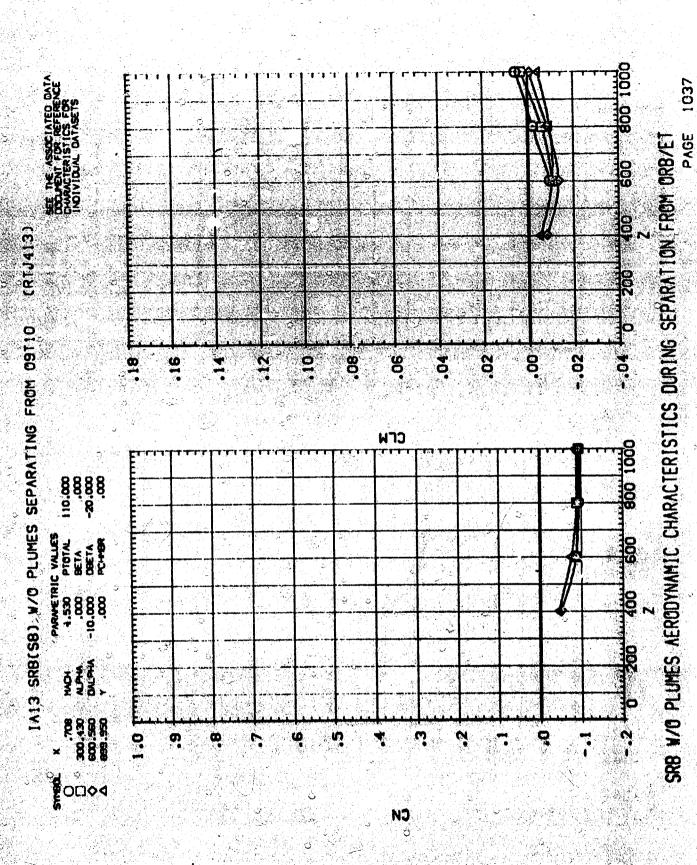
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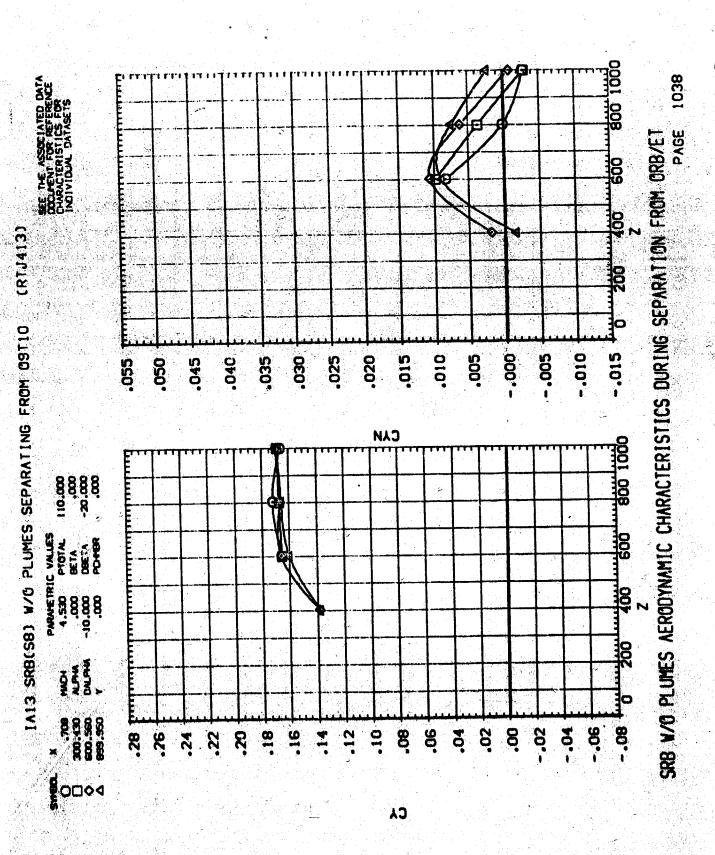


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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 100 8~ (RTJ415) 200 IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 -.04 8 90. .02 -.02 .18 .16 .12 2 80. 9 .14 CLM 800 1000 110,000 .000,05-000,000, PARMETRIC VALLES 4,530 PTOTAL .000 BETA -10.000 DBETA 400.000 PC-49R **100** 2 MOH ALPHA DALPHA .398 299.860 600.710 899.320 0.1 סָ œ Ģ ເຕ ų ? o . **№**0□**◊**4

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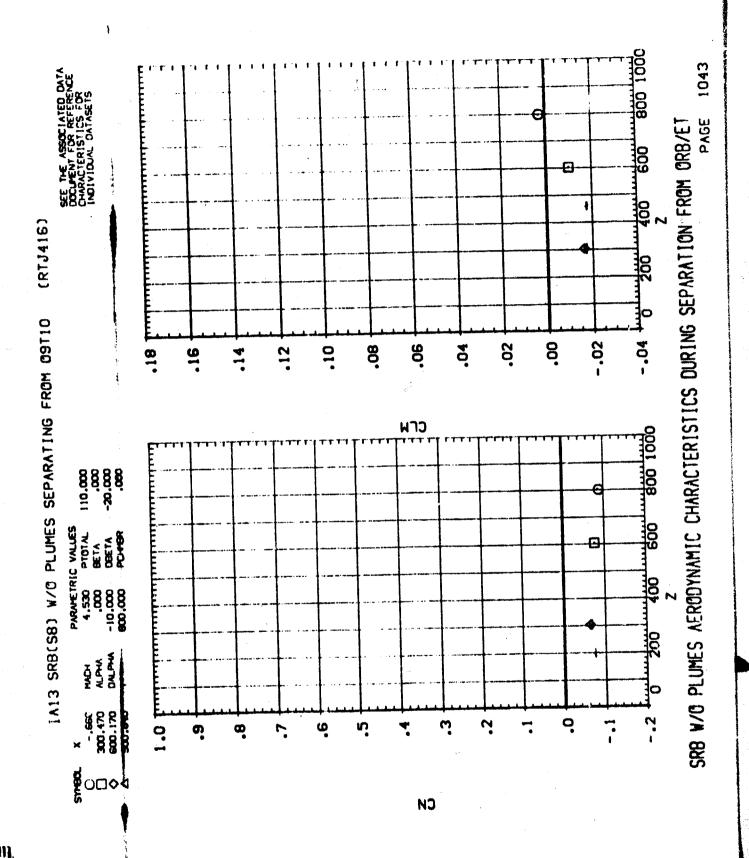
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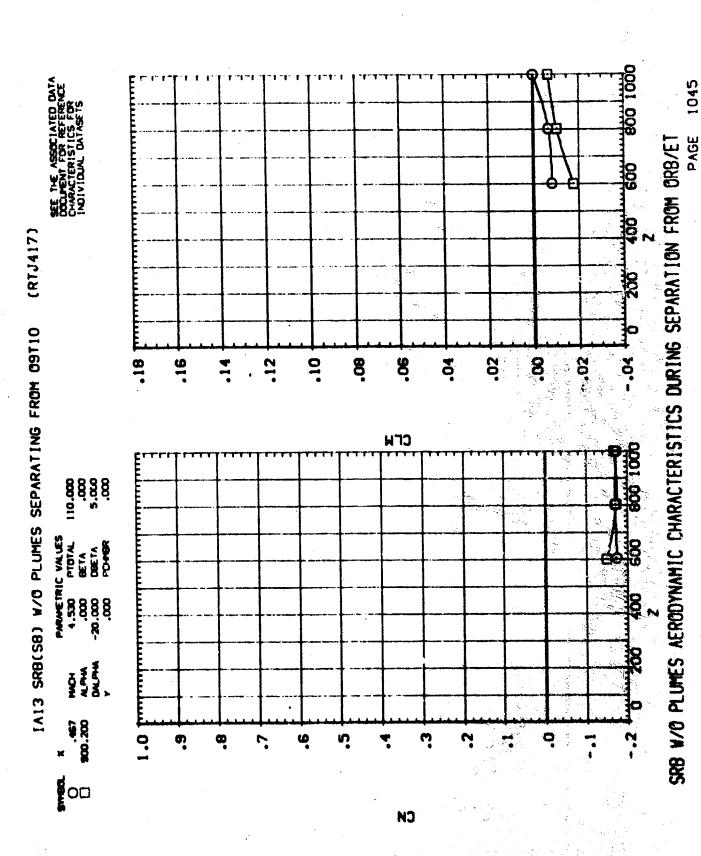
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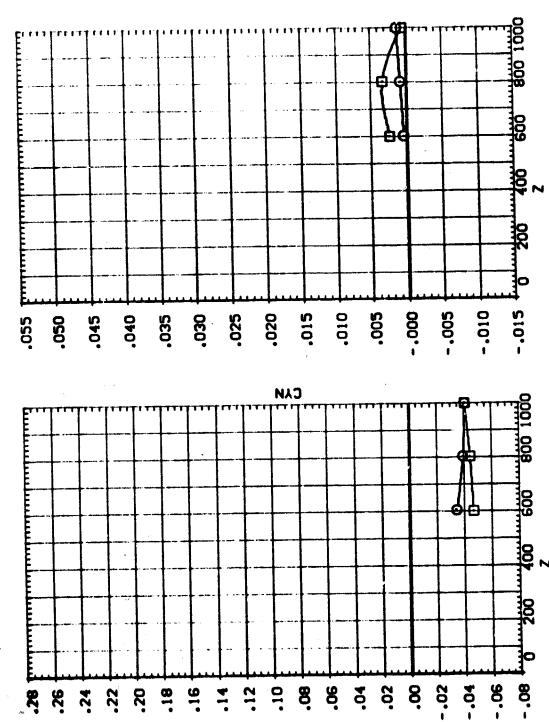
SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



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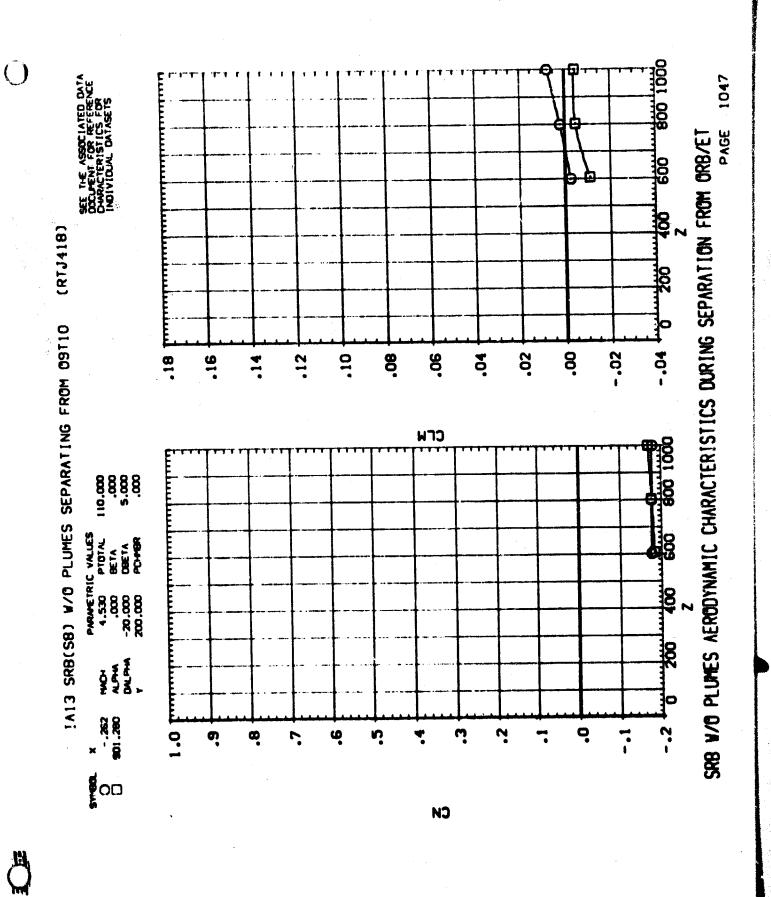


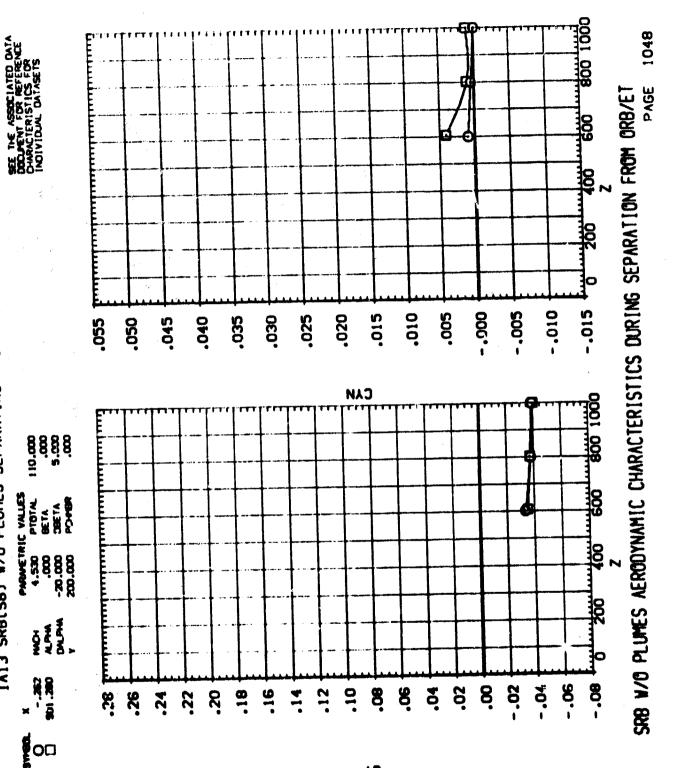


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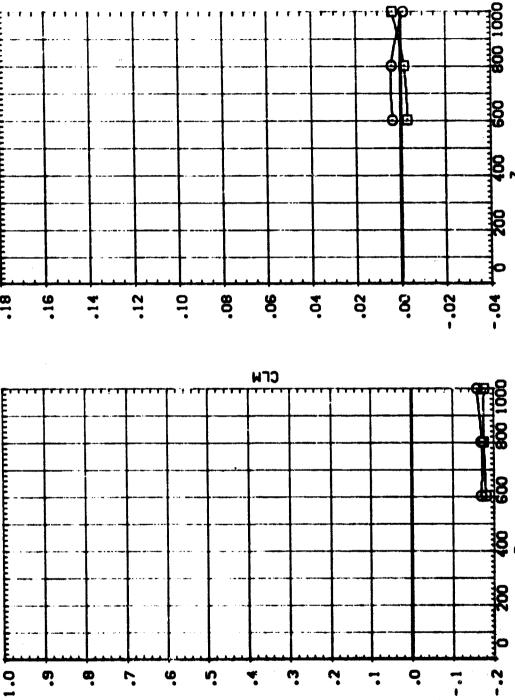


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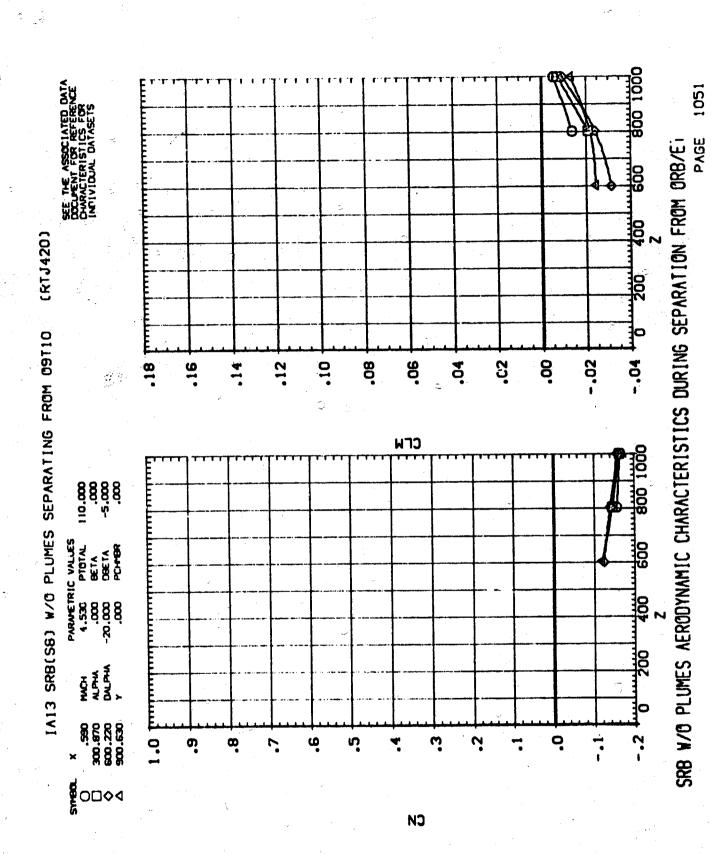
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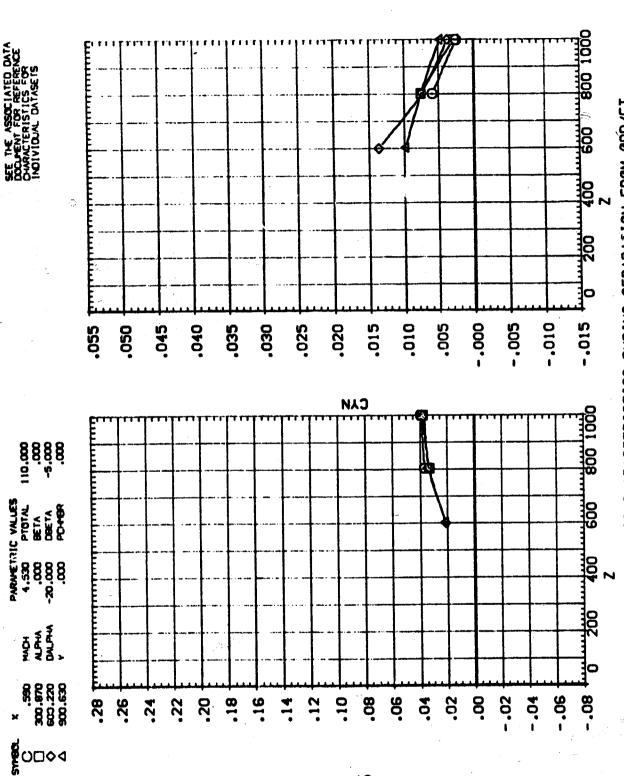
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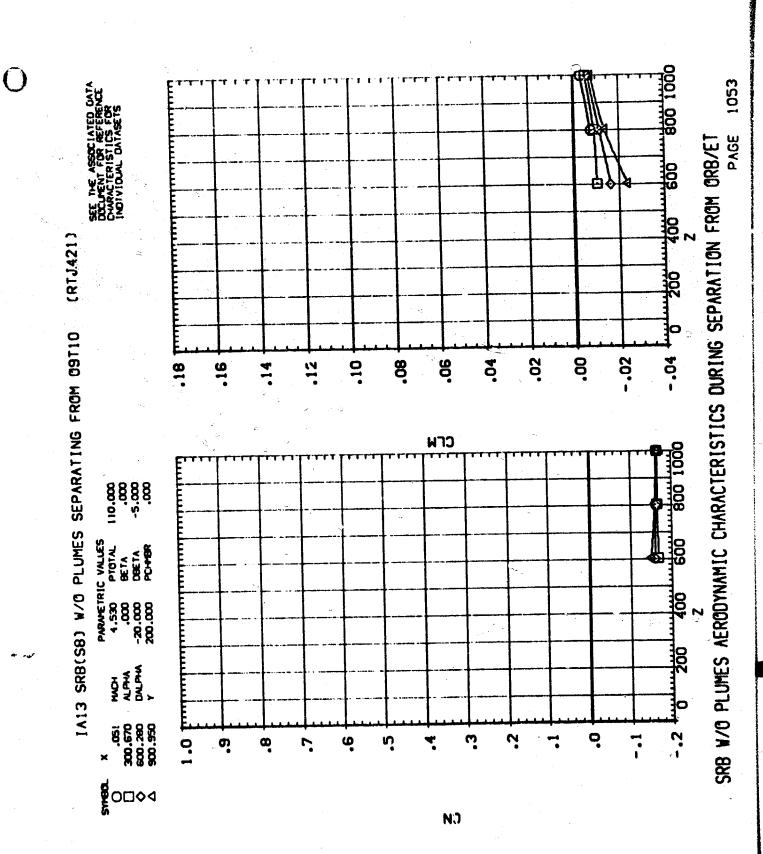


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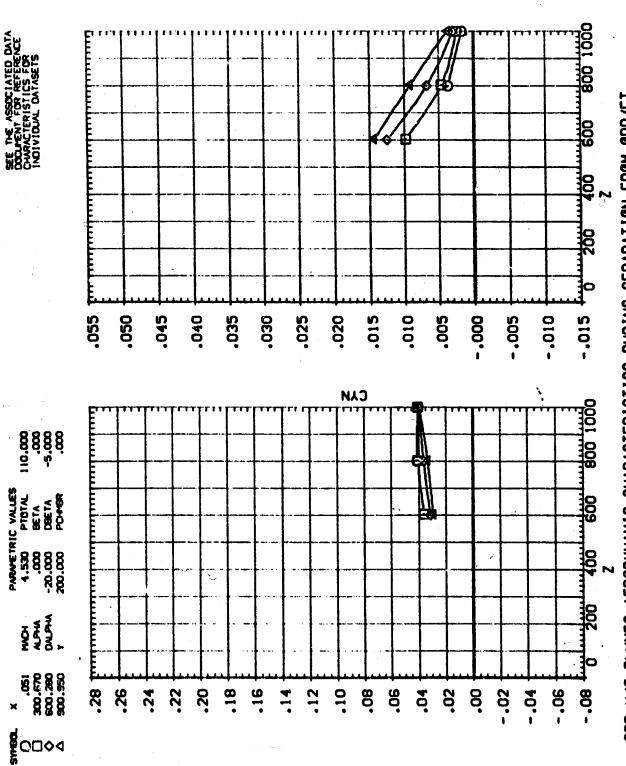


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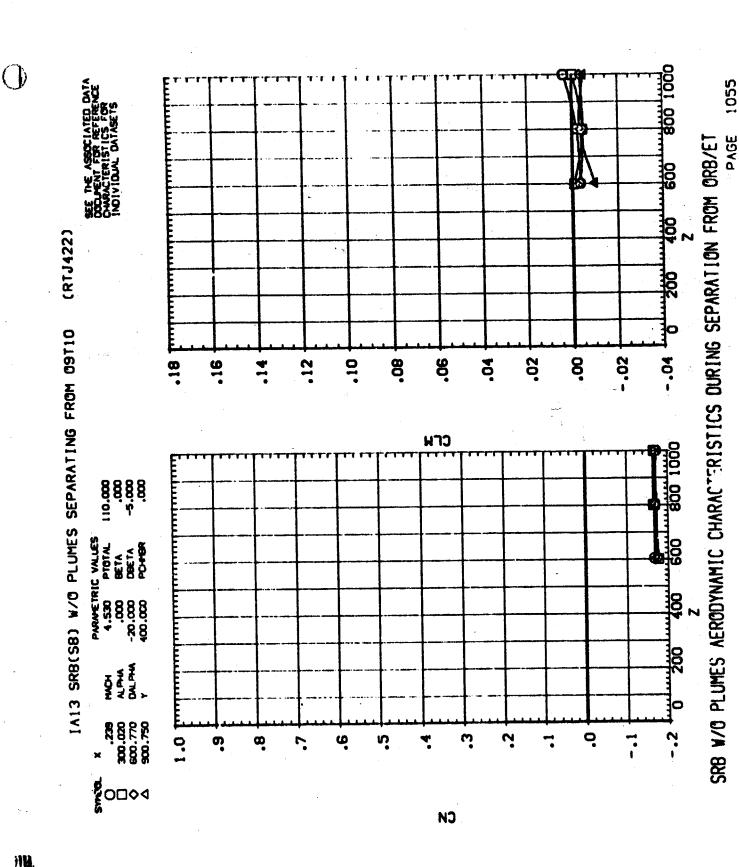


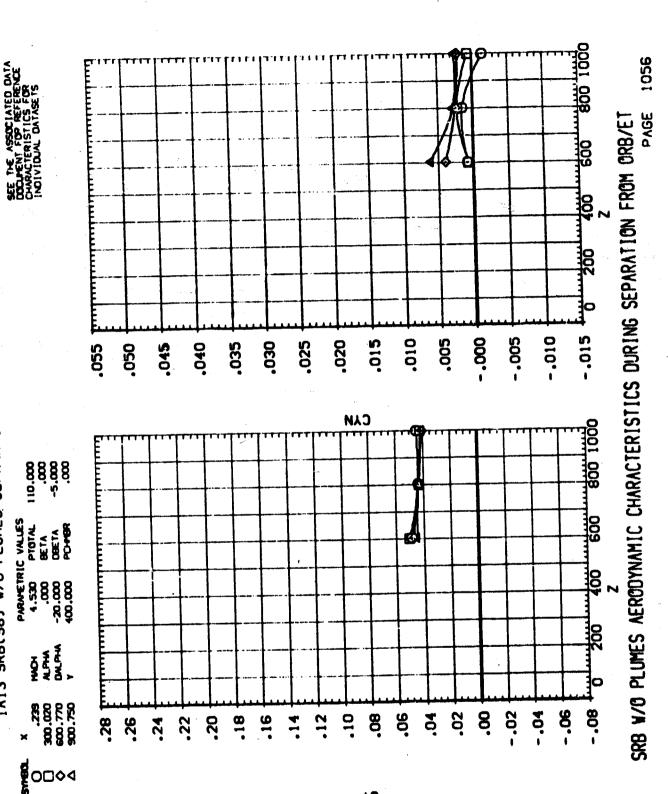
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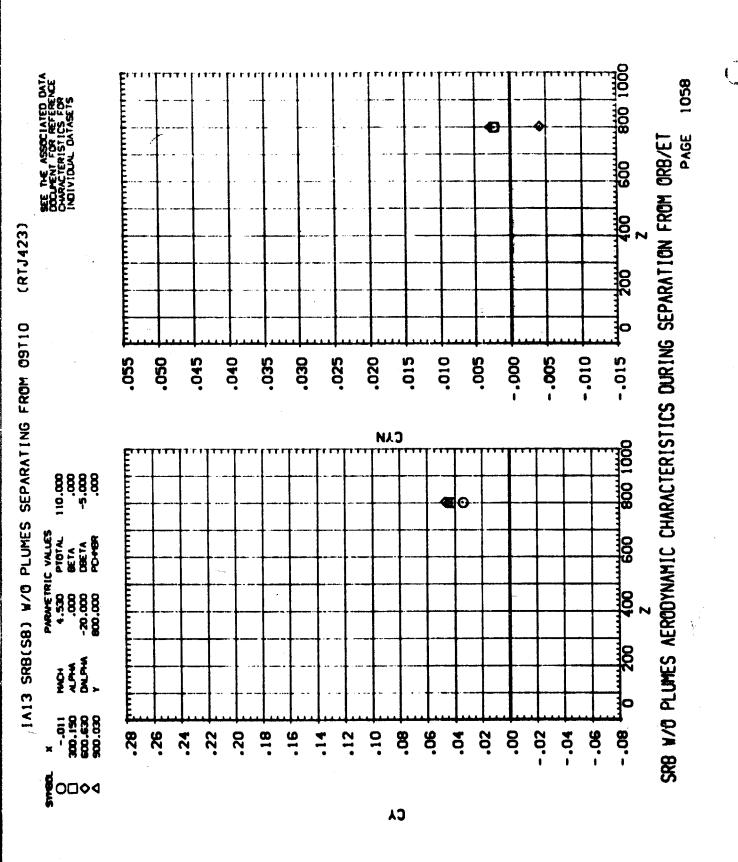
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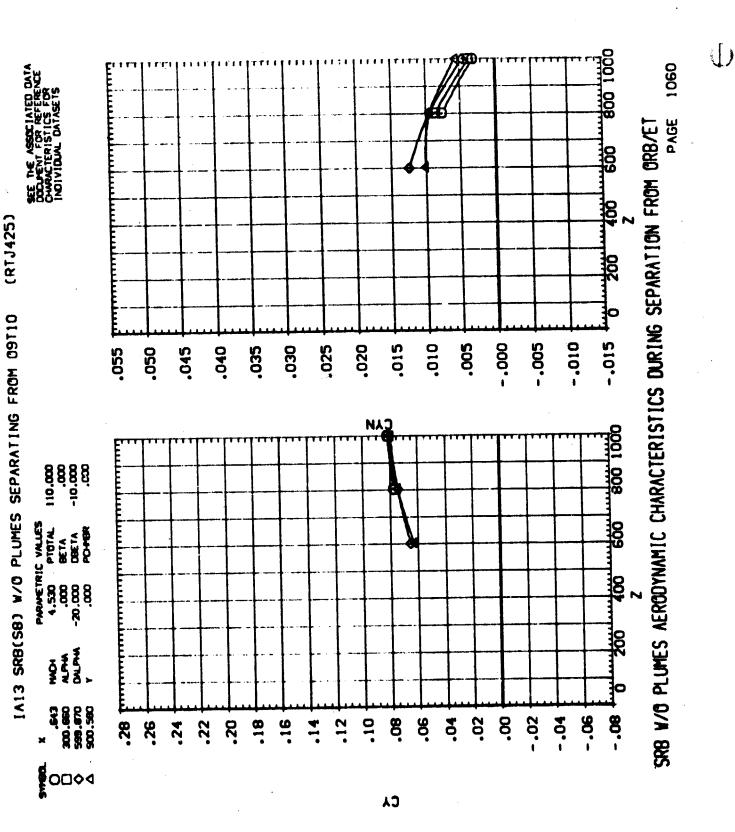


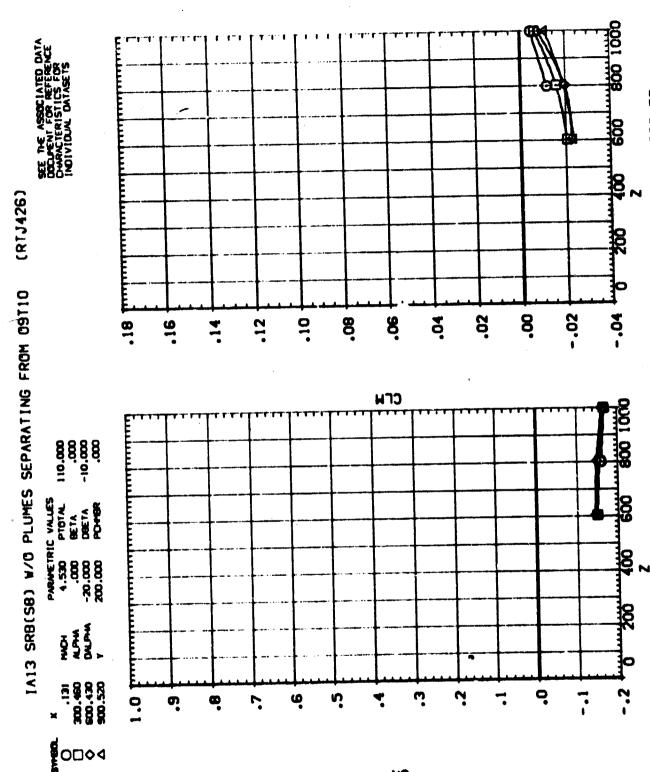
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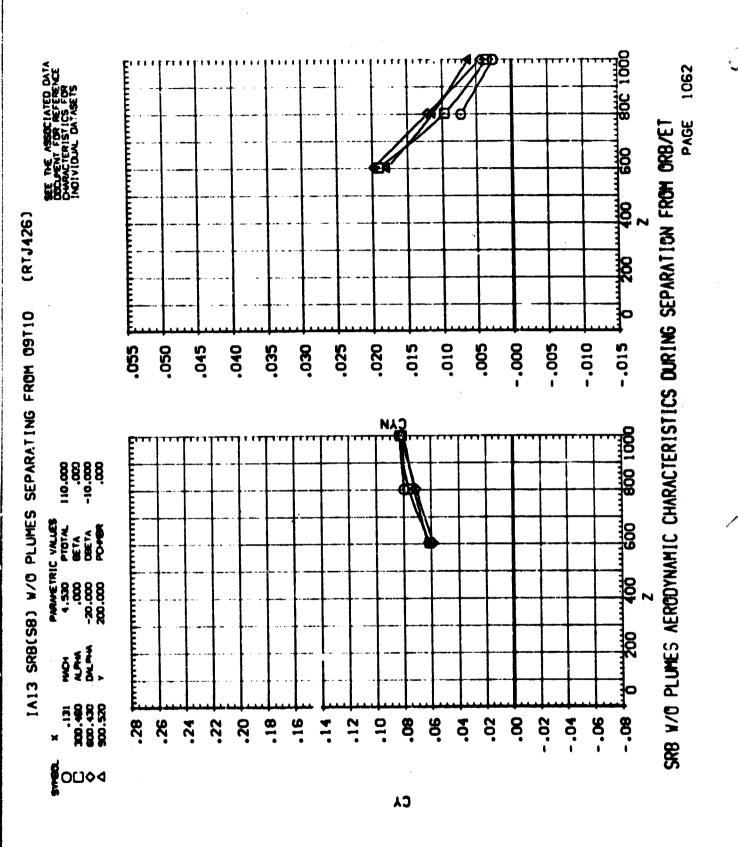
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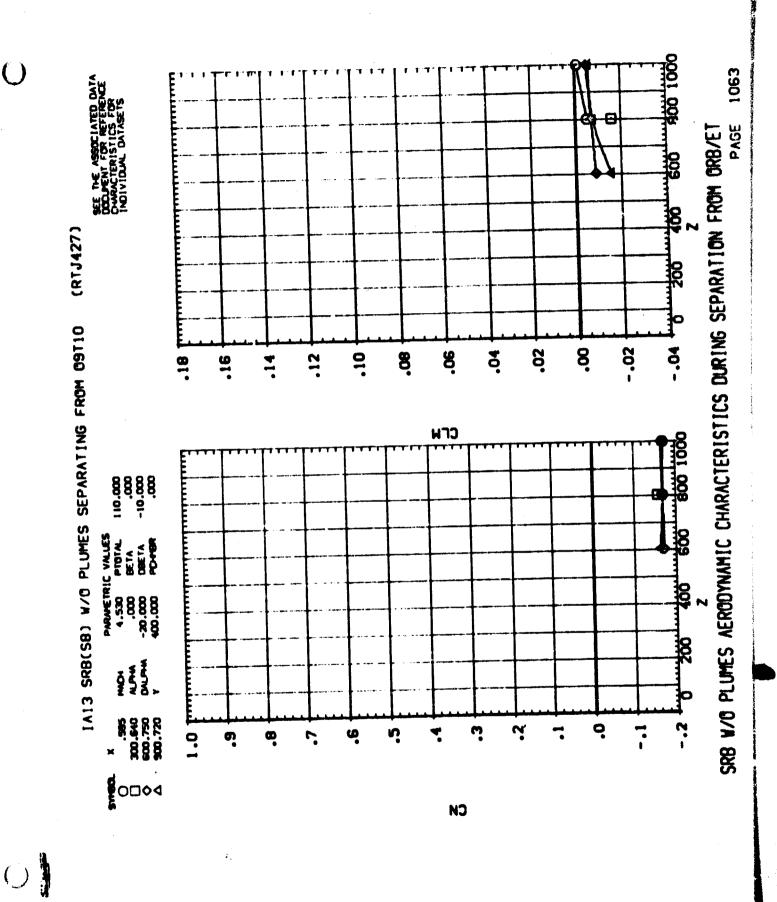
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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CAMPACTERISTICS FOR INDIVIDUAL DATASETS SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 800 82 (RTJ427) 18 IA13 SRB(SB) W/O PLUMES SEPARATING FROM 09110 -.015 .055 .045 -.010 020 .040 .035 .015 010. -.000 -.005 80. .025 .020 8 CAN **600 1000** PARMETRIC VALLES 4.530 PTGTAL .000 RETA -20.000 DRETA 400.000 PO-49R 8 18 2002 SER S 300.550 300.750 300.750 .28 97. .18 .16 0: 8 8 .02 .. .24 .22 8 .14 .12 ó 8 -.02 9. -.04 **€**0□**◊**4 CA

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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/EI 给 (RTJ428) 別 SEPARATING FROM 09110 .055 .050 .045 .040 .035 .030 .025 .020 .015 010 .002 -.000 -.005 -.010 -.015 CAN 800 1000 6.05 6.08 6.08 6.08 6.08 IA13 SRB(SB) W/O PLUMES PARAFERIC VALLES
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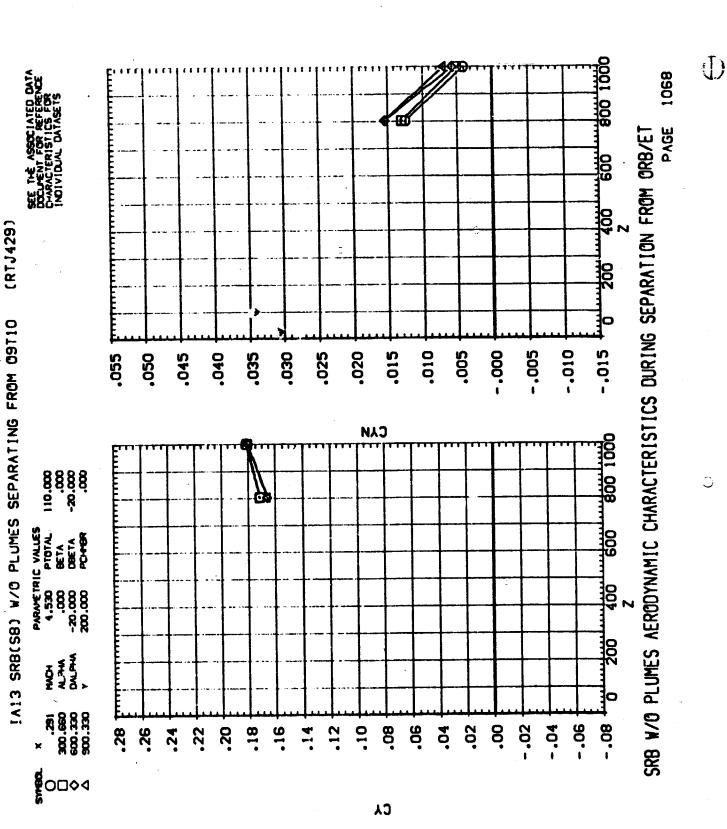
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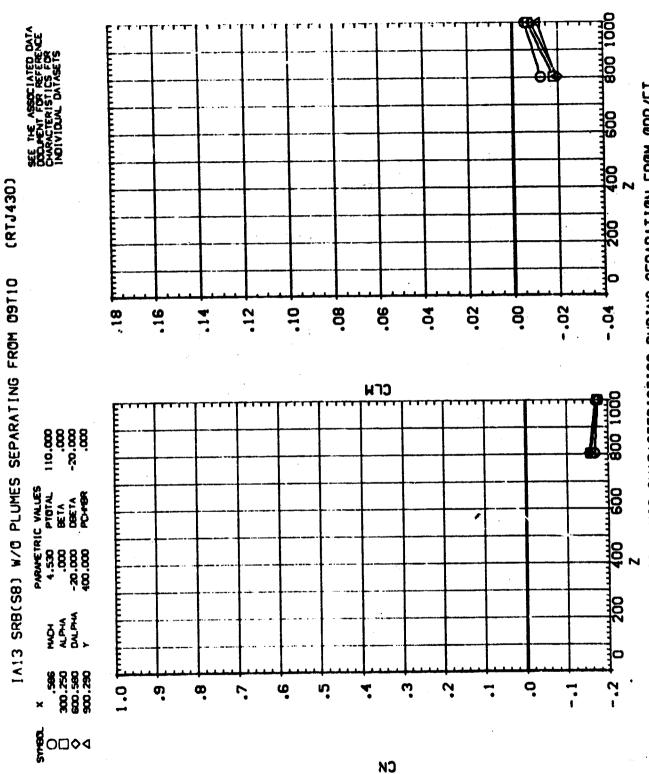
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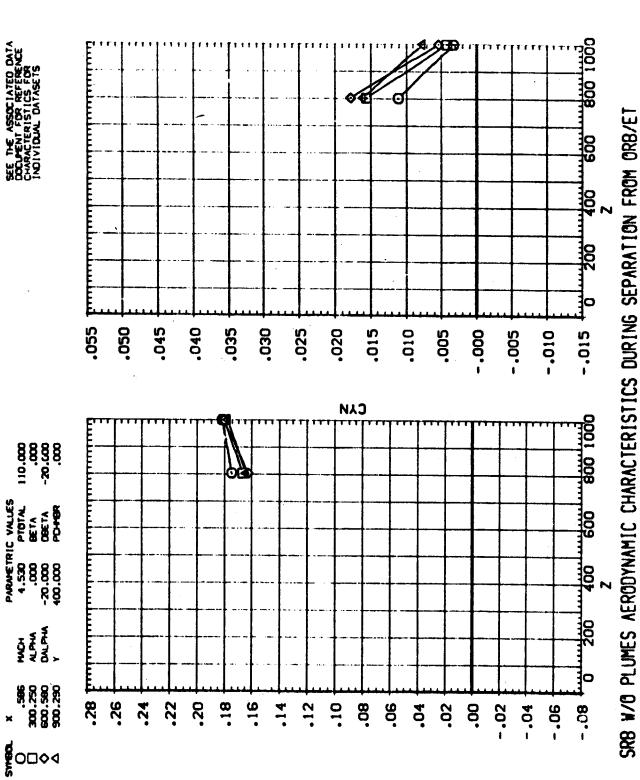
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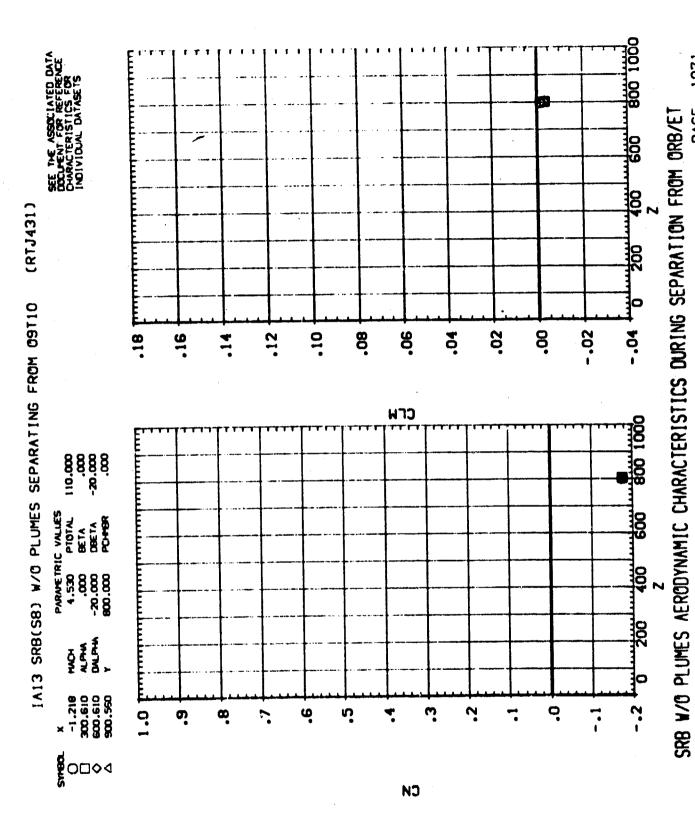
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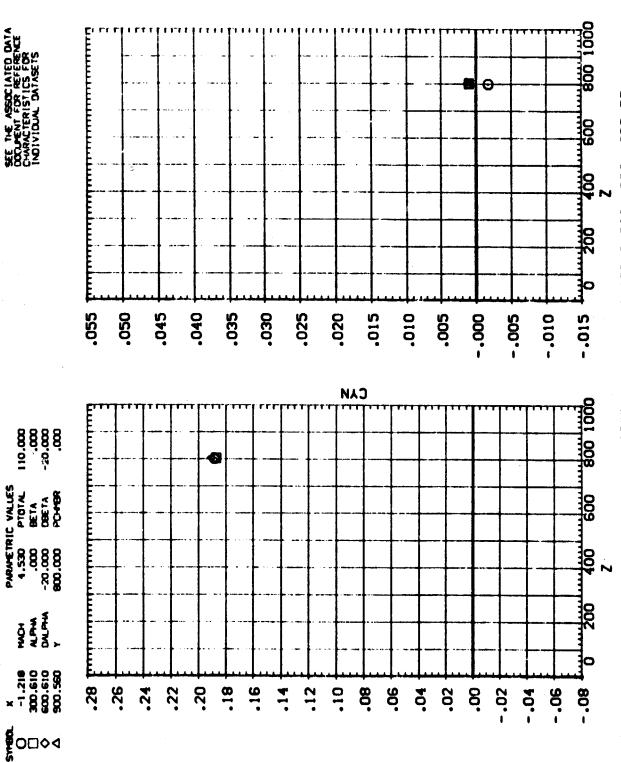
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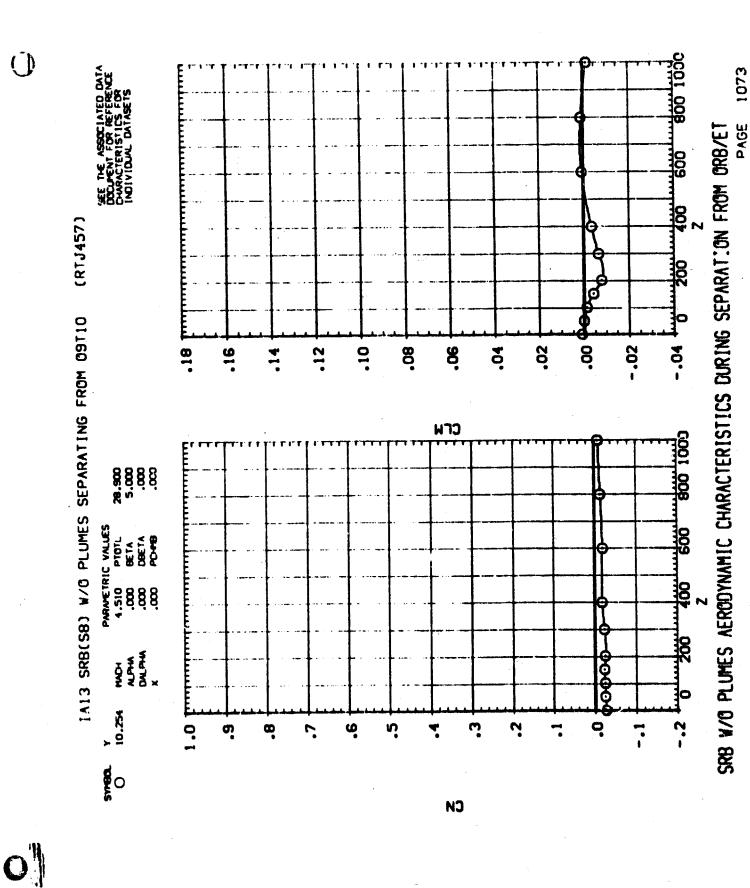


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SRB W/O PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



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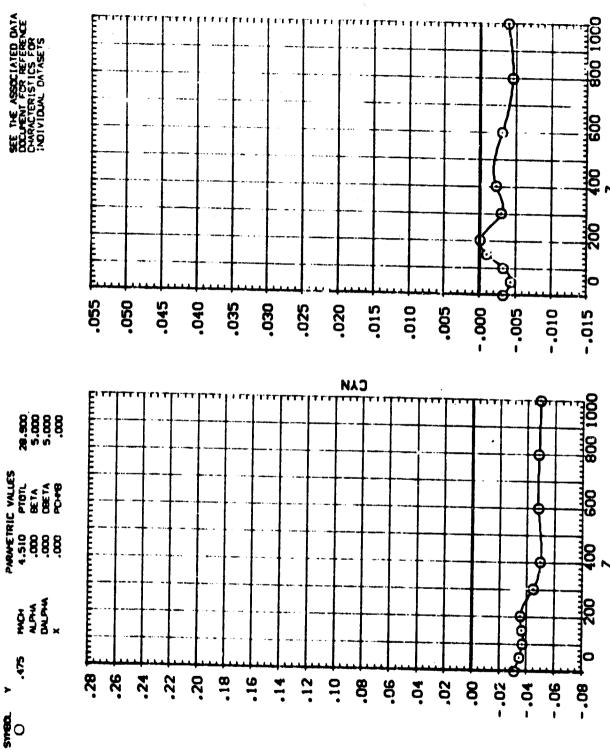
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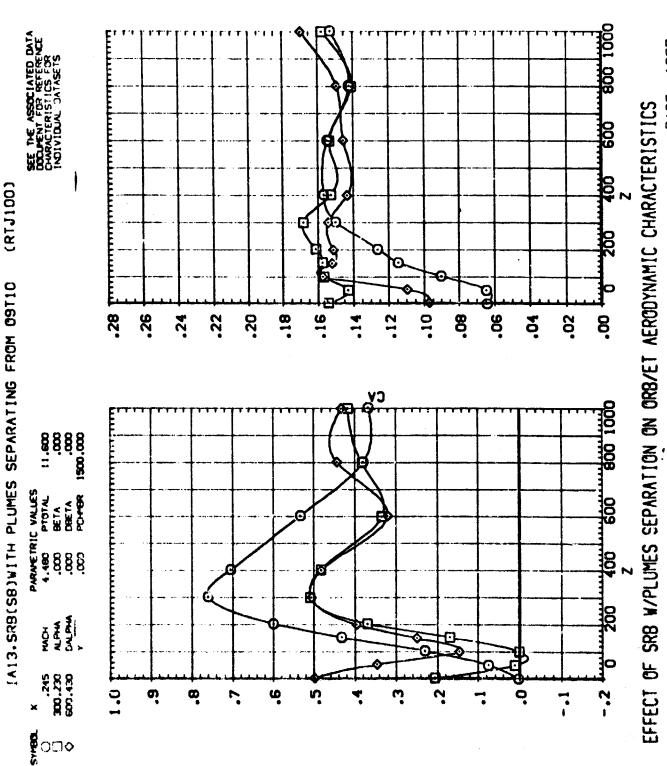


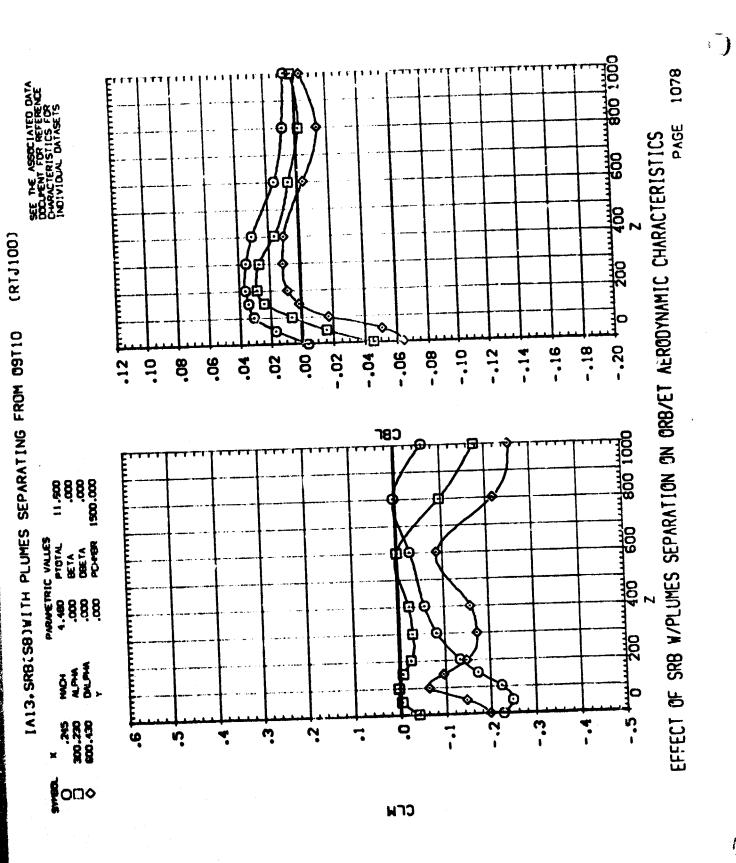
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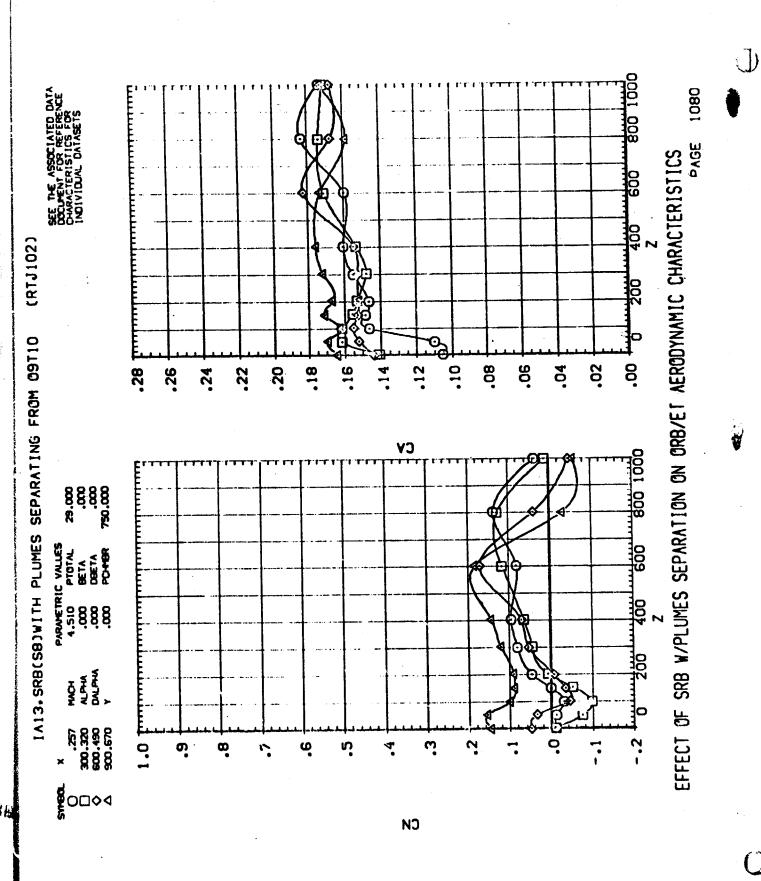
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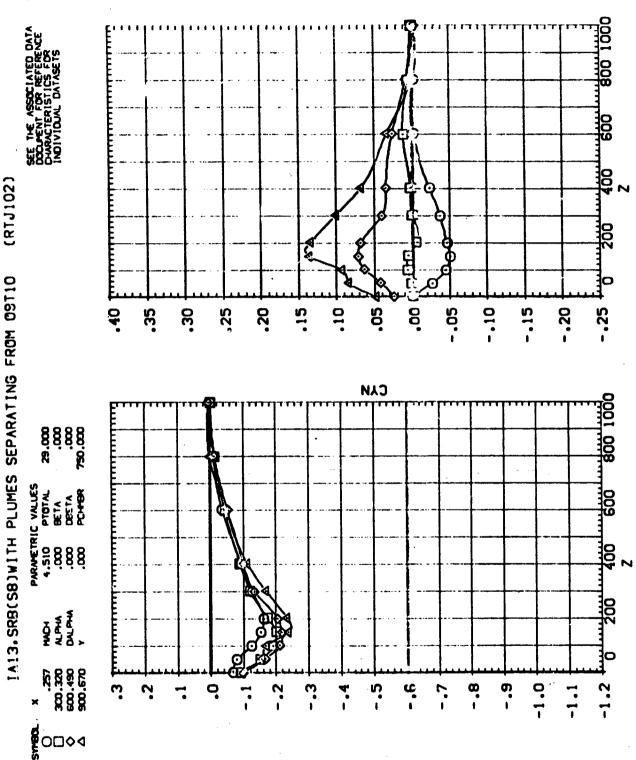
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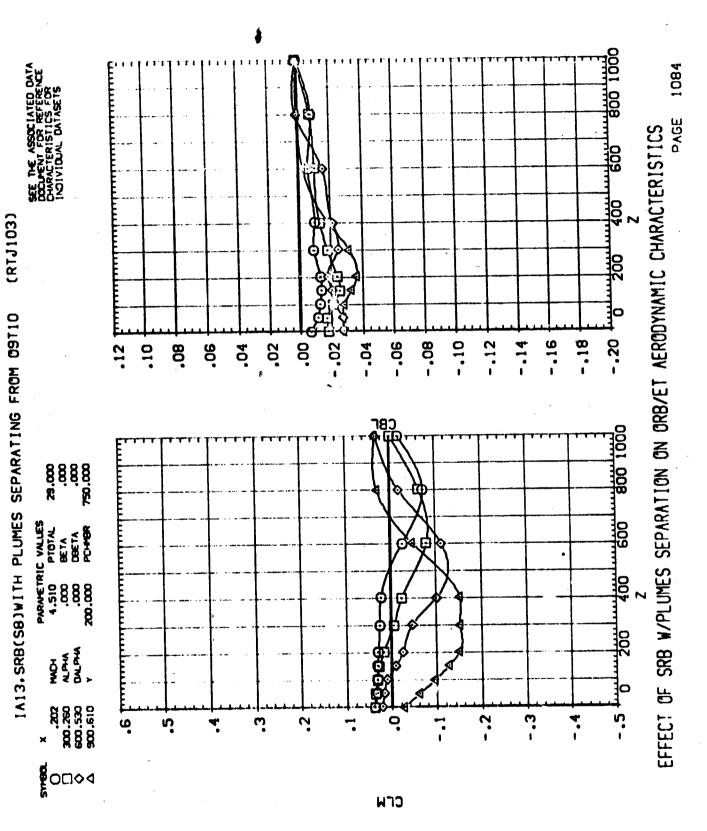
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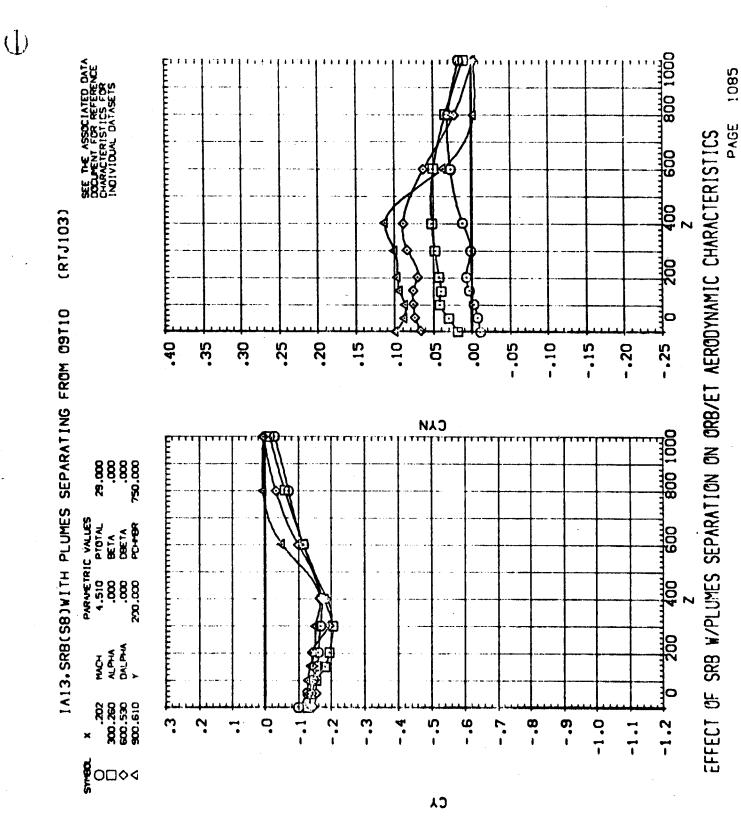
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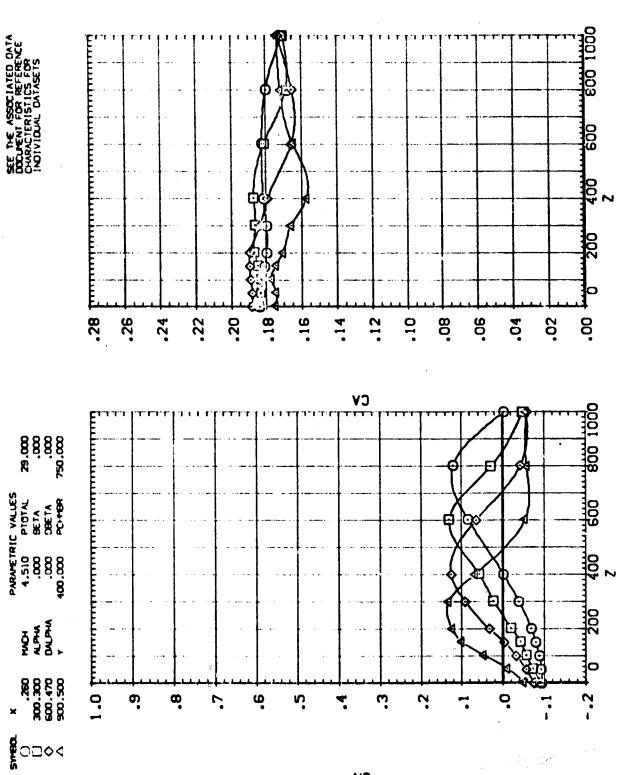
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IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 (RTJ104)



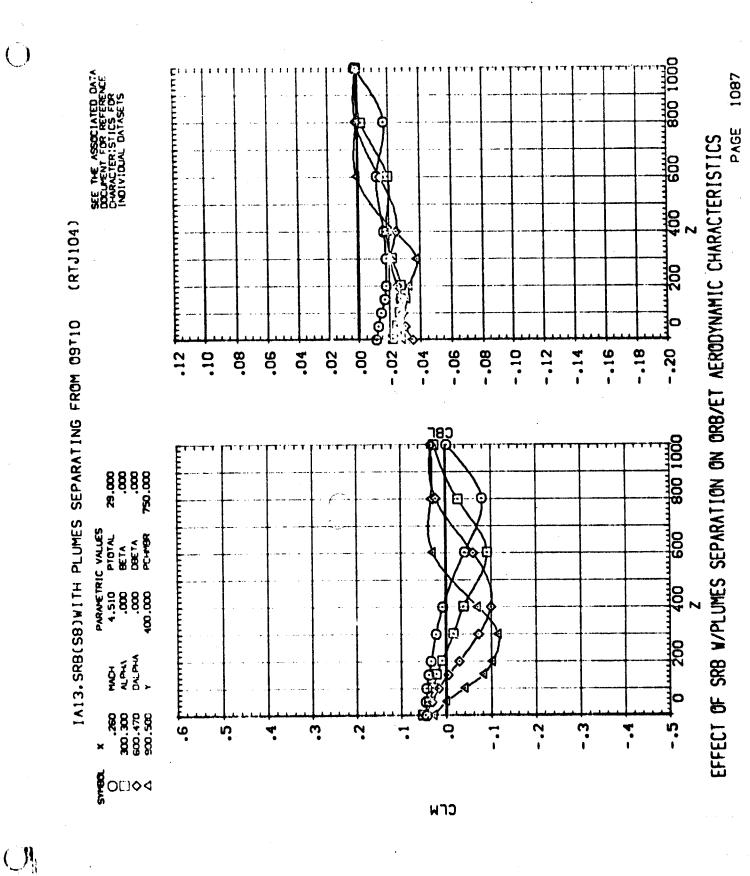
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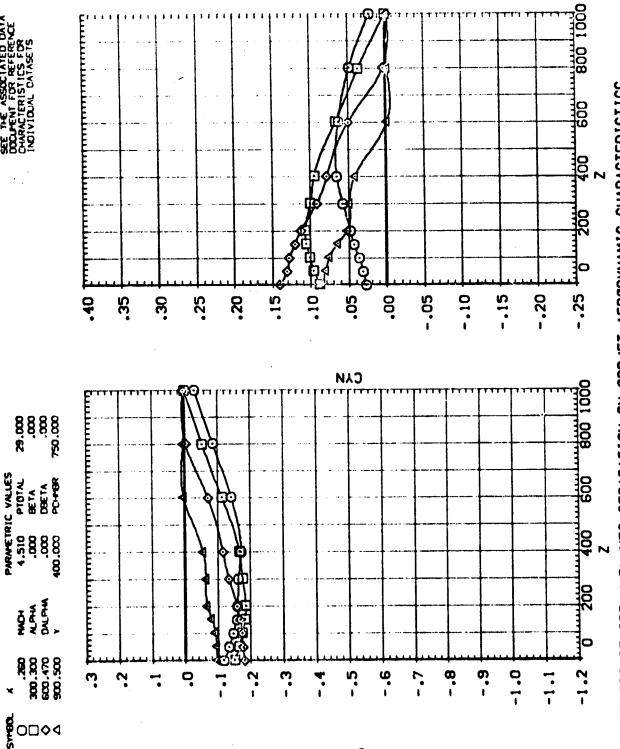
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710 (RTJ104)



PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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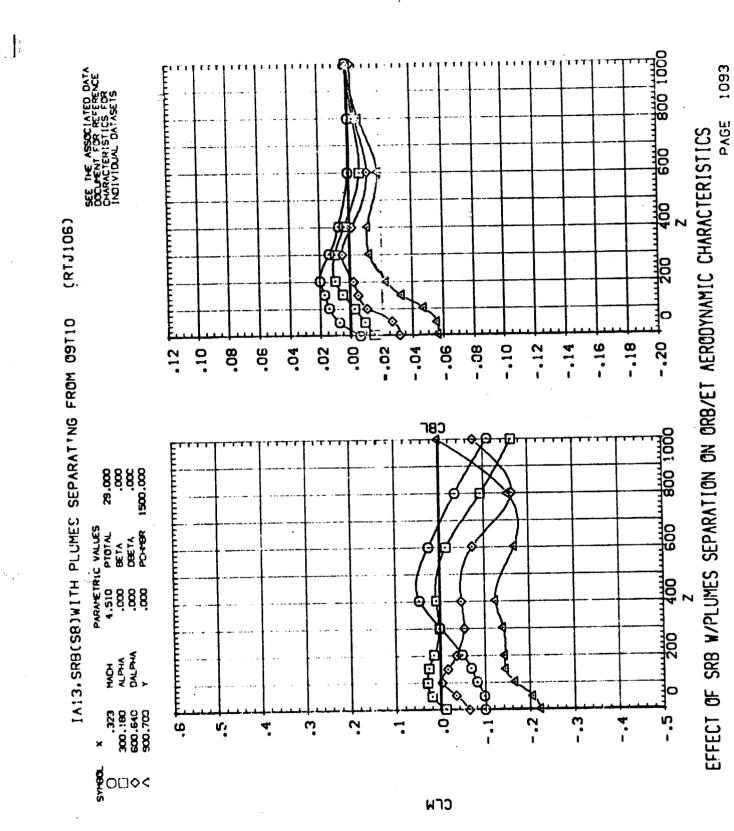
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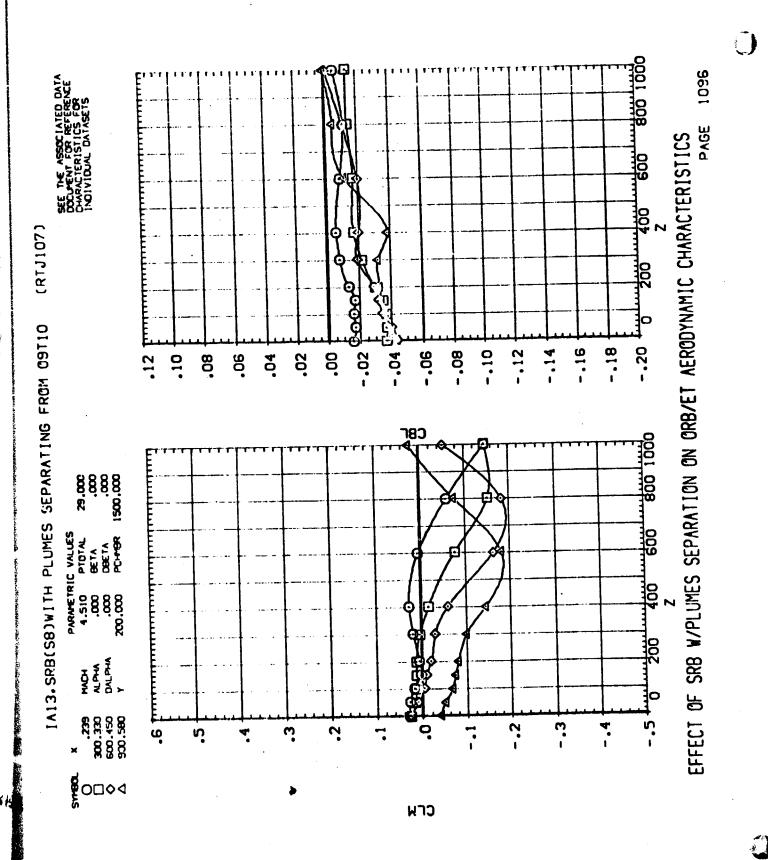
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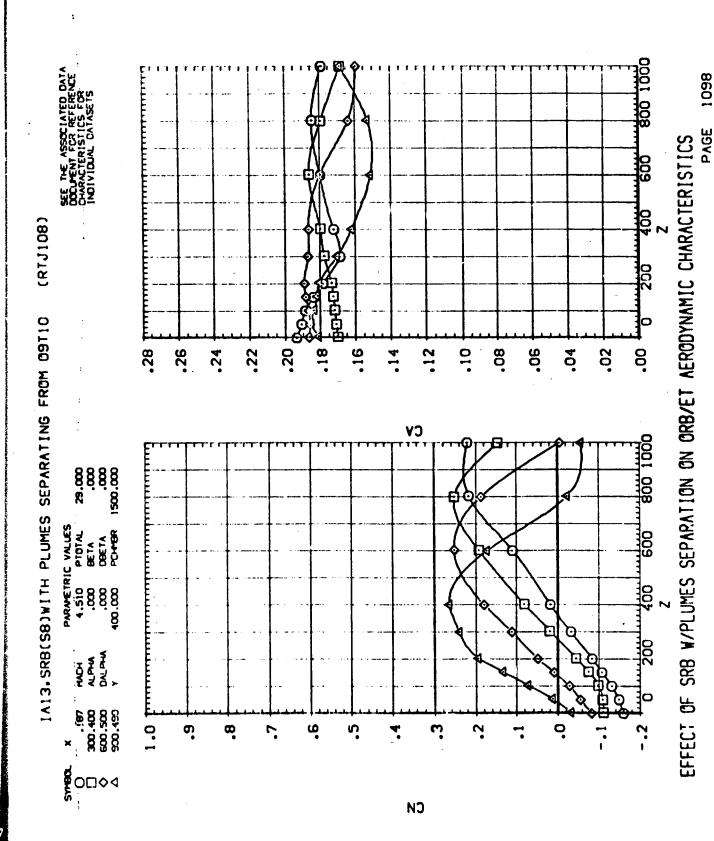
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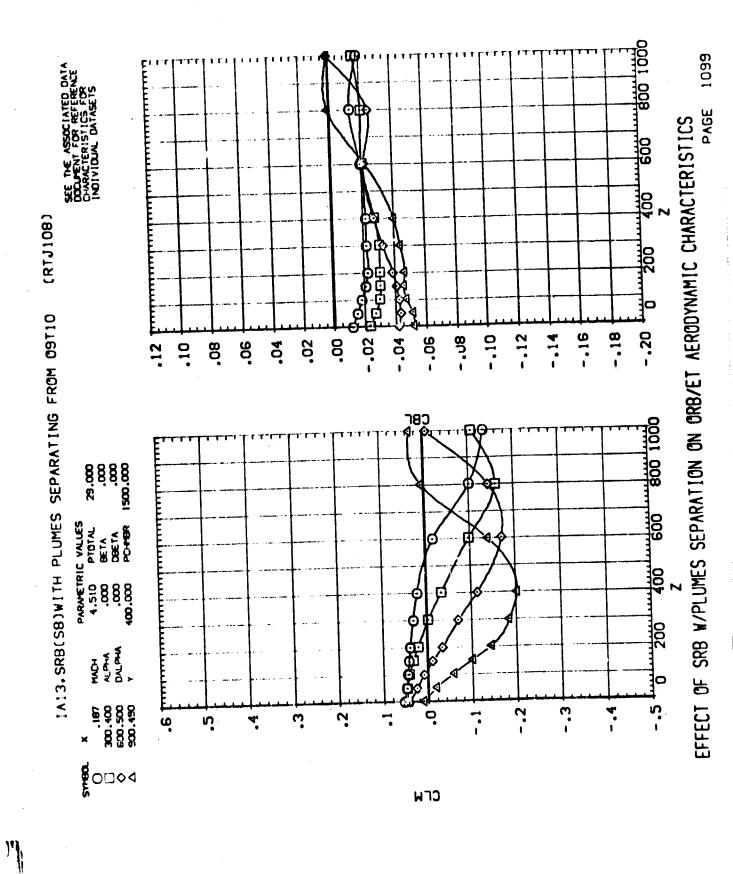
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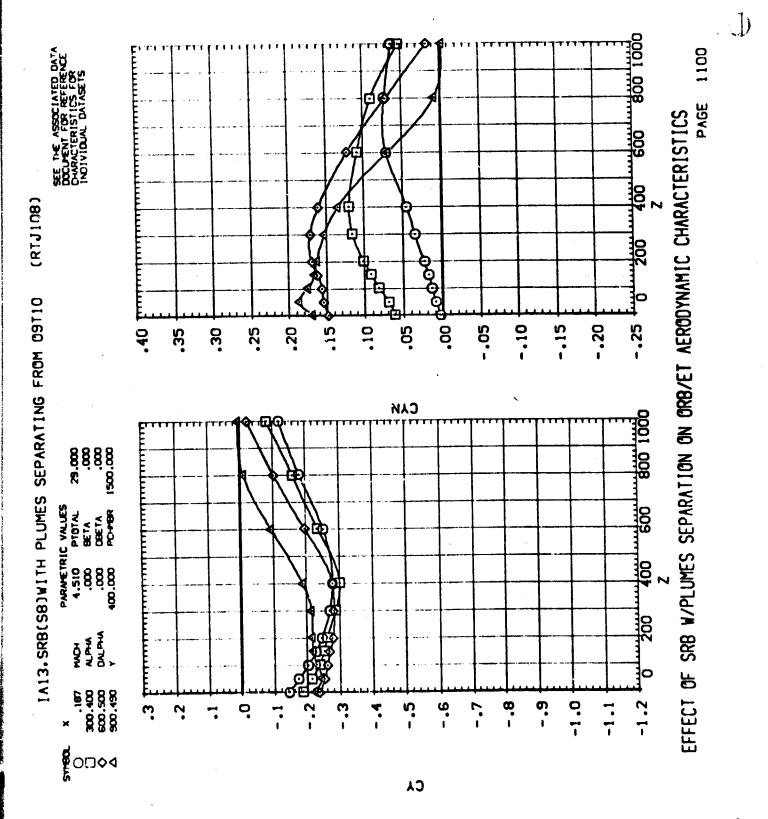
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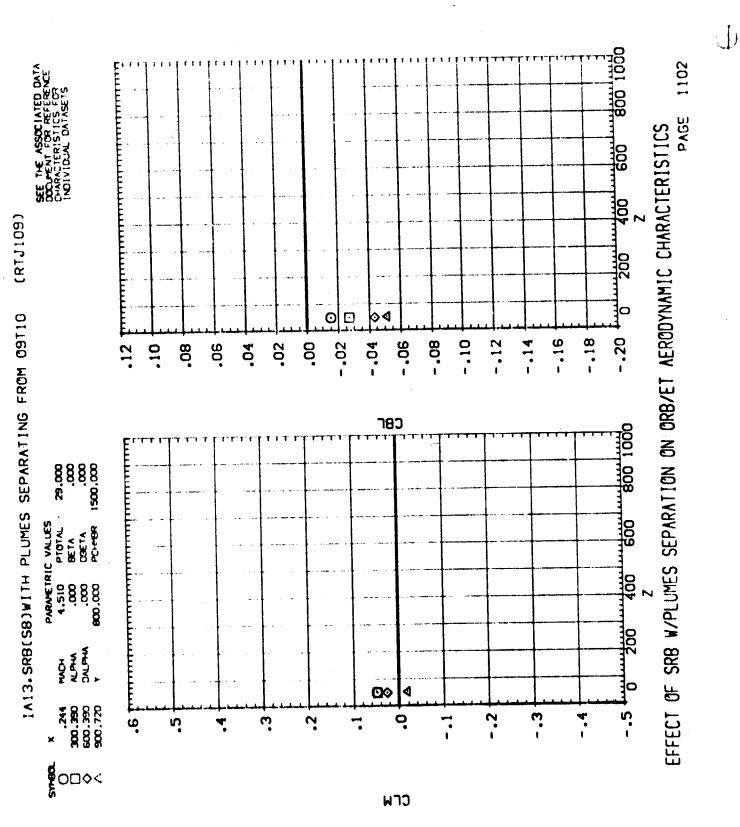


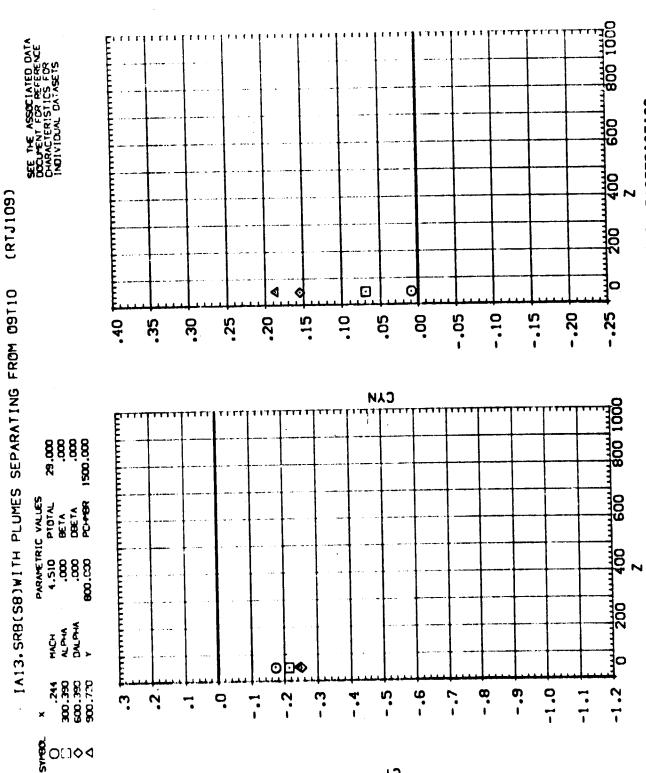


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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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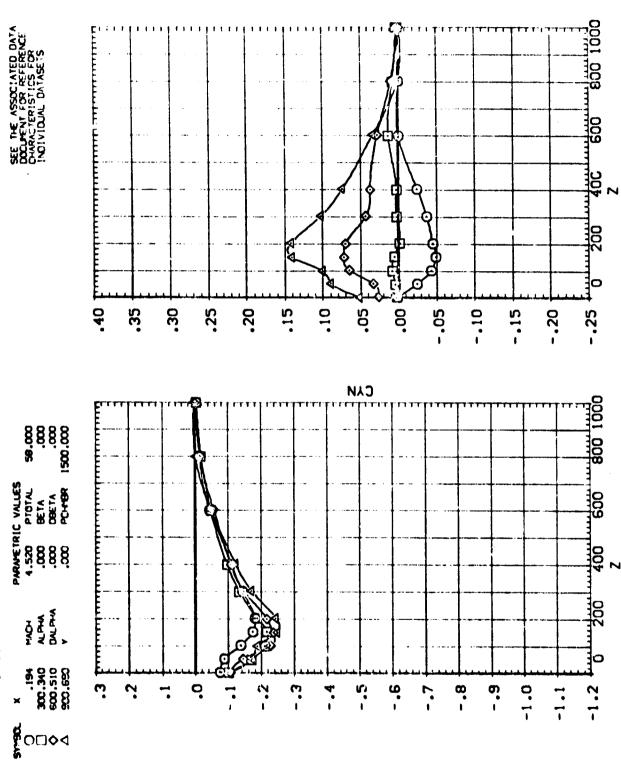
EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ110)

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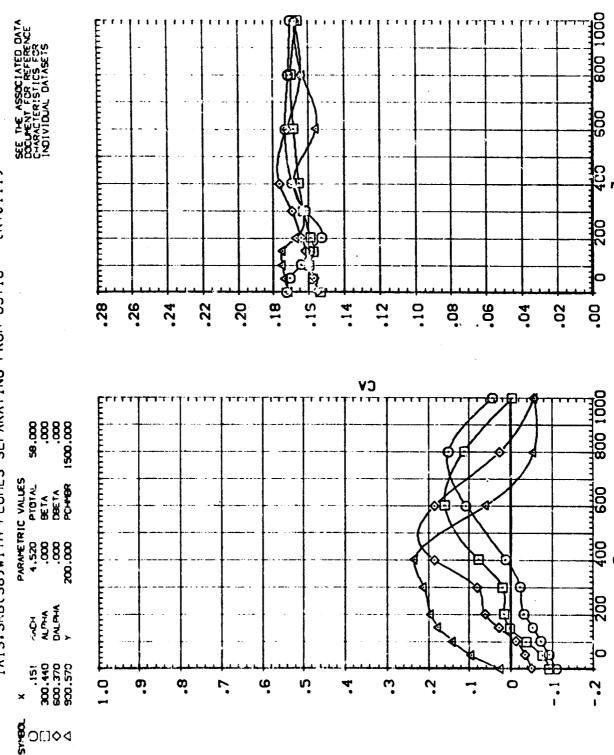
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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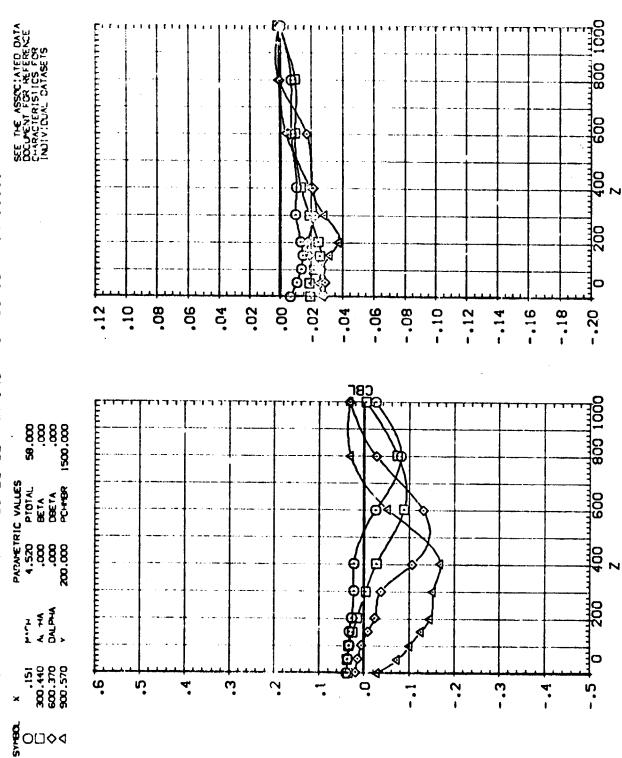
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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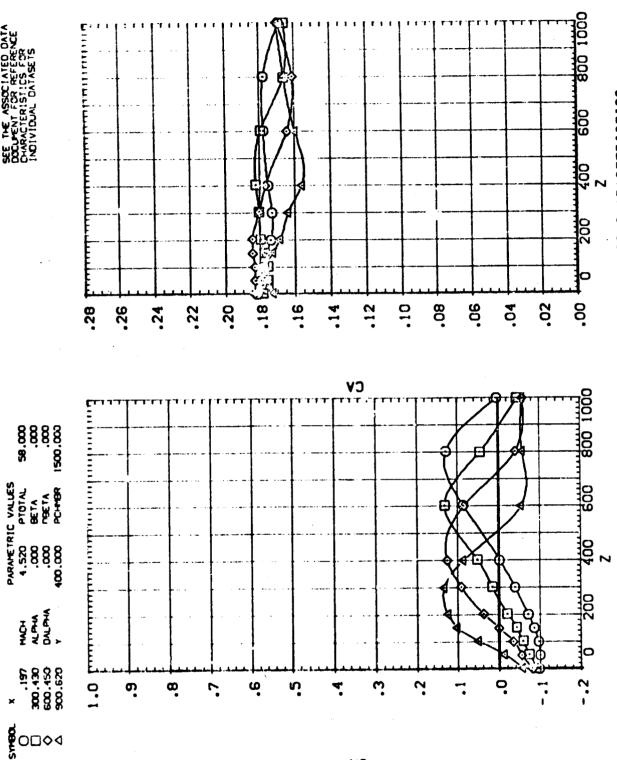
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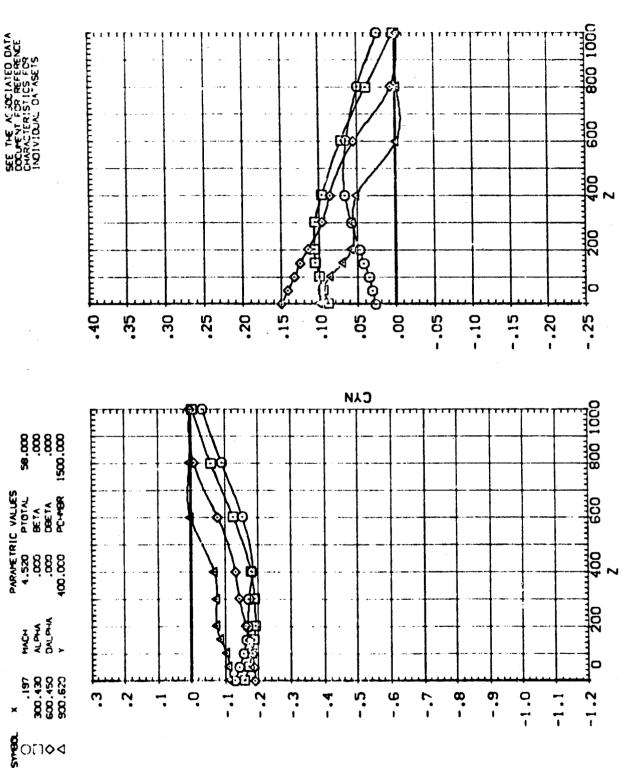
PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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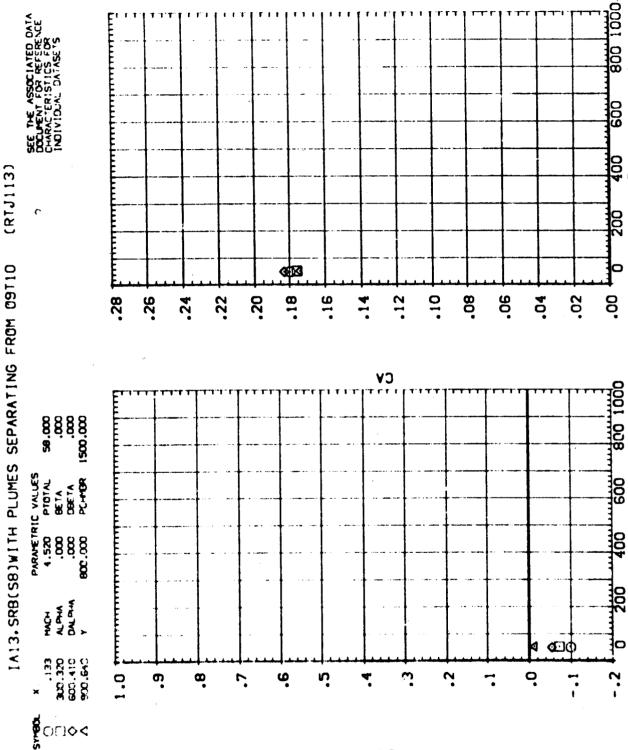
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ112)



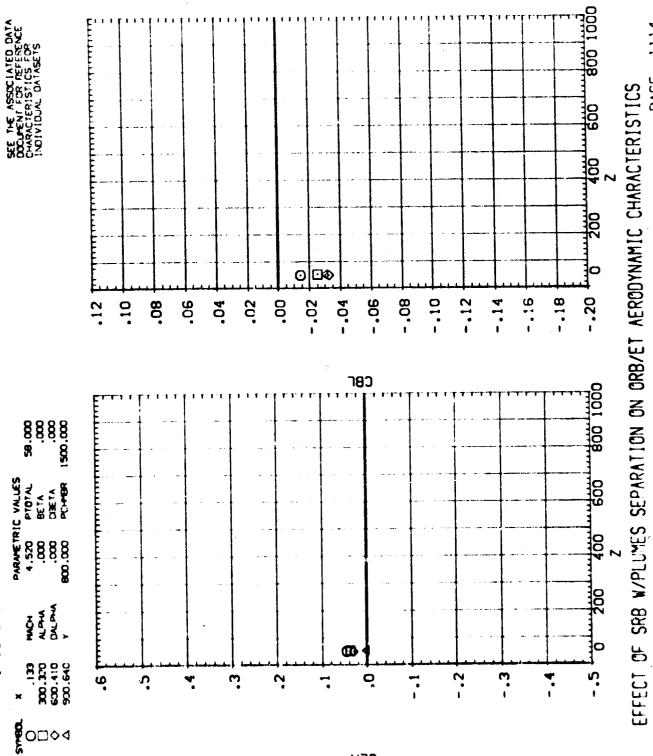
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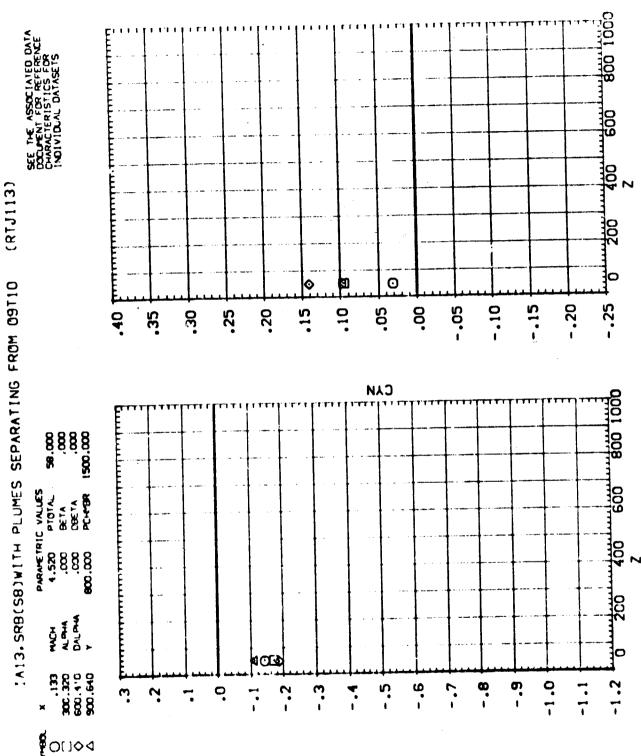
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ113)



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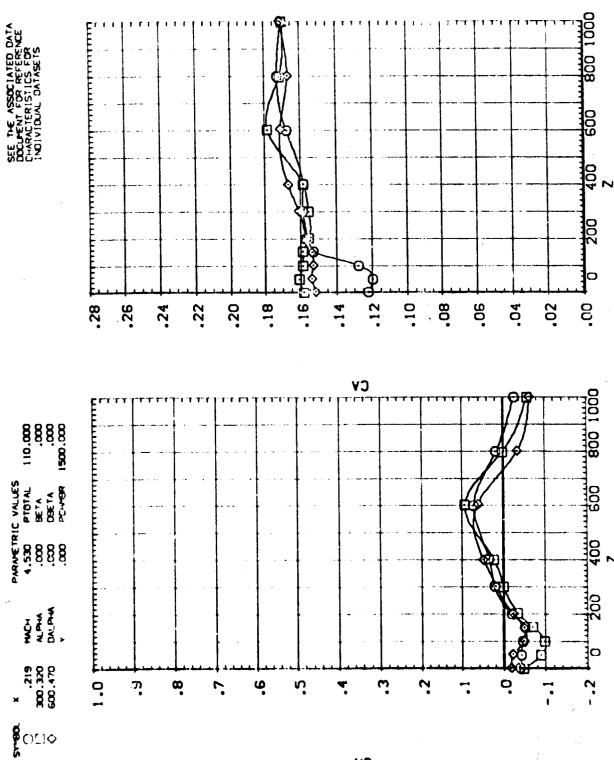
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 03110 (RIJ114)



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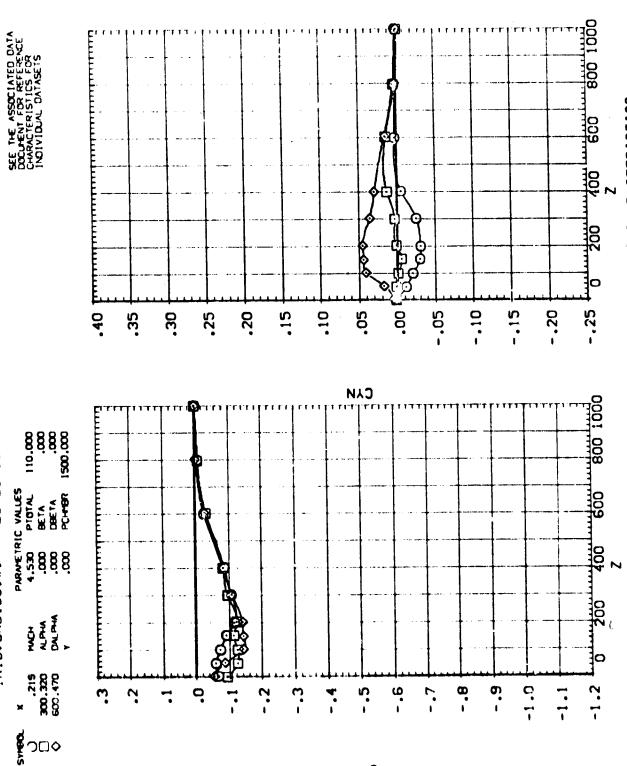
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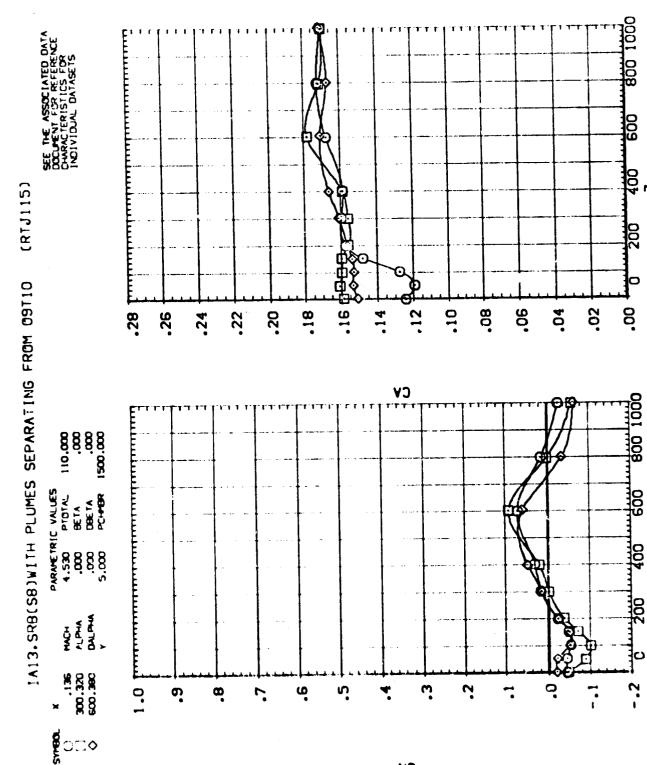
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IA13. SRB(S8)WITH PLUMES SEPARATING FROM 09110 (RTJ114)



1118 PASE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS



EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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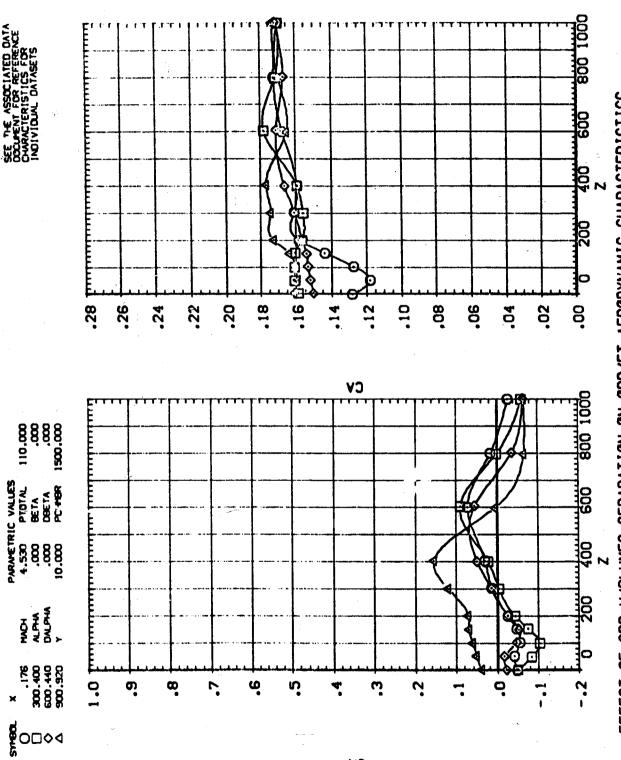
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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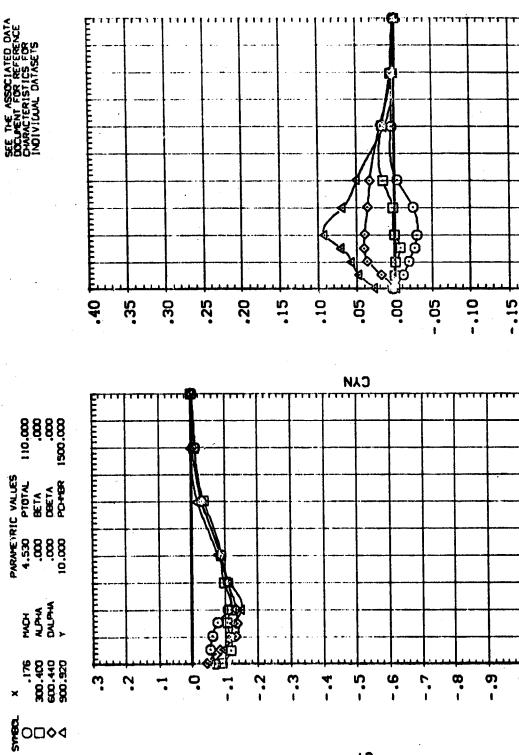
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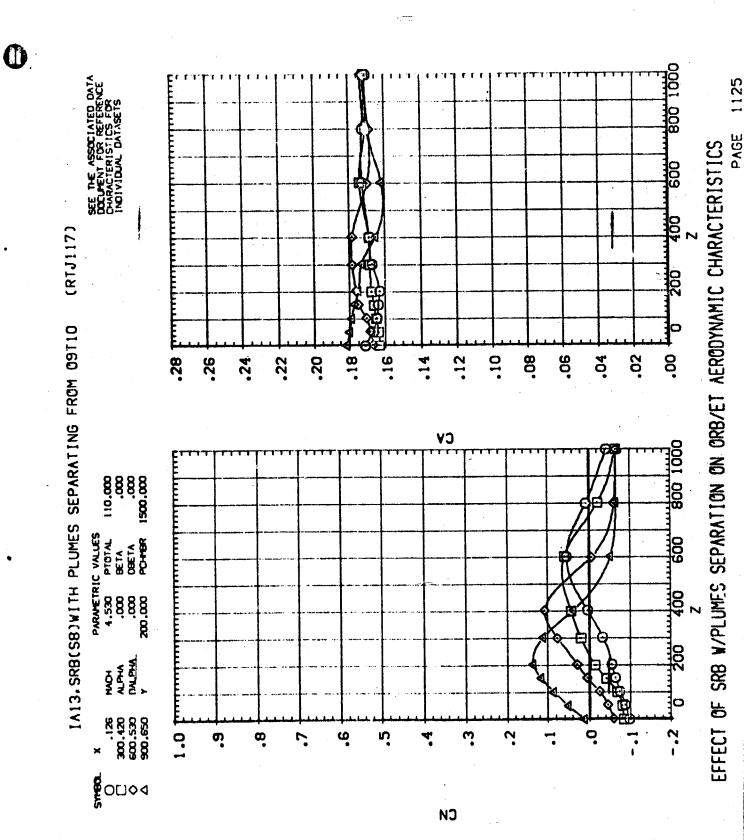
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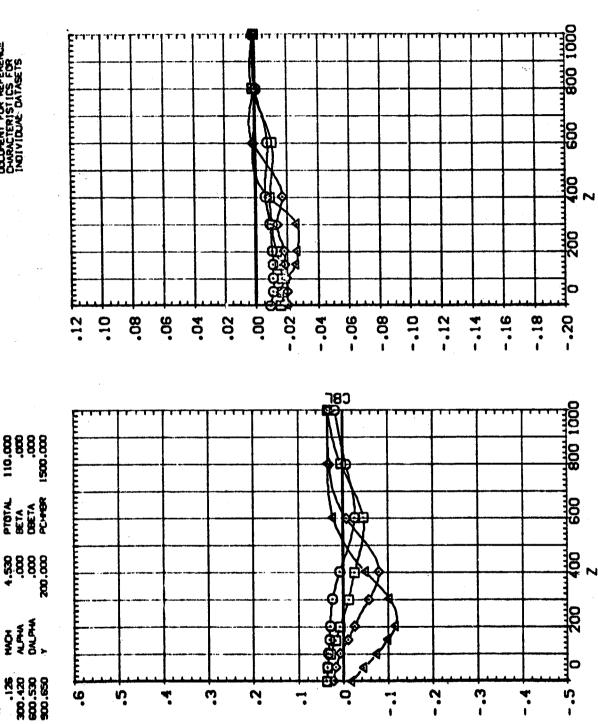
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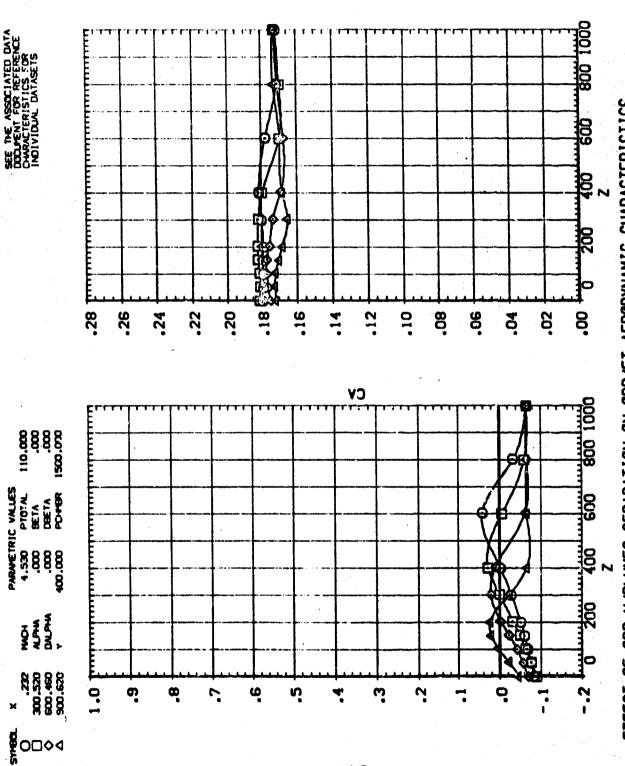
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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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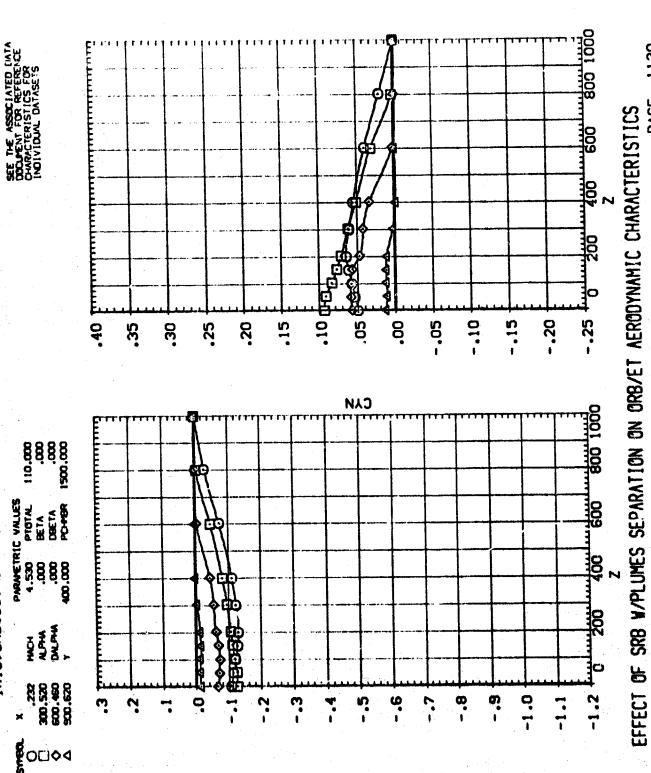
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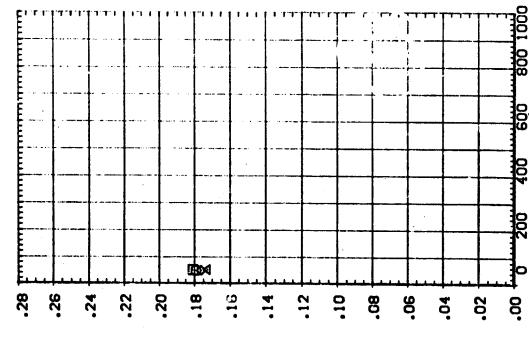
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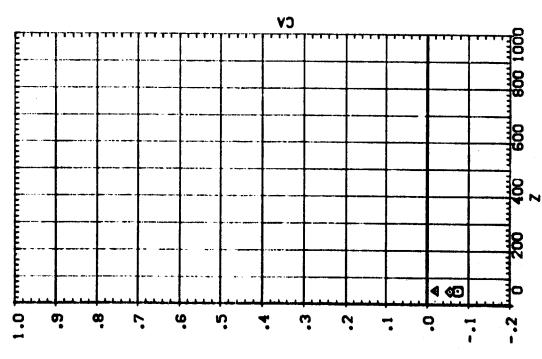
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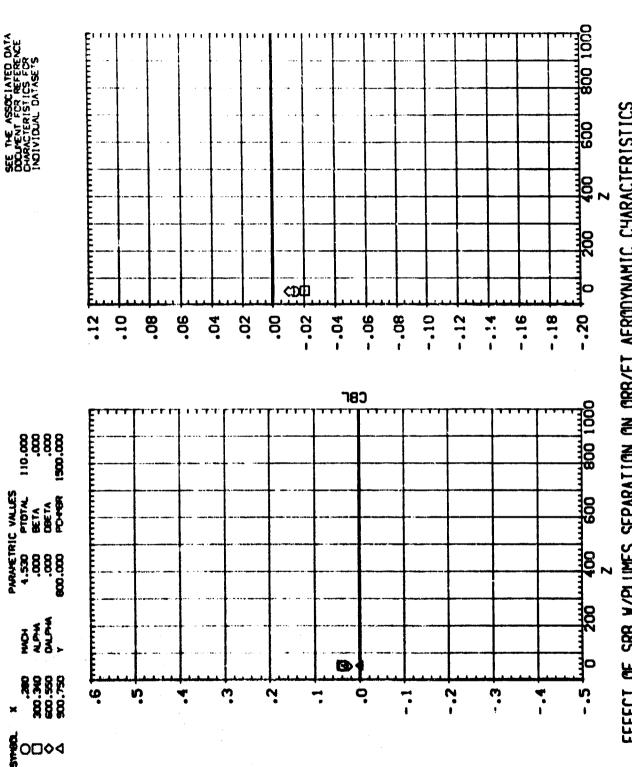


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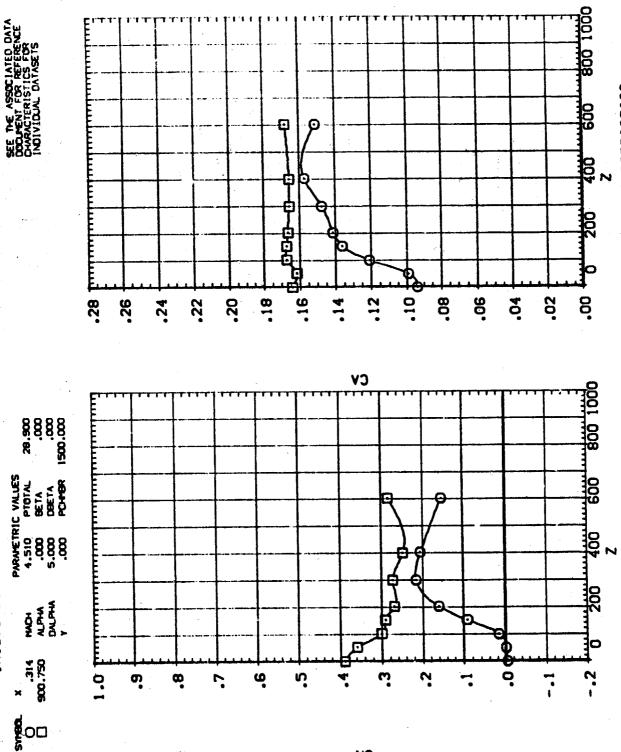
PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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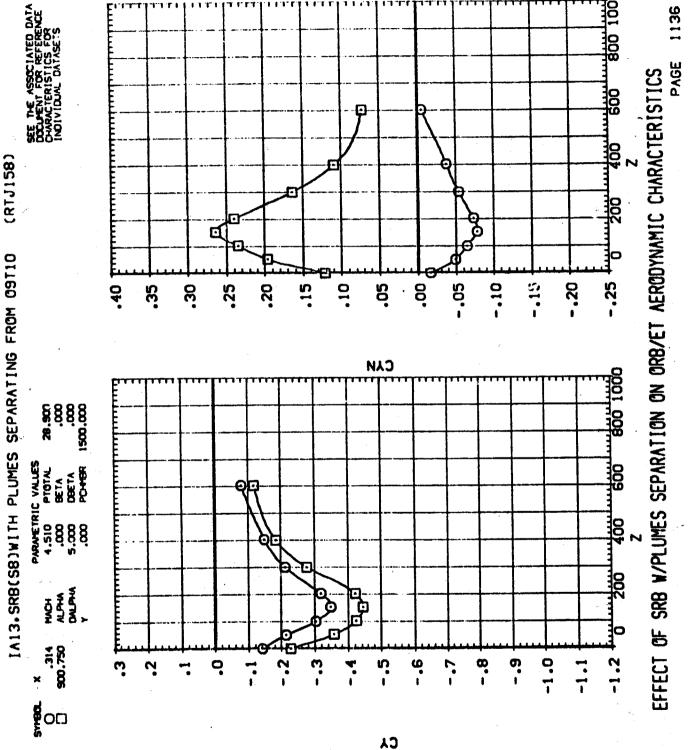
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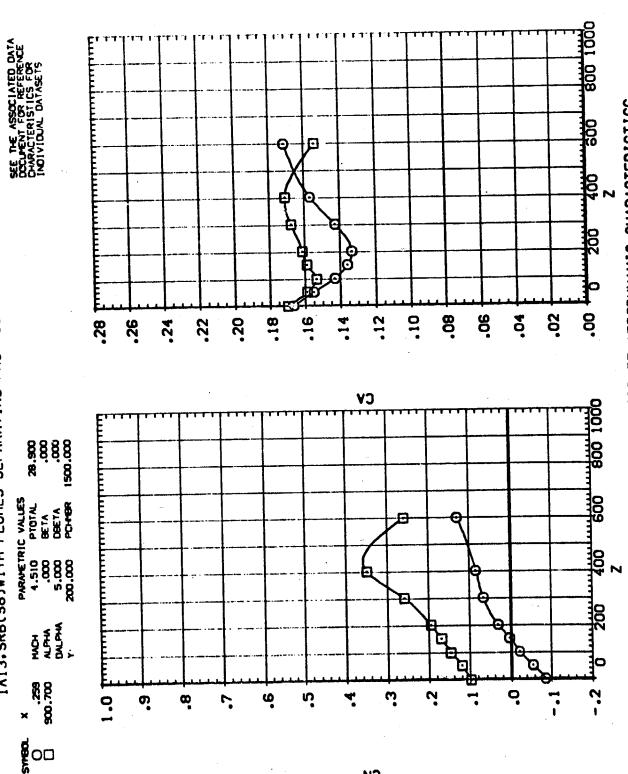
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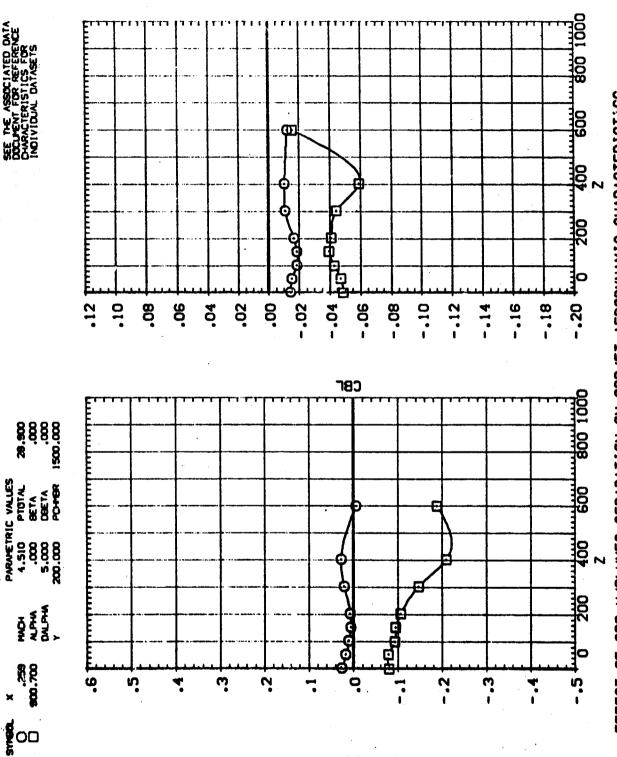
(RTJ159) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710



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(RTJ159) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



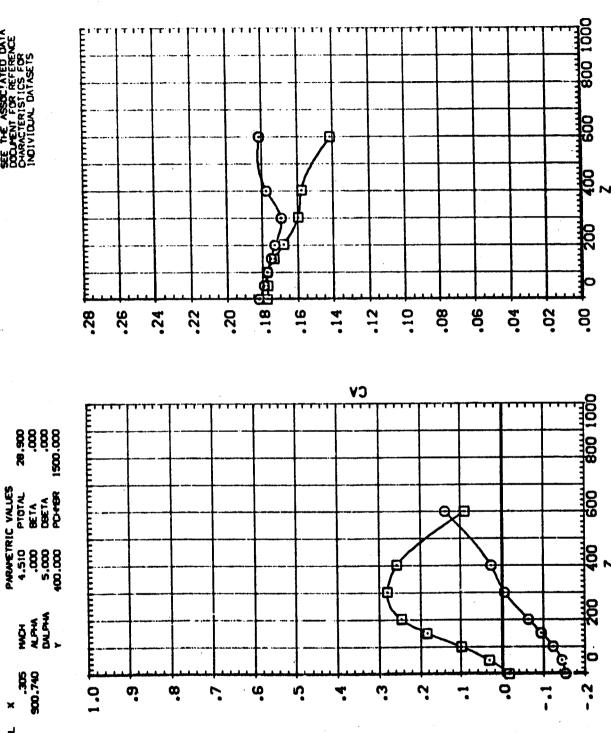
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EFFECT. OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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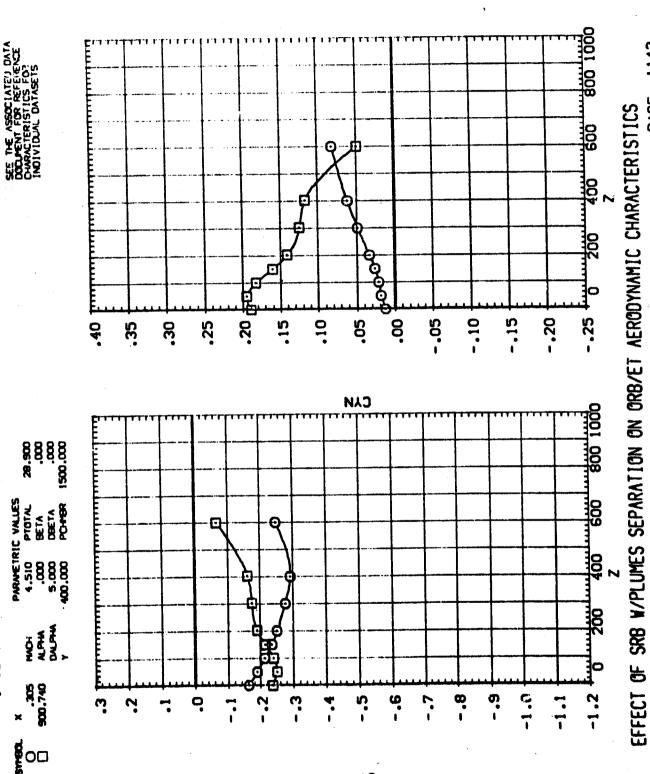




1140 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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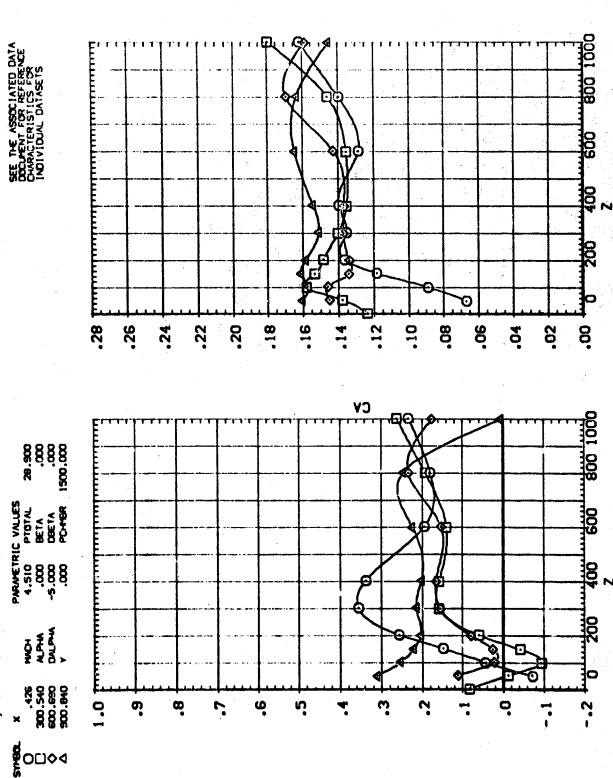


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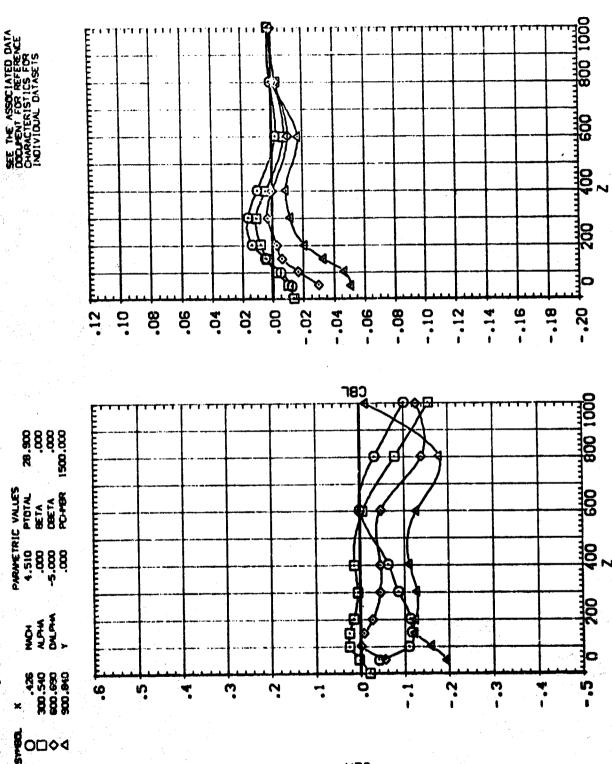
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(RTJ161) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110

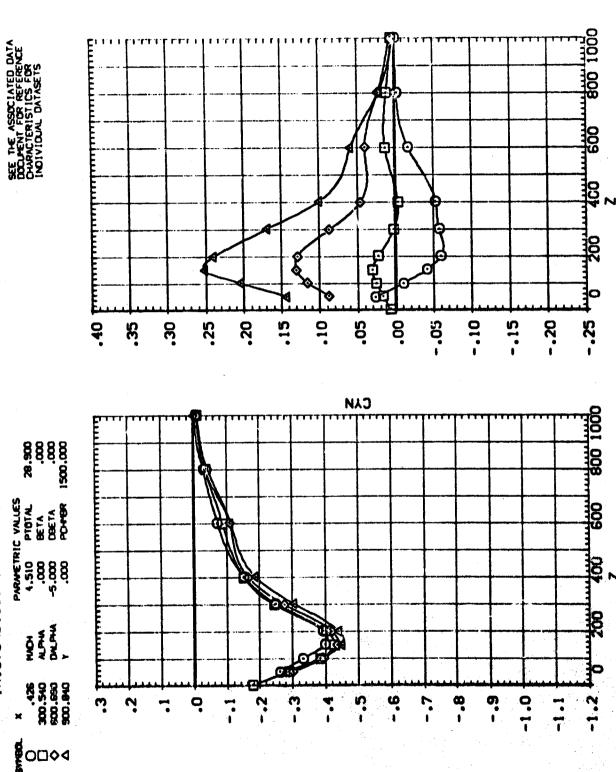


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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

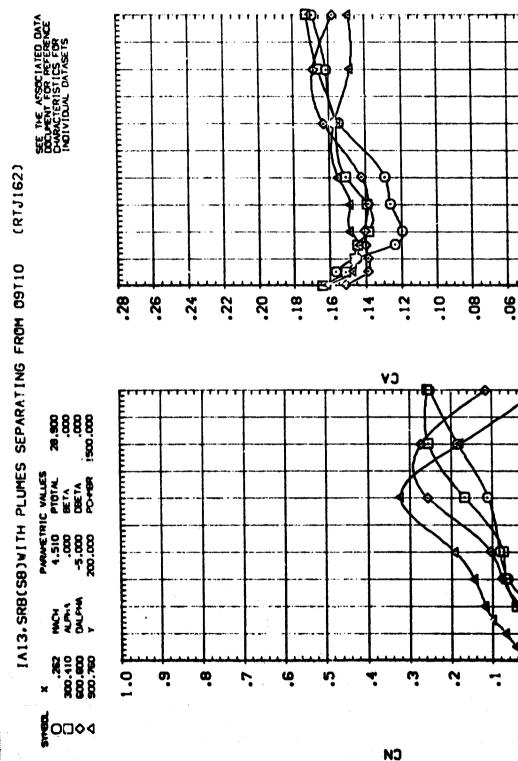
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09T10 (RTJ161)



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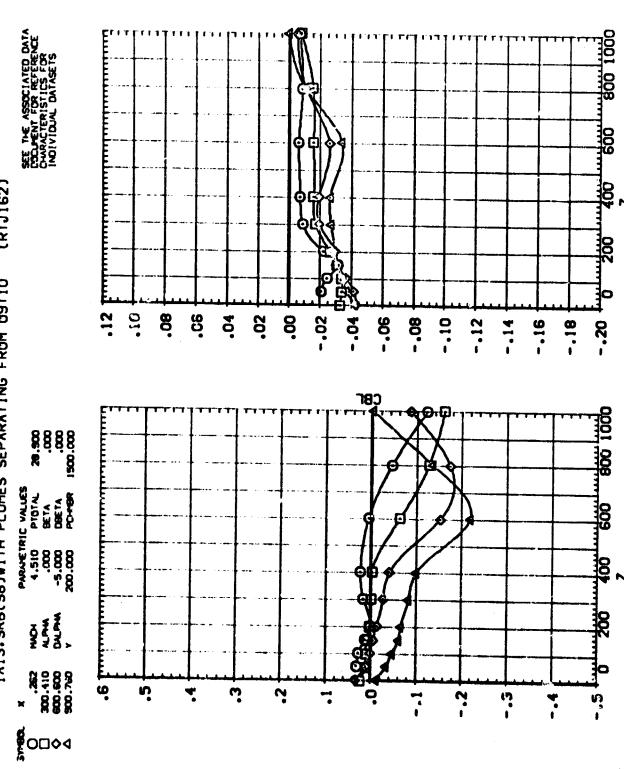
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(RTJ162) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110

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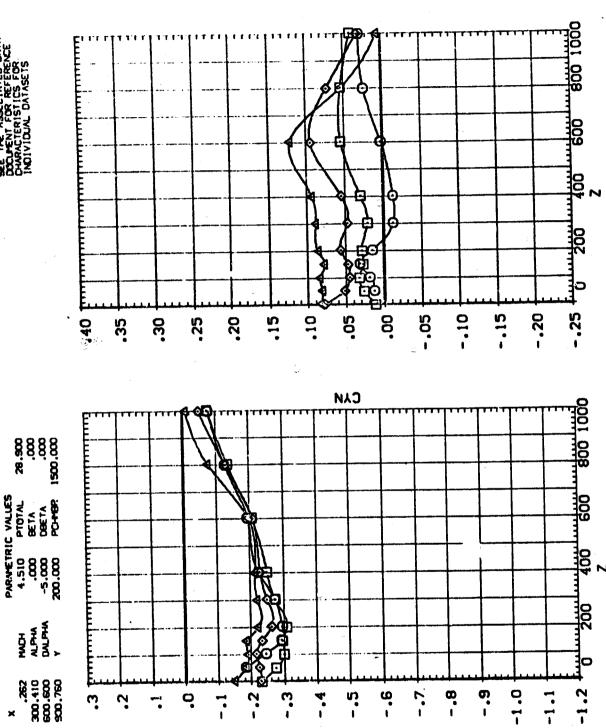
EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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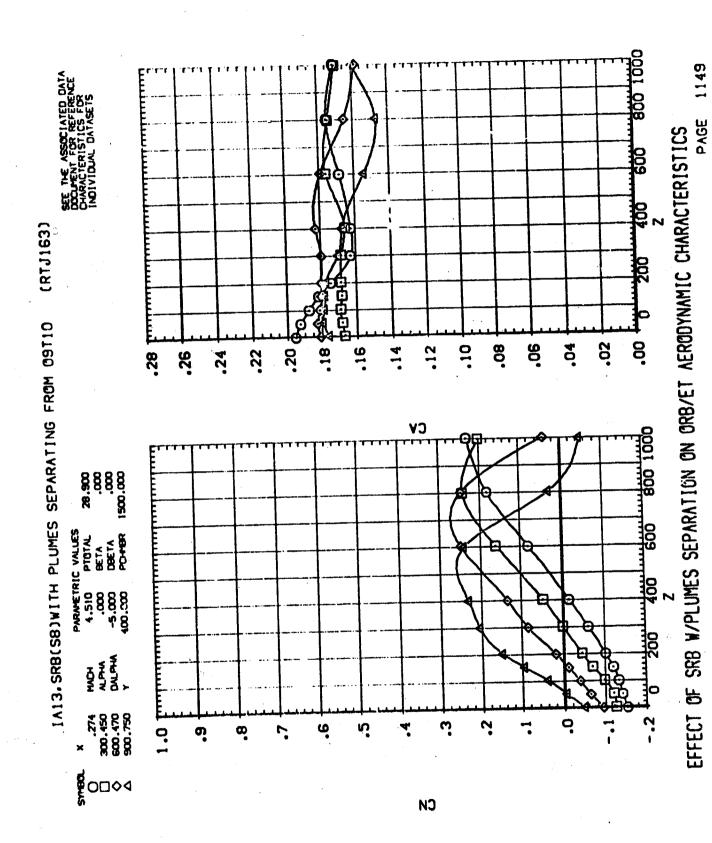
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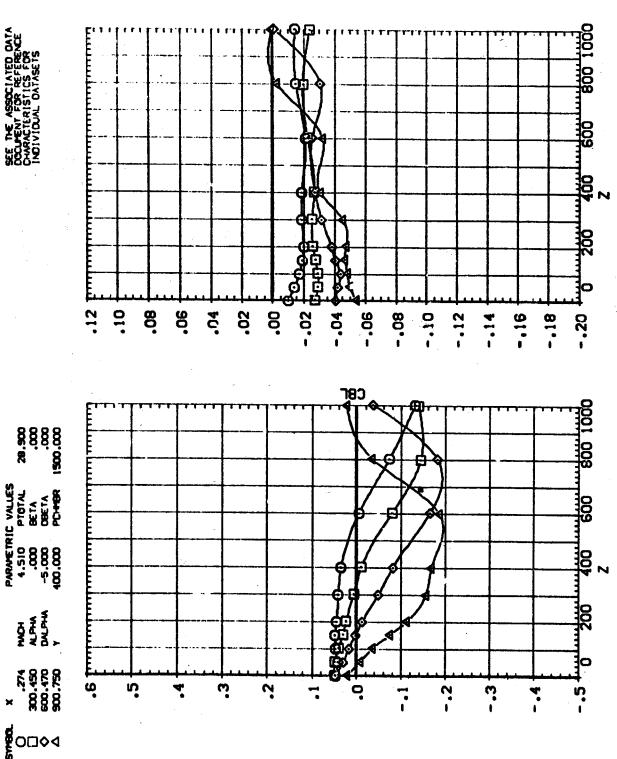




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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ163)

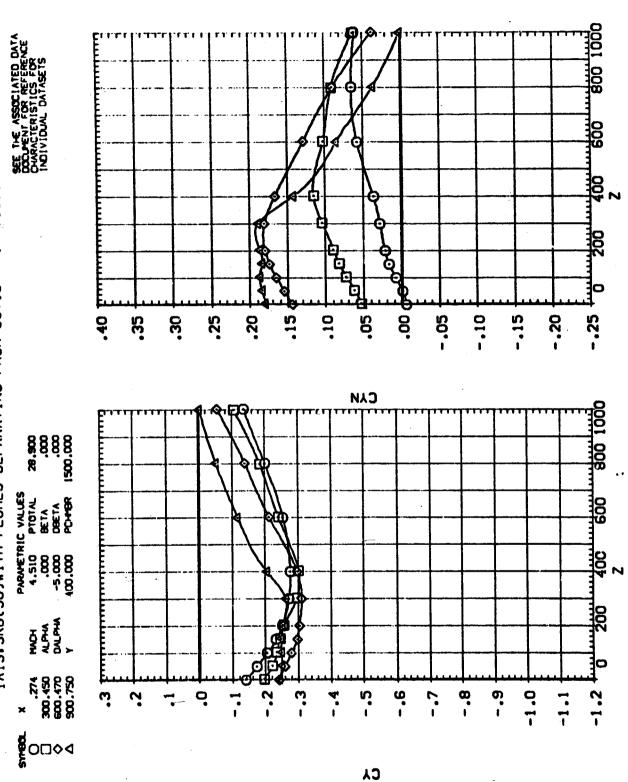


EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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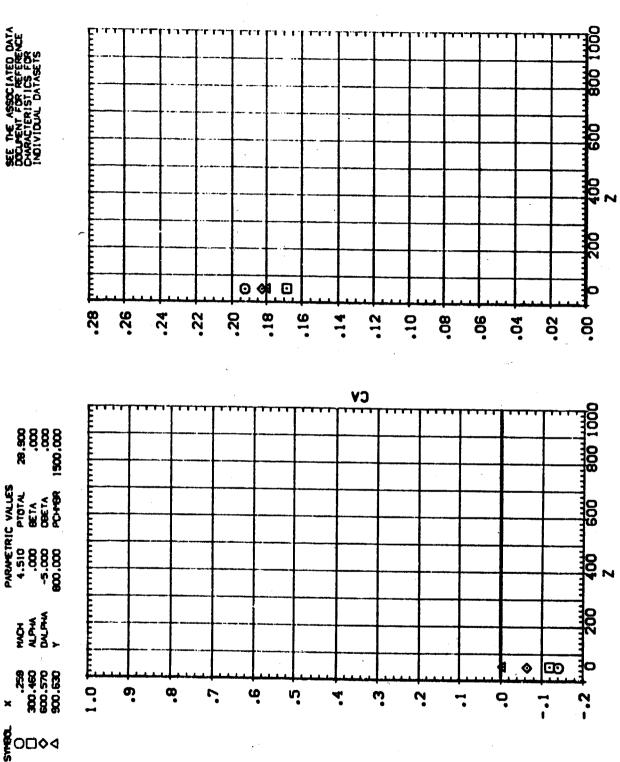
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(RTJ163) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710



EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

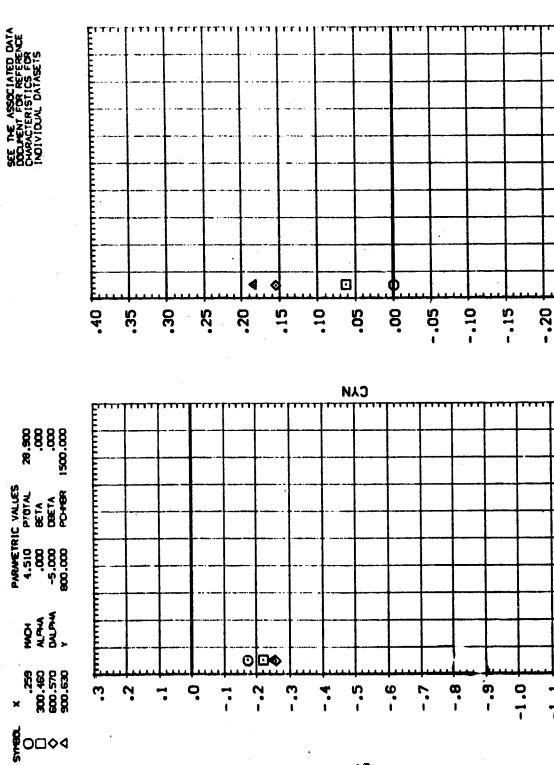
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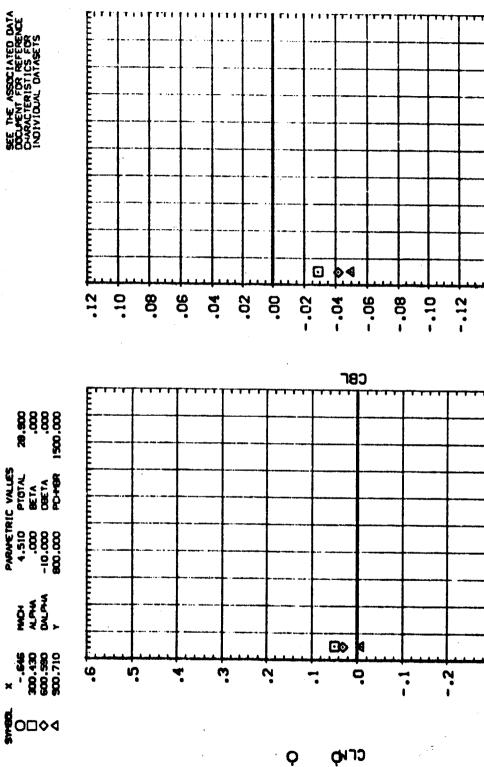
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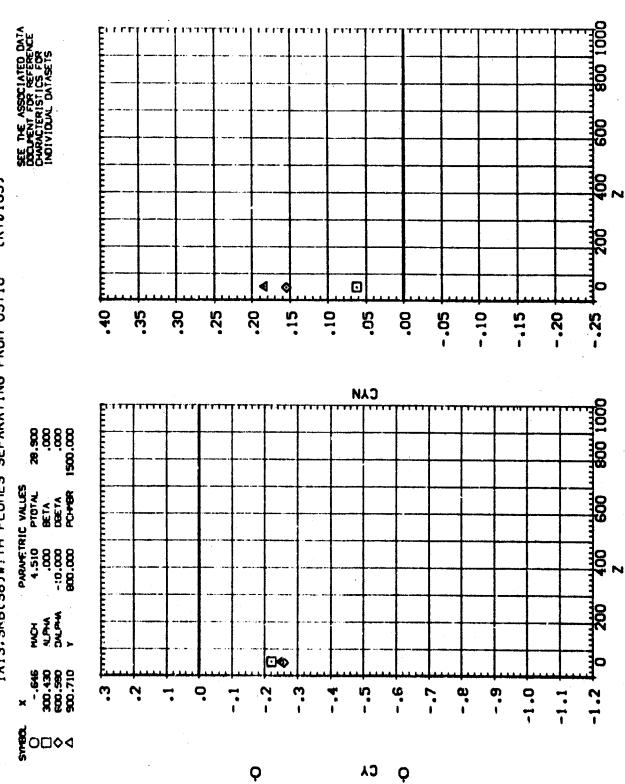
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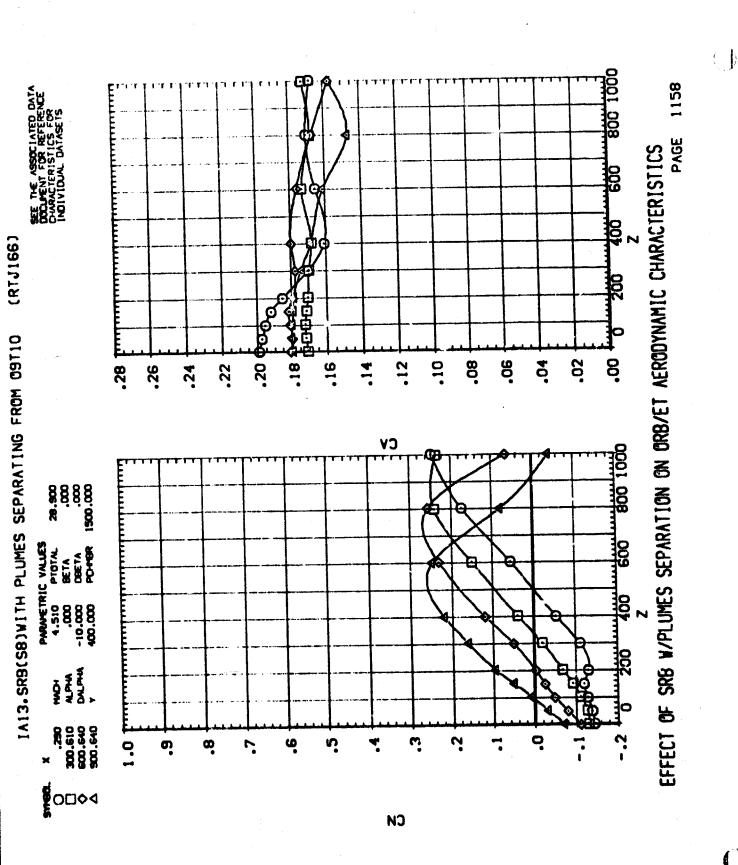
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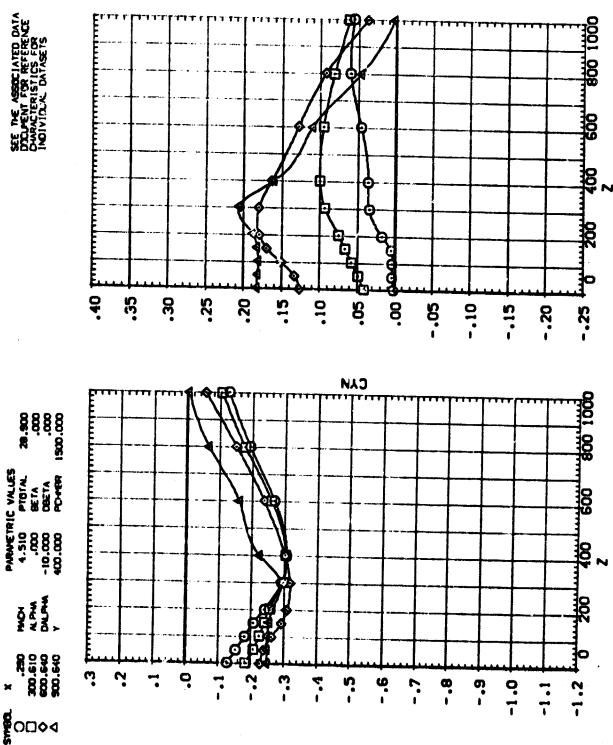
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS



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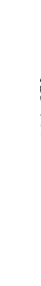


EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

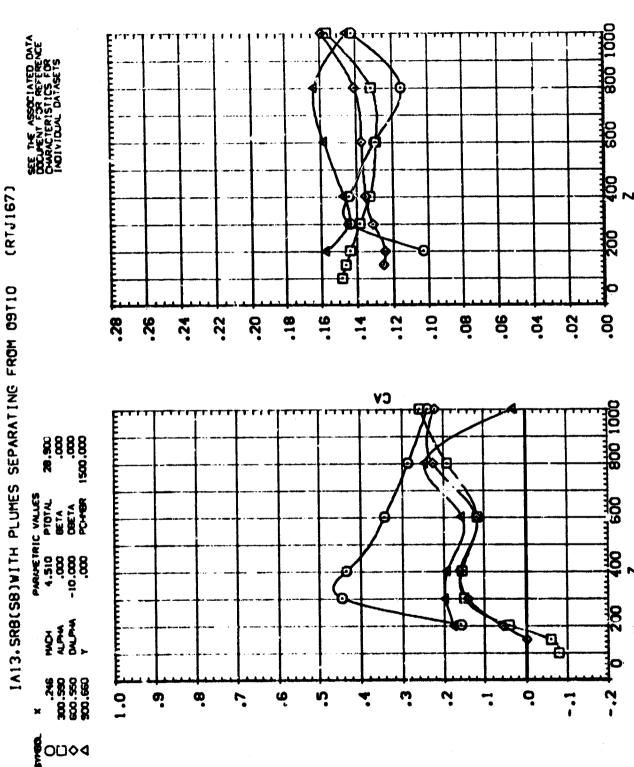
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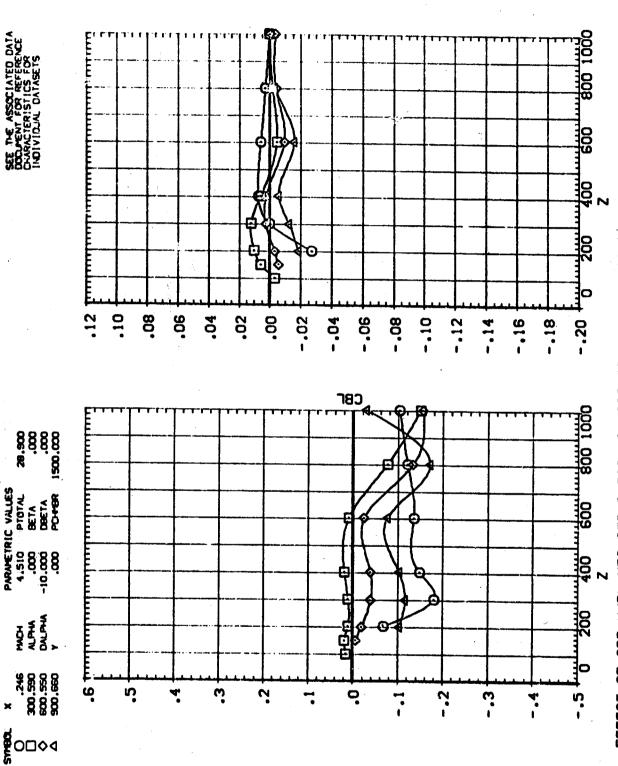


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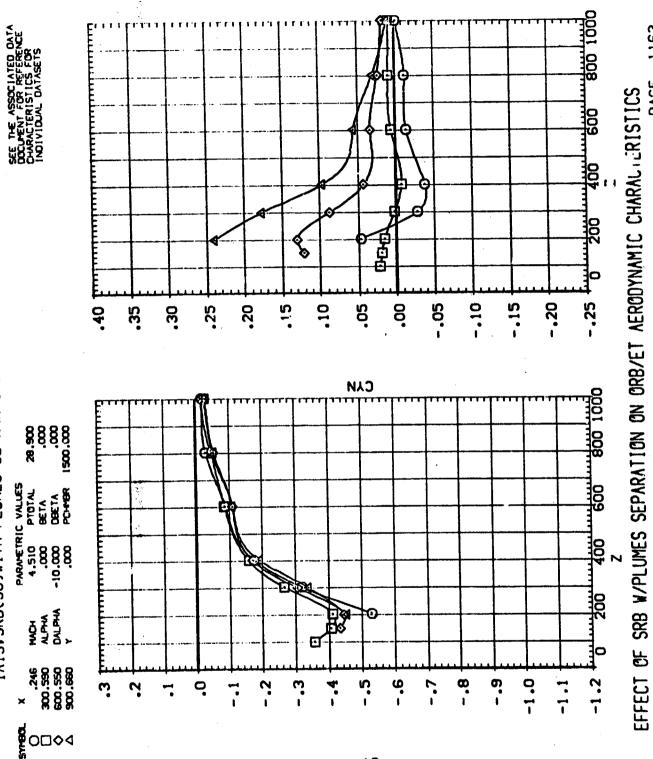
(RTJ167)

IA13. SRB(S8) WITH PLUMES SEPARATING FROM 09110

EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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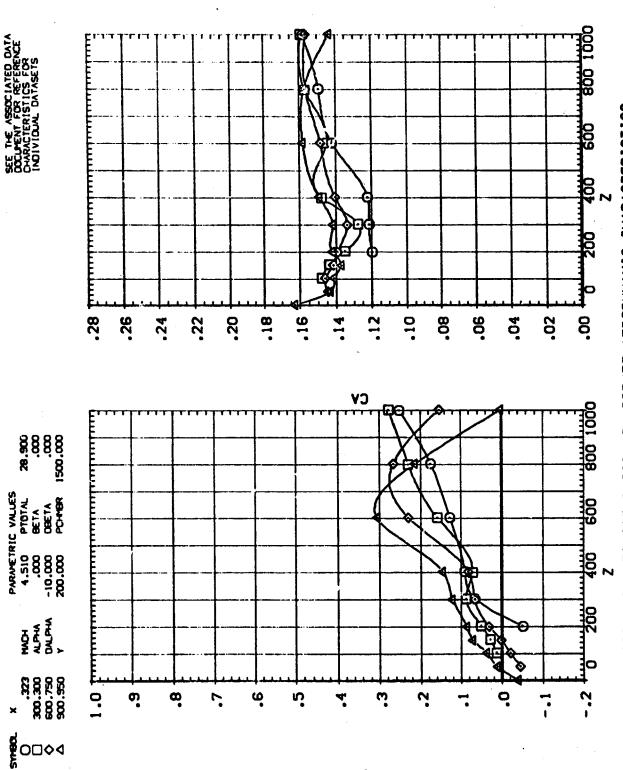
IA13, SRB(SB) WITH PLUMES SEPARATING FROM 09110 (RTJ167)



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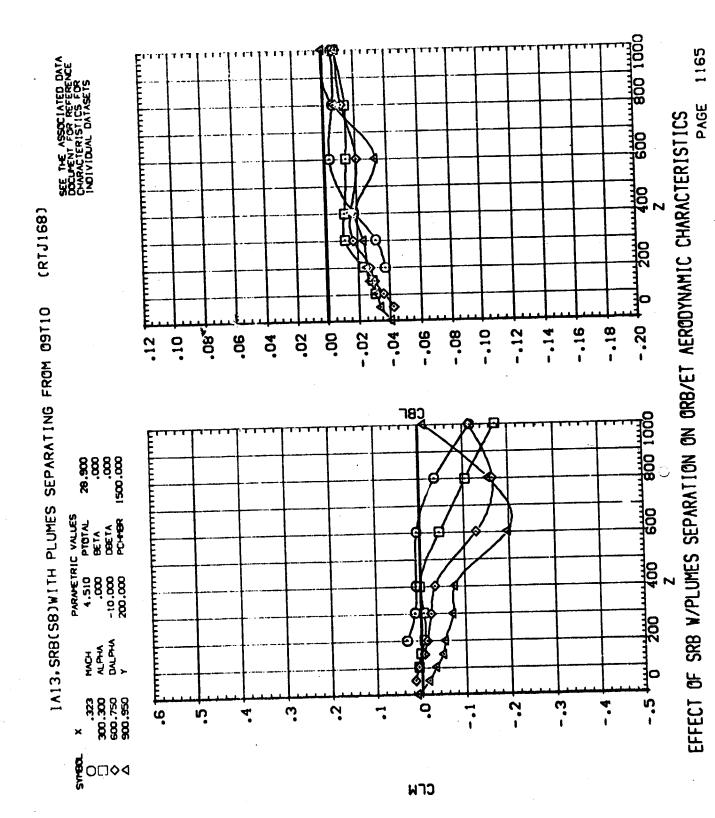
(RTJ168) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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800 1000 EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 400 (RTJ168) 邻 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 o 6 33 .15 -.25 8 29 So. -.05 -.15 .25 .10 8 -.10 -.20 CAM 800 1000 28.900 .000 .000 1500.000 PARAMETRIC VALUES 4.510 PTOTAL .000 BETA -10.000 DBETA 200.000 PCIMBR 900 400 7 MACH M.PHA PALPHA .323 330,300 600,750 900,950 m. .. 5 9. 8 ر. ف ? o -.2 د. 4. -.7 -1.0 -1.2 -: -1.1 **2**0□◊4

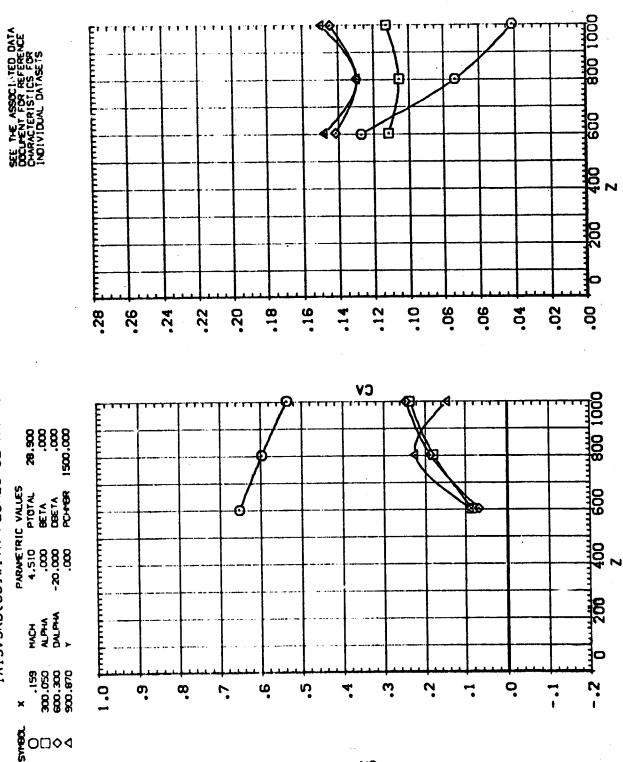
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ169)



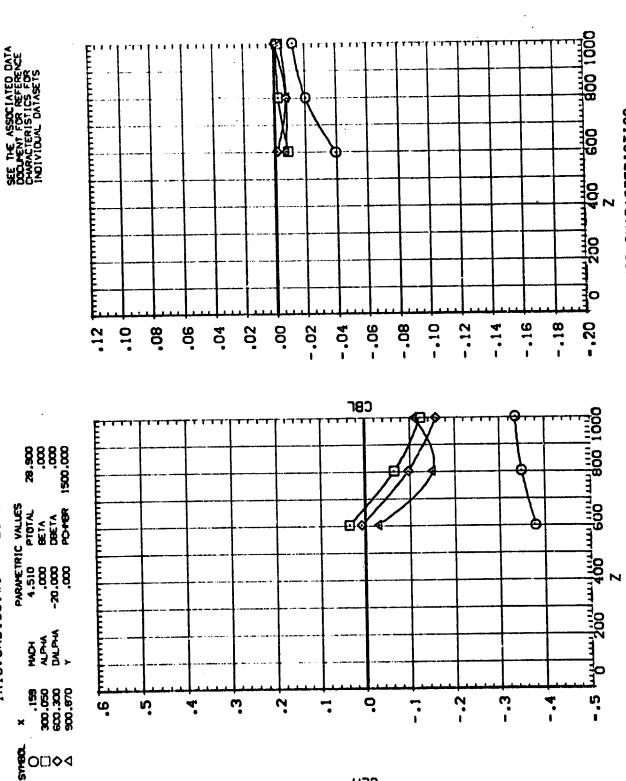
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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1168 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

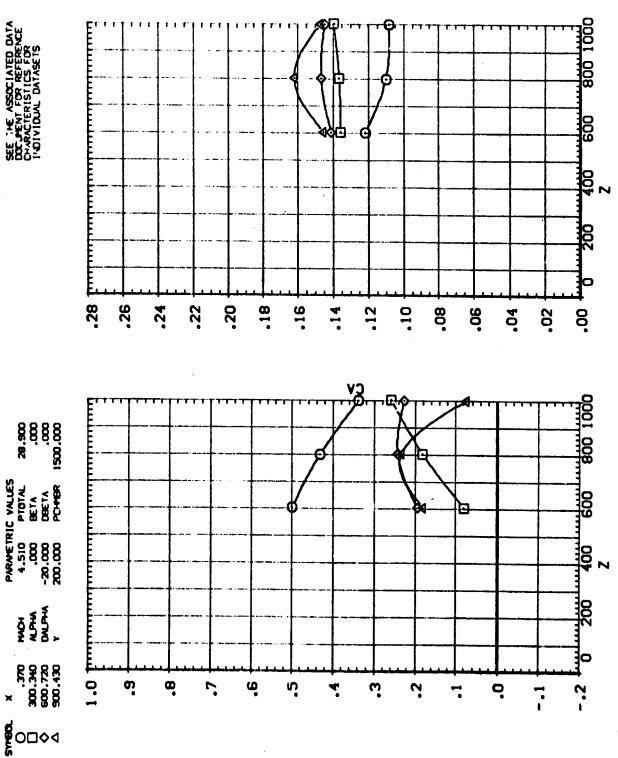
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS



EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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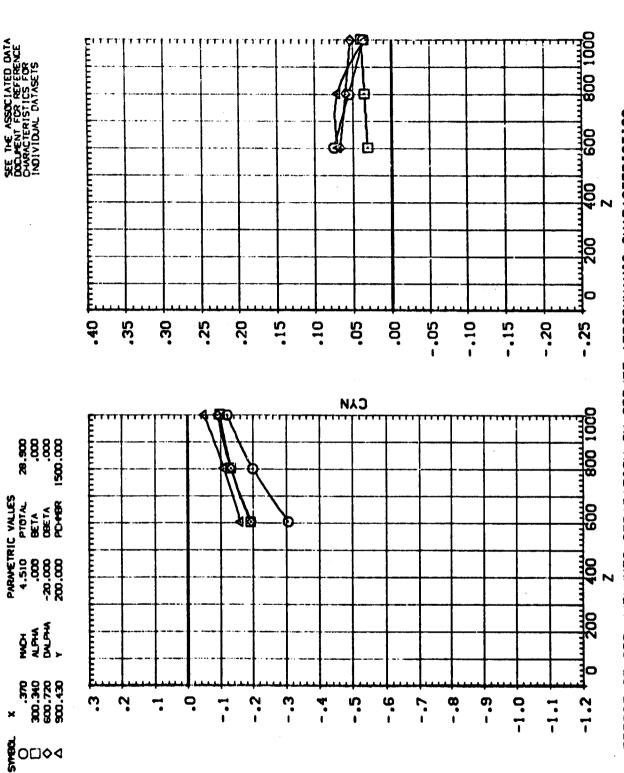
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

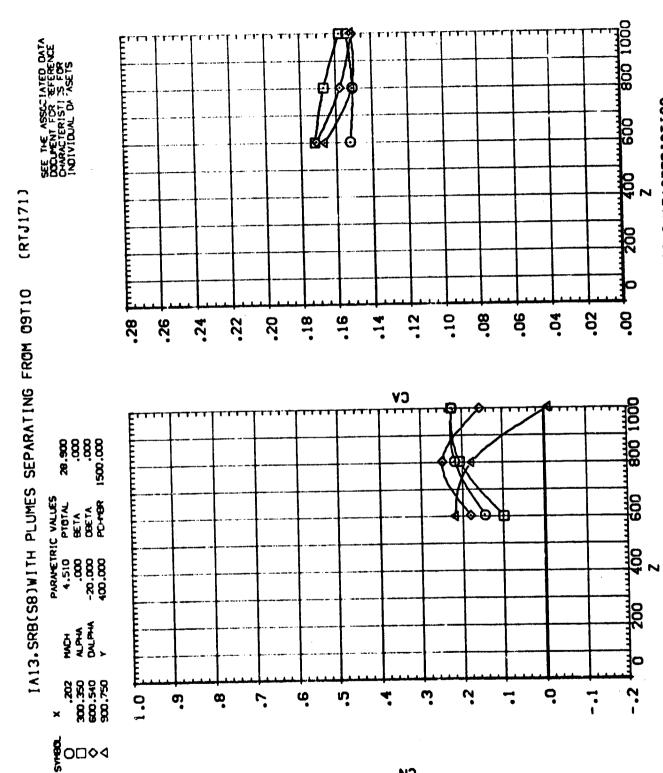
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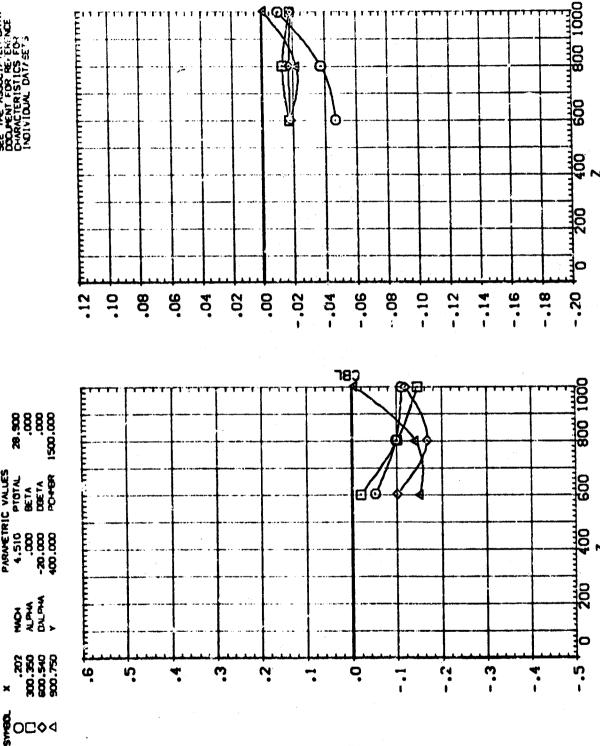


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1173 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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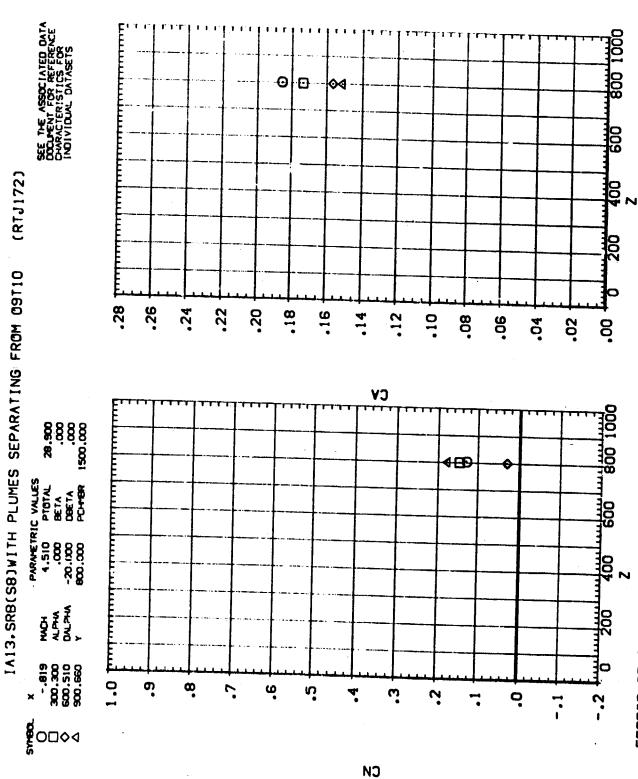
1174 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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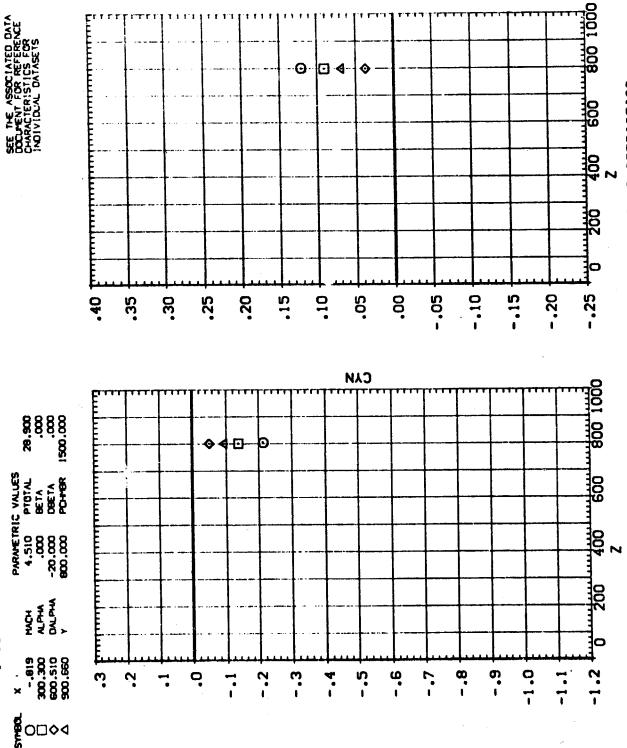
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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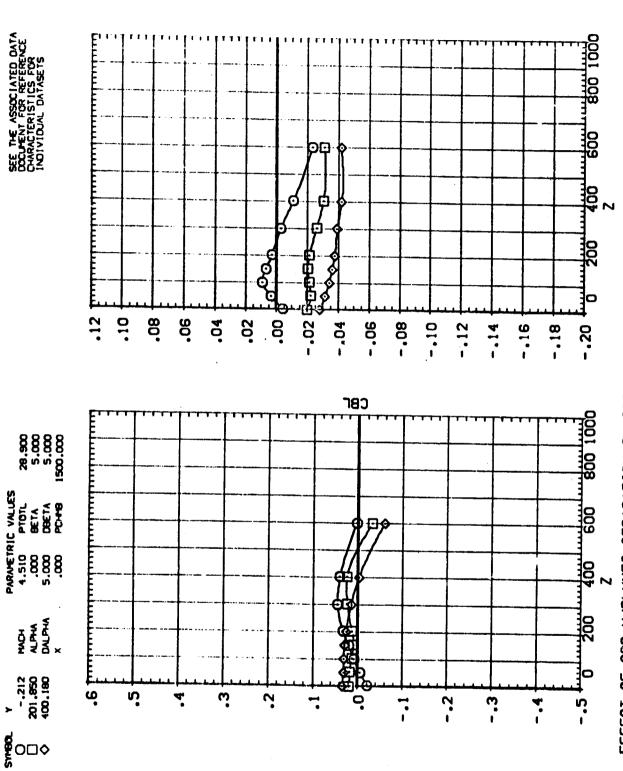
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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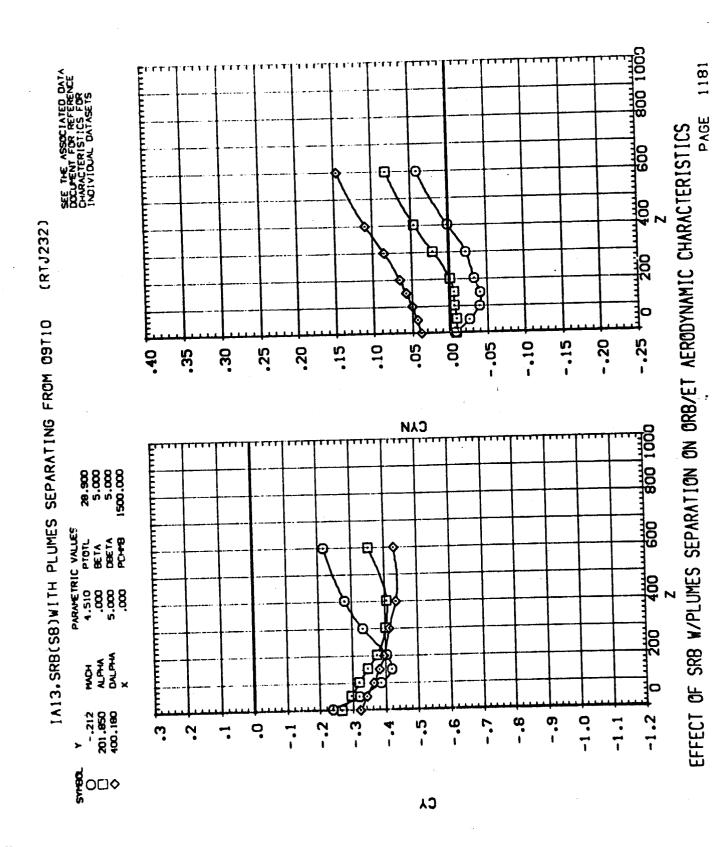


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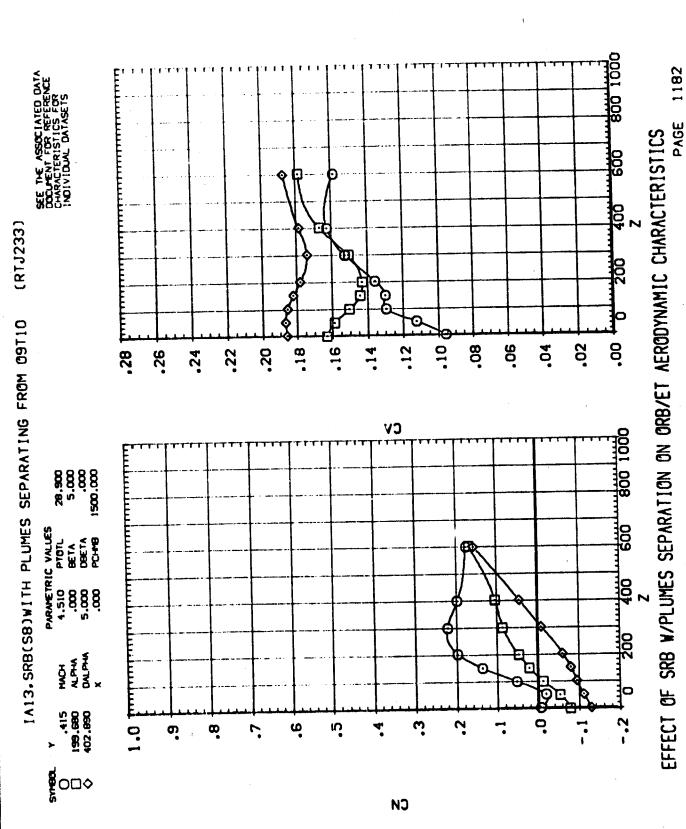
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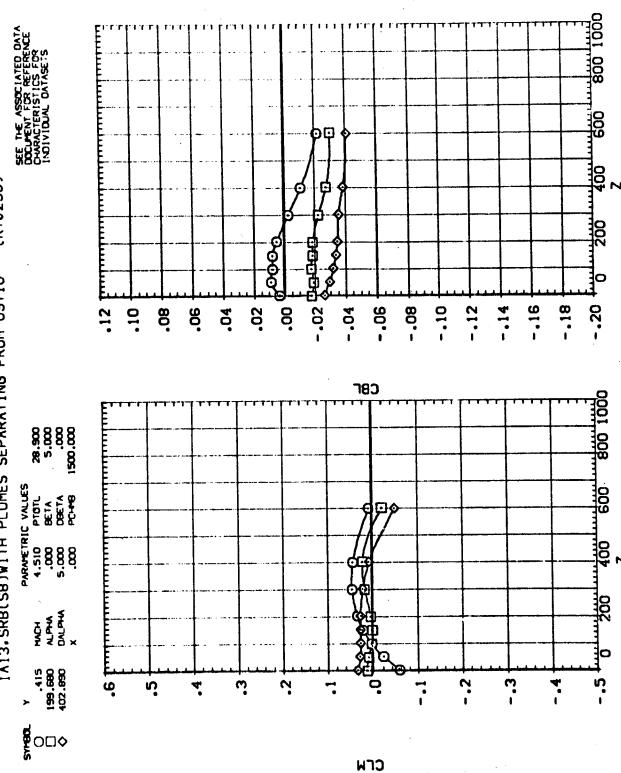


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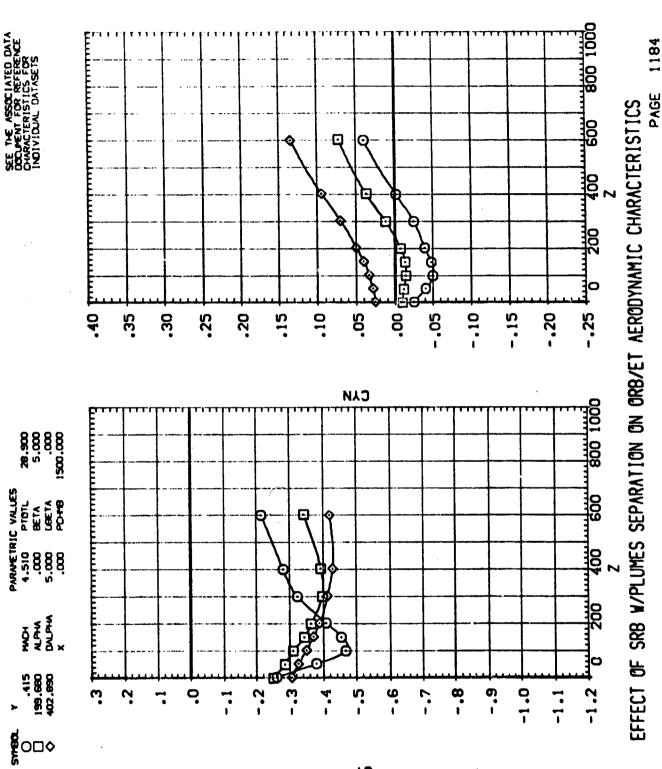
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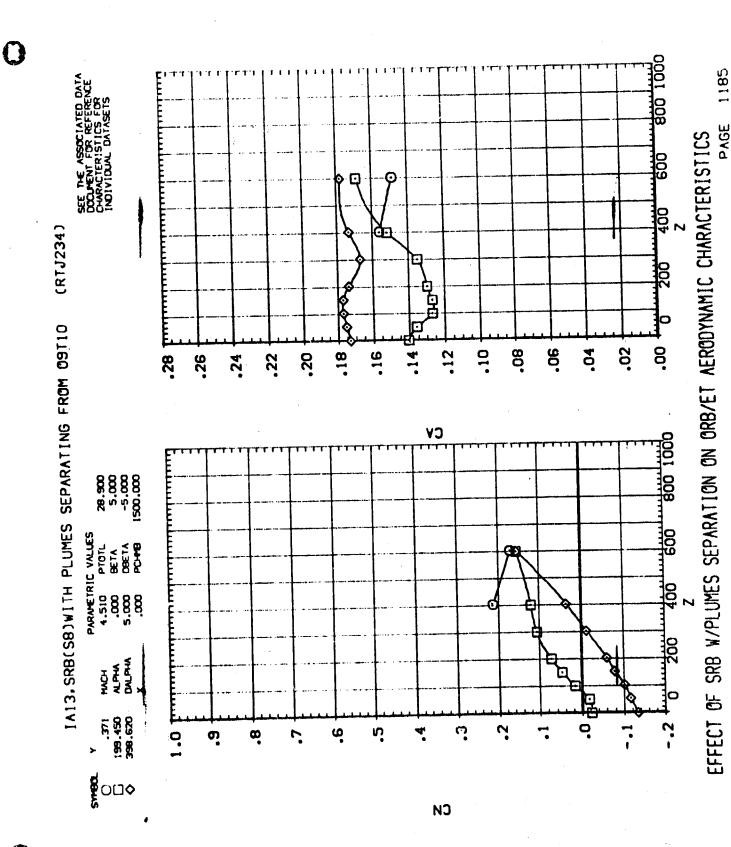
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS





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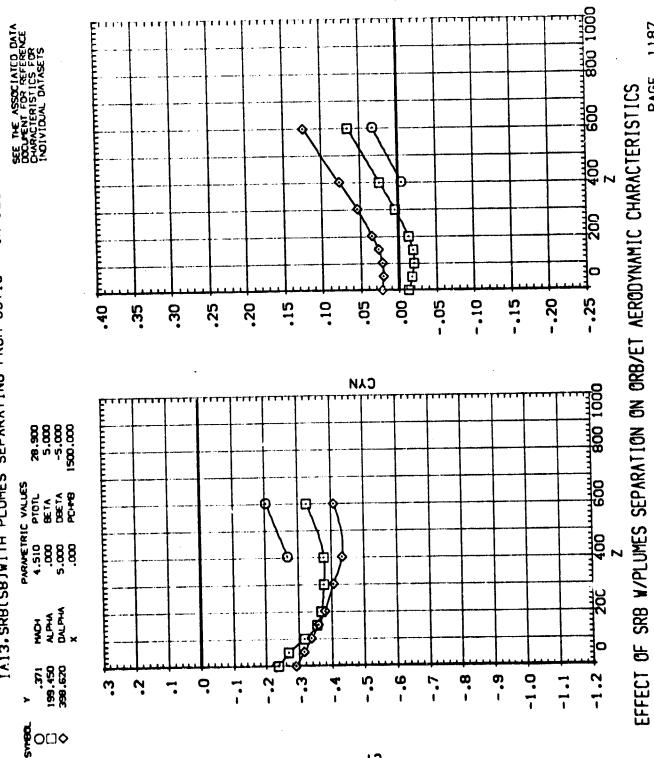
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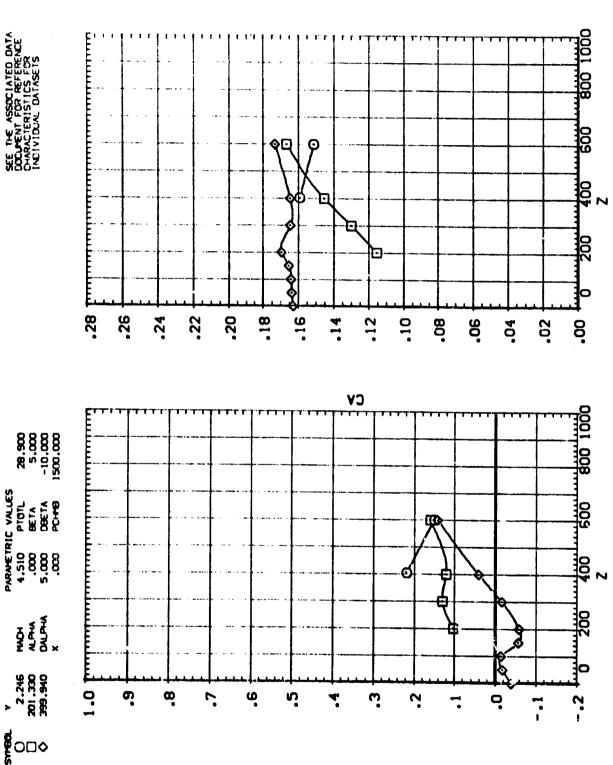
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110

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(RTJ234) IA13, SRB(SB)WITH PLUMES SEPARATING FROM 09110

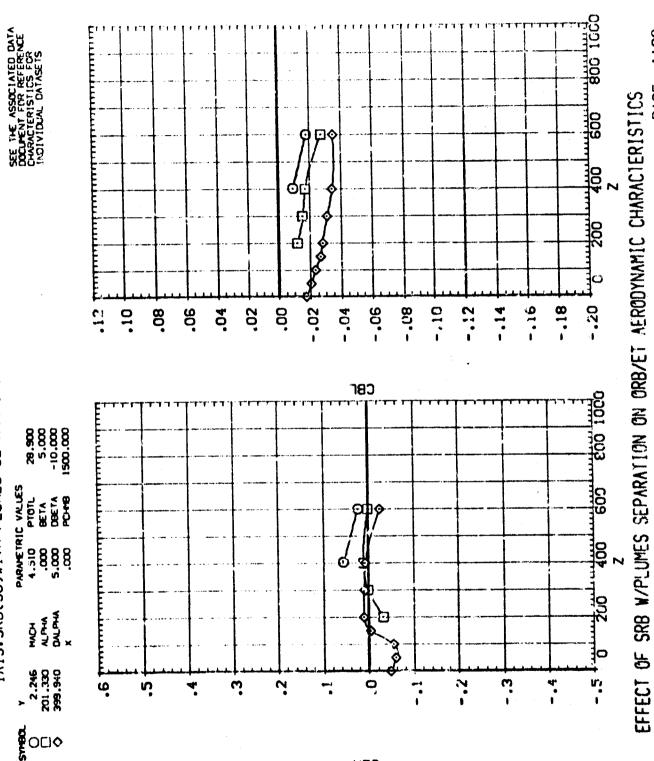


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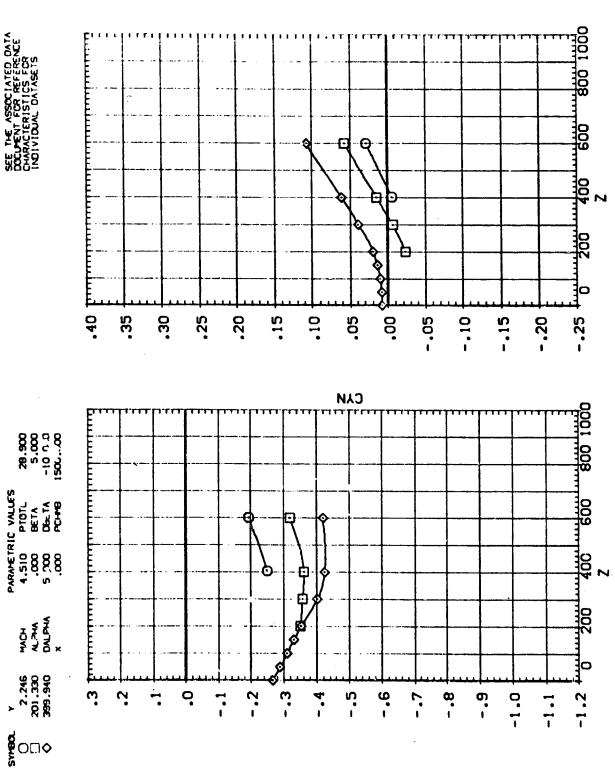


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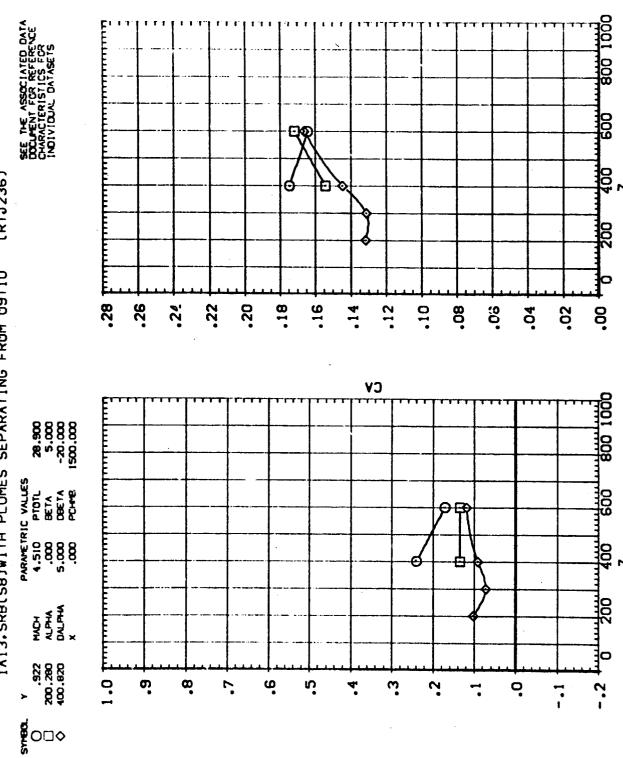




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(RTJ236) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110

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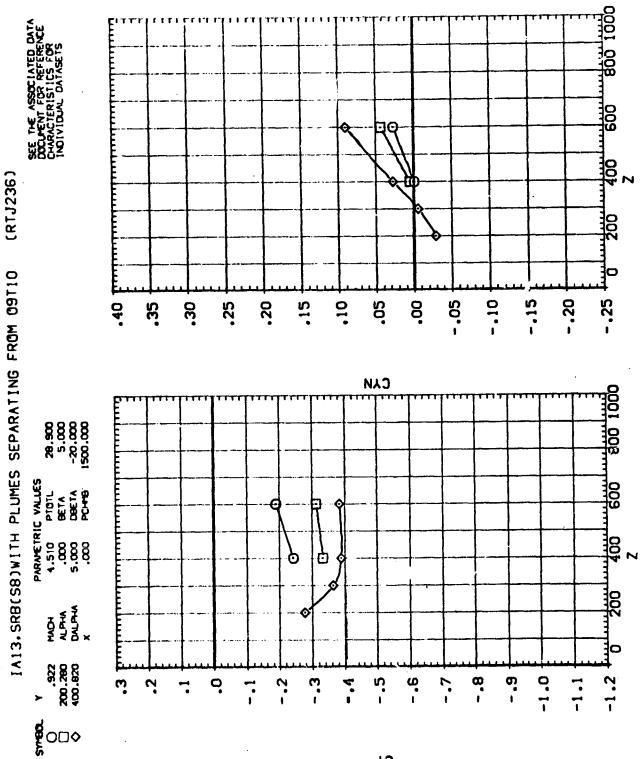
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5.000 DETA
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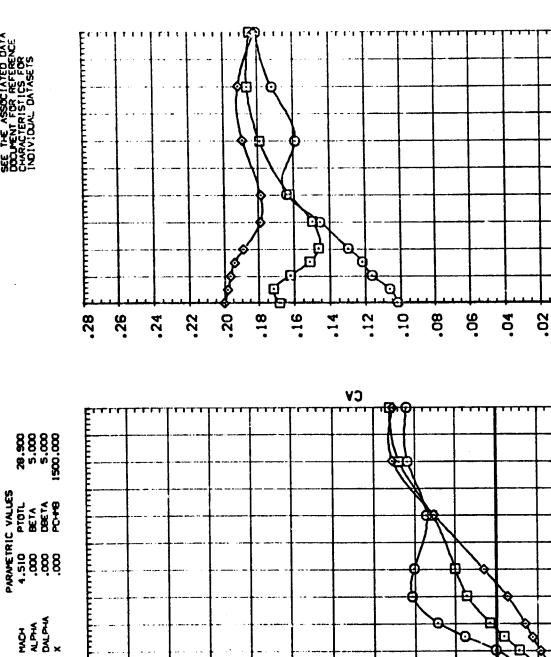
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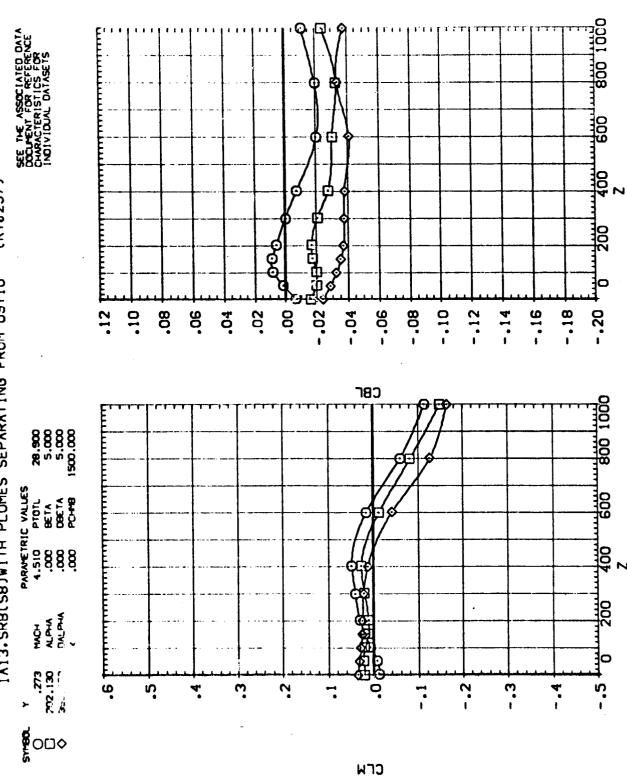
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(RTJ237) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110



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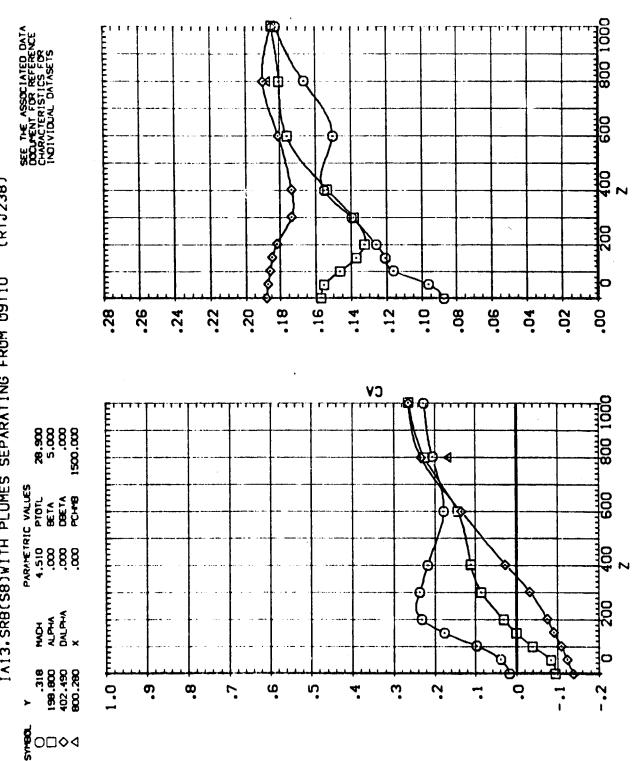
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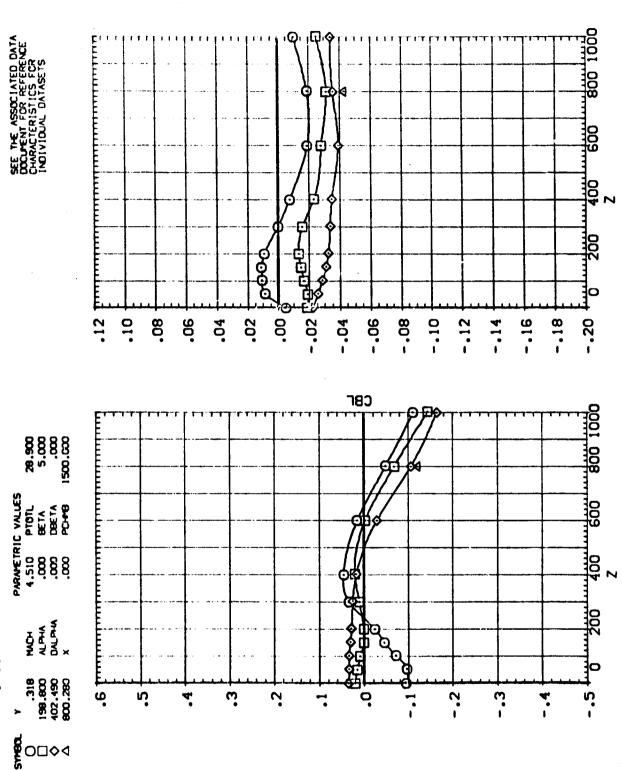
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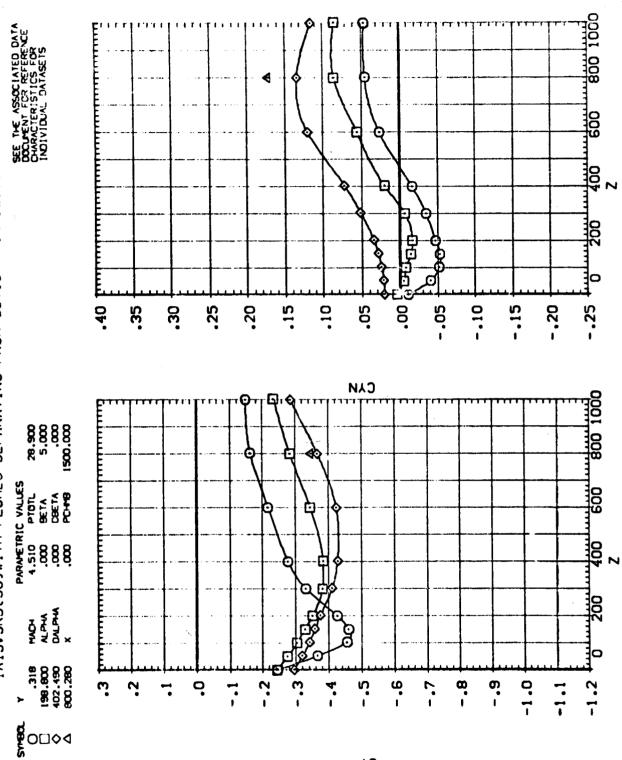
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ238)



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IA13.SRB(S8)WITH PLUMES SEPARATING FROM 09110 (RTJ238)

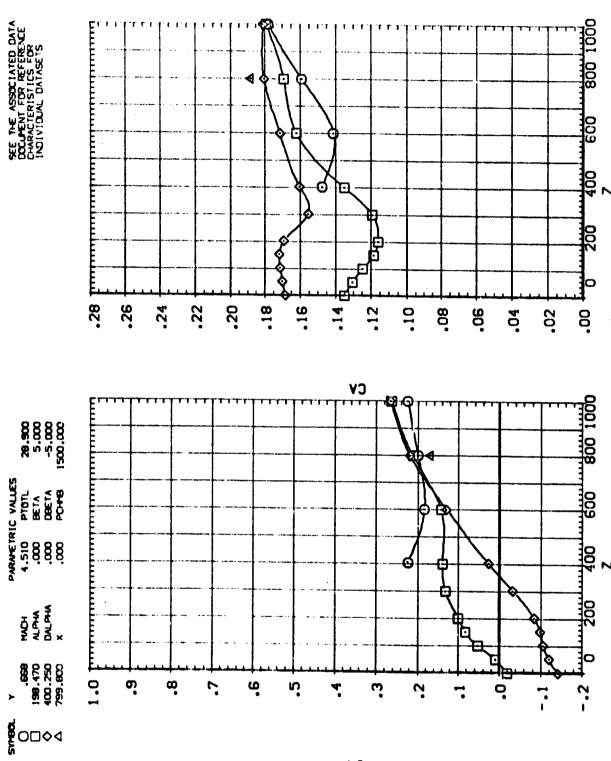


EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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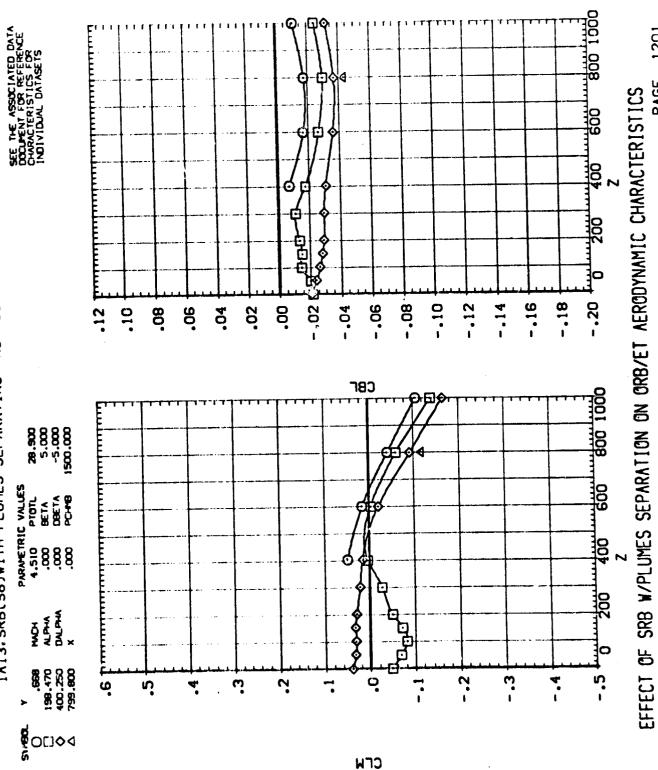
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ239)



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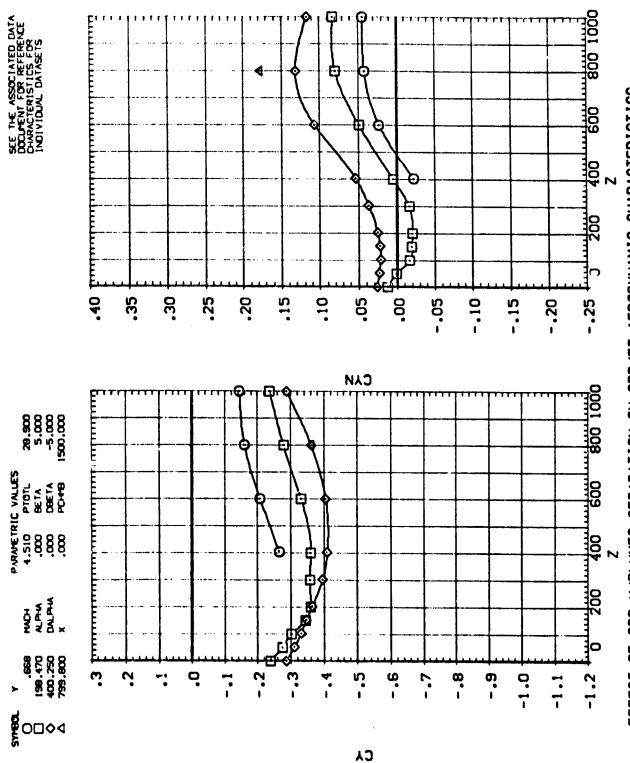
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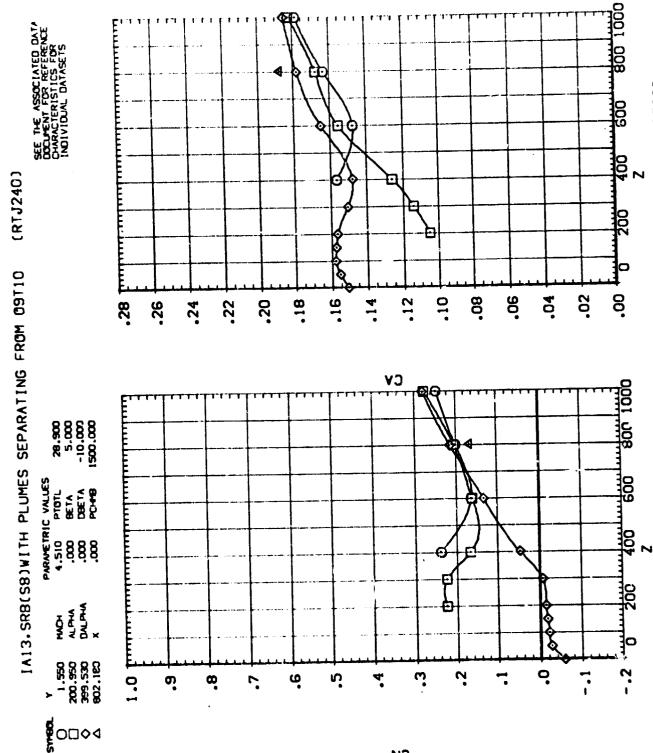
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(RTJ239) SEPARATING FROM 09110 IA13, SRB(SB)WITH PLUMES



EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

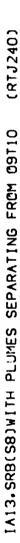


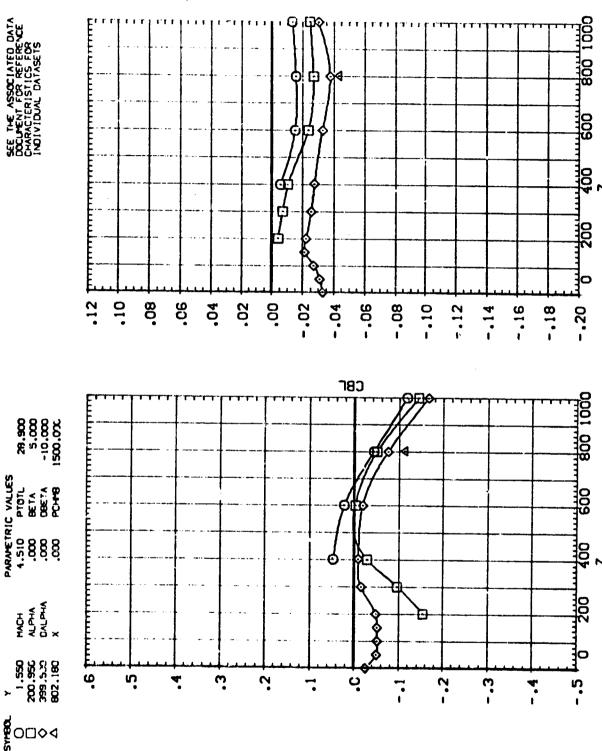
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CHARACTERISTICS FOR
INDIVIDUAL DATASETS 400 2 (RTJ240) 200 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 -.25 -.20 . S -.15 .15 .10 8 -.05 -.10 40 .35 8 .25 .20 CAM 800 1000 28.900 5.800 -10.900 1500.000 PARAMETRIC VALUES
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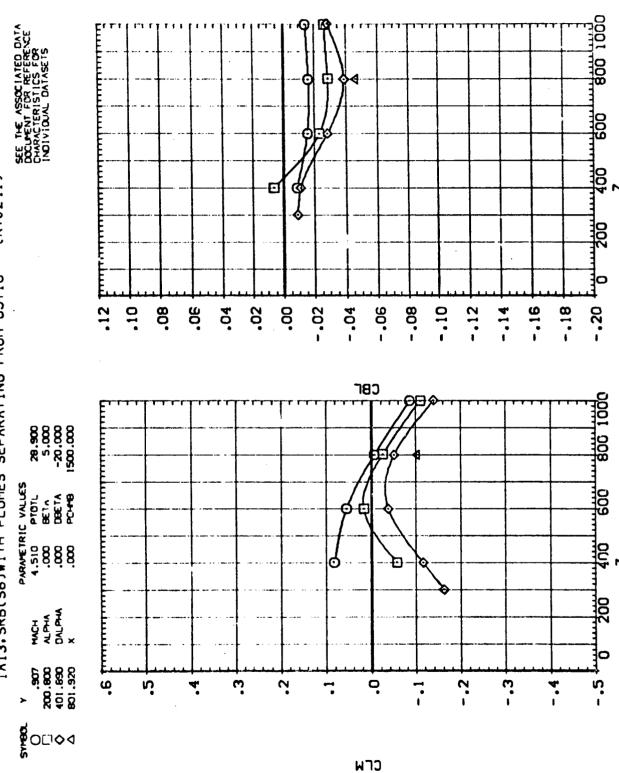
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ241)



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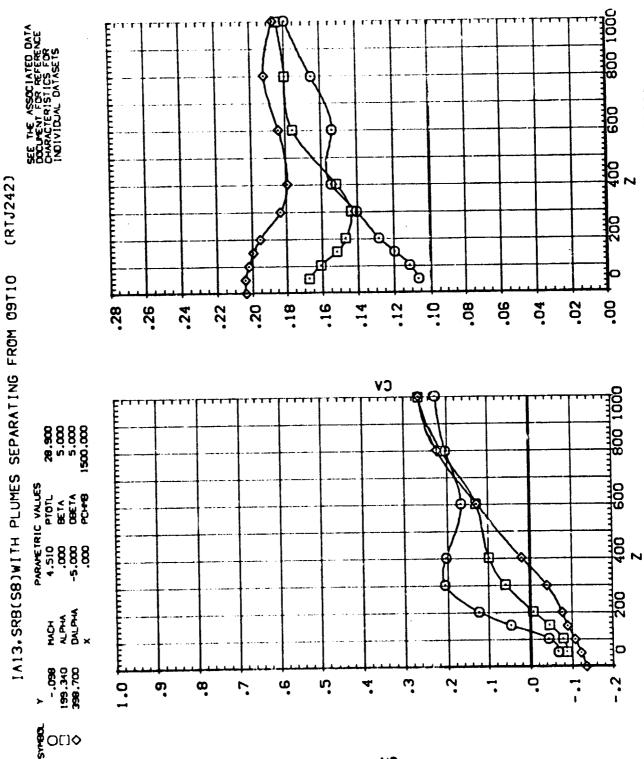
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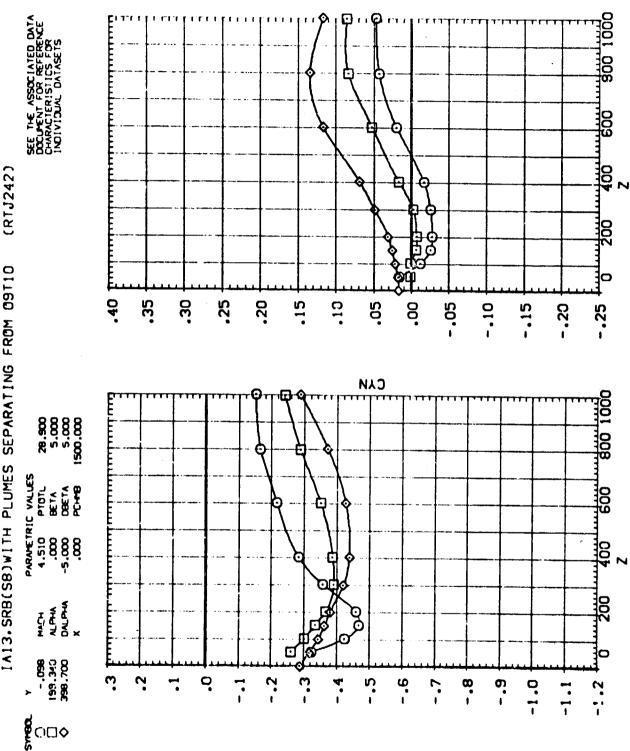
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IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110



EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

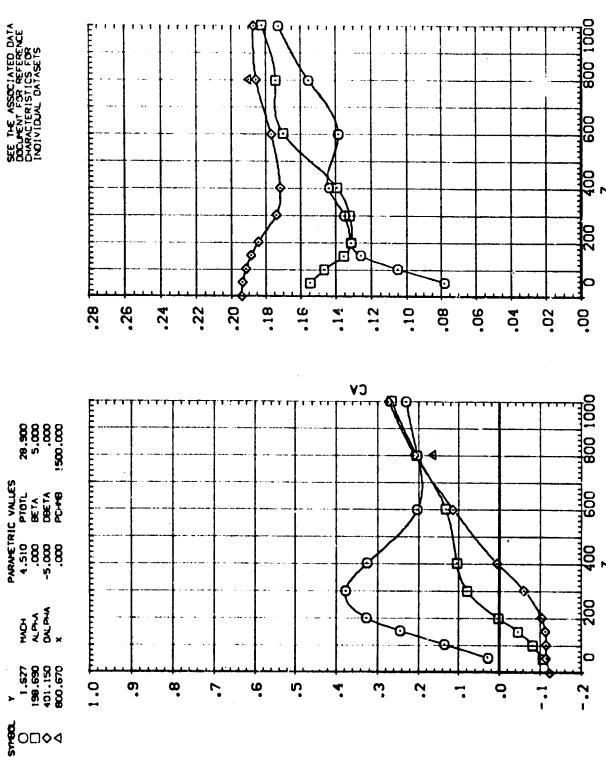
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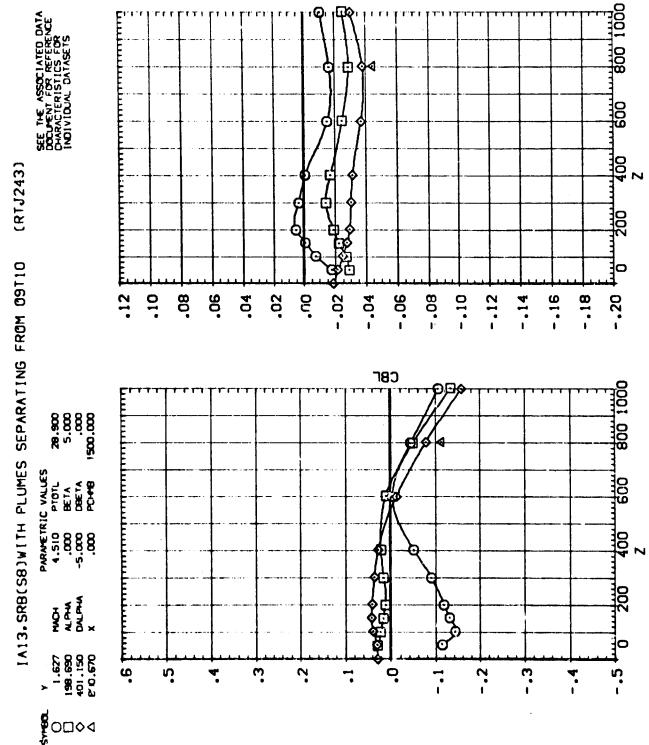
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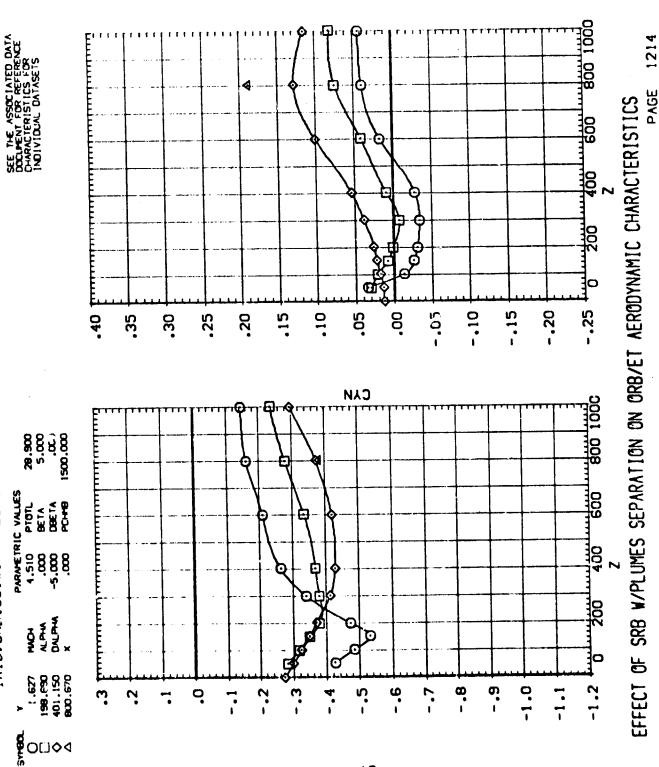


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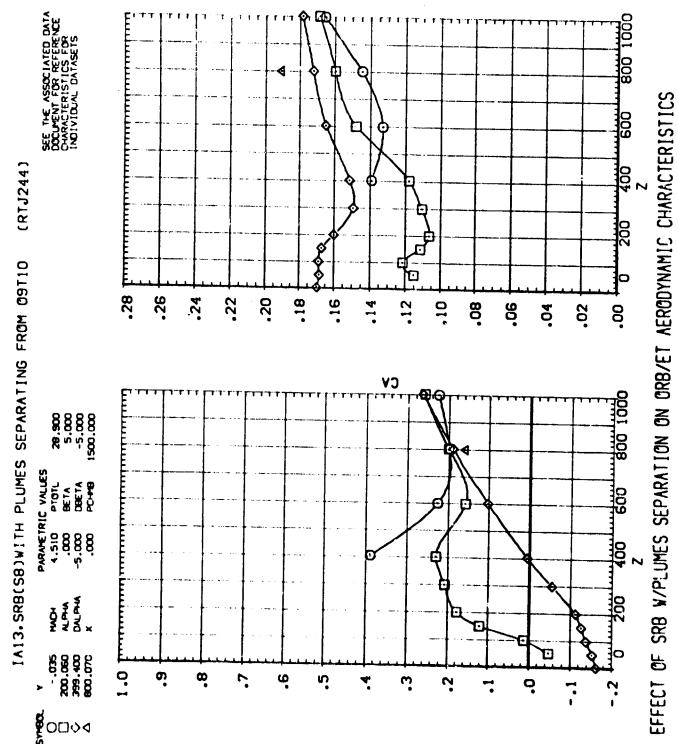
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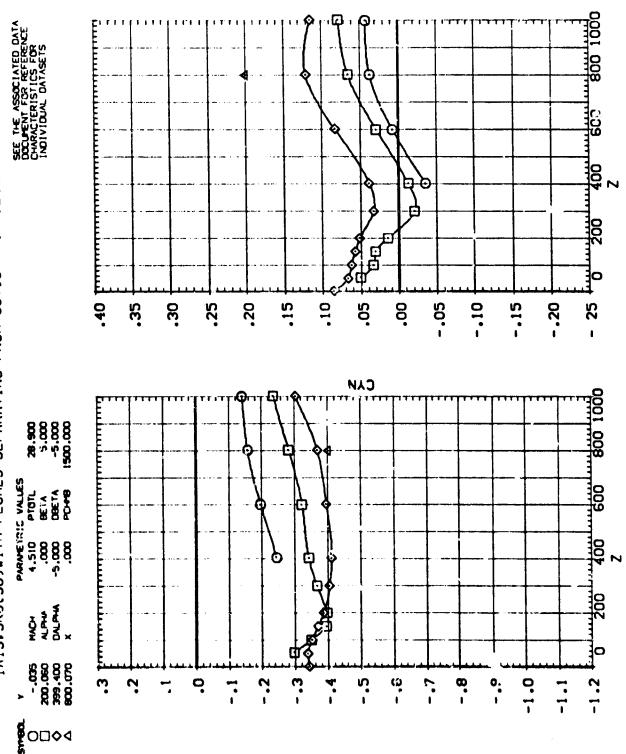
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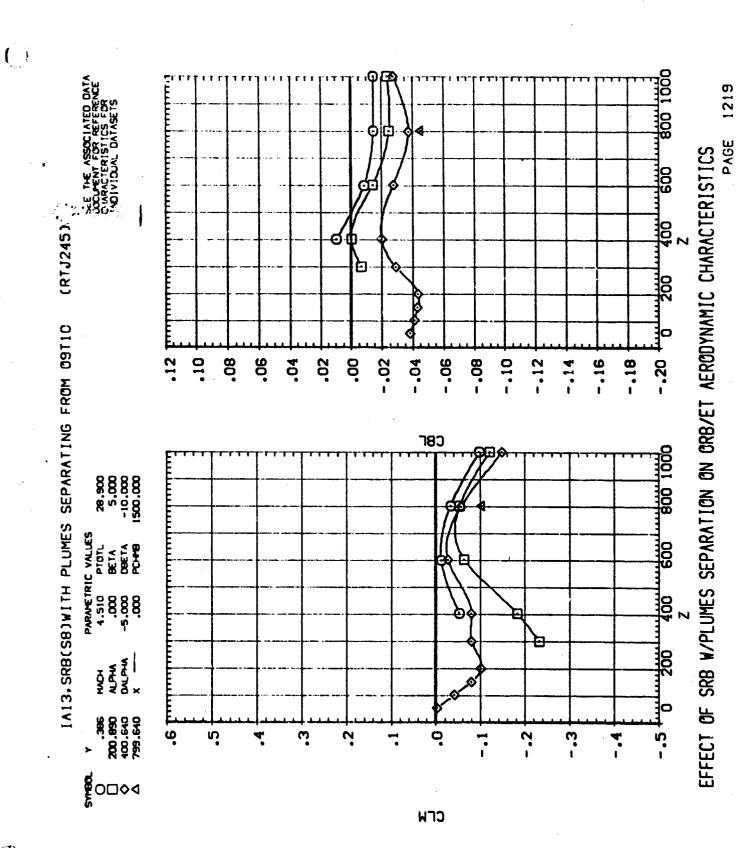
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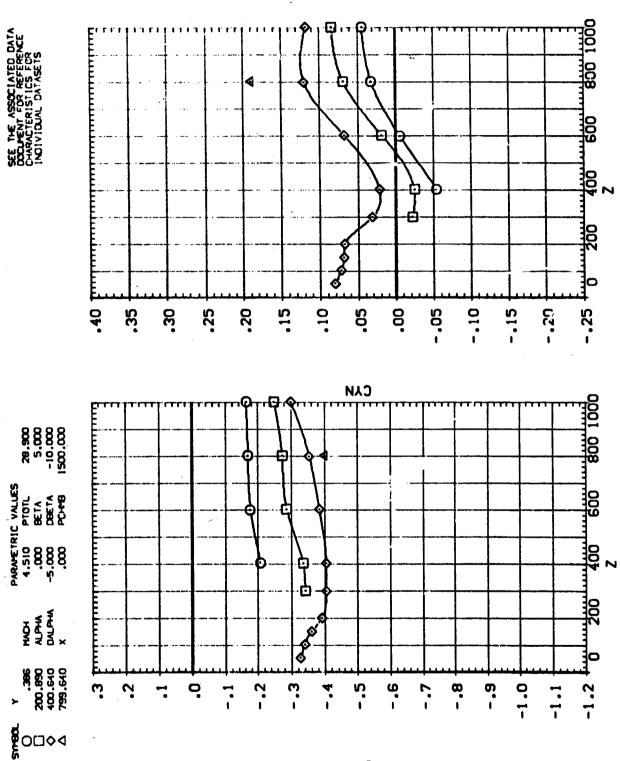
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ245)



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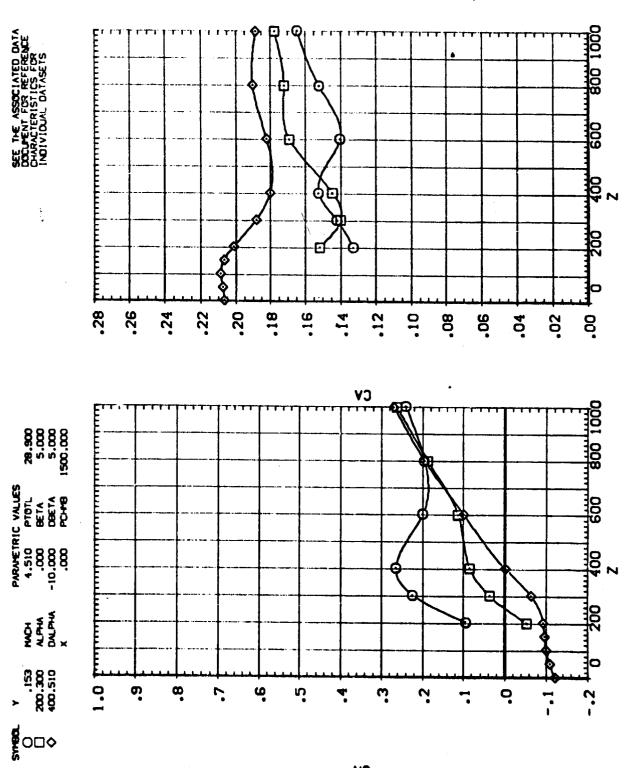
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

AGE 1223

IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ247)



1224 PAGE EFFECT of SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

800 1000 SEE THE ASSOCIATED DATA
DOCUMENT FOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS 100 18₇ (RTJ247) 器 0 SEPARATING FROM 09110 90.--.12 -.16 - 08 -.10 -.14 -.18 .02 -.02 .10 8 .06 **.** 8 .12 -.04 CBF 800 1000 5.000 5.000 5.000 1500.000 IA13, SRB(SB)WITH PLUMES PARAYETRIC VALUES
4.510 PTOTL
.000 BETA
-10.000 OBETA
.000 PCH48 600 99 祭 KACH ALPHA DALPHA **19**000 0 .153 200.300 400.510 Ö 6.1 ທຸ -.2 4. ဖ္ 'n က္ 7 -٦. **%** ○□◇ CLM

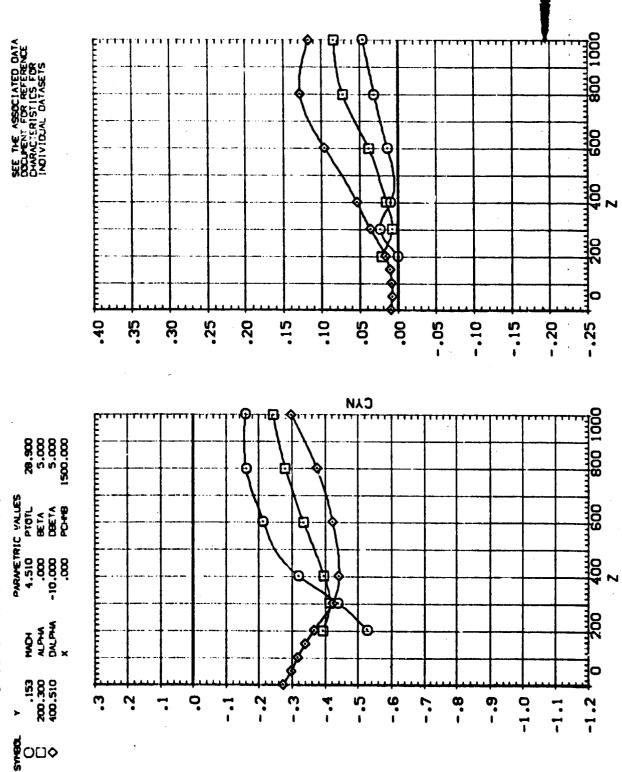
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ247)

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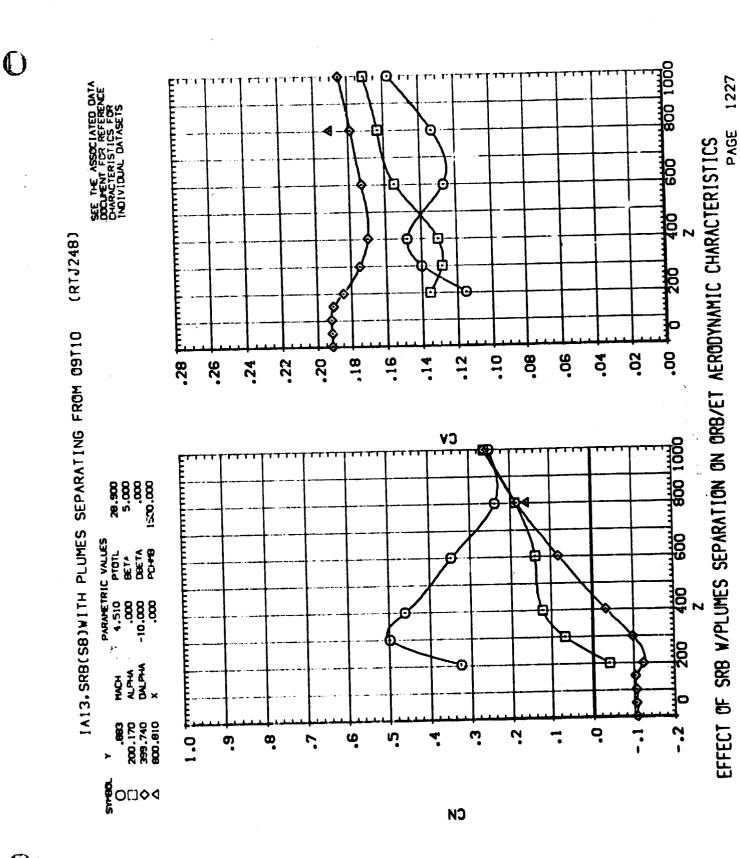


EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

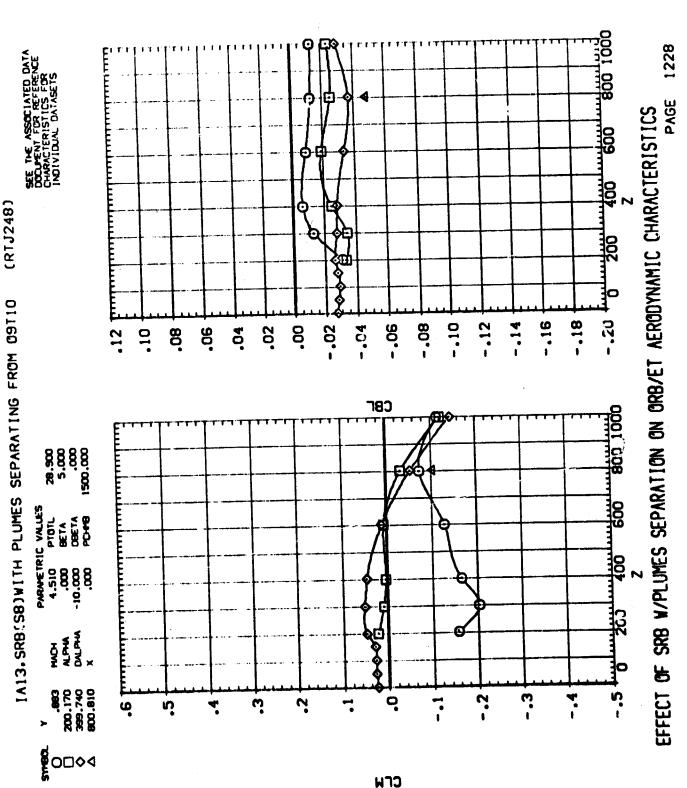
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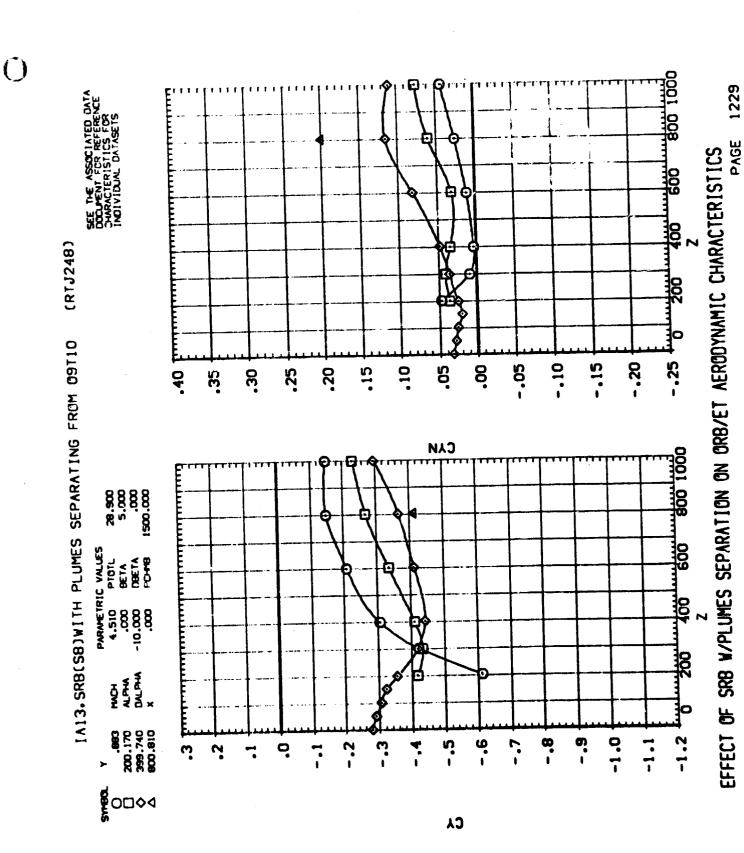
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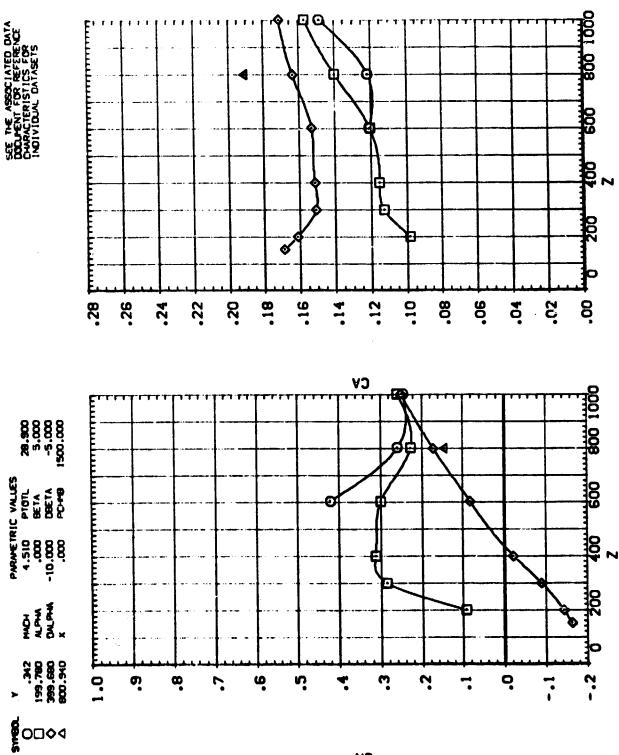
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(RTJ248)



IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ249)



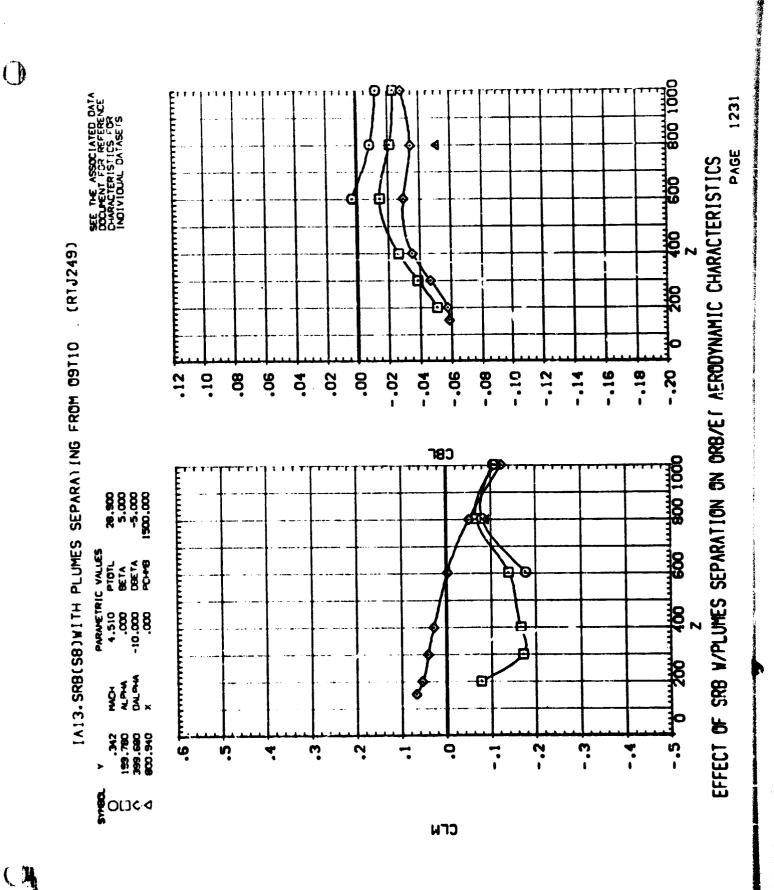
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EFFECT OF SRB N/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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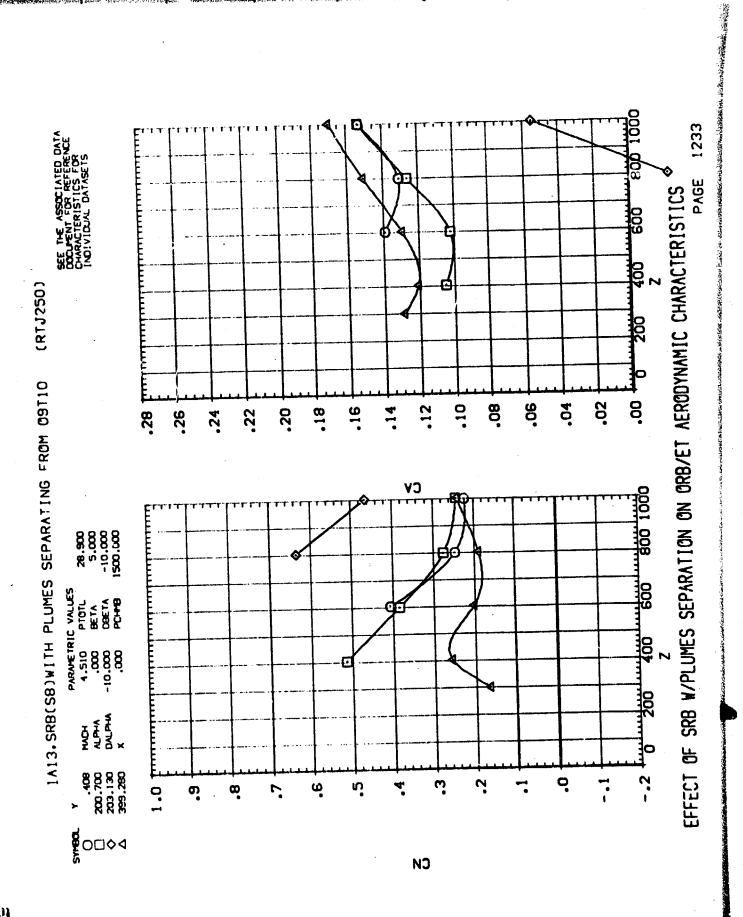


800 1000 SEE THE ASSOCIATED DATA
DOCUMENT FOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS 909 **8** 2 (RTJ249) 388 Q SEPARATING FROM 09110 -.25 -.20 -.10 -.15 10 . S 8 -.05 .15 .25 .20 ဗ္ဗ 64. 35 CAM 800 1000 28.900 5.000 5.000 5.000 5.000 IA13. SRB(SB)WITH PLUMES PARAMETRIC VALUES 4,510 PTOTL ,000 BETA -10,000 DBETA ,000 PCHRB 88 19 88 MACH ALPHA DALPHA * .342 199.780 399.680 800.940 -1.0 8. -6. -1:1 9. -.7 ı. -.2 **ا.** က 7 -: **8**0□◊4 CA

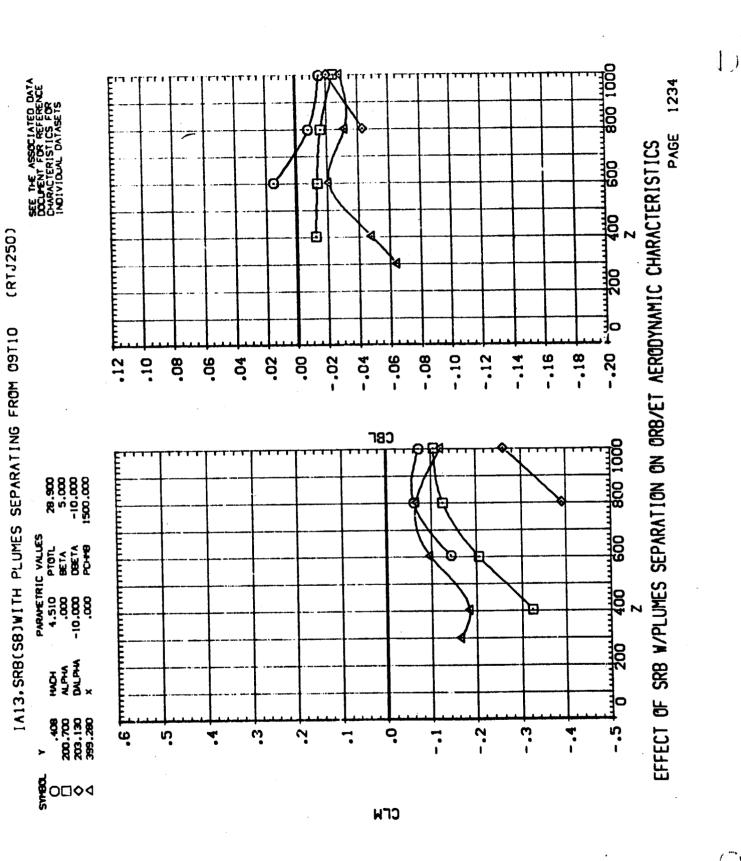
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1232 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

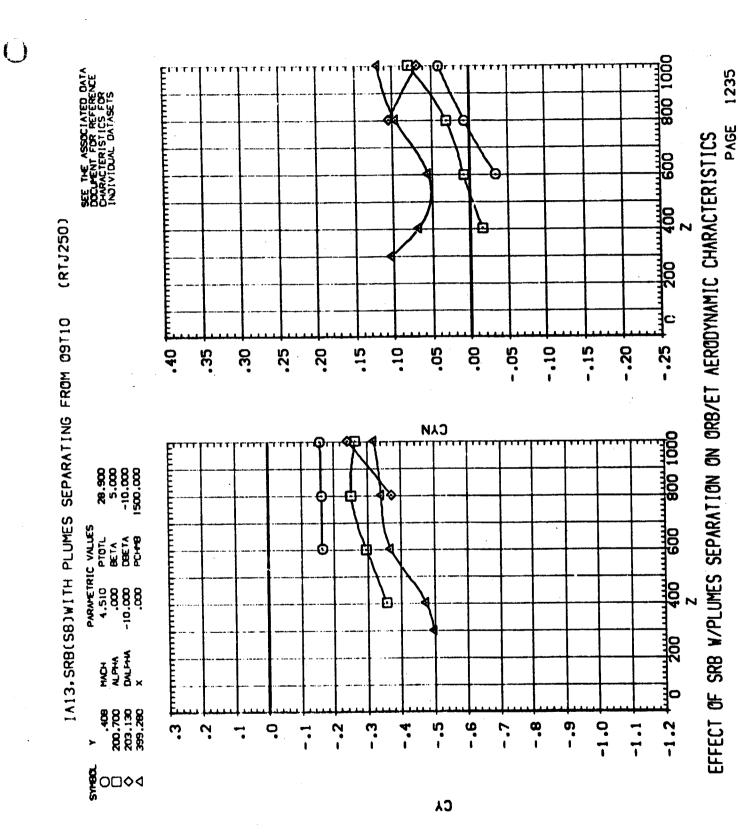
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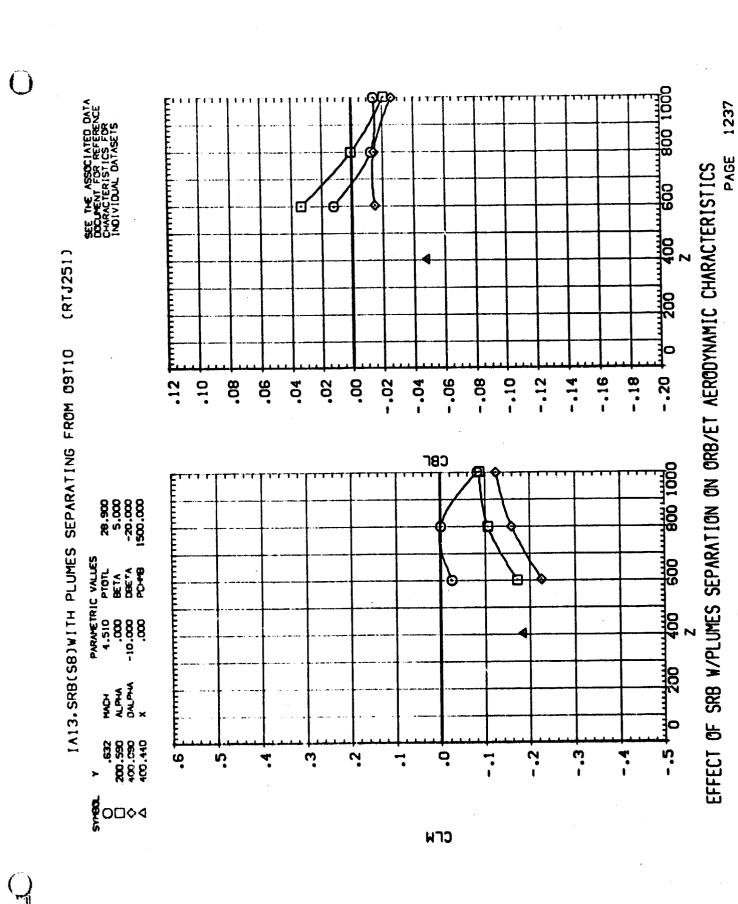
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800 1000 1236 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 900 400 18 18 FROM 09110 **.**08 90. **.** .03 8 .10 .18 .16 .12 .28 .26 .22 .20 .24 SEPARATING CV 800 1000 28.900 5.000 -20.000 1500.000 **PLUMES** PTOTL BETA CBETA PCHIB 909 IA13. SRB(SB)WITH 400 MACH ALPHA VALPHA X 0 200.530 200.530 400.030 400.440 4 ó ų. ທຸ c) Ģ 4 ດ œ -**№**0□**◊**4 CN

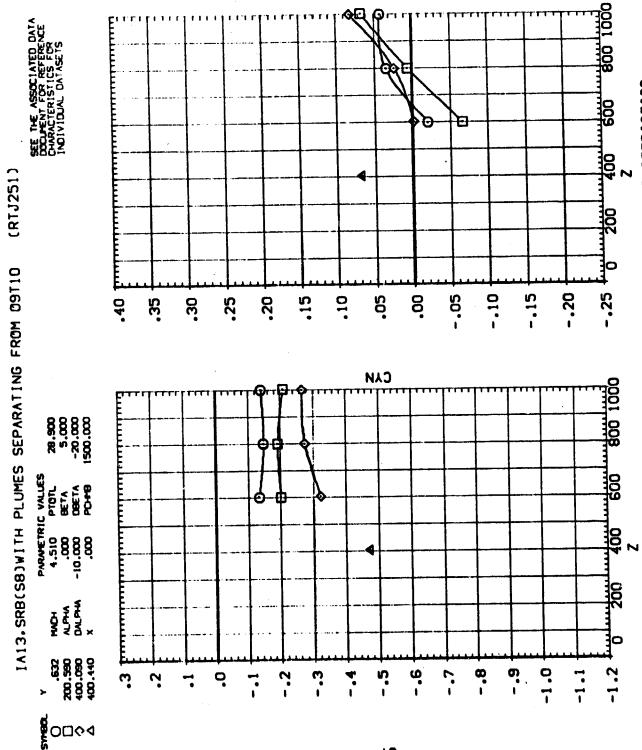
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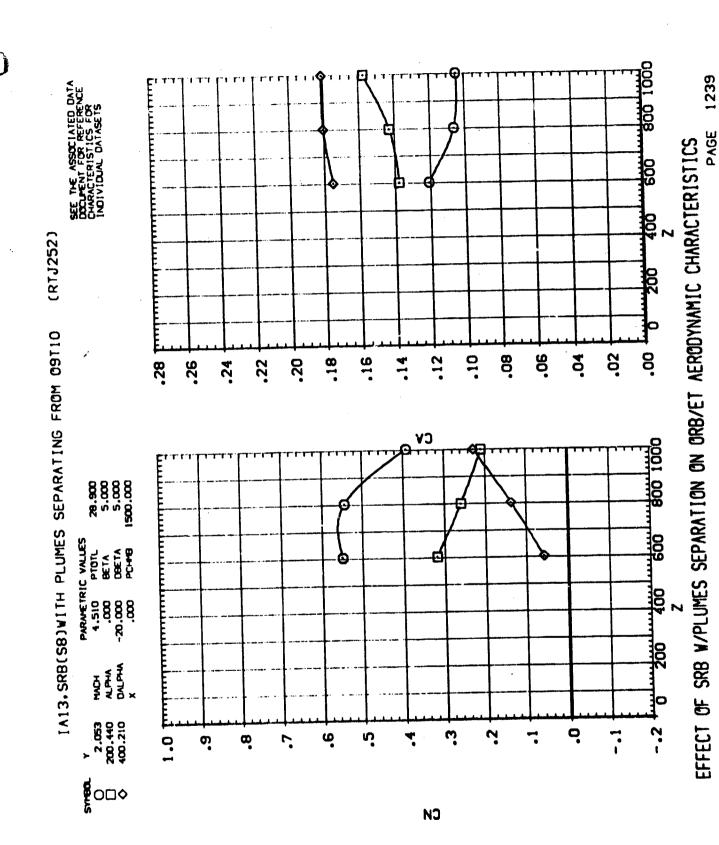


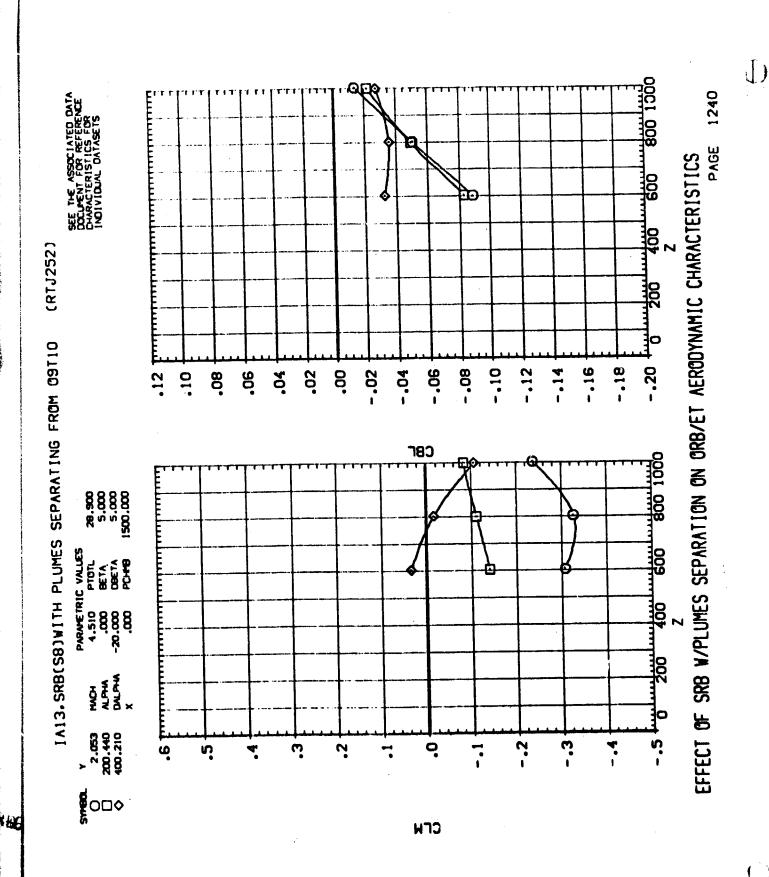
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1238 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS





SEE THE ASSOCIATED DATA
DOCUMENT FOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS (RTJ252) IA13. SRB(SB)WITH PLUMES SEPARATION FROM 09T10 -.15 8 .20 .15 01. **.** 8 -.05 -.10 **4** .25 35 CAN 28,900 5,000 5,000 1500,000 PARAMETRIC VALLES
4.510 PTOTL
.000 BETA
-20.000 DBETA
.000 PCH-8 MACH ALPHA DALPHA 2.053 200.440 400.210 6. ٠. 3. 9. o. -.2 4. ۳, 4 --**¥**0□**◊**

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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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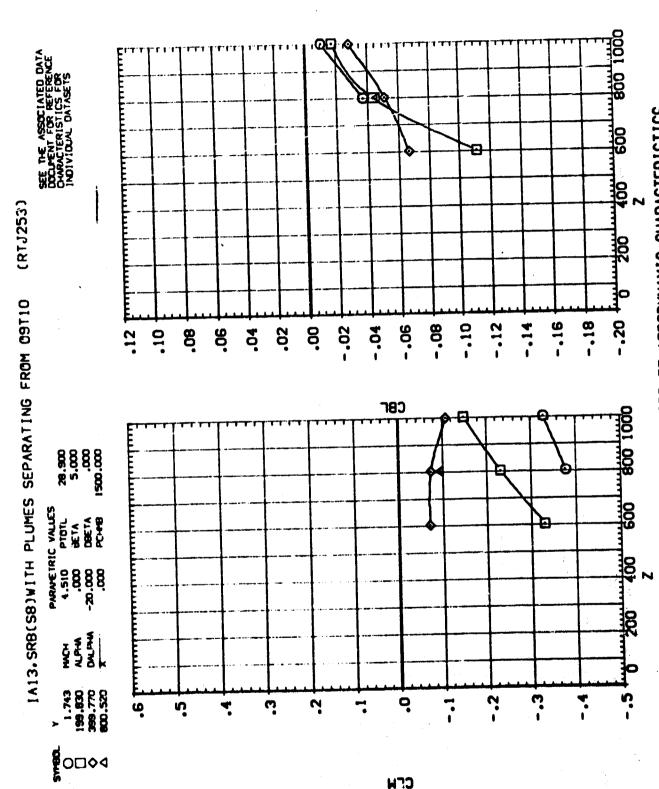
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800 1000 1242 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 009 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 8 9: 90. 90. .02 58 .18 .16 .14 .12 9 2 .26 .24 .22 CV 800 1000 28.900 5.000 .000 .000 .000 PARAYETRIC VALLES
4.510 PT01.
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-20.000 GBETA
.000 PC-48 8 454 854 854 1.743 199.830 399.770 800.520 0.1 o. n ? -: 9 'n 4 æ o. **₹**0□**◊**4 CM

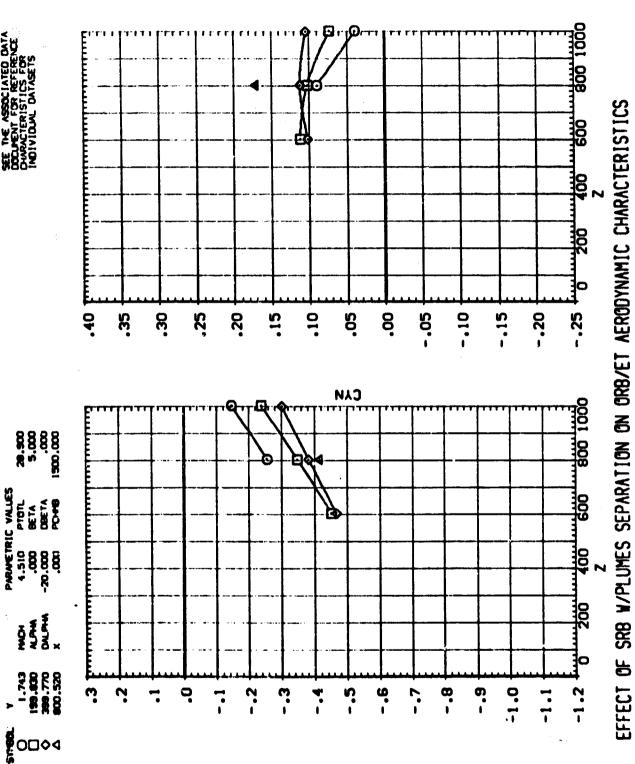
(RTJ253)



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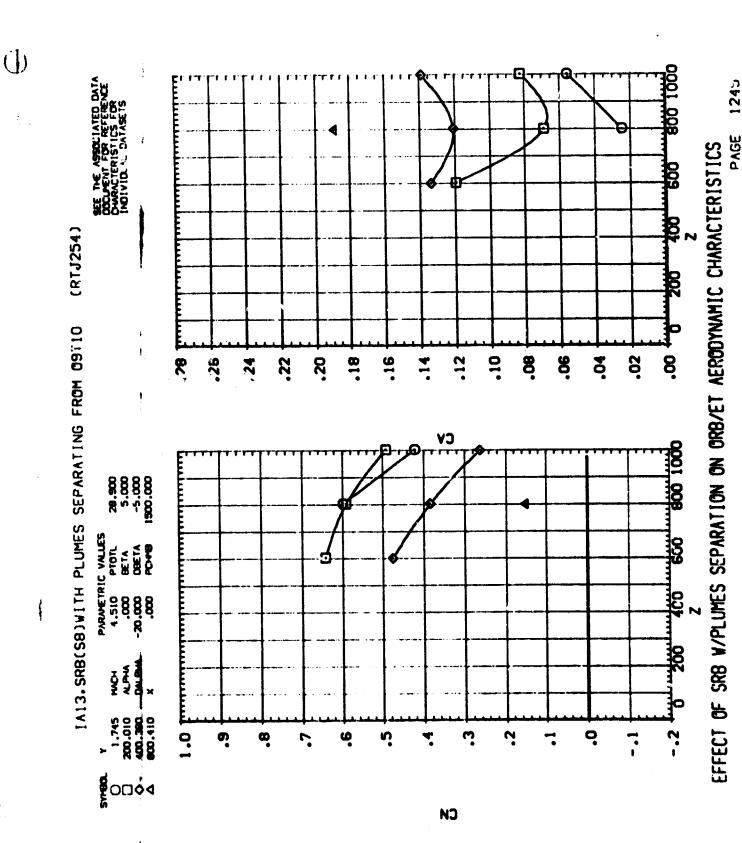
PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

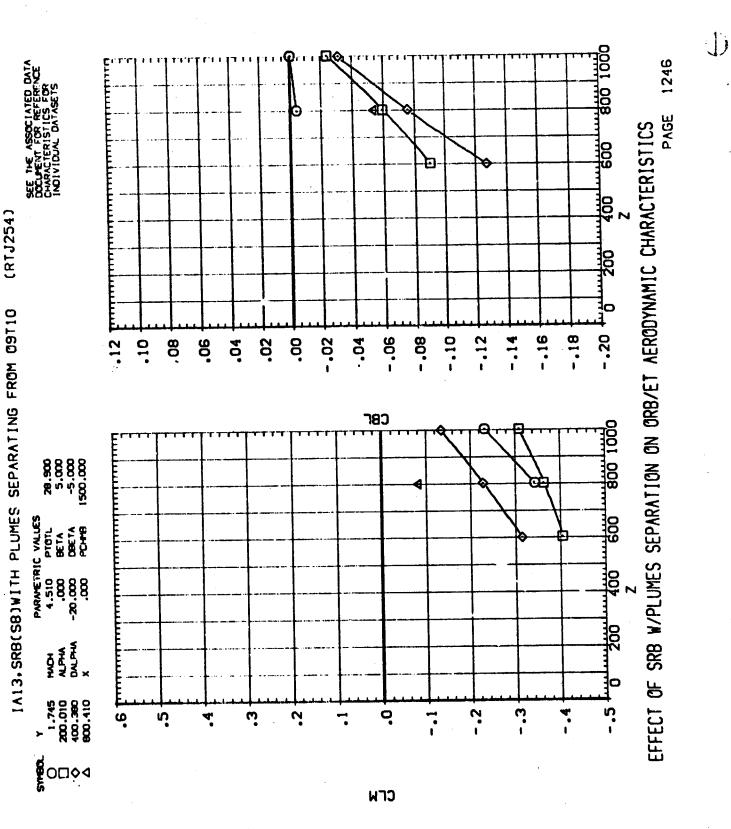
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ253)

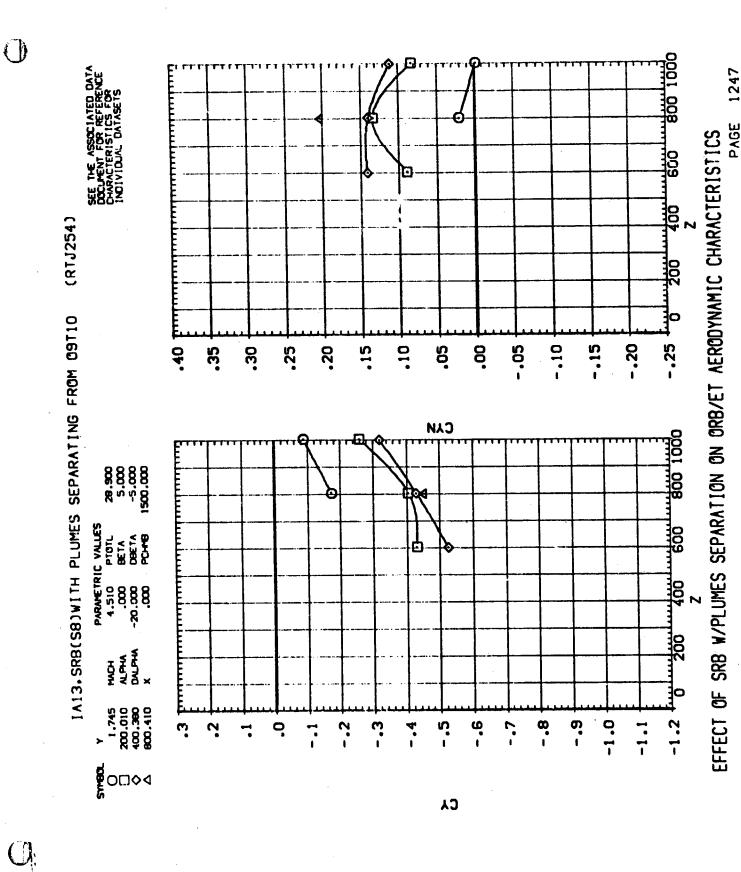


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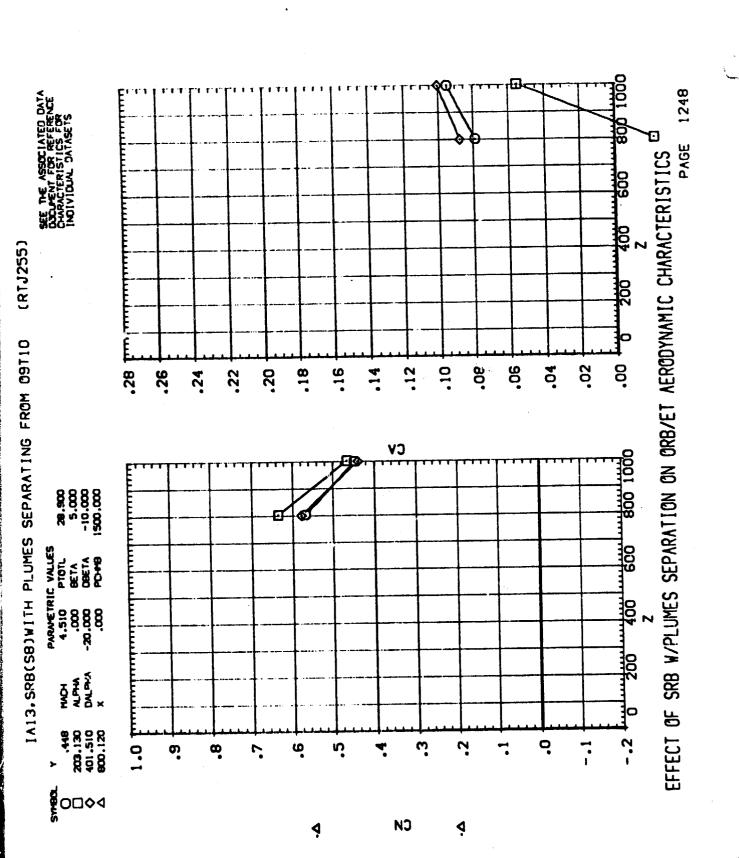
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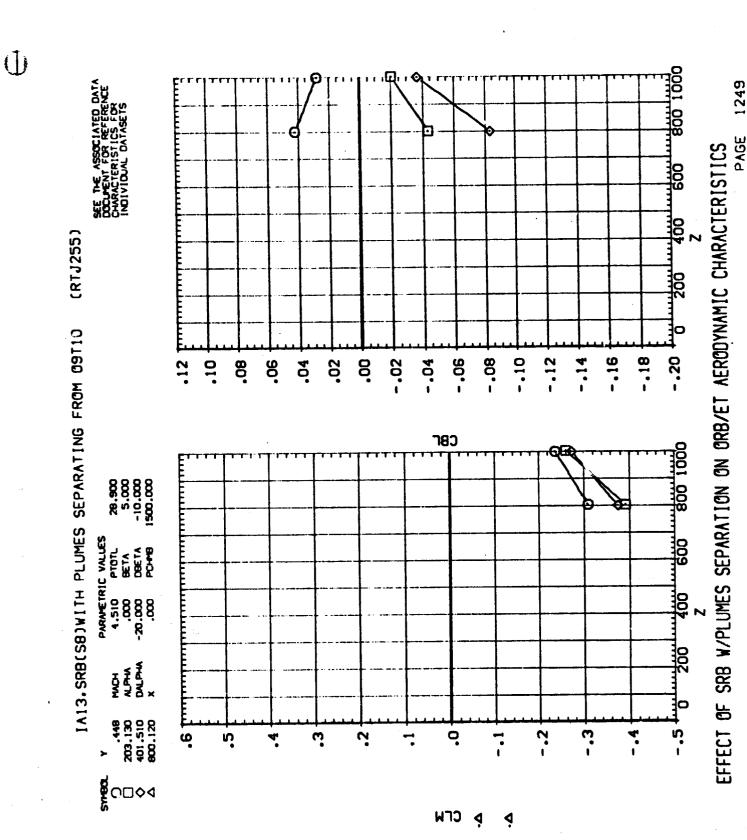


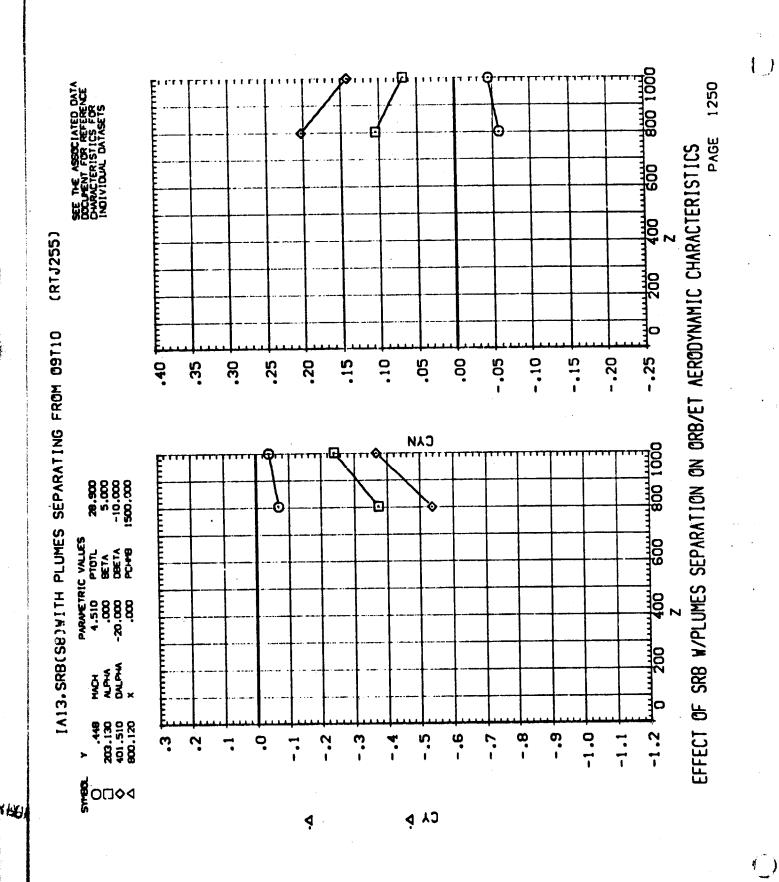


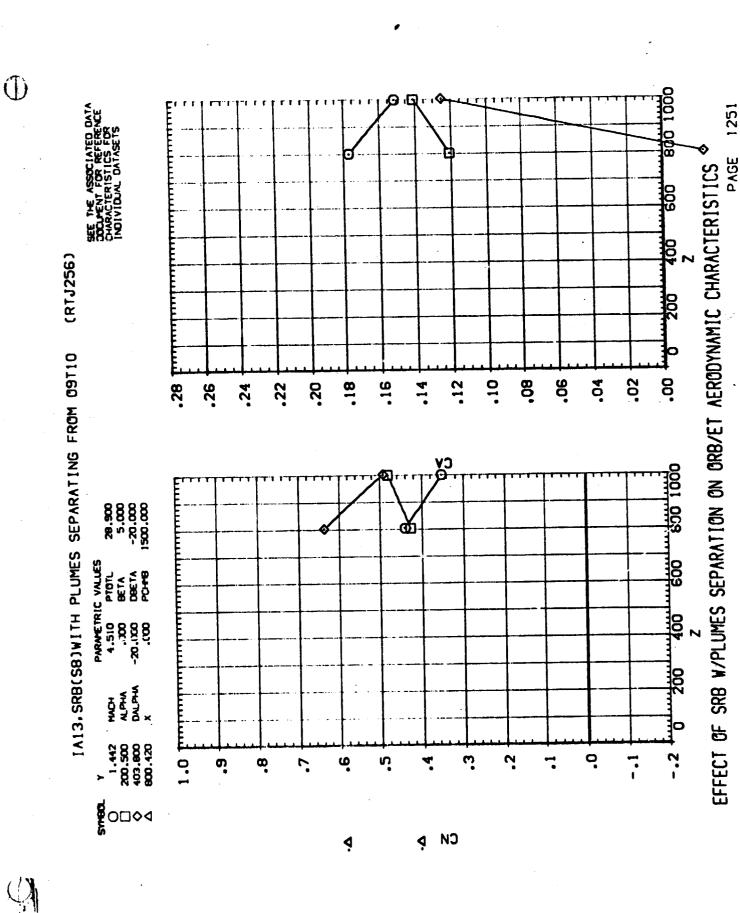


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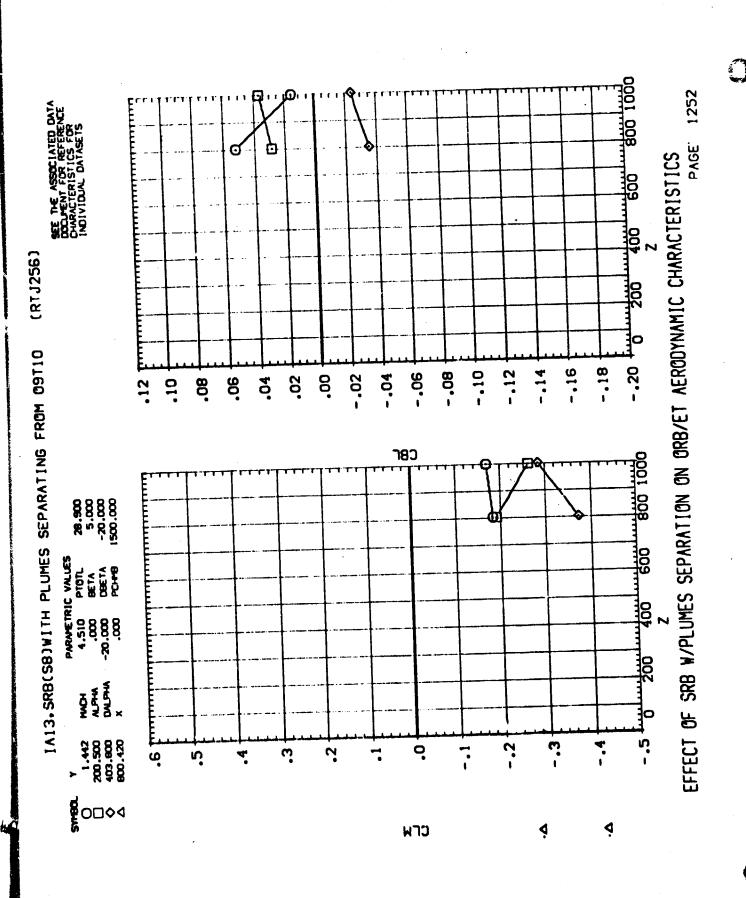


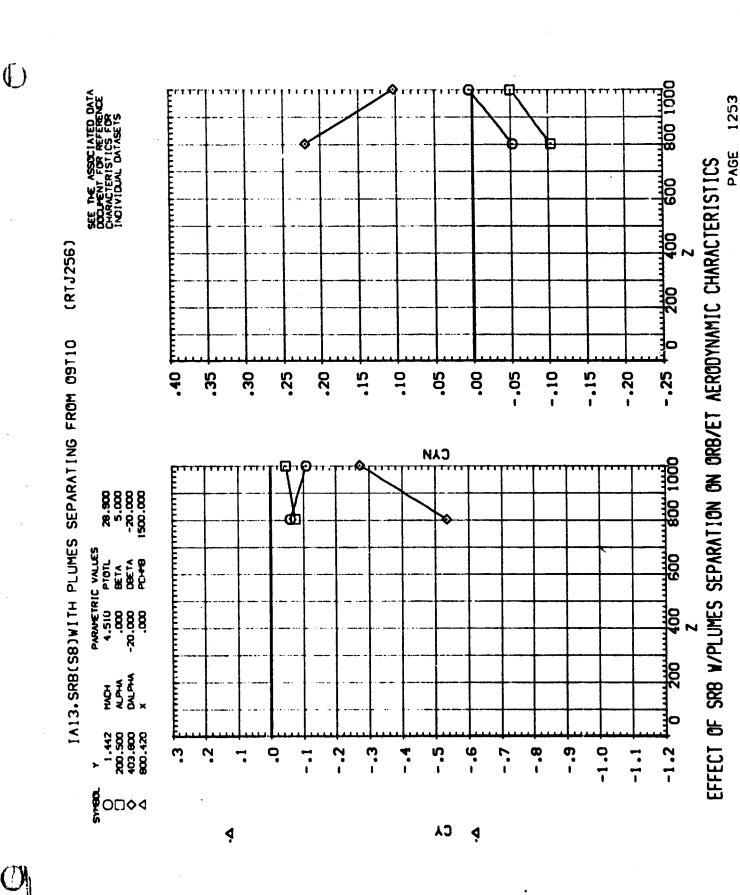




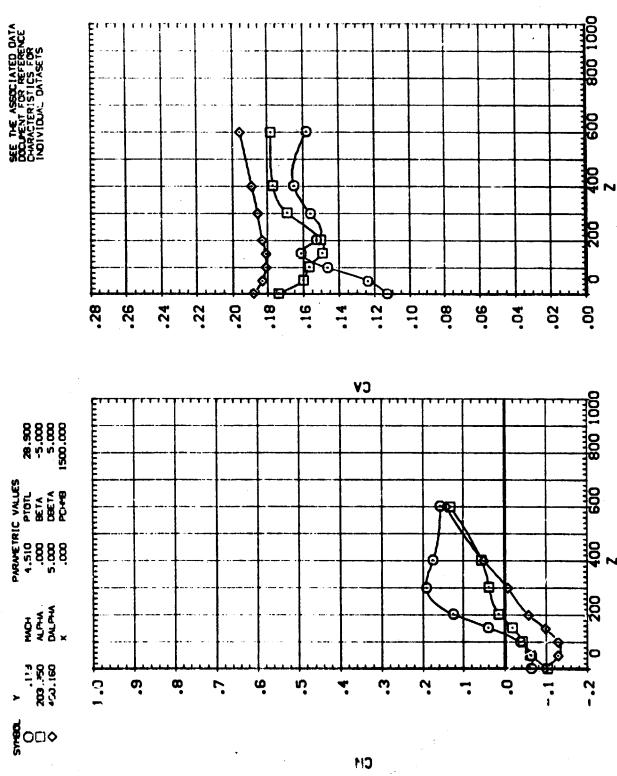


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IA13, SRB(SB)WITH PLUMES SEPARATING FROM 0911() (RTJ259)

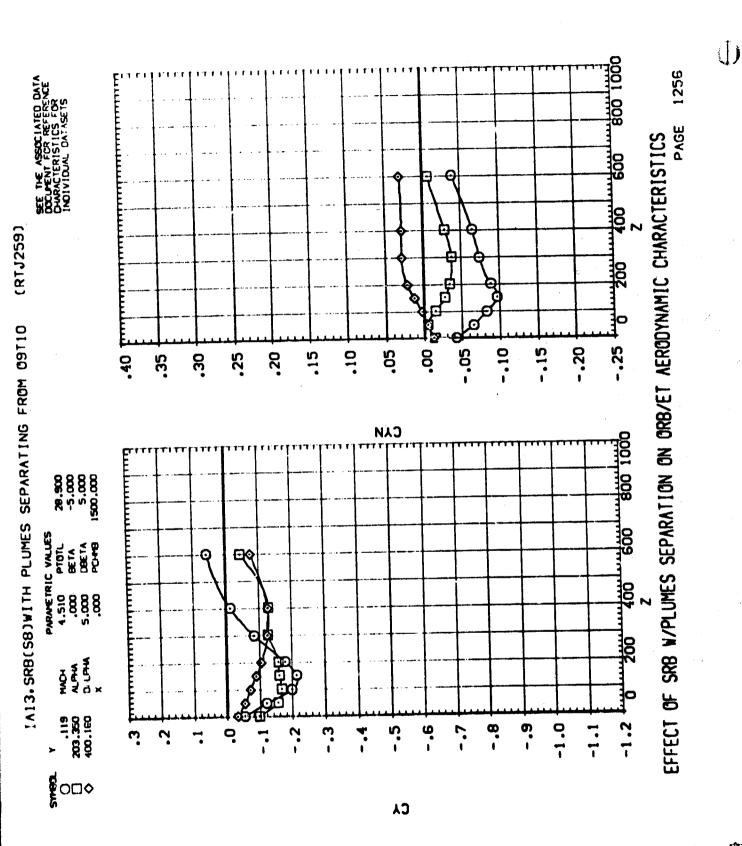


1254 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

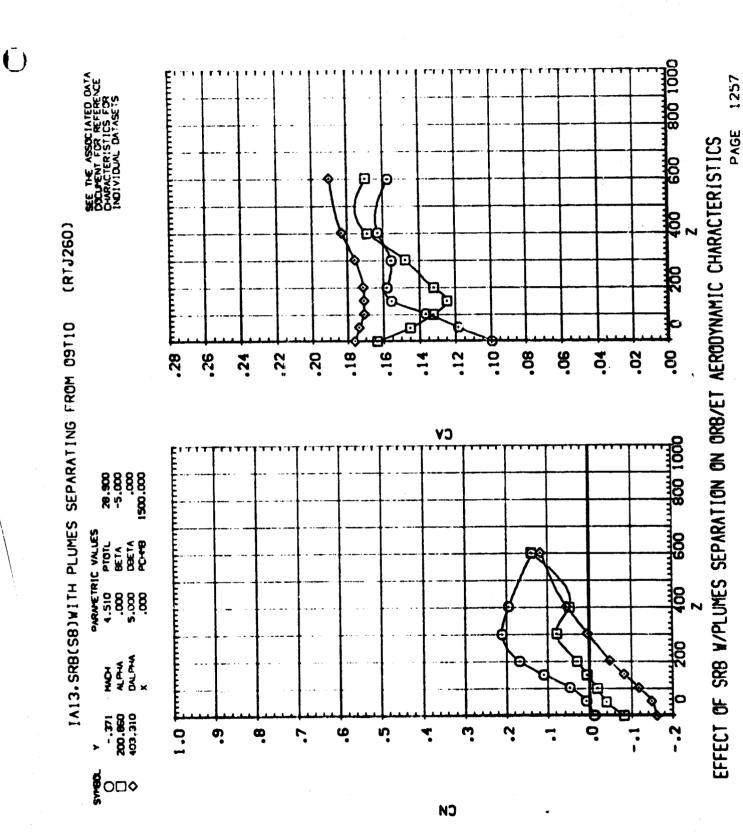
800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 600 400 (RTJ259) 認 IA13, SRB(SB) WITH PLUMES SEPARATING FROM 09110 01: 90. 90. .02 -.06 - 08 -.10 -.12 -.16 -.18 -.20 .04 8 -.02 -.04 -.14 CBF 800 1000 28.300 -5.000 5.000 1500.000 PARAFERIC VALUES
4.510 PTOT.
.000 BETA
5.000 DBETA
.000 PCP-8 ф 8 MACH ALPHA DALPHA 203.350 0 ٠. 'n 'n 'n

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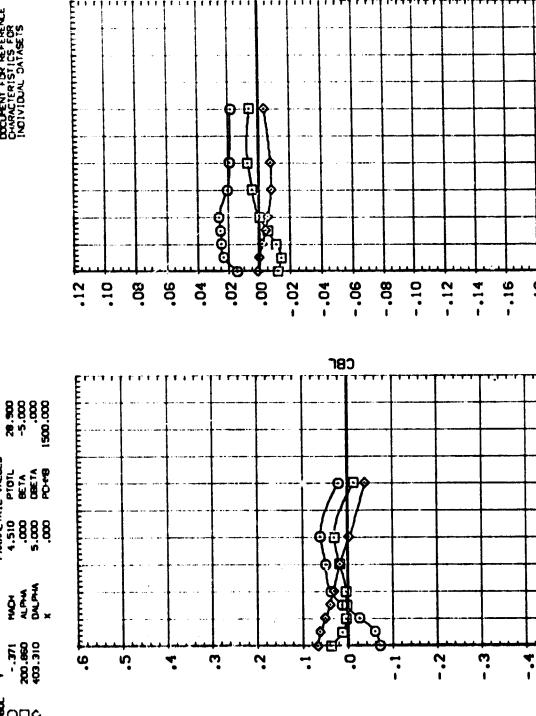


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SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ260) SEPARATING FROM 09110 IA13. SRB(SB)WITH PLUMES PARAYETRIC VALUES
4.510 PTOTL
.000 BETA
5.000 OBETA
.000 PC+49 1 ¥ O□¢



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1258 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE DARACTERISTICS FOR INDIVIDUAL DATASETS 900 8 (RTJ260) **100 100** 1A13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 0 9 8 -.10 -.15 -.20 -.25 9: S 2 .13 -.05 ĸ. 8 23 CAM 800 1000 28.300 -5.000 1500.000 1500.000 # 510 PTOT.

4.510 PTOT.

.000 BETA
5.000 DBETA
.000 PCP-8 88 **⊡**Q Θ * DEPT -.371 200.860 403.310 8. 6. -1.0 -1.2 Ġ -1.1 -.7 n ? o 7. **©**O□◊

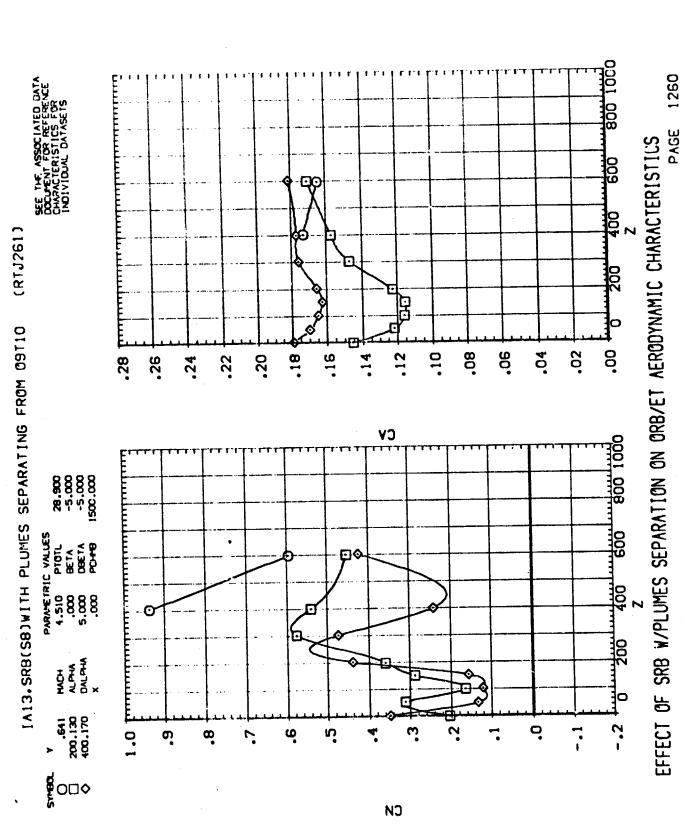
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PAGE

EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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800 1000 SEE THE ASSOCIATED DATA
DOCUMENT FOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 909 QΦ 9 (RTJ261) 90 200 今のの中 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 -.16 -.18 -.20 8 90.--.08 -.10 -.12 -.14 .02 -.02 .04 80. 90. .04 .12 .10 ายว 800 1000 28.900 -5.000 -5.000 1500.000 PARAMETRIC VALUES
4.510 PTOTL
.000 BETA
5.000 DETA
.000 PCMB MACH ALPHA DALPHA × 8 Oţ 200.130 400.170 Ź 'n -.2 د. 4. 7 o ဖ 'n (.) 4.

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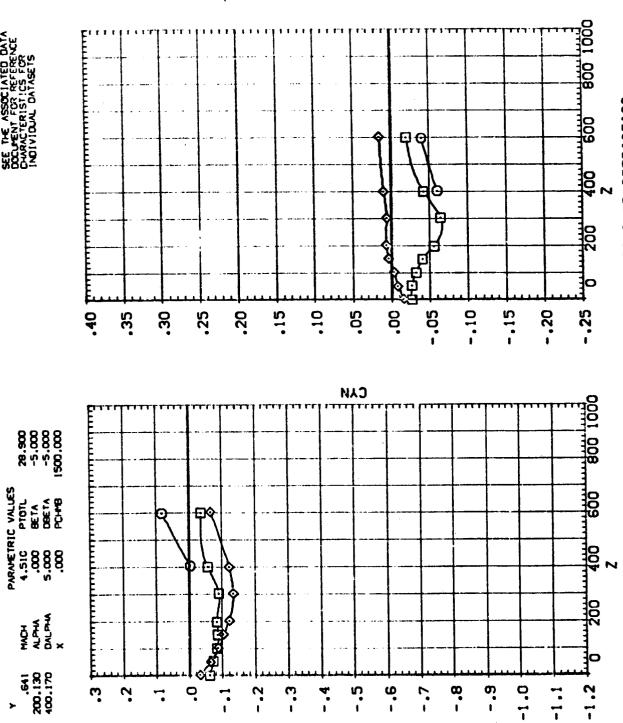
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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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DOCUMENT HOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS 9 **න**ロ 400 (RTJ262) 200 IA13.SRB(SB)WITH PLUMES SEPARATING FROM 09110 2. 01. 80. 90. • .02 8 .18 .16 .12 .20 .14 .28 .26 .22 CV 800 1000 28.900 -5.000 -10.000 1500.000 PARAMETRIC VALUES
4,510 PTÖTL
,000 BCTA
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,000 PCHPB 100 Θ. 18 MACH ALPHA SALPHA 200.490 399.140 3.036 o. 7 --က္ 0. o, æ ~ ယ္ ល 4 ¥O∏♦

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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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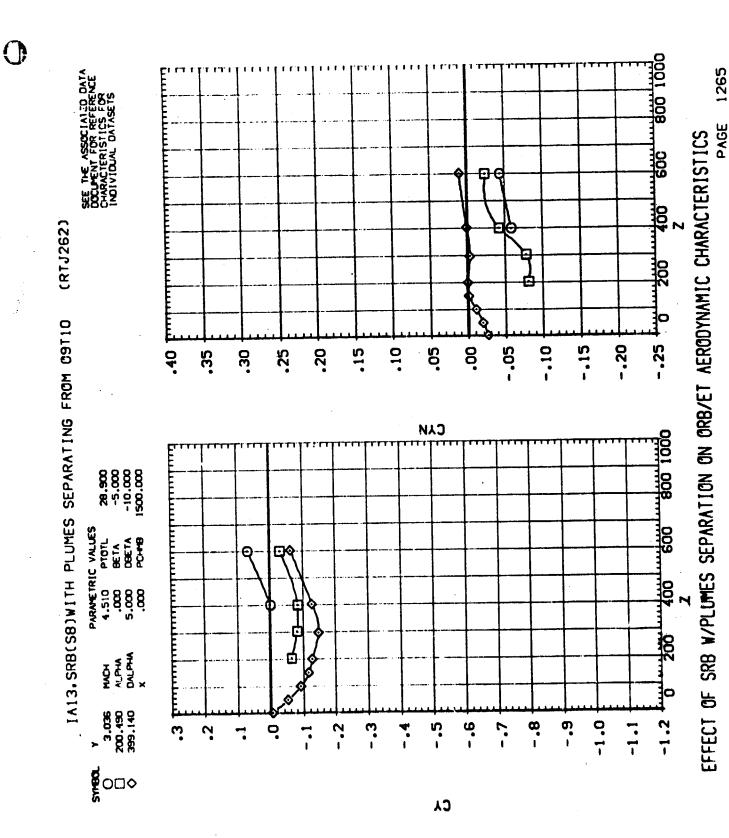
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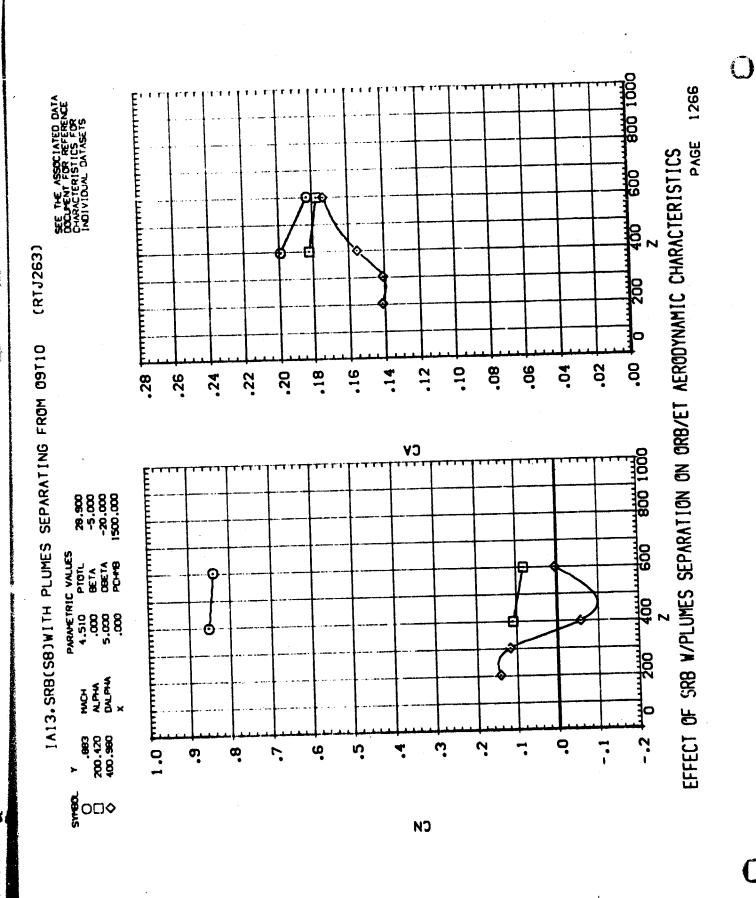
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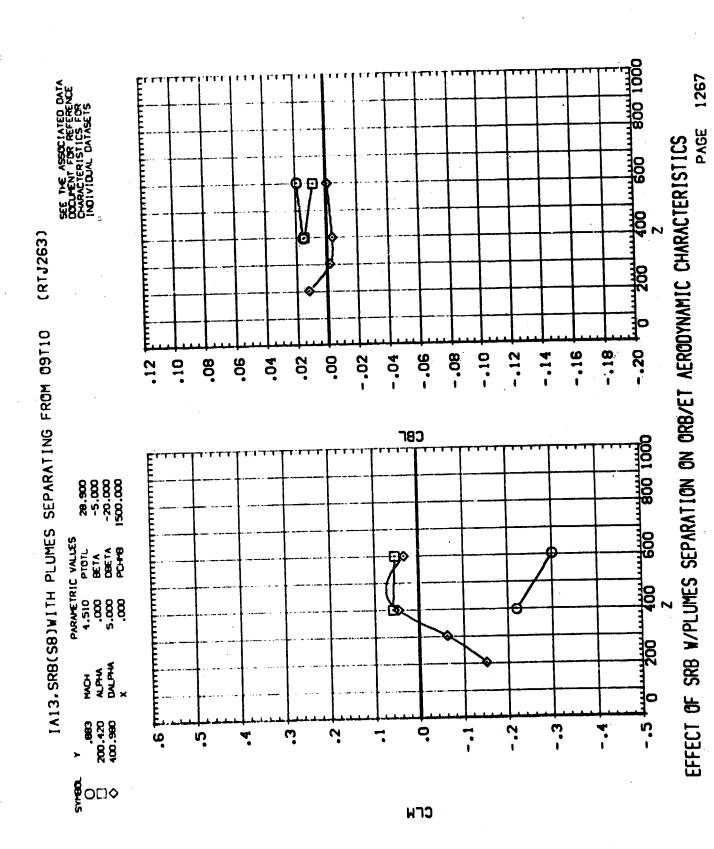
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4.510 PTGTL
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5.000 DBETA
.000 PCHB 1 900 460 200 MACH ALPHA X CALPHA 3.036 200.490 399.140 ò -.2 ė. S. 'n r, -: ဖ ٧. ? **§**O□◊ CLM

(RTJ262)

IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110



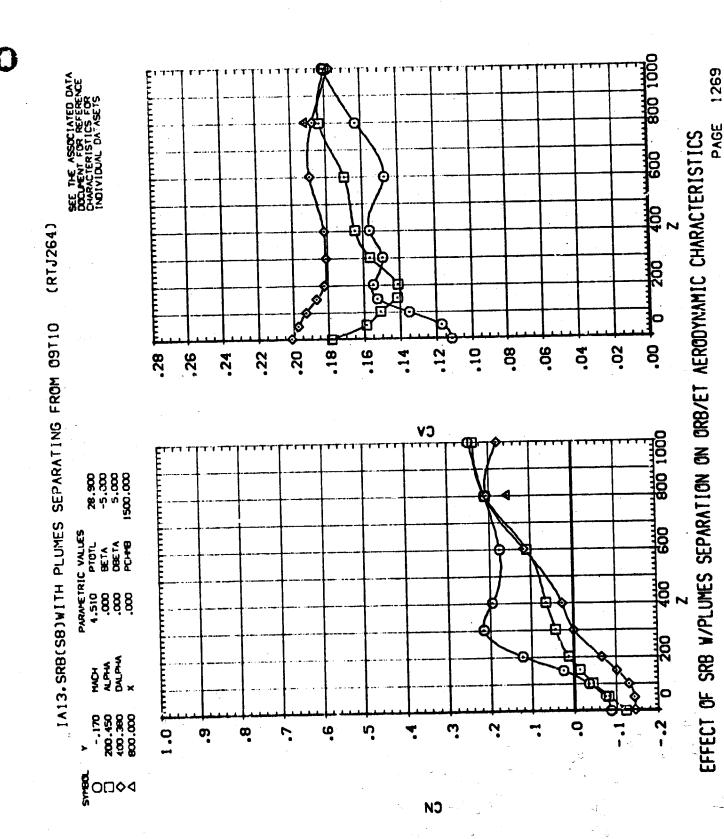


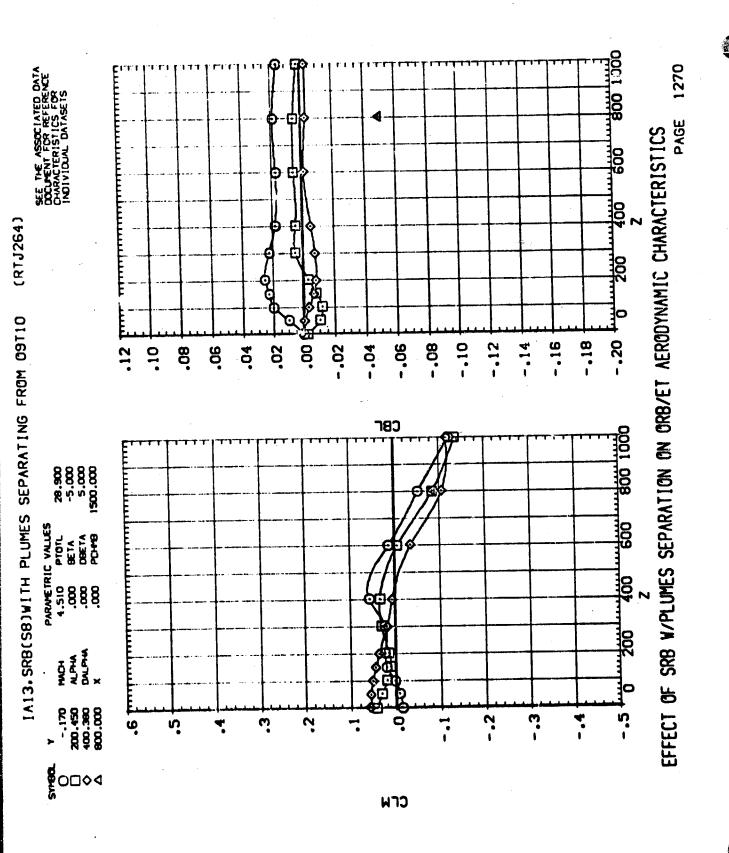


800 1000 1268 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INJIVIDUAL DATASETS PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 器 488 (RTJ263) 色 器 þ IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09T10 -.25 -.05 -.10 -.15 -.20 .20 .15 5. so. 8 .25 64. .35 8.30 CAN 800 1000 86.85 86.85 86.88 86.88 86.88 86.88 PARAMETRIC VALLES
4.510 PTOTL
.000 BETA
5.000 DEETA
.000 PCM-8 909 8 b 200 MCH ALPHA VALPHA .883 200.420 400.980 -1.2 8. 6. -1.0 -1.1 9. -.2 e. -٠. ئ -.7 4. က 4 0 -: **₹**0□**◊** L

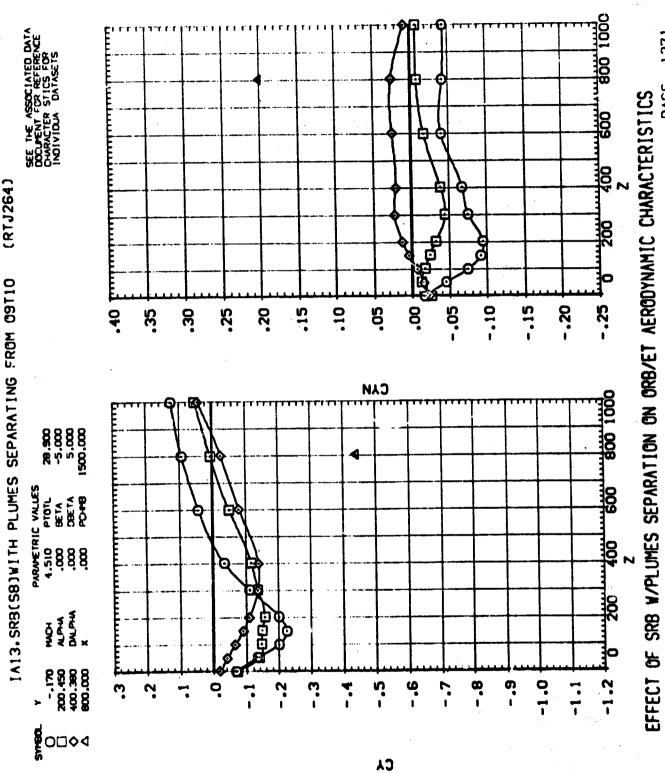
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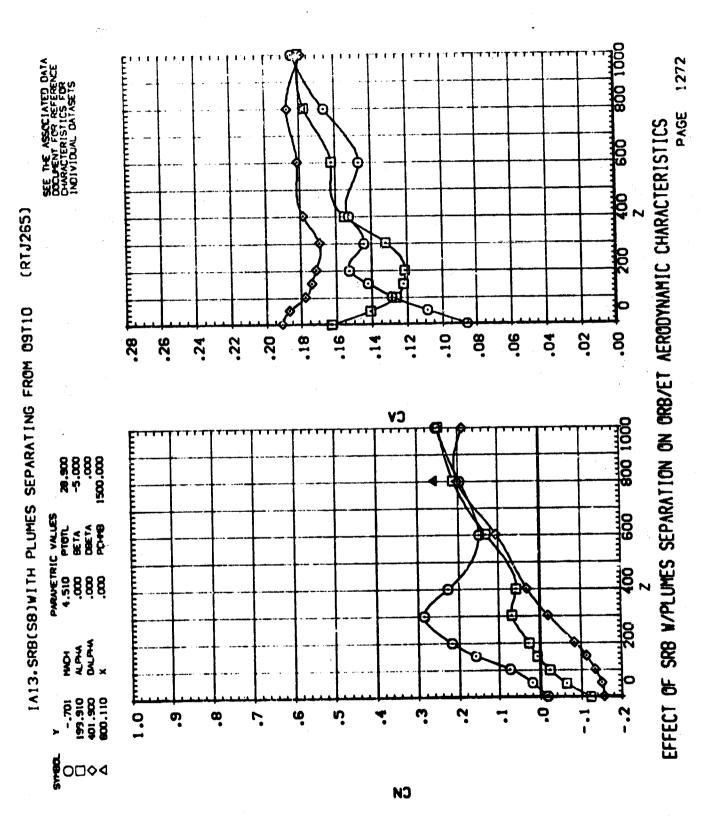
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(RTJ264)





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4.510 PTOT.
.000 BETA
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

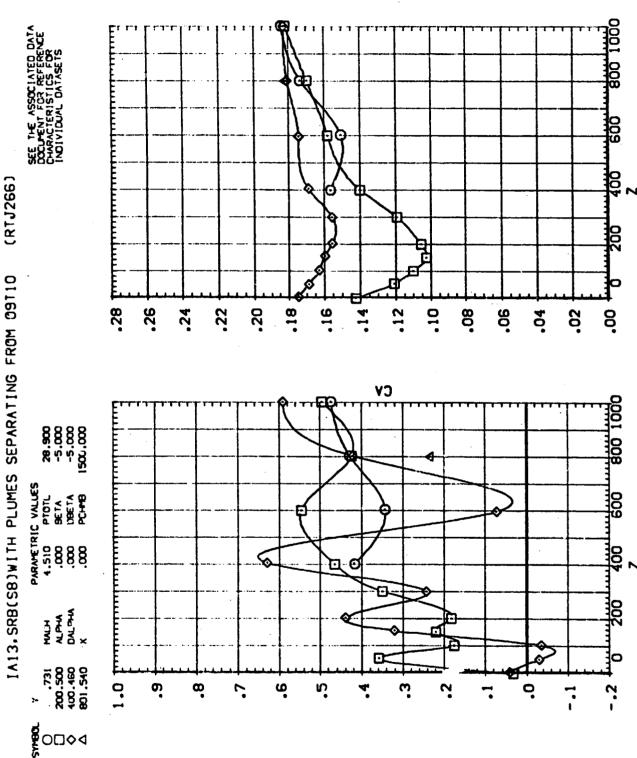
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4.510 PTOTL
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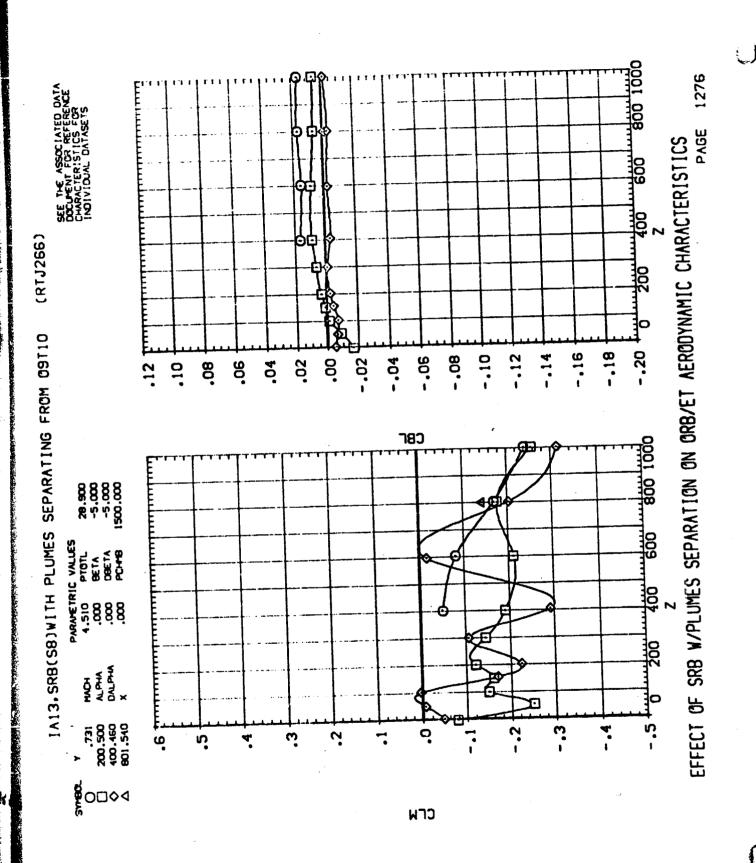
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PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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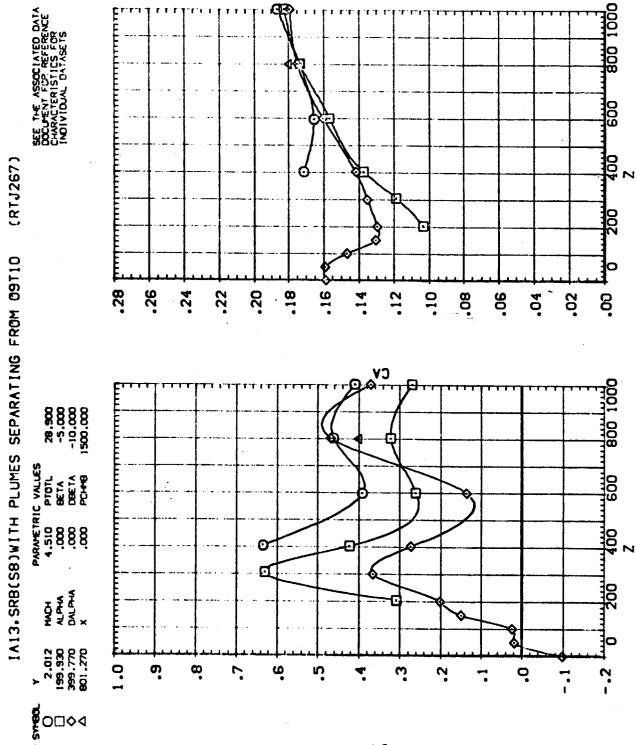
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4.510 PTÖTL
.000 BETA
.000 DBETA
.000 PCMB **4**8 認 MACH ALPHA DALPHA × .731 200.500 400.460 801.540 -1.2 -1.0 <u>თ</u> 9. 8 -.2 e. -٠. ئ - .7 က o -1.1

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1278 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ267) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 -.10 -.02 8 90.--.08 -,12 -.16 -.18 .12 90. 90. 9. . 02 -.04 -.14 .10 CBF 28.900 -5.000 -10.000 1500.000 PARAMETRIC VALUES
4.510 PTOTL
.000 BETA
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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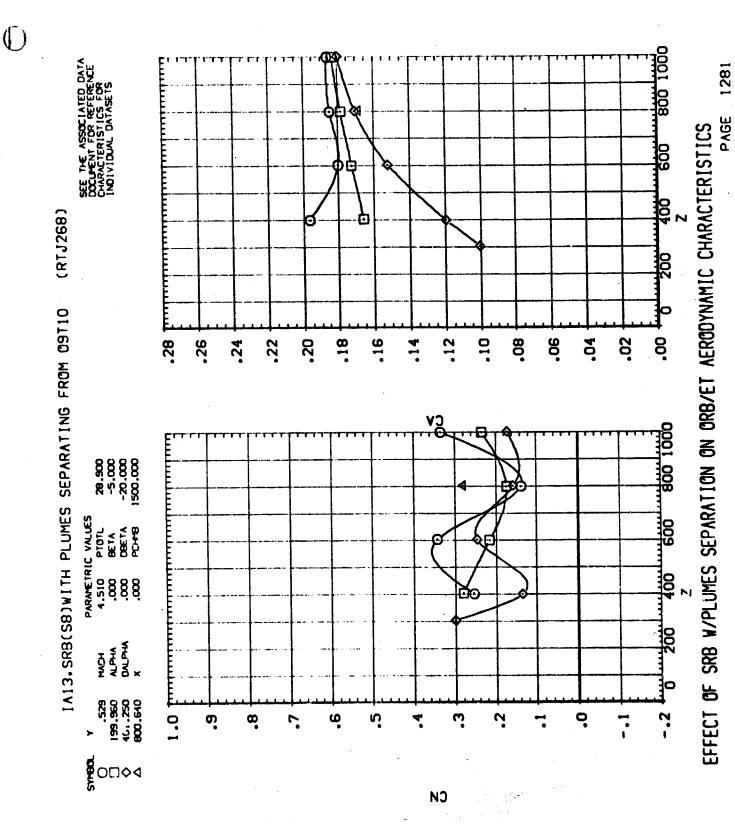
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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 1280 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS 100 18 (RTJ267) 紹 0 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 -.25 -.15 -.20 .15 . S 8 -.05 -.10 40 .35 30 .25 .20 9: CAN 800 1000 28.900 -5.000 -10.000 1500.000 PARAMETRIC VALUES
4.510 PTOTL
.000 BETA
.000 DETA
.000 PCMB 999 99 ø 200 MACH ALPHA DALPHA X \Box 0 2.012 199.930 399.770 801.270 8. -1.0 -1.2 ن -1.1 က္ o -.2 e. -4. ٠. 9. -.7 ? **№** 0□ ◊ 4

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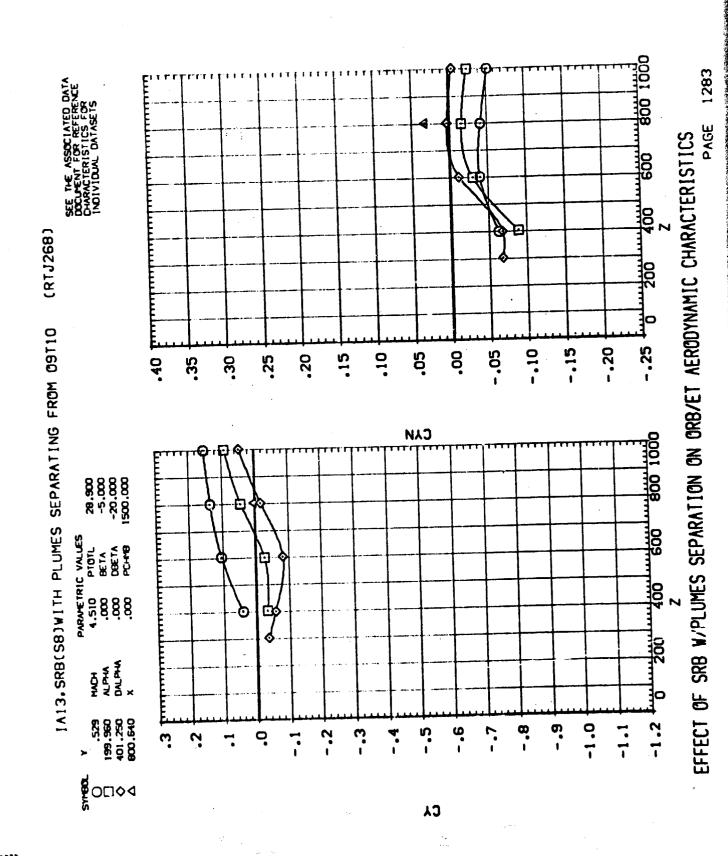
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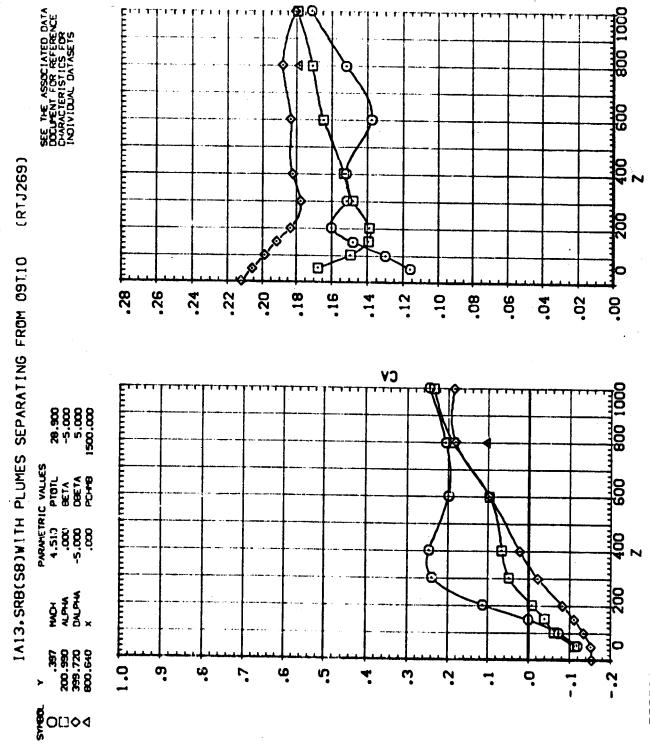
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

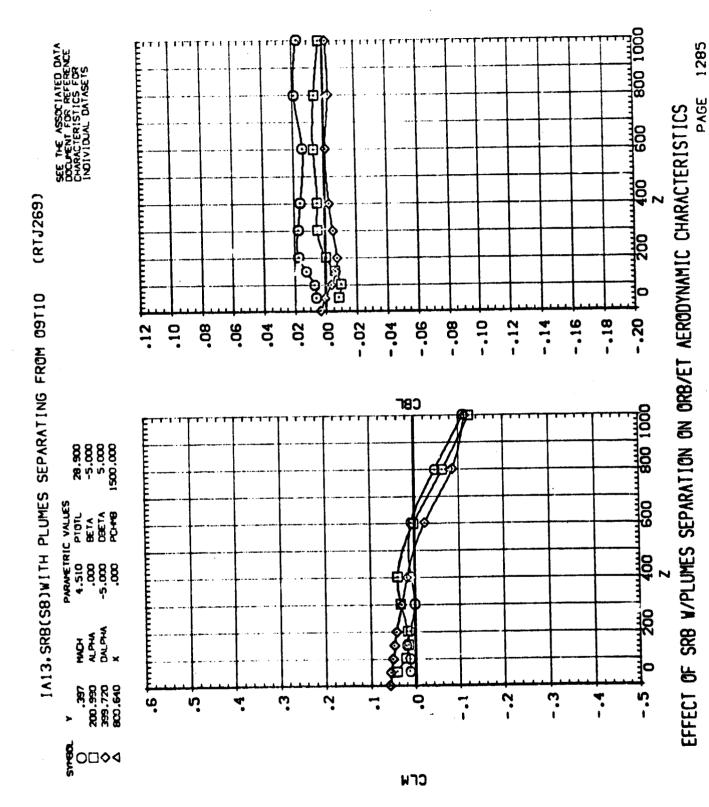
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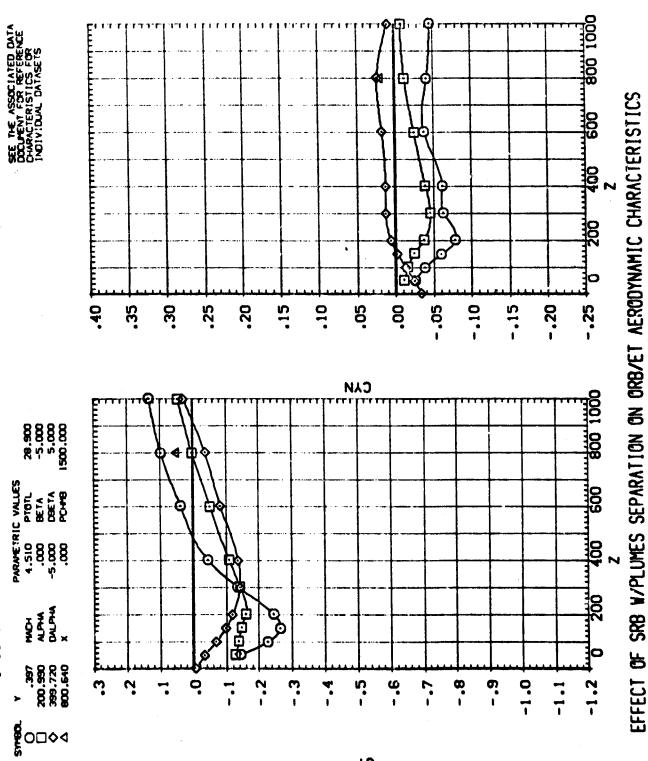
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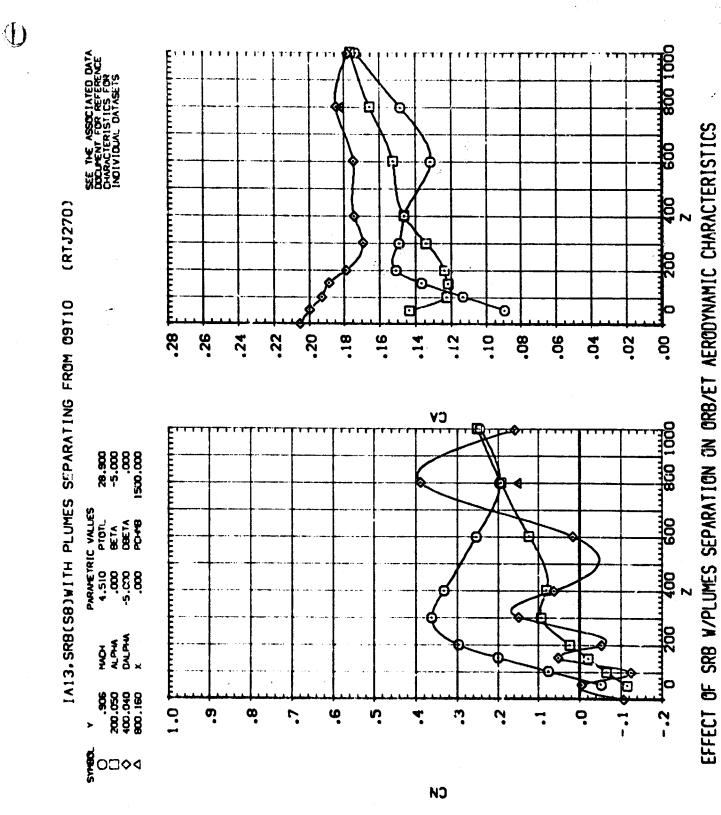
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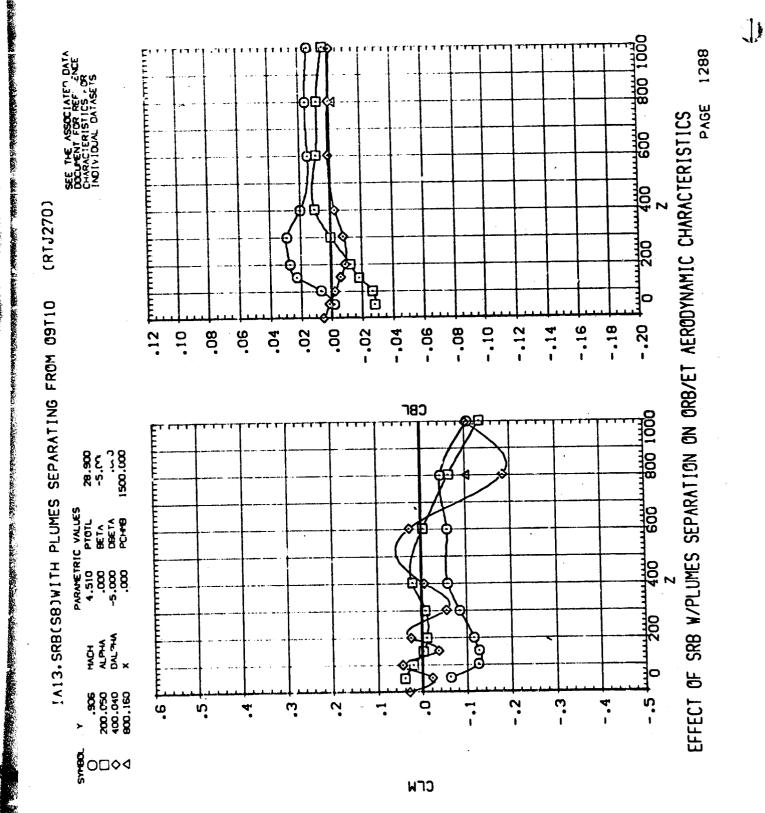


(RTJ269) SEPARATING FROM 09110 IA13. SRB(SB)WITH PLUMES



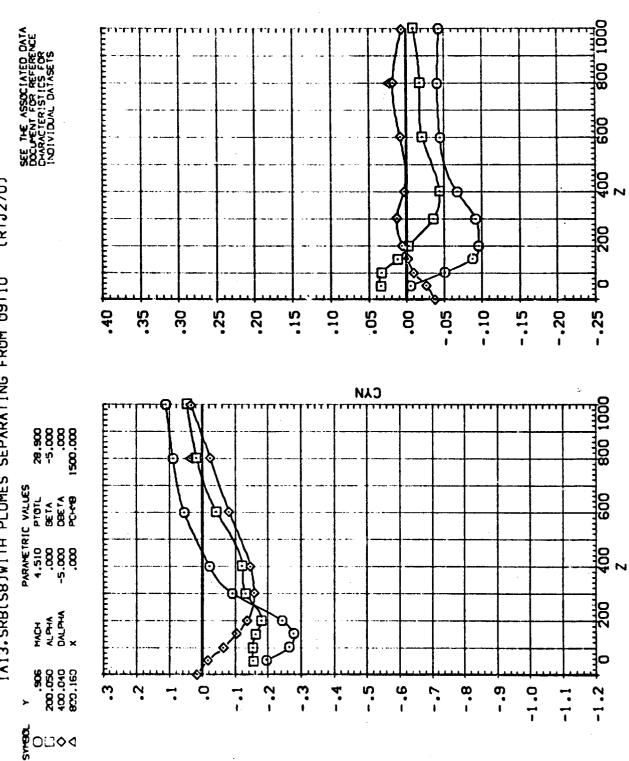
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(RTJ270) 1A13.SRB(S8)WITH PLUMES SEPARATING FROM 09110

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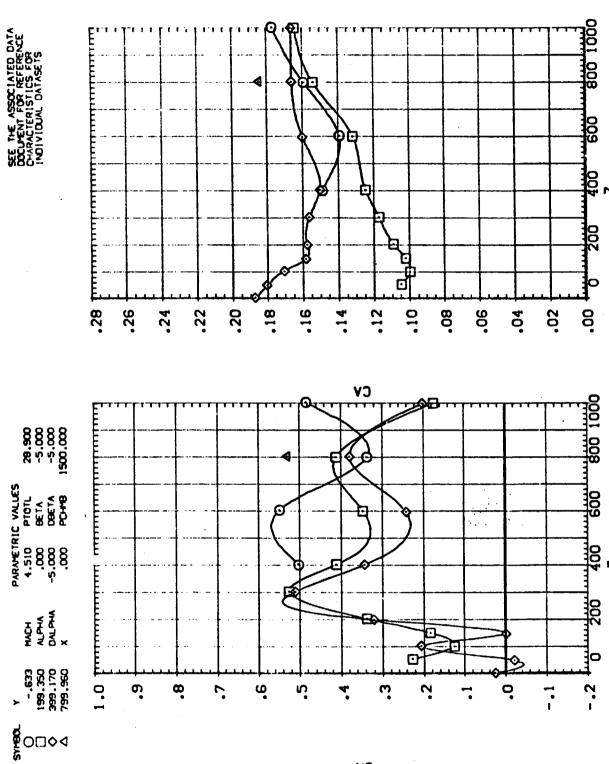
EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09T10 (RTJ271)



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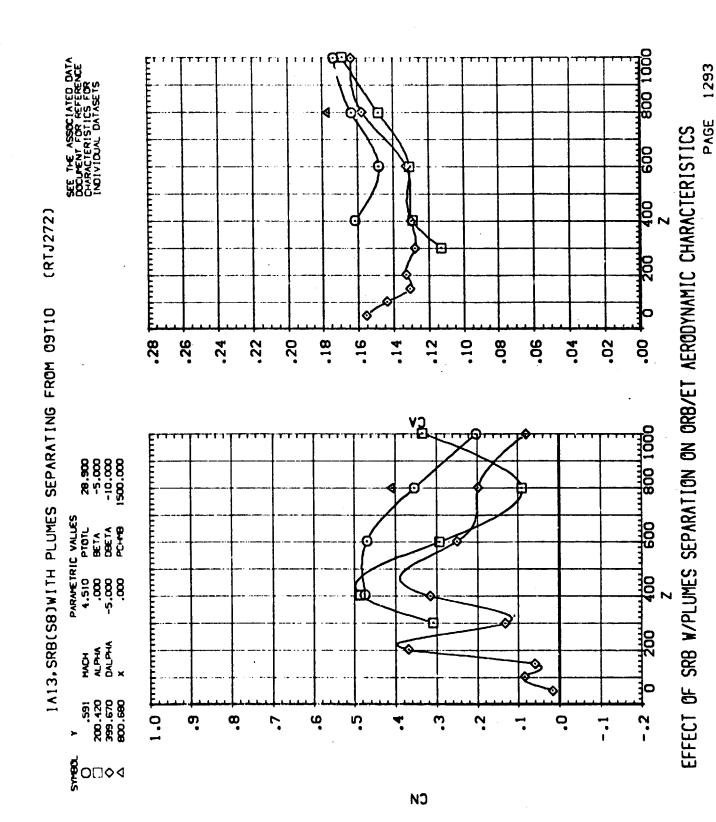
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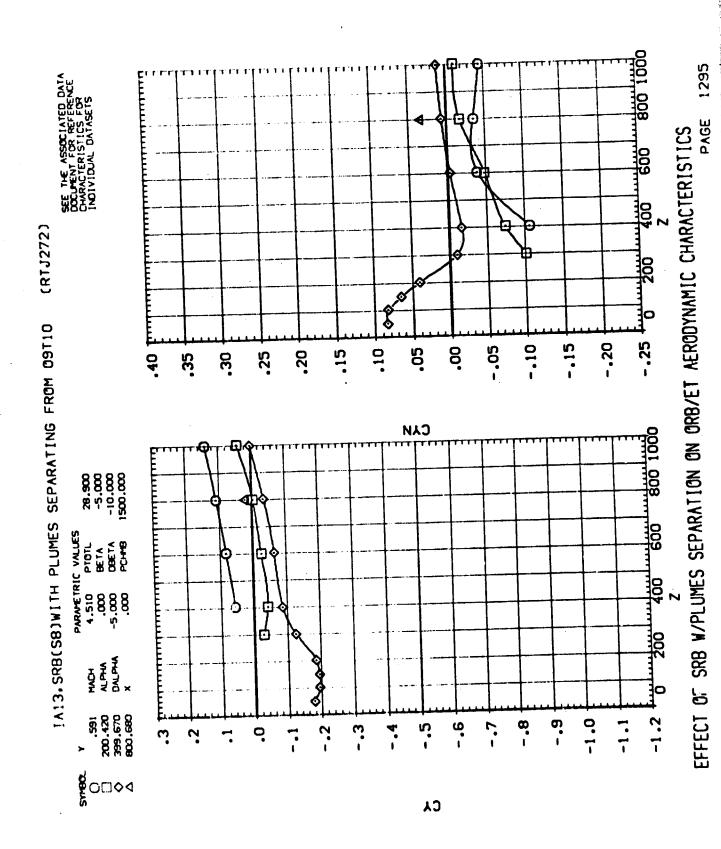
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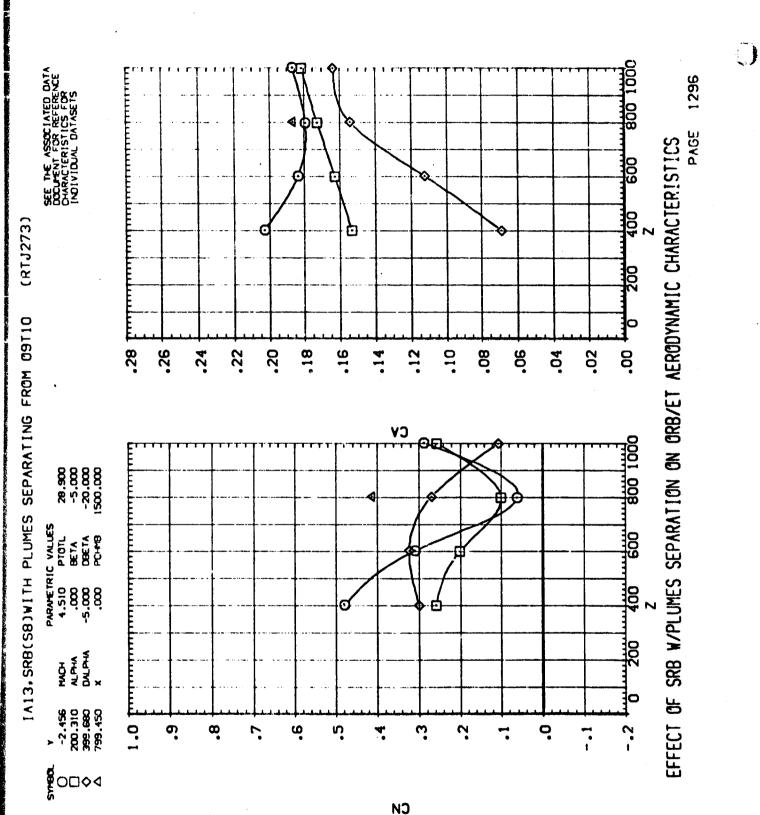


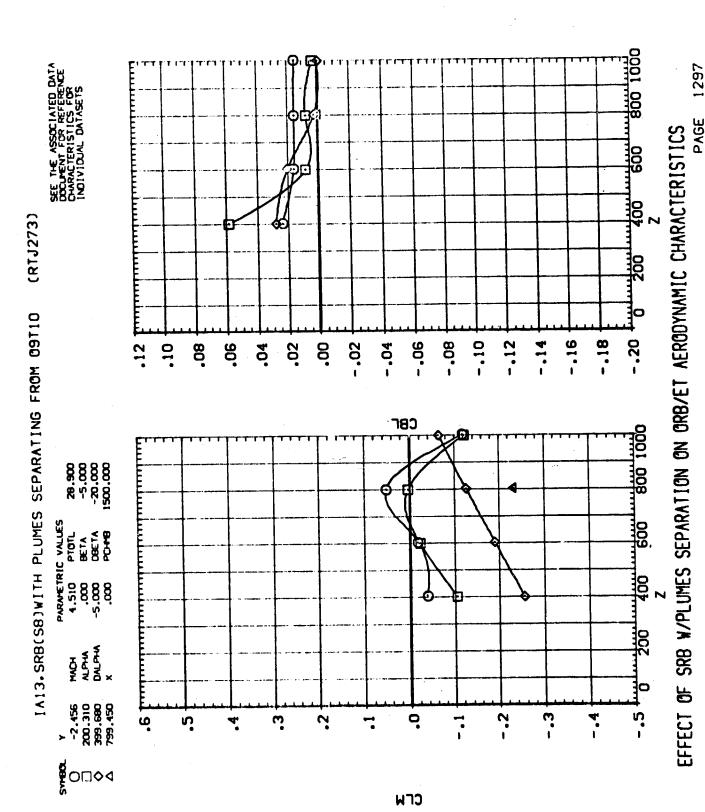
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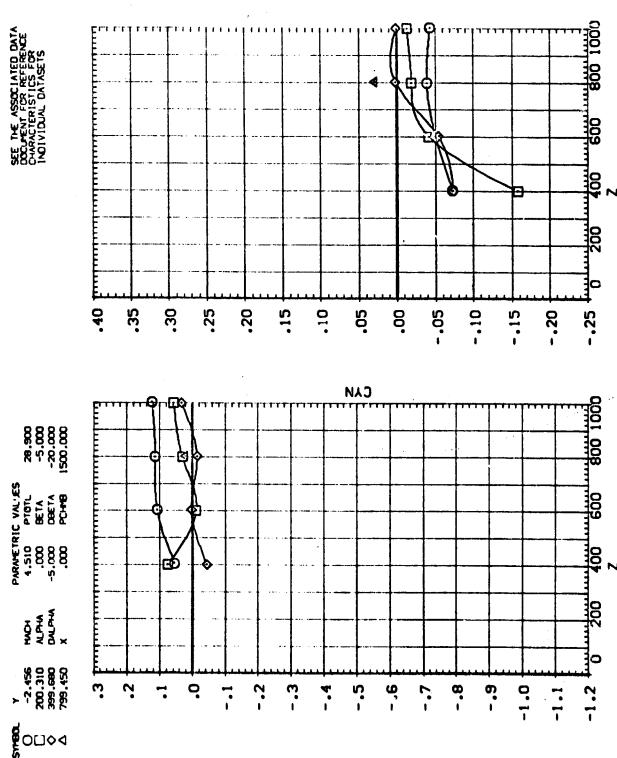
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ273)



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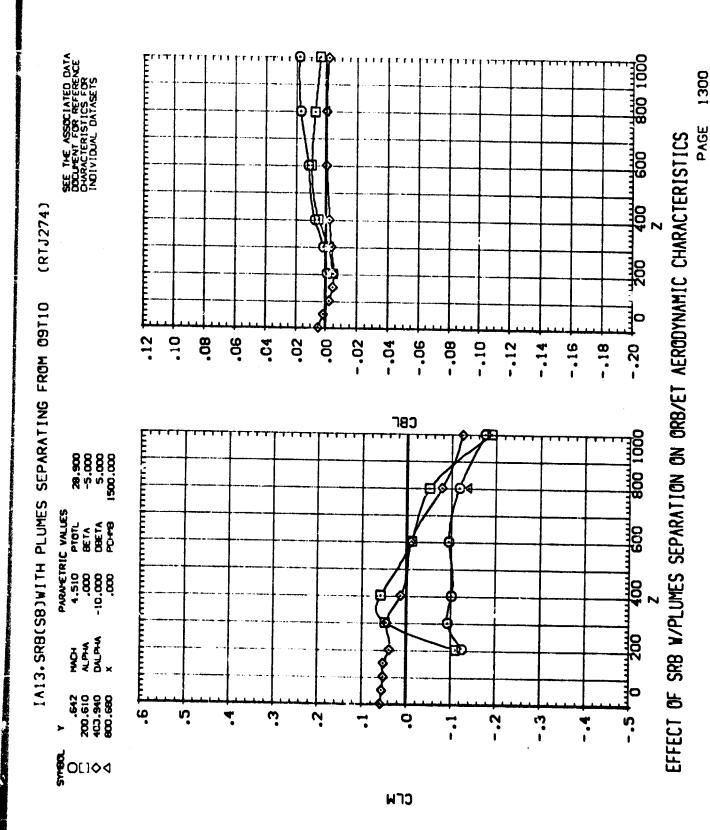
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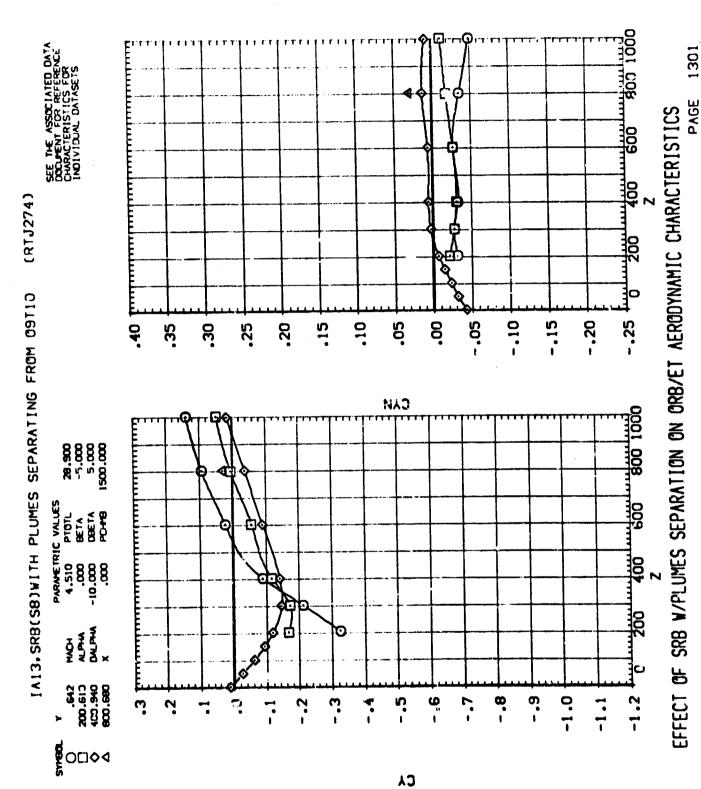
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EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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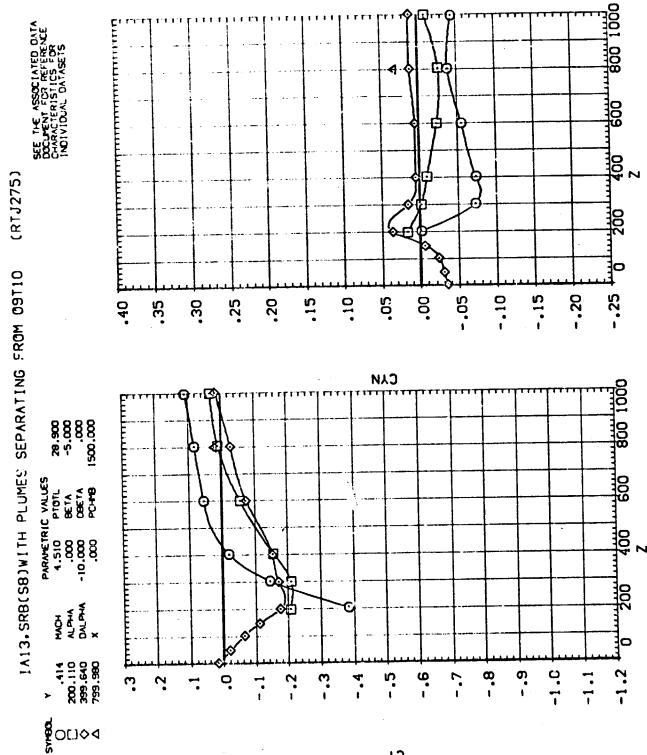
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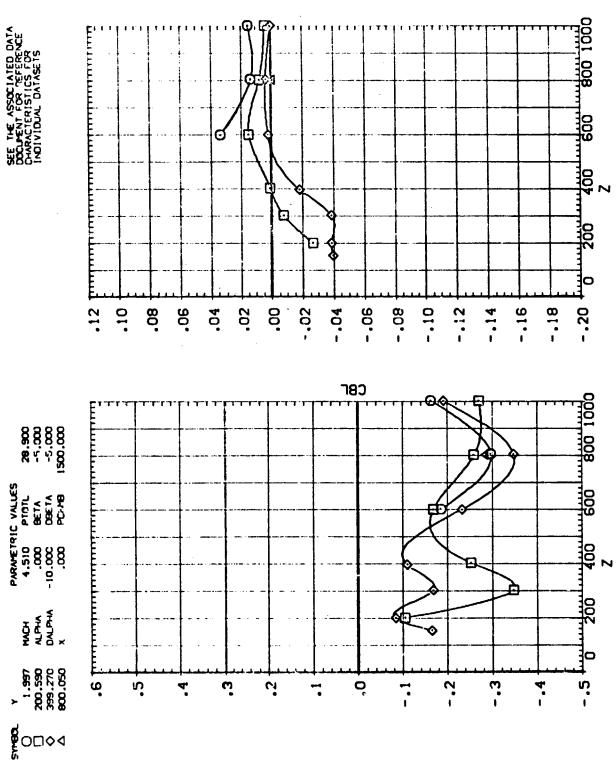
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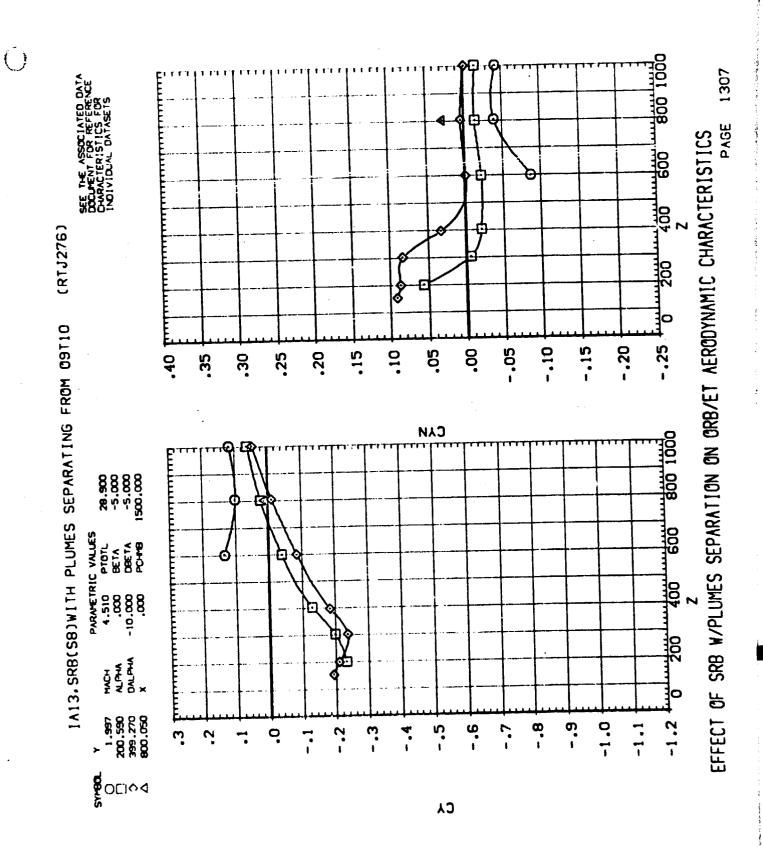
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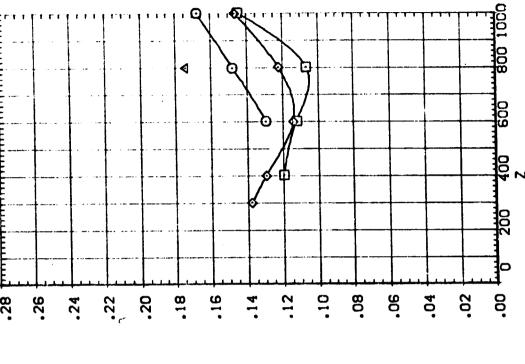
(RTJ276) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



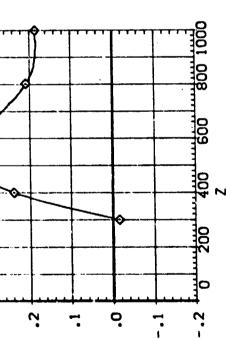
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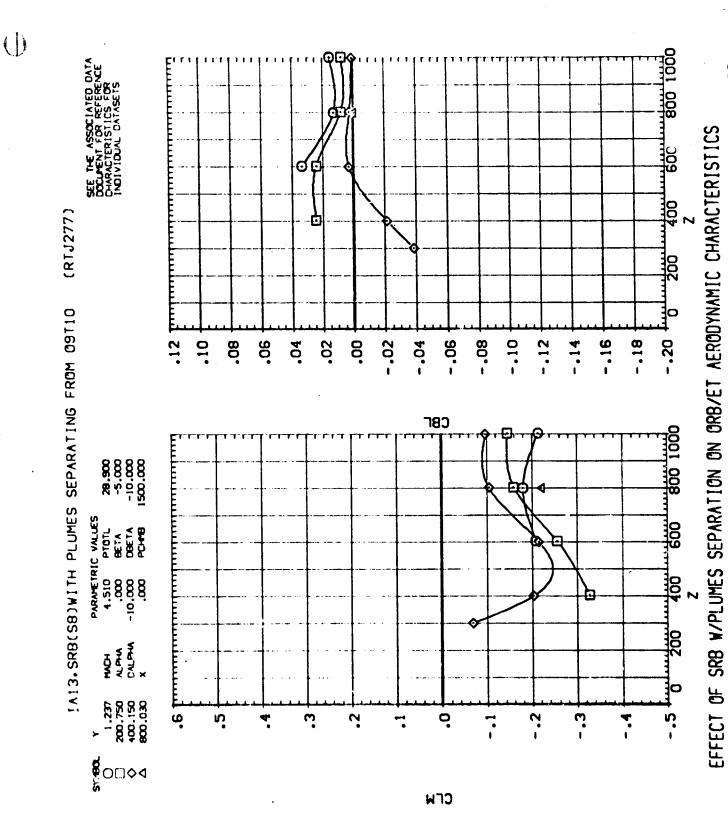
1308 PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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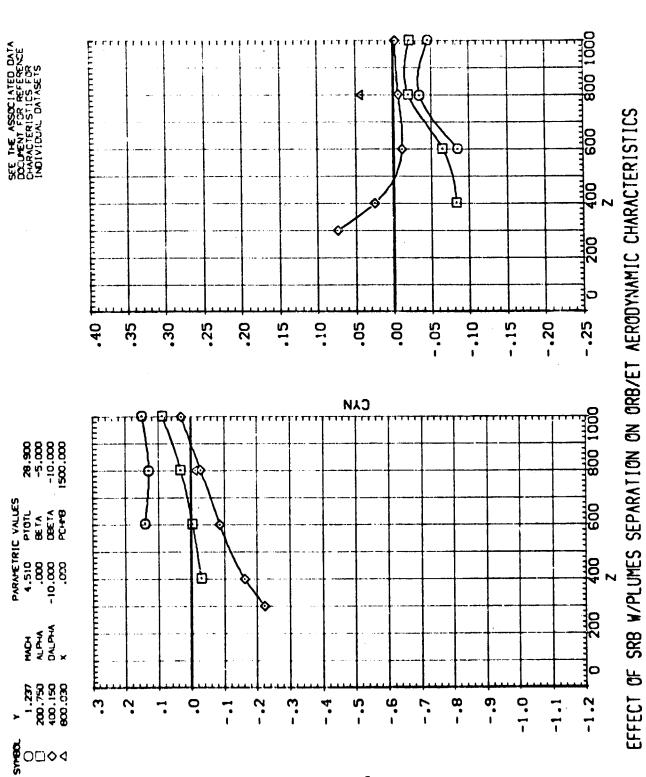
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ277)

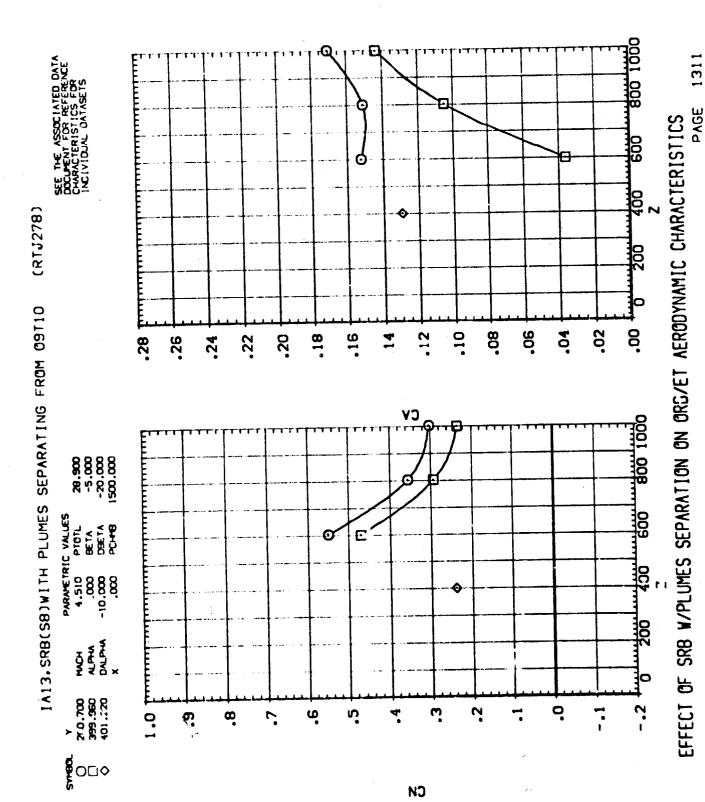


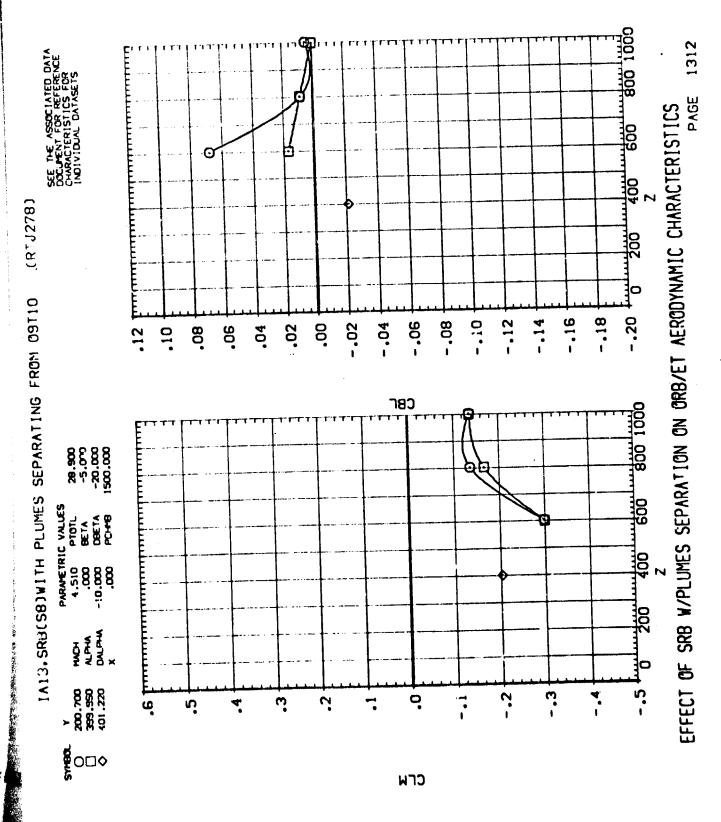
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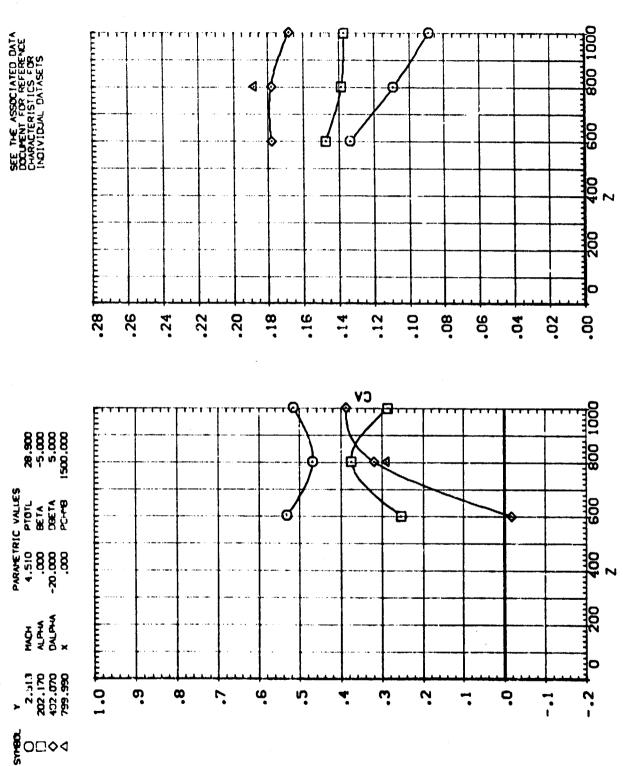
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ279)



PAGE EFFECT OF SRB W/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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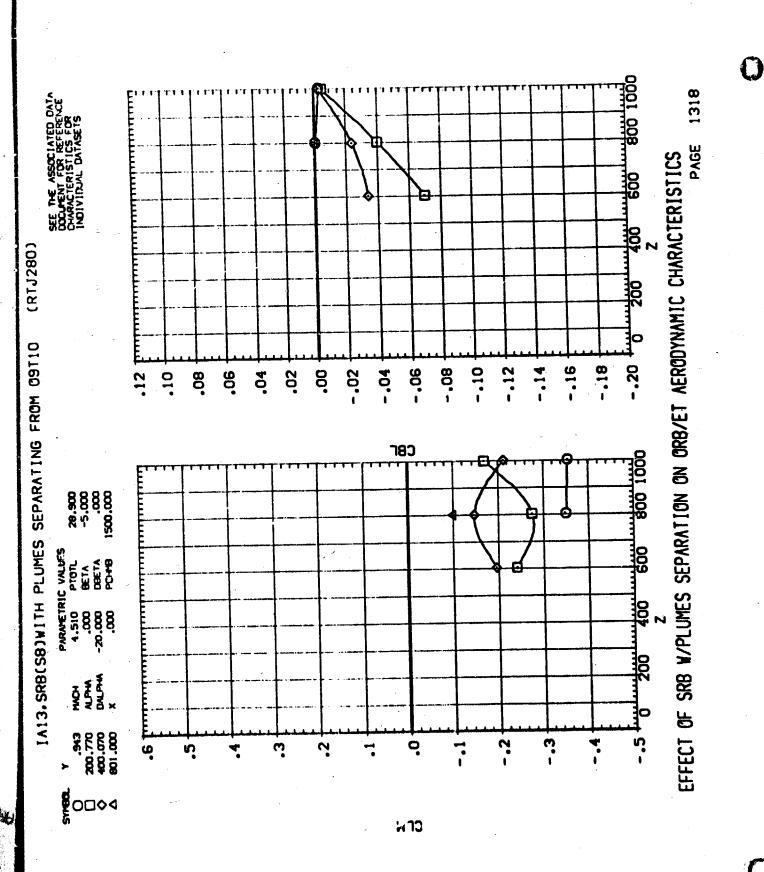
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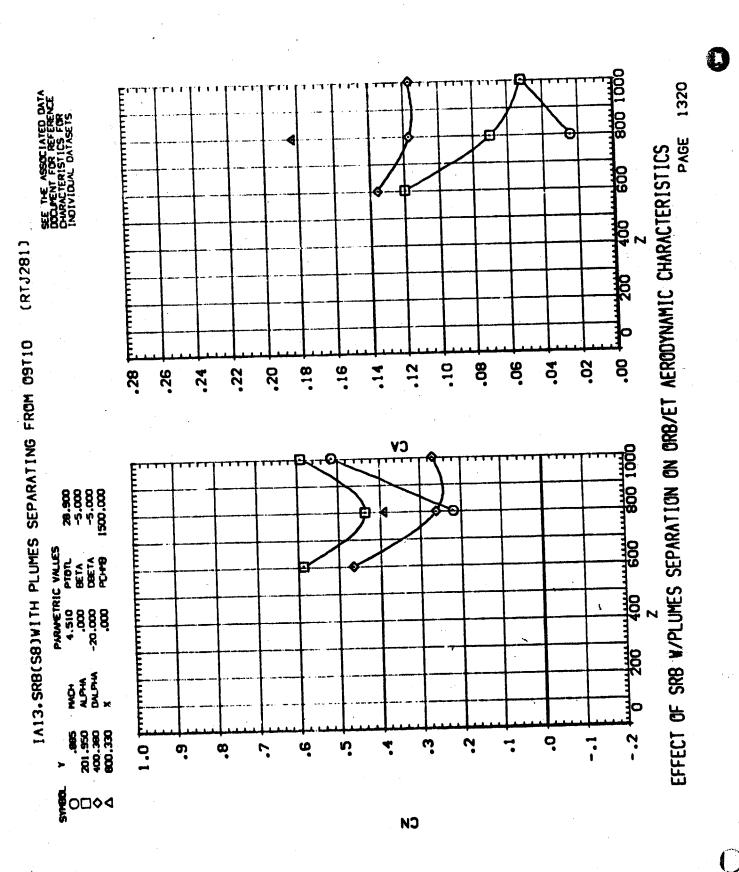


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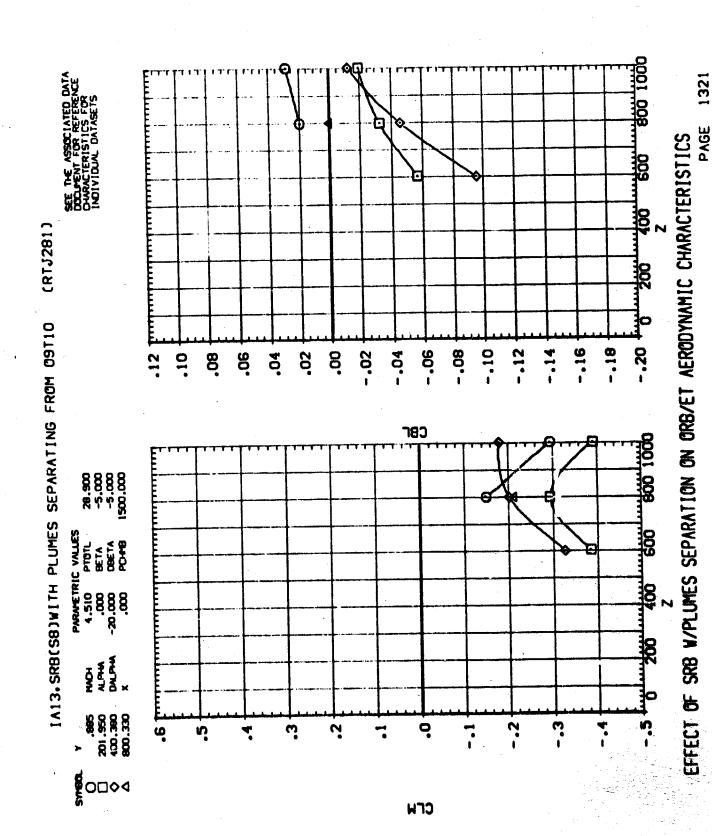
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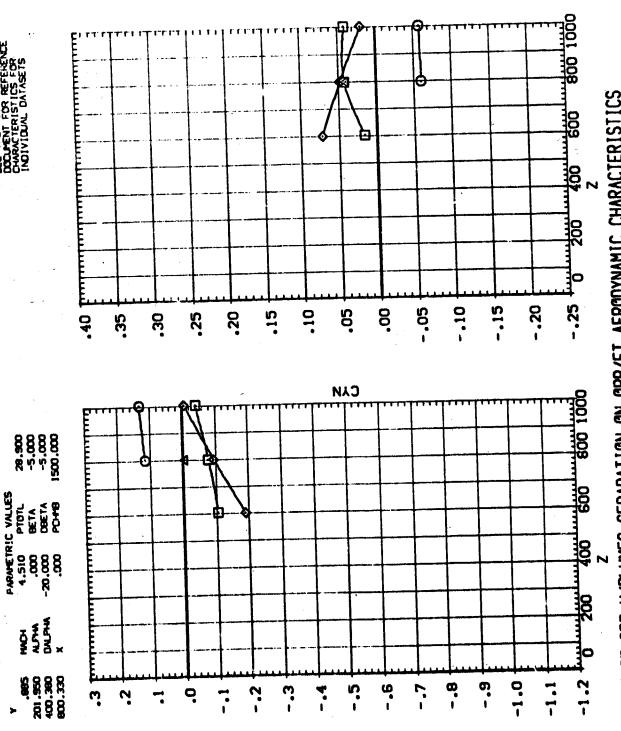
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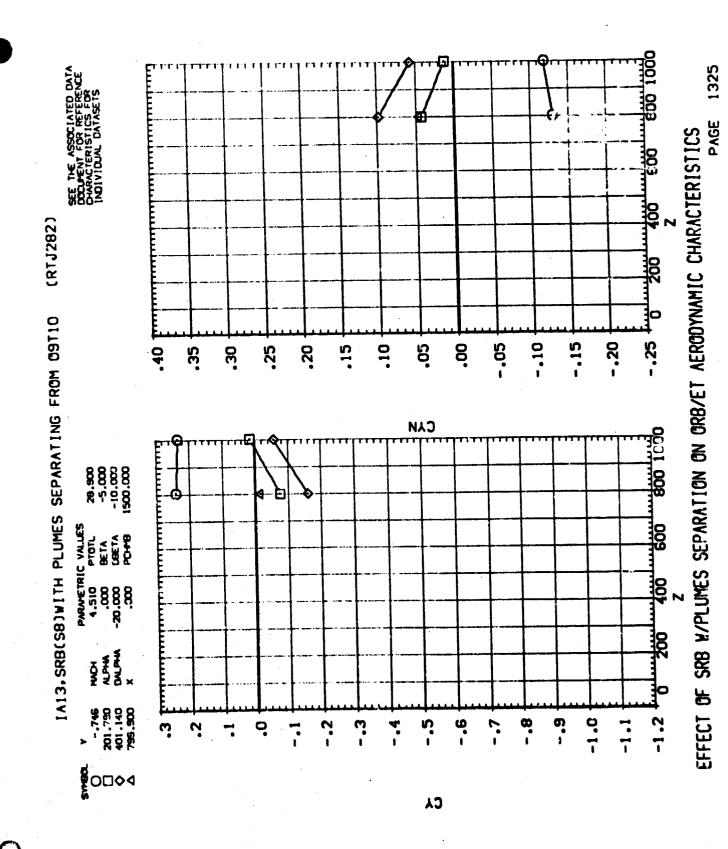




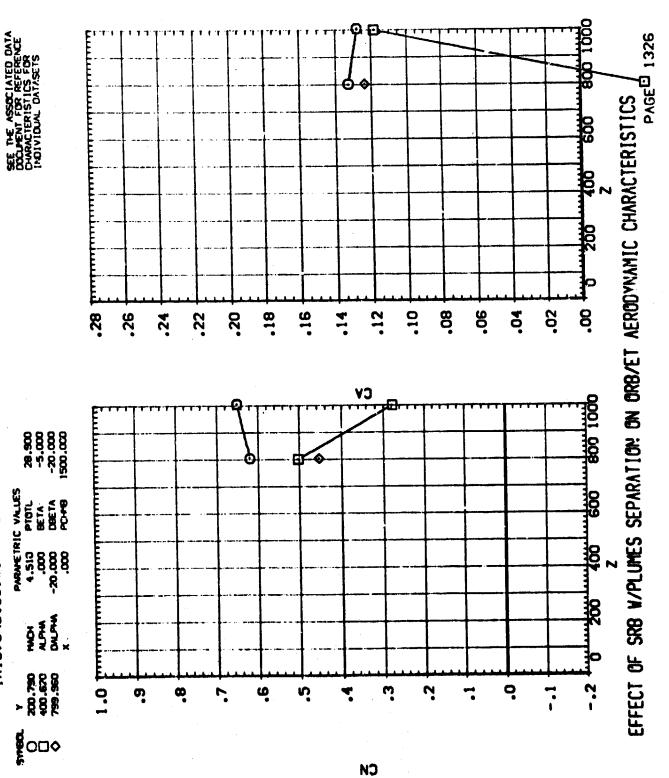
1322 PAGE EFFECT OF SRB #/PLUMES SEPARATION ON ORB/ET AERODYNAMIC CHARACTERISTICS

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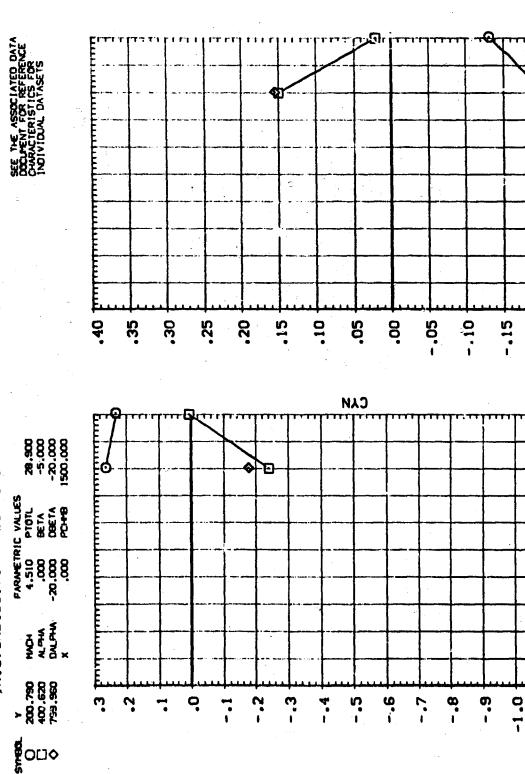
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IA13. SRB(SB)WITH FLUMES SEPARATING FROM 09110 (RTJ283)



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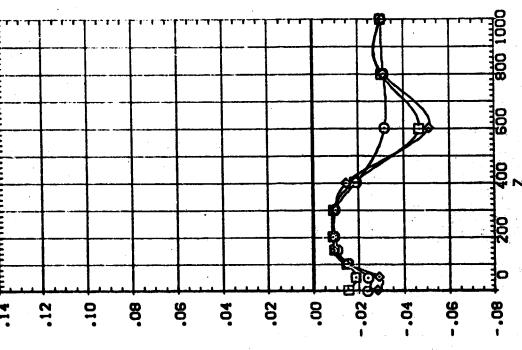
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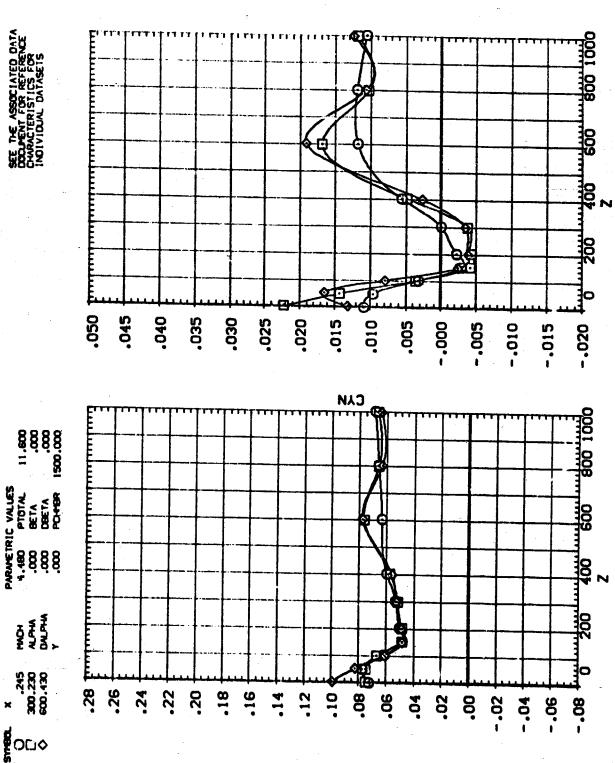
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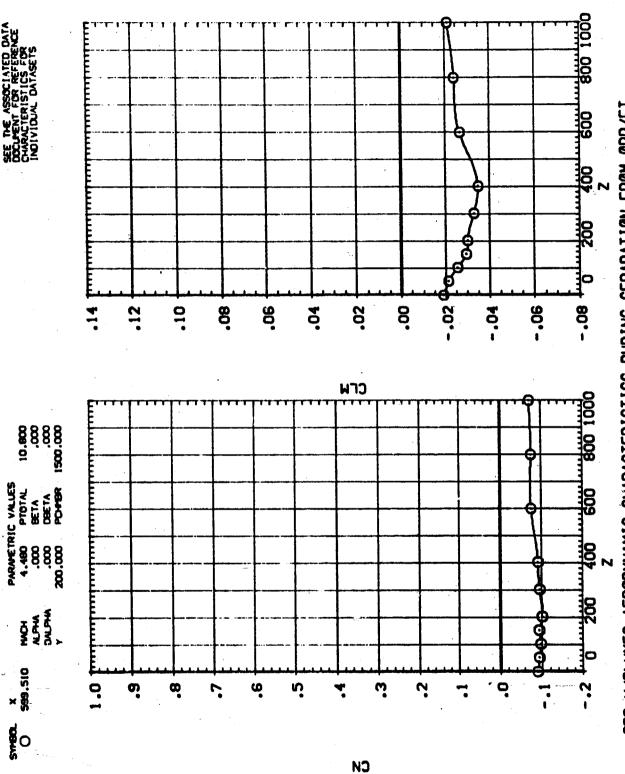
IAI3. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ300)



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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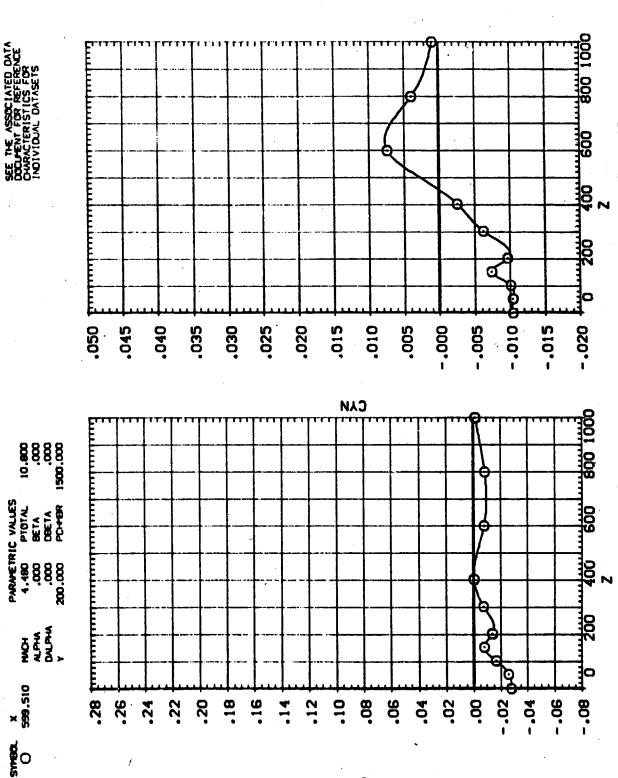
IA13, SRB(SB)WITH PLUMES SEPARATING FROM 09T10 (RTJ301)



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

PAGE 1331

(RTJ301) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE DAMPACTERISTICS FOR INDIVIDUAL DATASETS (RTJ302) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09TIO .12 .10 8 9 9 .02 8 -.02 -.04 CLM 2. % 8888 A.510 PTOTAL.
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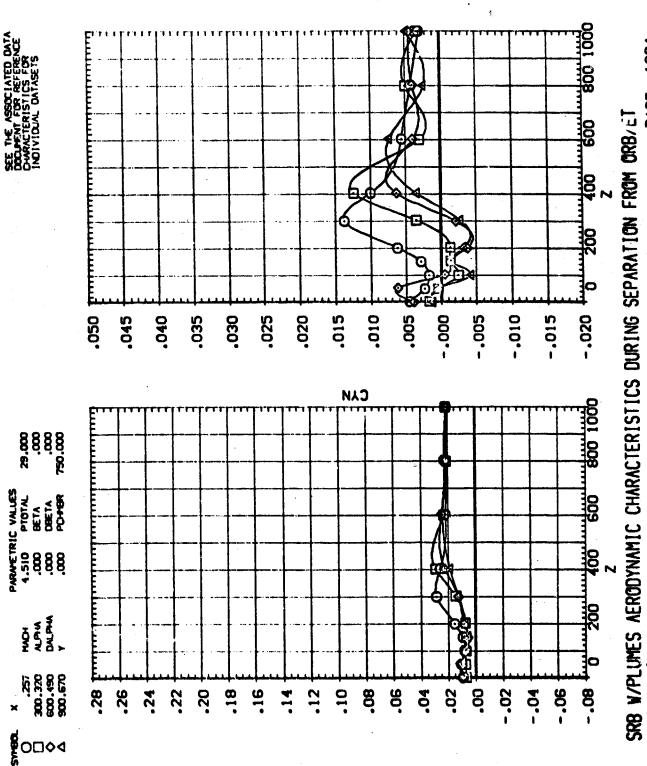
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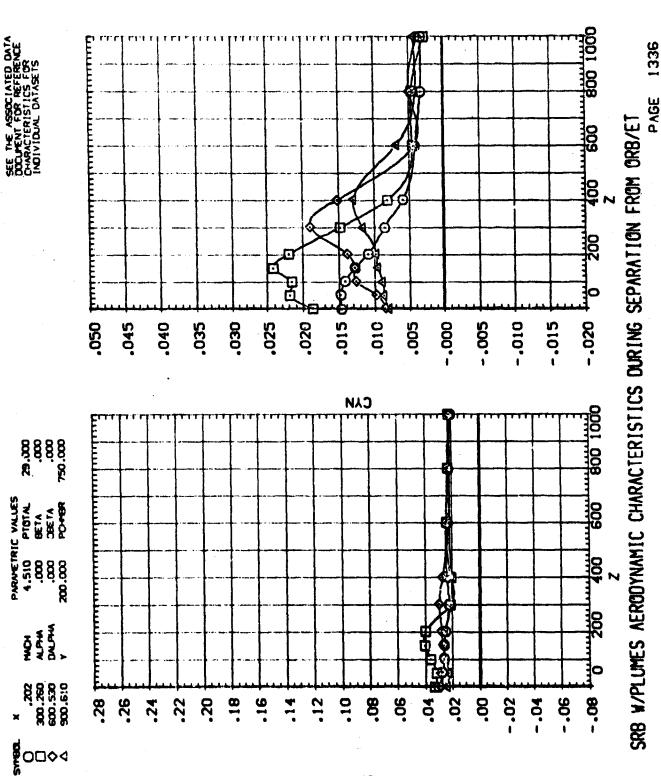
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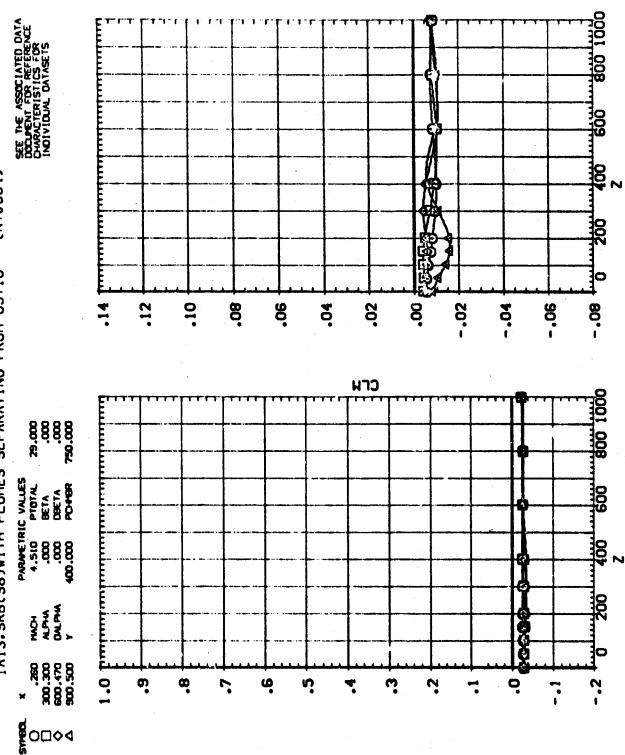
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ303)



(RTJ304) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



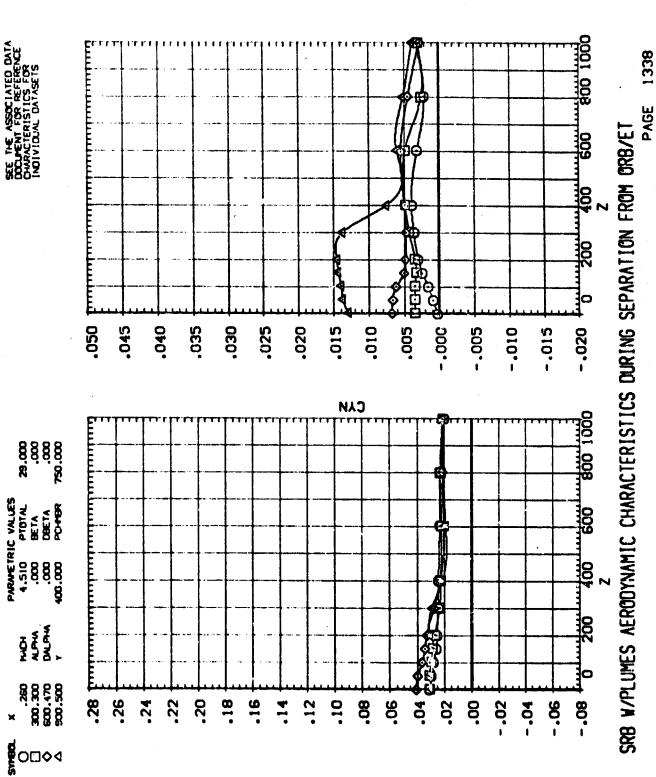
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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IA13.SRB(S8)WITH PLUMES SÉPARATING FROM 09110 (RTJ304)

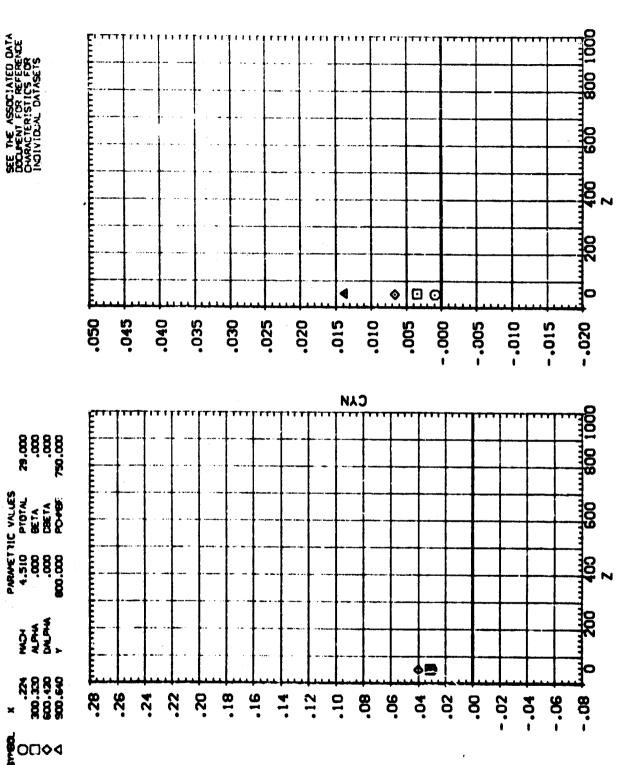


SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 18 (RTJ305) 以以 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 **B**4 .12 .10 80. 90. .02 2 . 90. 8 -.02 -.04 -.06 CLM 800 1000 PARAMETRIC VALUES 4.510 PTOTAL .000 BETA .000 DBETA .000 PDM-9R 1.0 minimization 18 MOT MOTE DALPHA .224 300.320 600.420 900.640 ò ດ 8 . ທຸ ó n 4 -:1 **E**O□ ◊< CM

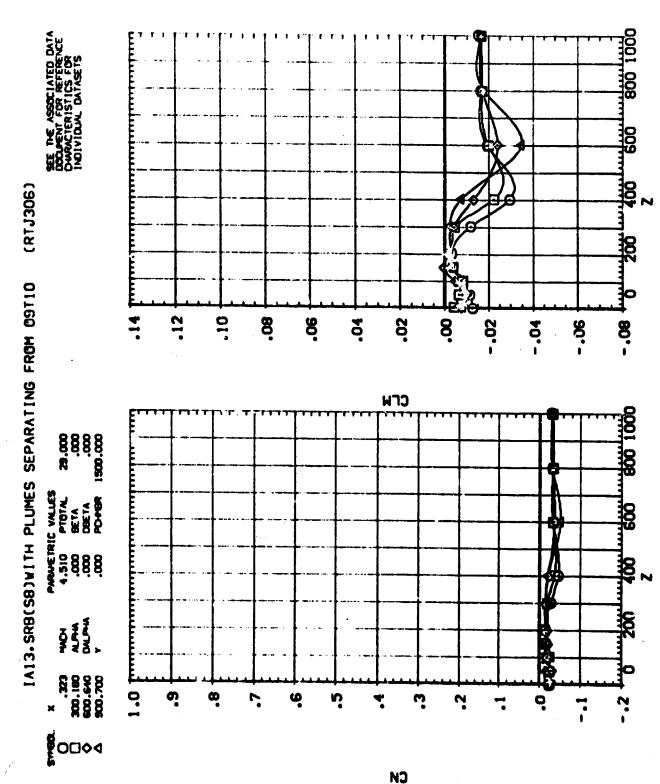
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ305)



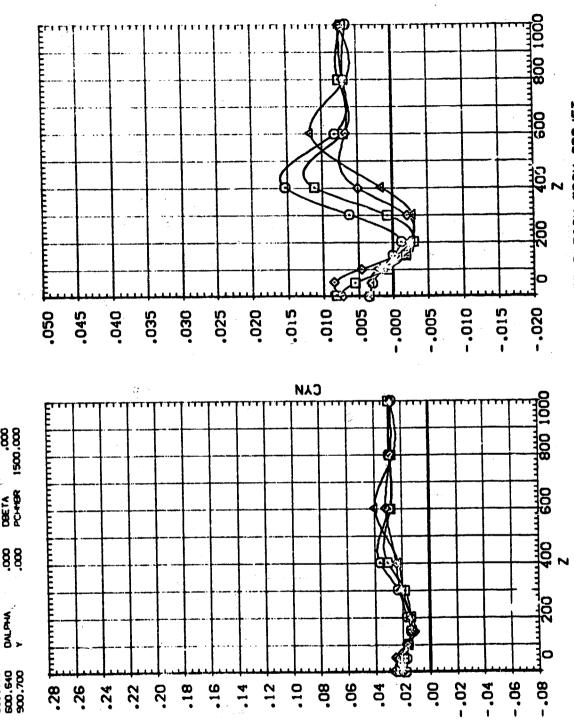
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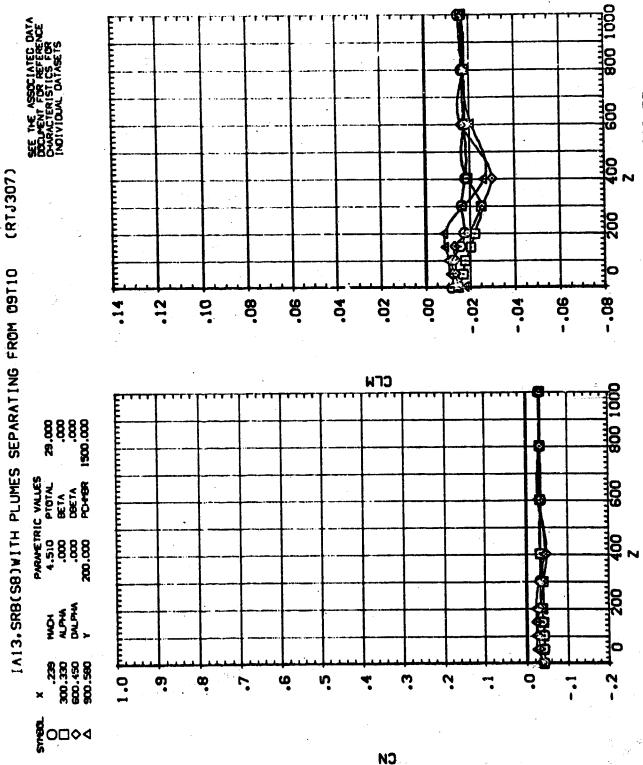
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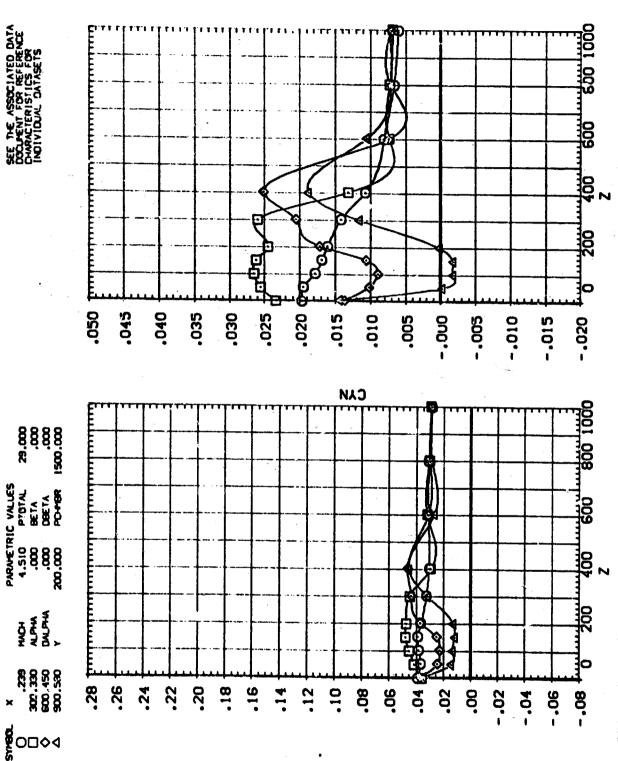




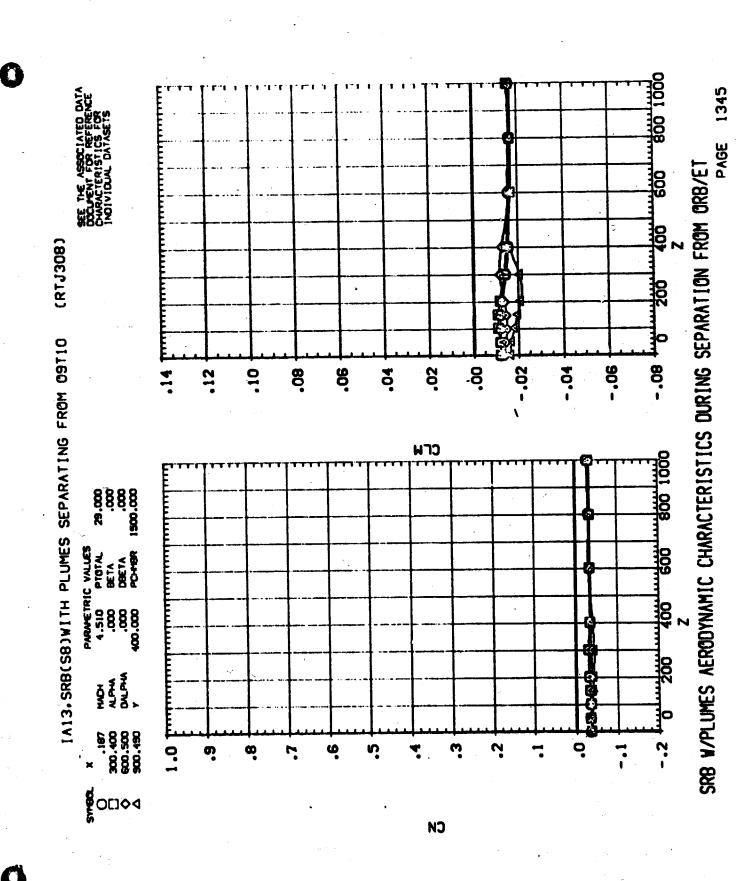
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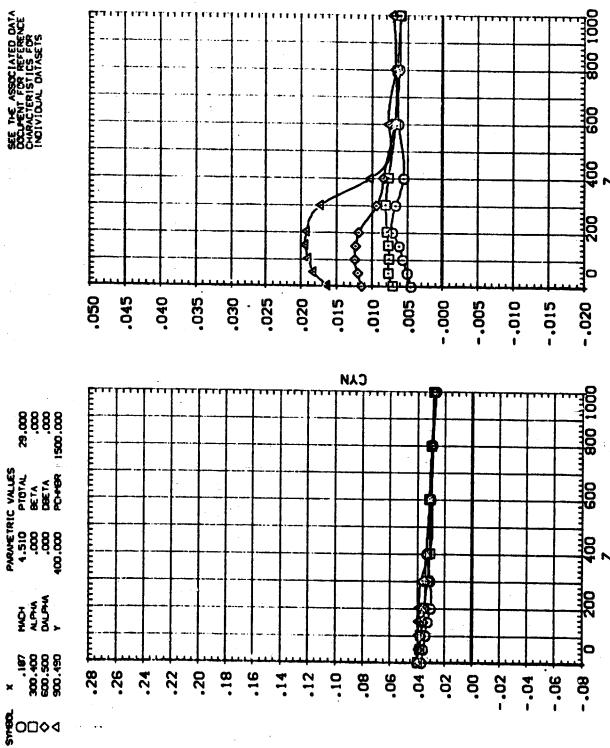


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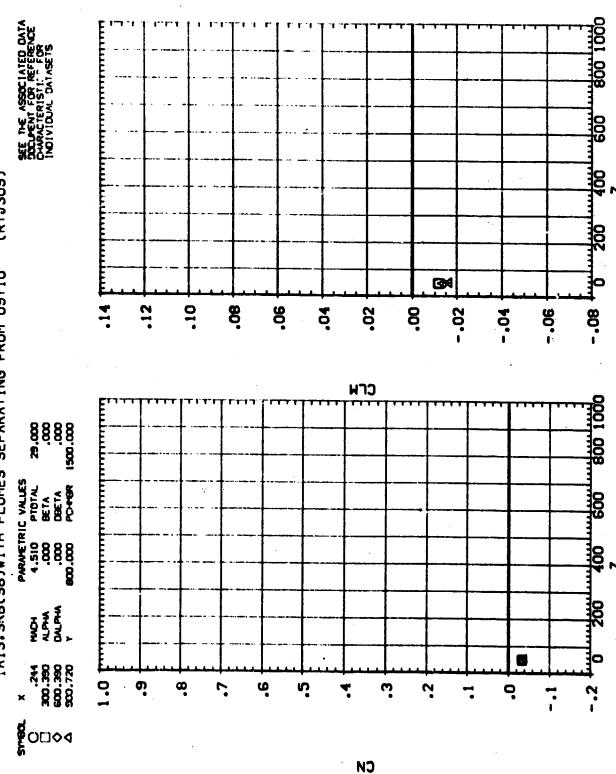




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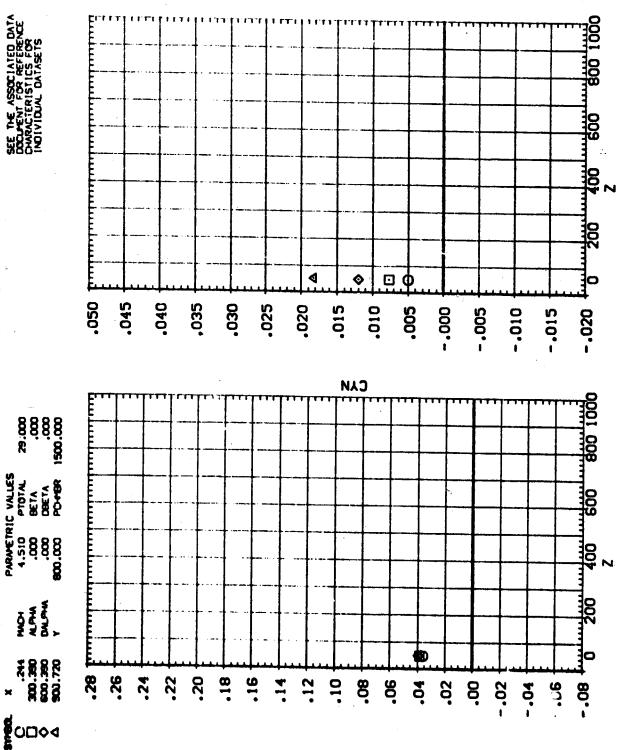
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(RTJ309) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110

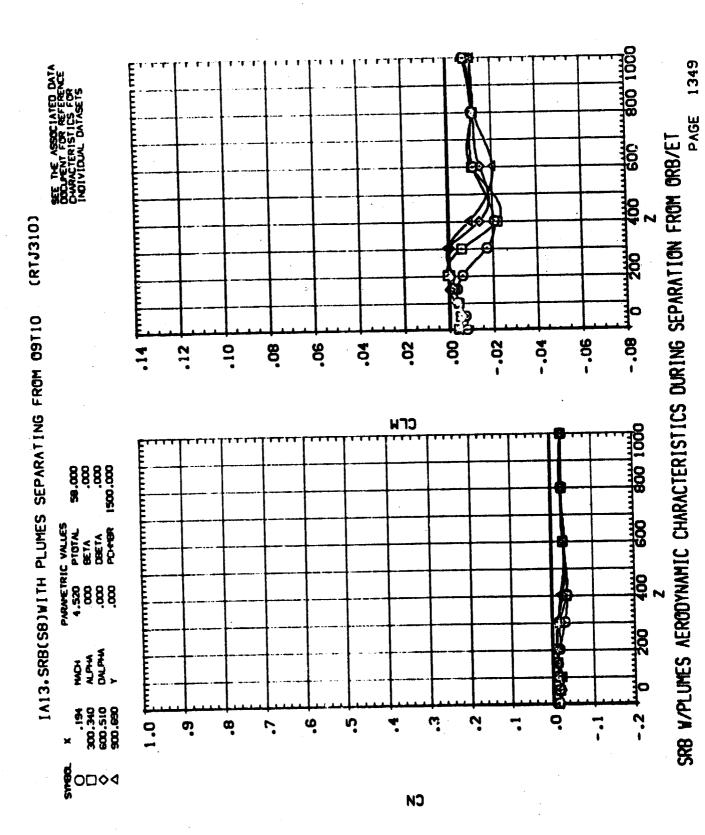


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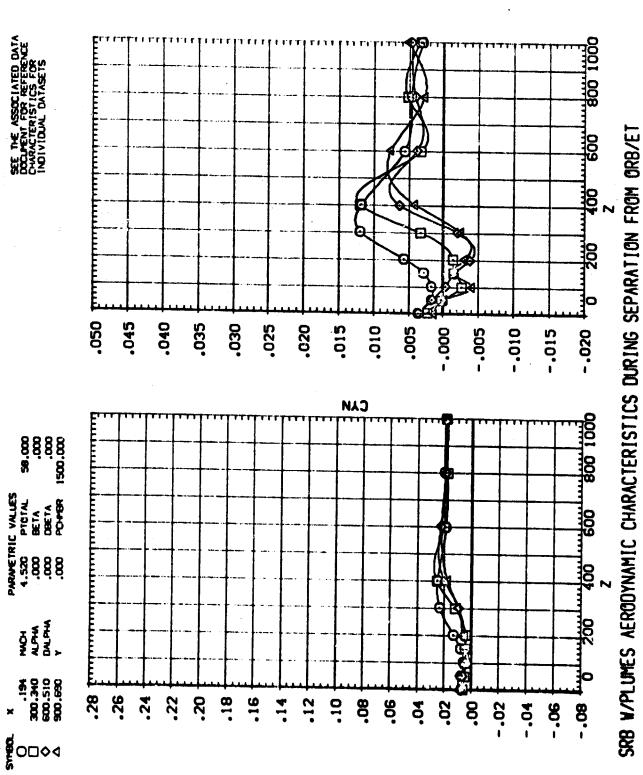
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET





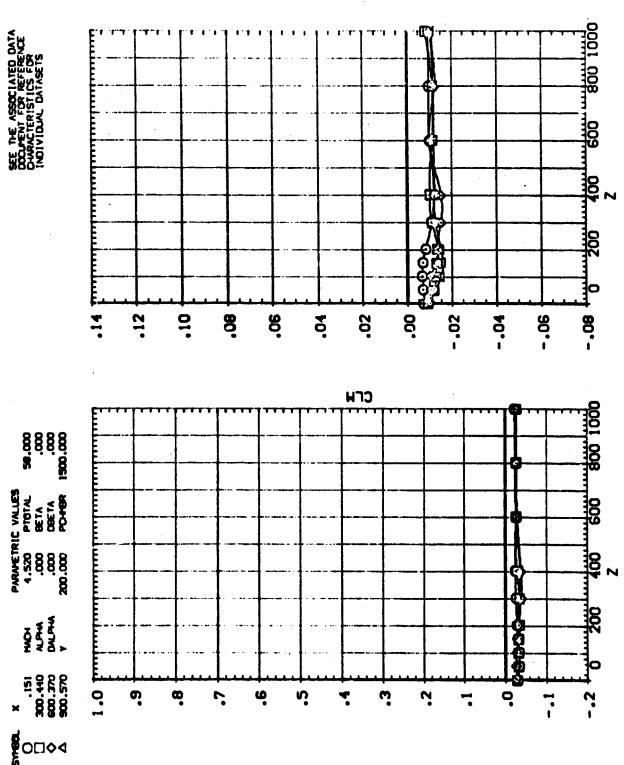


(RTJ310)

IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09T10 (RTJ311)

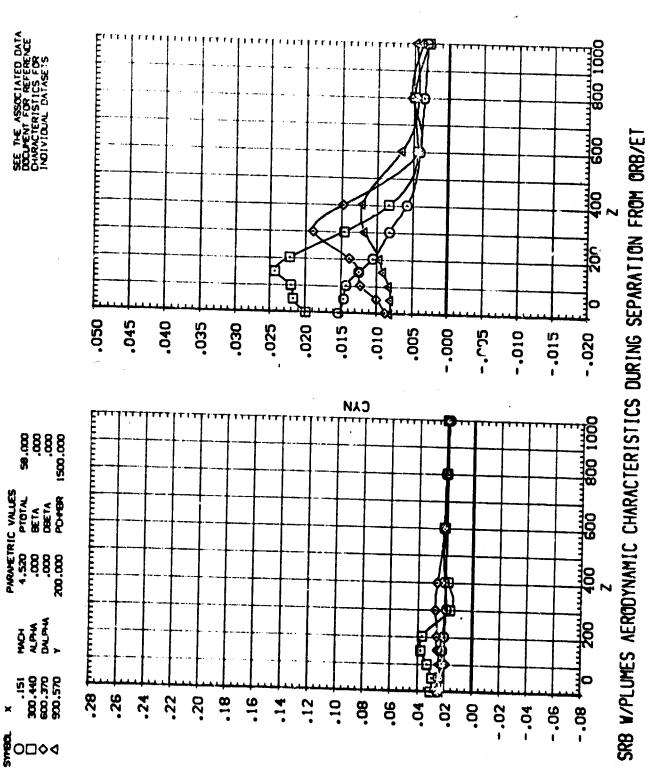


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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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IA13.SRB(SB)WITH PLUMES SEPARATING FROM 09710 (RTJ311)



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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 909 (RTJ312) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710 -.08 .14 .12 .10 8 9 .02 8 -.06 o. -.02 -.04 СГН 800 1000 95.000 000.000 000.000 PARAMETRIC VALLES
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400.000 POMBR 1 100 18 MON APA PACPER 600 .197 300.430 600.450 900.620 0.1 ທຸ ຄຸ ? 0 œ Ģ 'n . -: **E**O□◊4

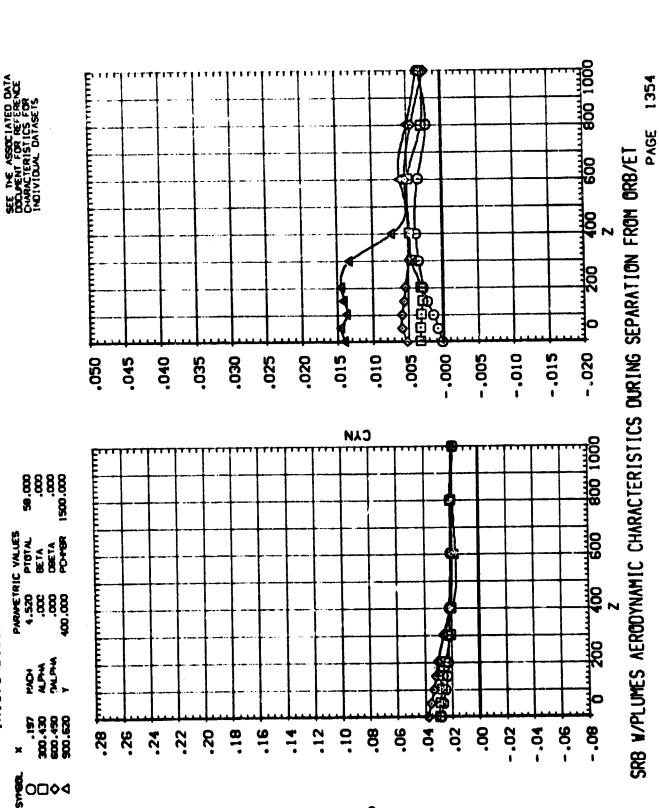
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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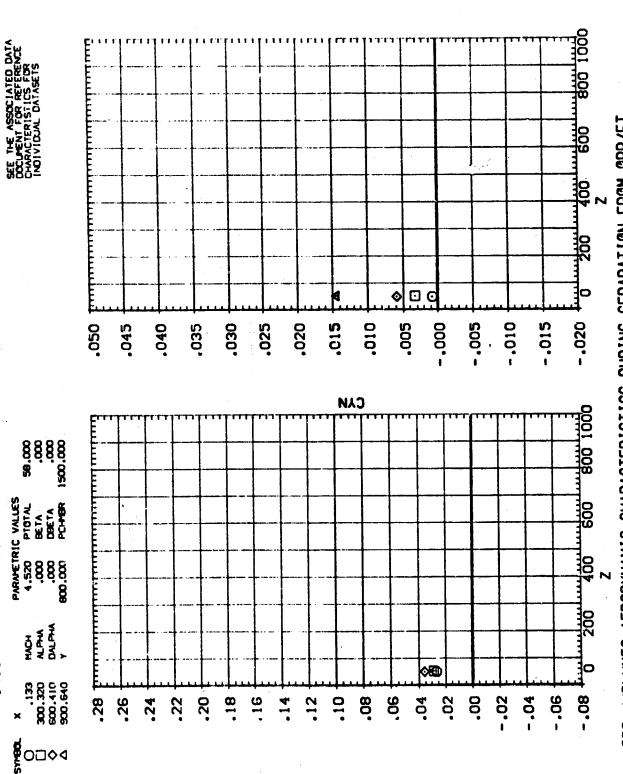


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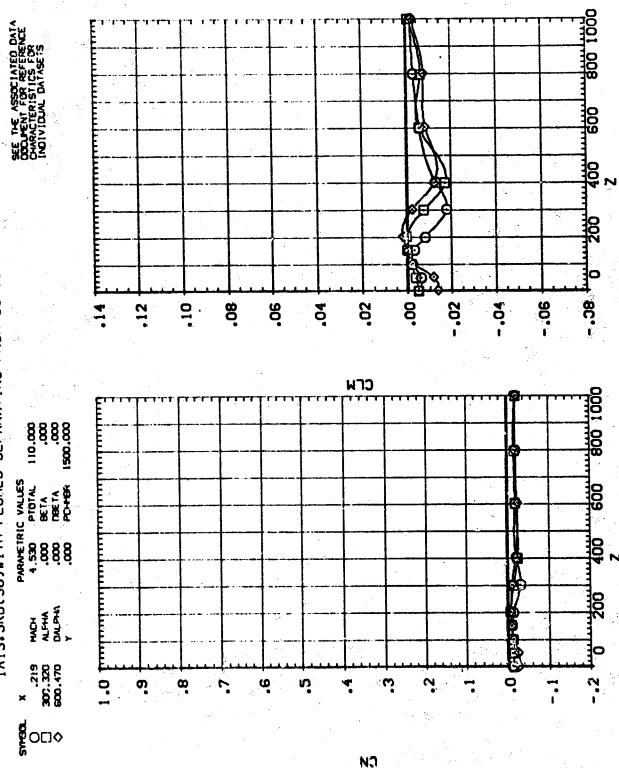
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ313)



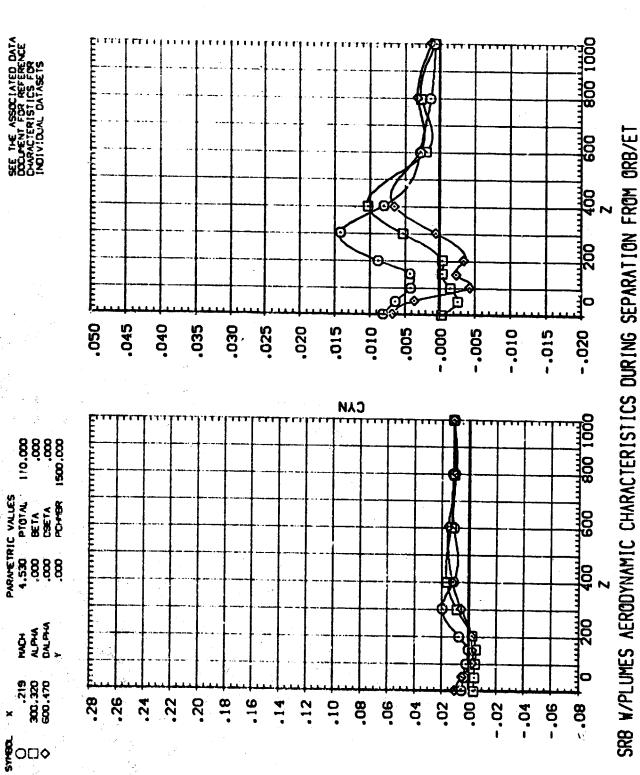
PAGE SRB W./PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ314)

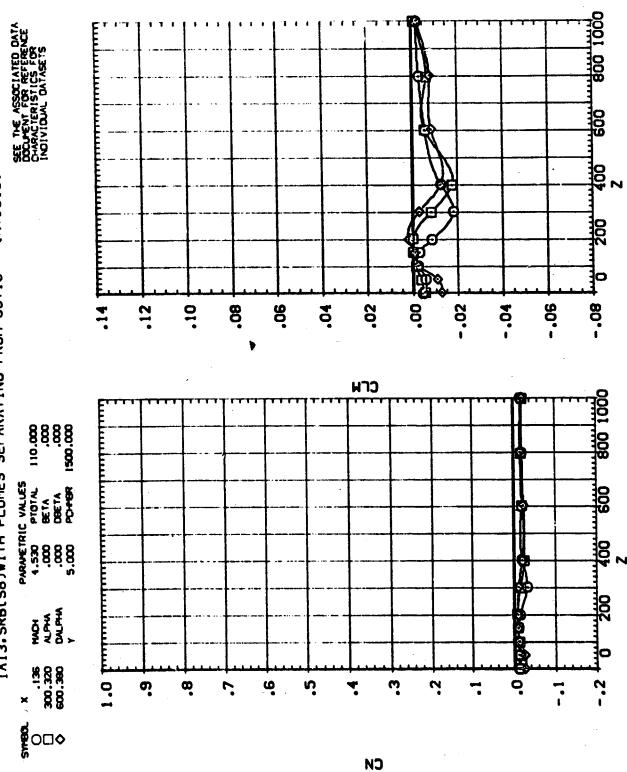


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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ314)



IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ315)



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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 600 18 (RTJ315) 器 SEPARATING FROM 09110 .050 .045 .040 .035 .030 .025 .020 .015 .010 .003 -.000 -.010 -.005 -.015 -.020 CAM 800 1000 150.000 000.000 000.000 IA13. SRB(SB)WITH PLUMES PARMETRIC VALUES
4.530 PTOTAL
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5.000 PCM-98 200 MACH ALPHA DALPHA 300.320 . 6 .26 .22 .24 .20 .18 .16 14 80.-.12 .10 90. 90. **.** .02 -.02 -.04 -.06

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SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ316) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 .12 .10 90. 90. 9 .02 8 -.02 СГЫ 00.02 00.002 00.002 A.530 P107AL .000 BETA .000 DBETA .000 DBETA ACPA DALPHA .176 300.400 600.440 900.920 0. ດ œ . ø ທຸ e. 7

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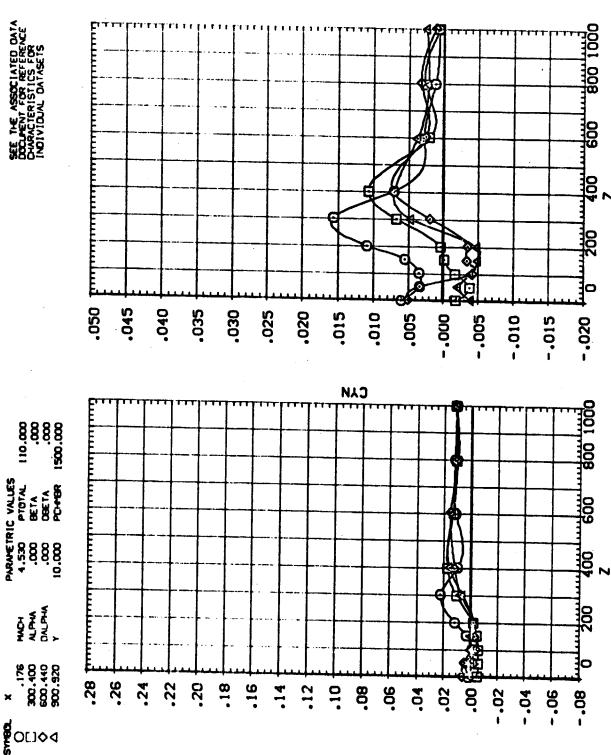
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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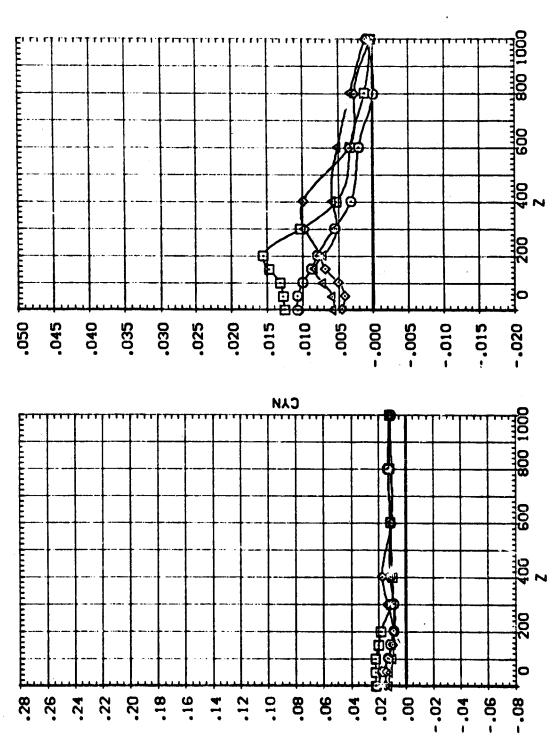
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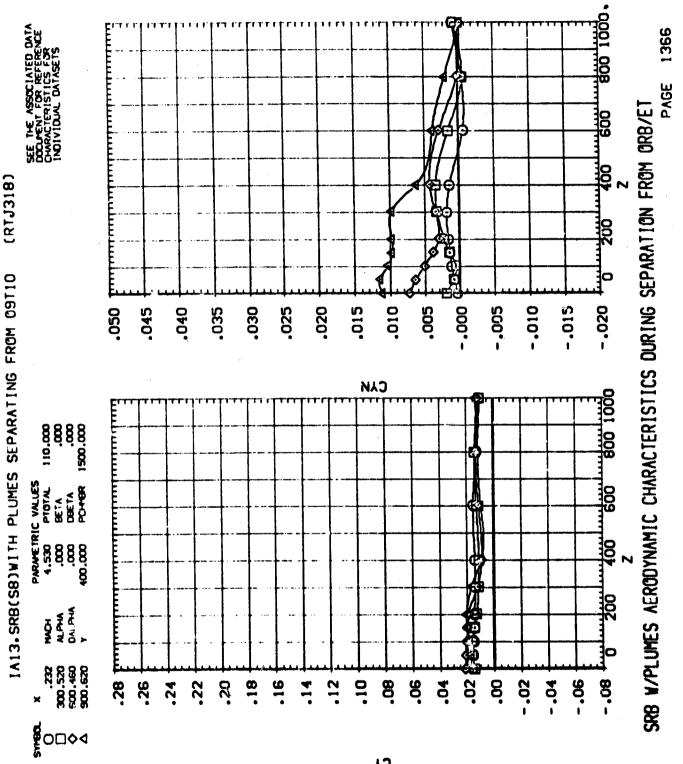
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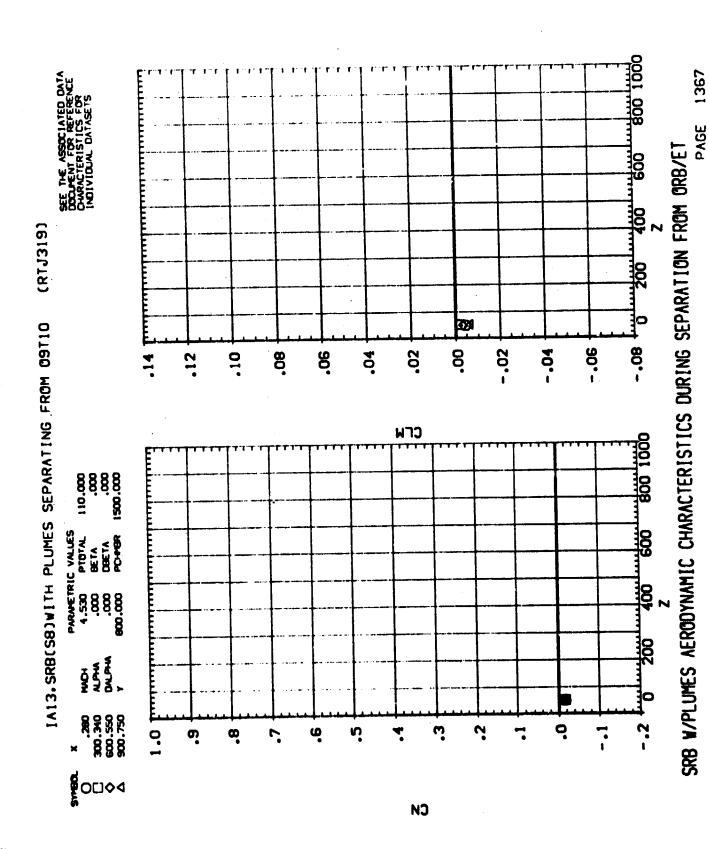




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800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 89 **2**2 (RTJ318) 200 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710 - 08 8 90.--.02 90. .02 .12 .10 8 **.** -.04 CLM 800 1000 000.001 000.0021 PARAWETRIC VALUES
4.530 PTOTAL
.000 BETA
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400.000 PC-PBR 600 **6** ′ 200 MACH ALPHA DALPHA .232 300.520 600.460 900.620 0.1 Ģ ស က တ œ **8**0□◊4 СИ





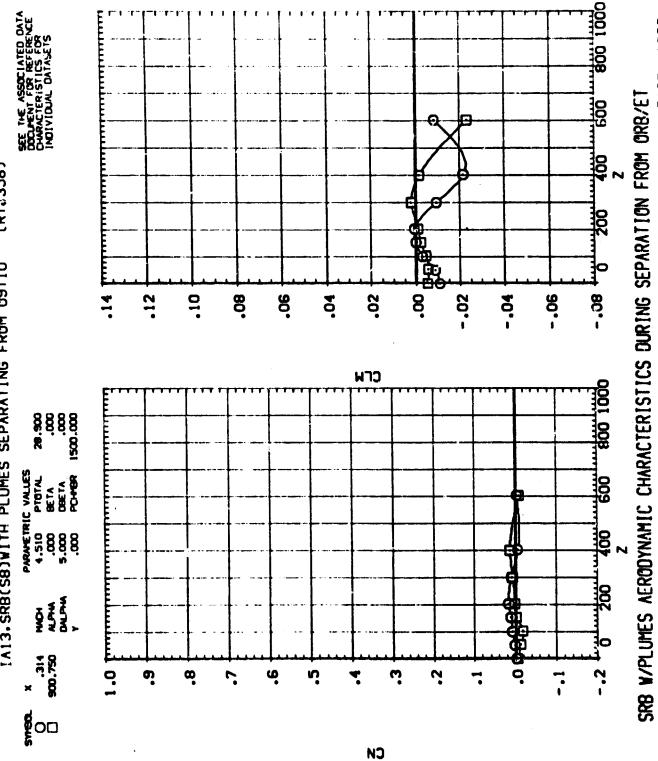
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800 1000 1368 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 100 200 2 (RTJ319) 200 o 0 0 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710 -.020 -.000 -.010 -.015 -.005 .010 .005 .015 .050 .045 .040 .035 .030 .025 .020 CAM 800 1000 10.00 00.00 1500.000 PARAMETRIC VALUES 4,530 PT0TAL .000 BETA .000 DBETA .000 DPETA 466 200 MOT N.P.E. 300.340 600.550 900.750 90.--.08 01. 80. 90. 0. .02 8 -.02 -.04 .18 .16 .14 .12 .26 .22 20 LJ

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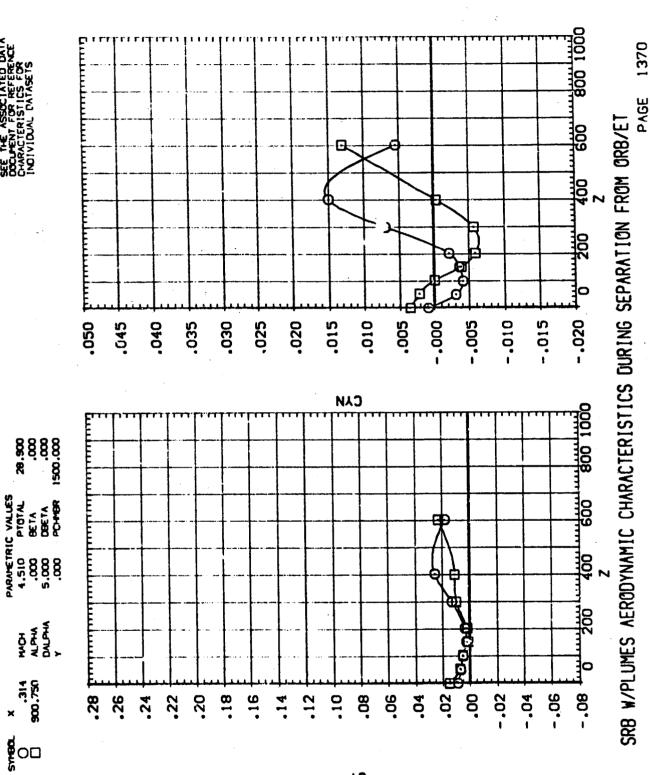
(RTJ358) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110

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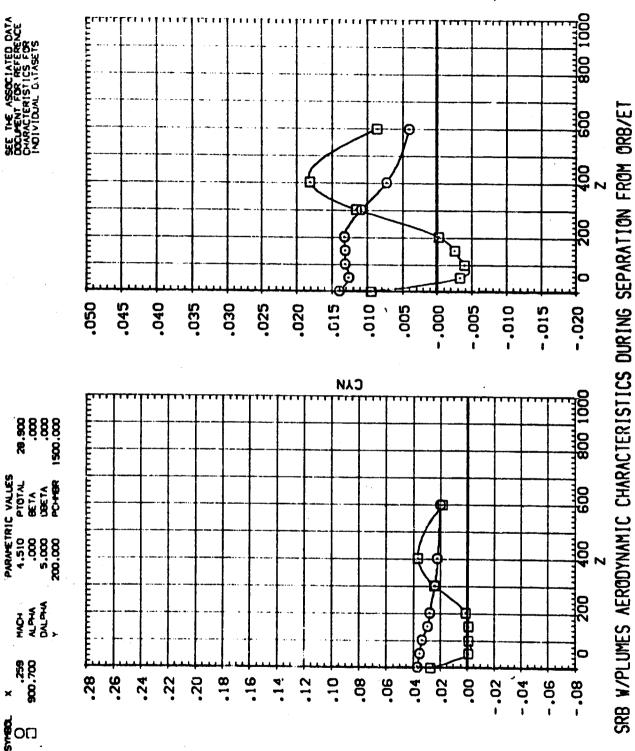
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4.510 PT0TAL
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ359)

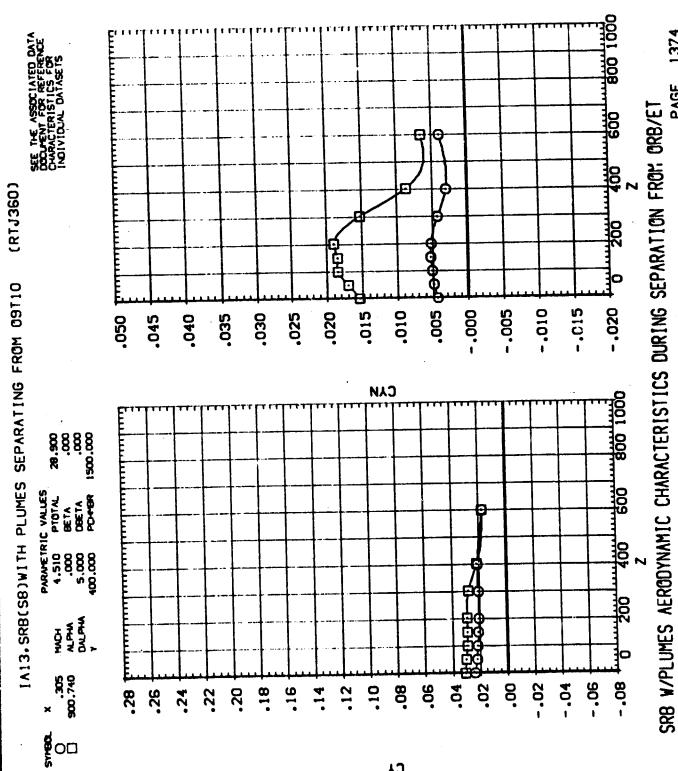


SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 800 1000 89 400 2 (RTJ360) 200 Hopor Hopor IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 O .12 9: 80. 90. **.**04 .02 8 -.02 -.04 -.06 -.08 CLM 800 1000 28.900 .000.0021 .000.0021 PARMETRIC VALLES 4.510 PTOTAL .000 BETA 5.000 DBETA 400.000 PO-PBR 80 400 200 A PAGY PAGA PAGA .305 1.0 o. œ ဖ္ ທຸ 'n 7 ó 1:--.2 ¥ o□

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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

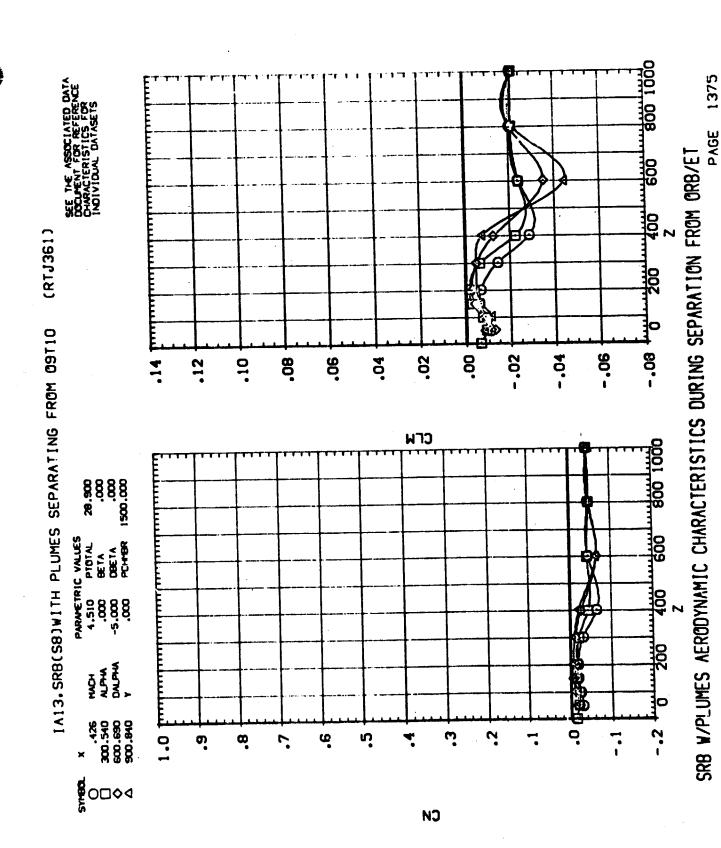


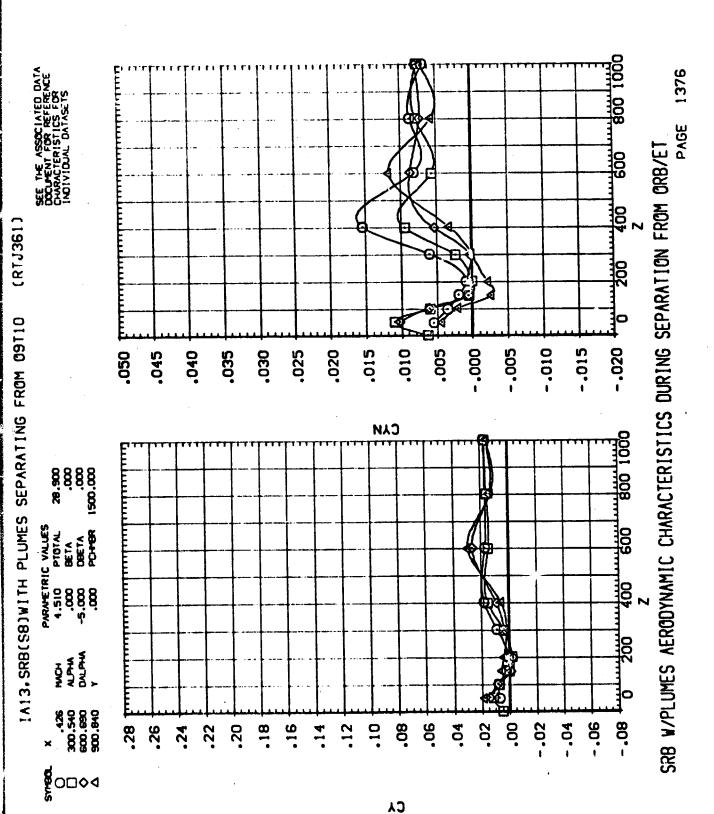
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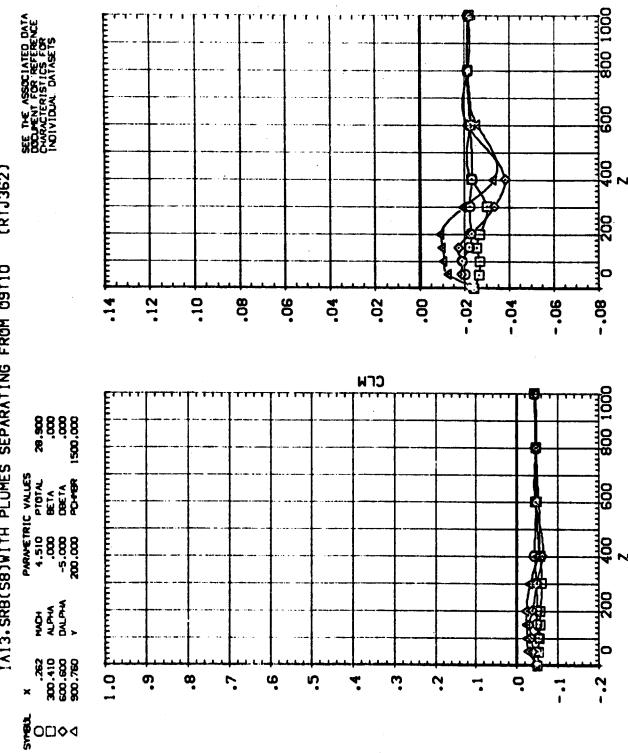




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(RTJ362) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710

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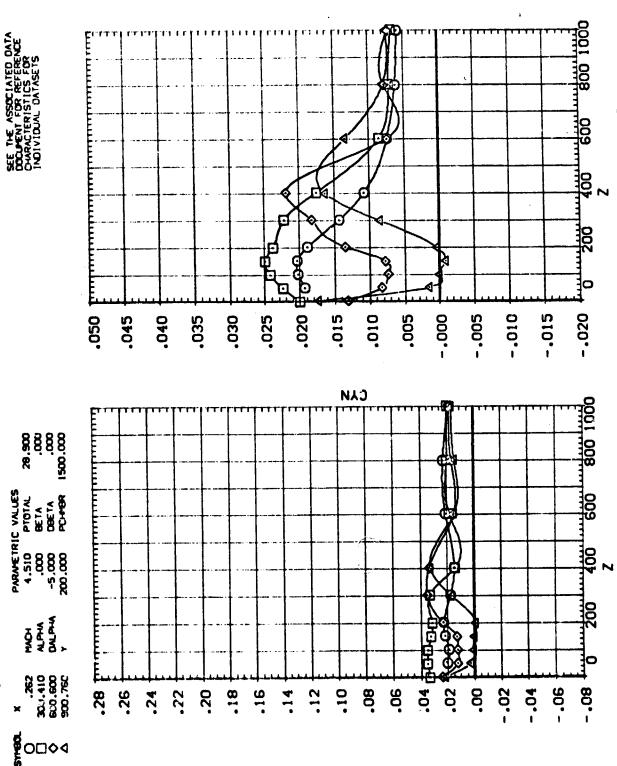


SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ363) IA13, SRB(SB) WITH PLUMES SEPARATING FROM 09110 -.02 (90.-90. .02 8 -.04 .12 2 80. •04 СГЫ 28.90 .000. .000. .000. .000. PARAFERIC VALLES
4.510 PTOTAL
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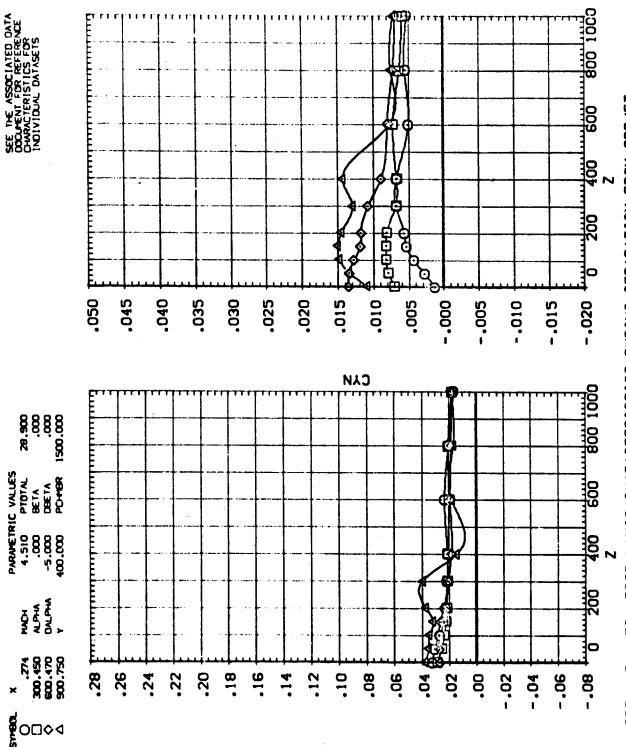
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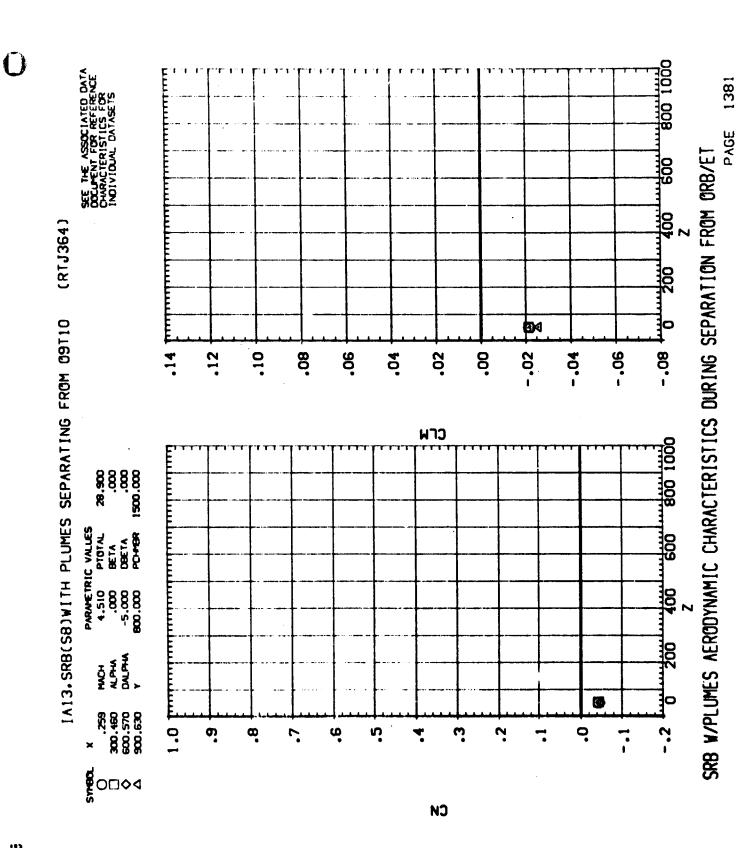
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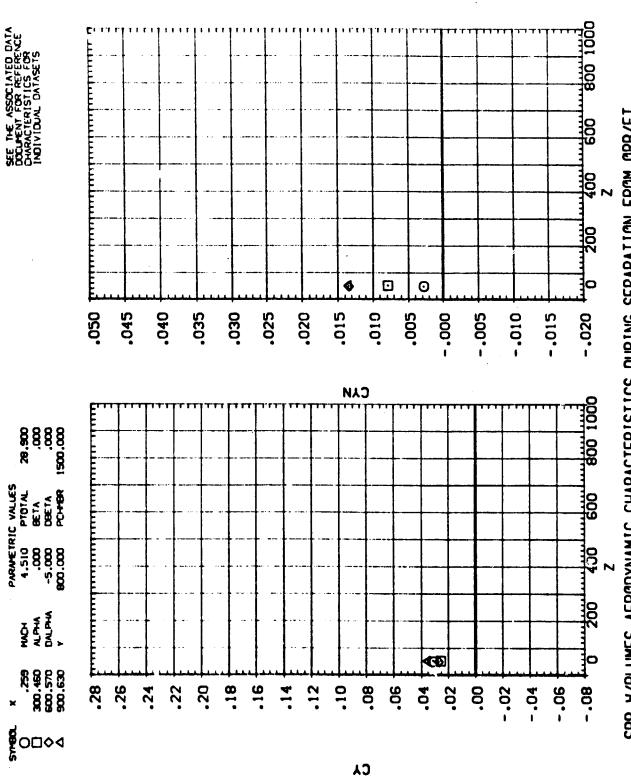
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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(RTJ364) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710



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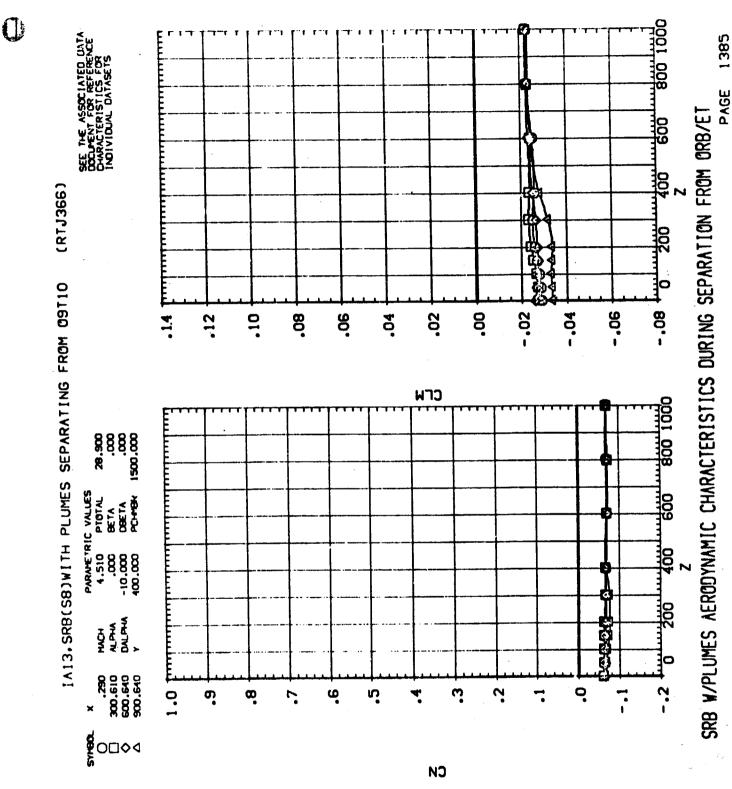
800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS 600 400 (RTJ365) 200 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 P 24 - .03 .12 .10 90. 90. .02 -.02 **.** 8 -.04 90.-כרש 800 1000 28.90 .000. .000. .000. PARAVETRIC VALUES
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4.510 PTOTAL
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600.000 PC-PBR 200 MACH ALPHA OALPHA **]**0 **4 0** -.646 300.430 600.590 300.710 - 08 -.06 .10 80. 90. .02 8 -.02 -.04 .16 .18 . 14 .12 .26 20 .24 .22 C ф

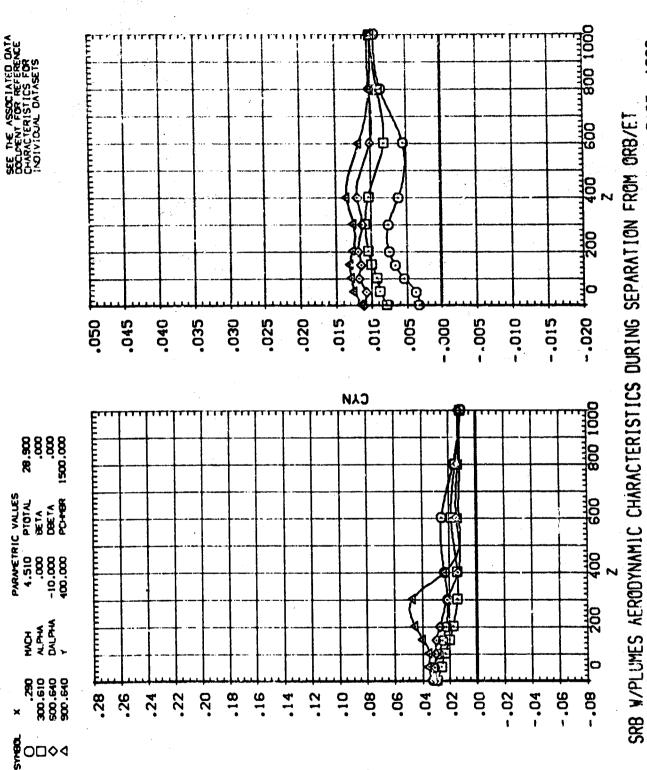
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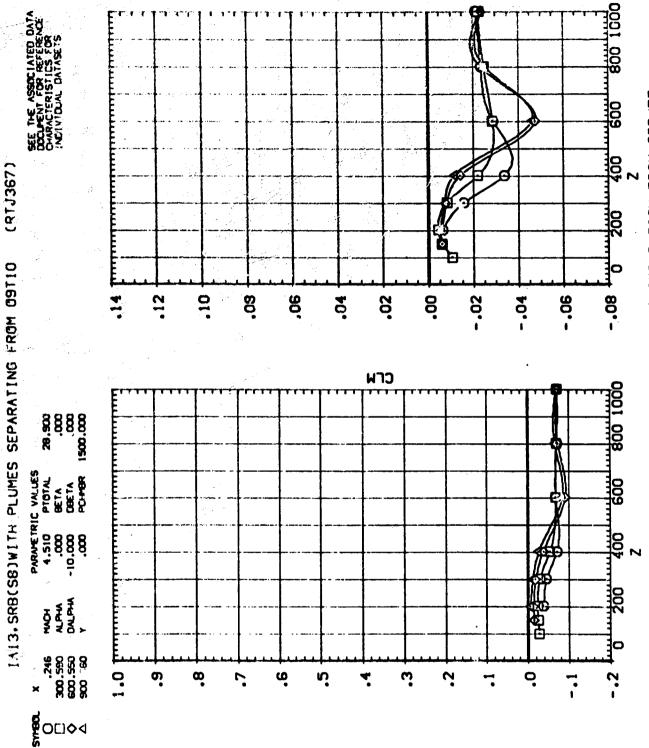
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710 (RTJ366)



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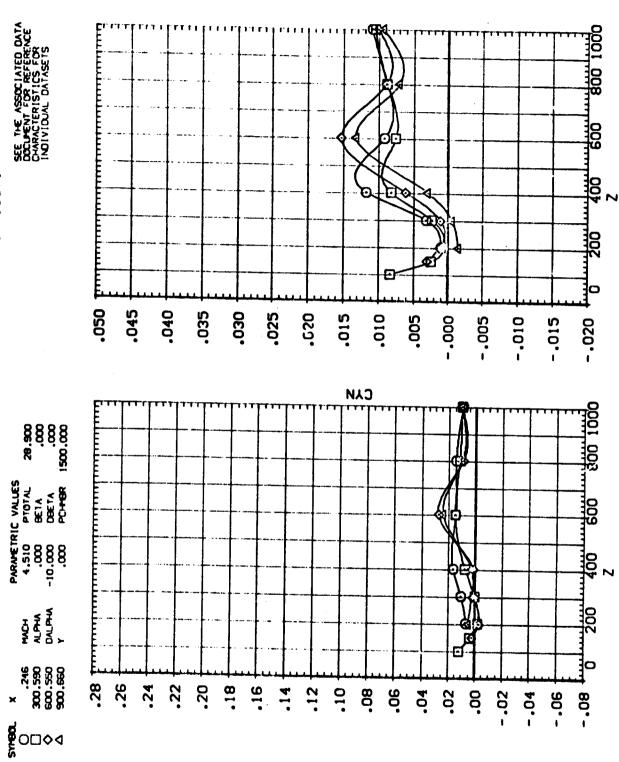


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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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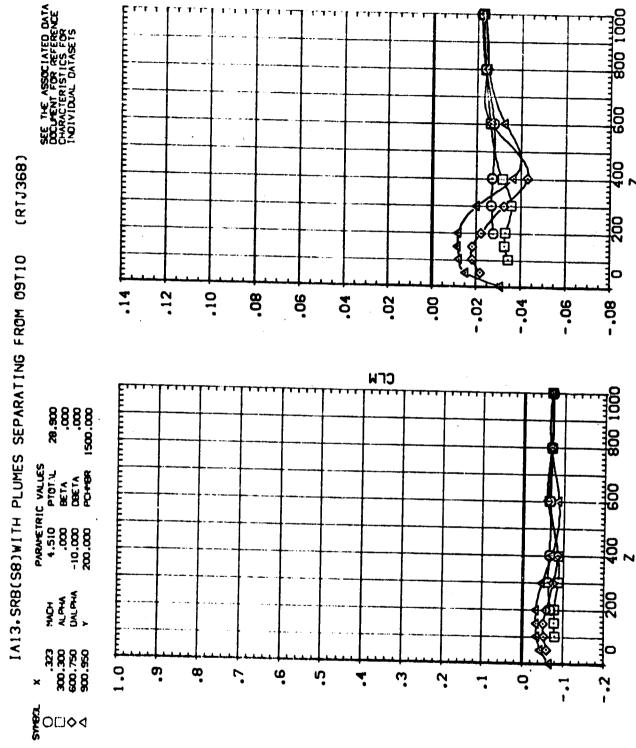
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ367)



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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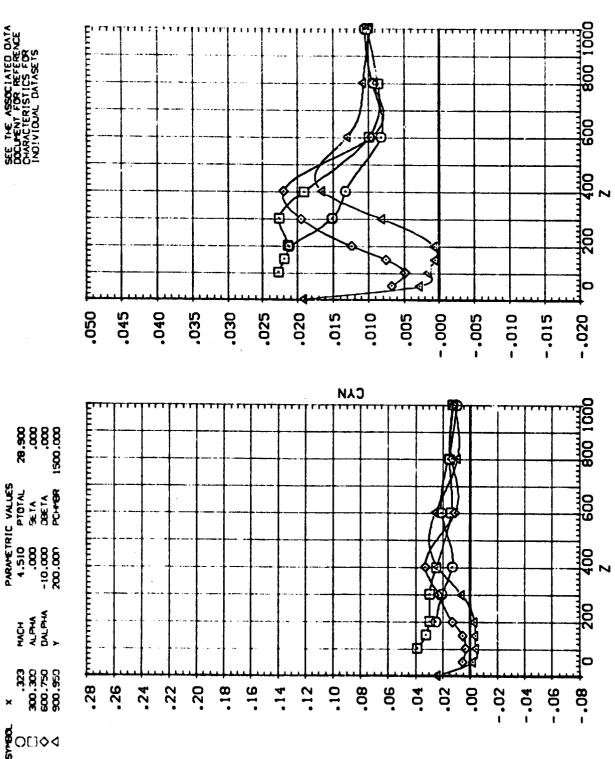
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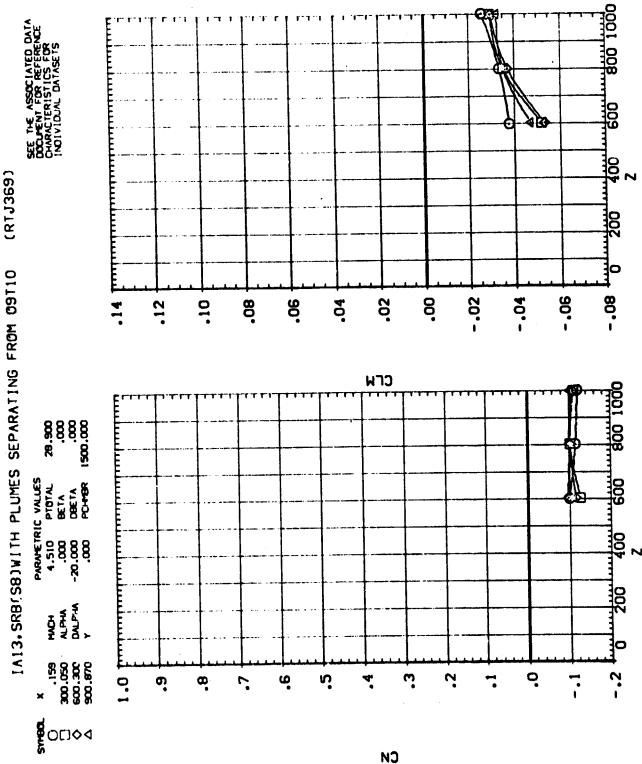
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IA13, SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ368)



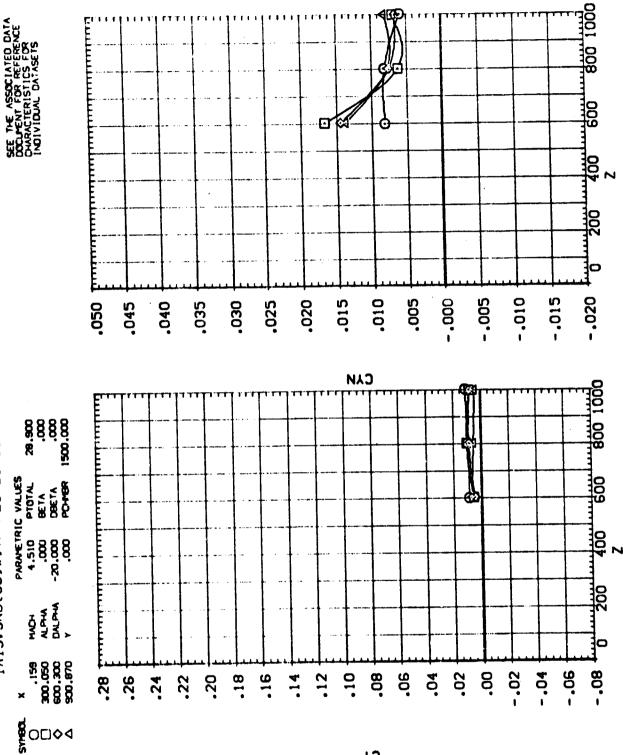
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



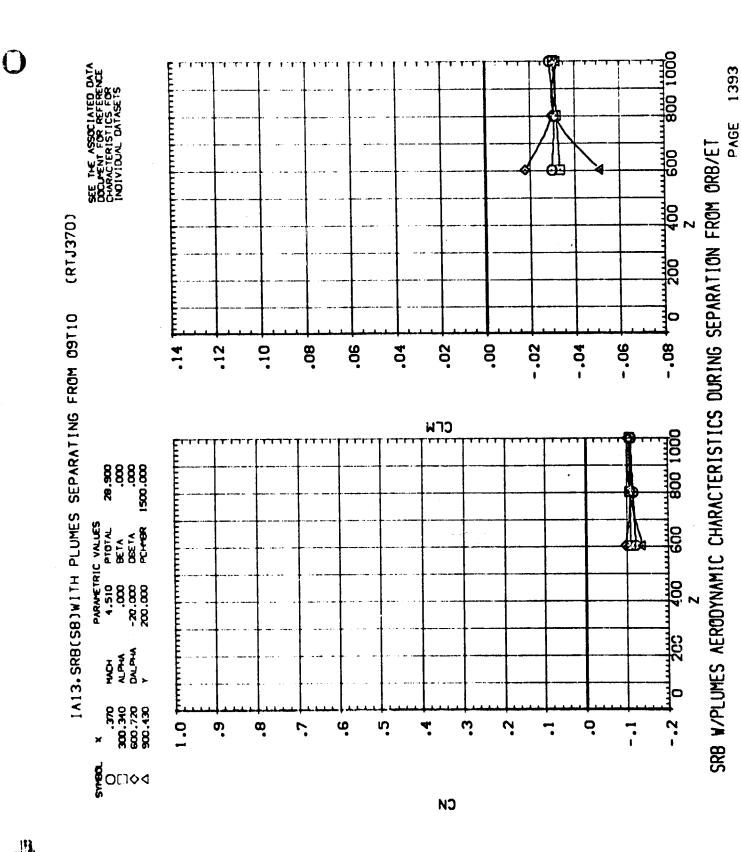


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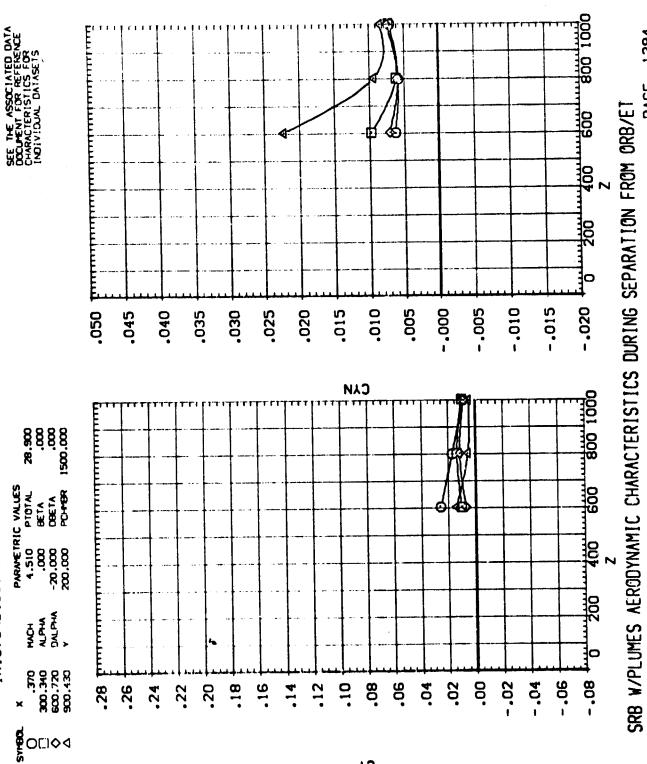
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(RTJ370) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110



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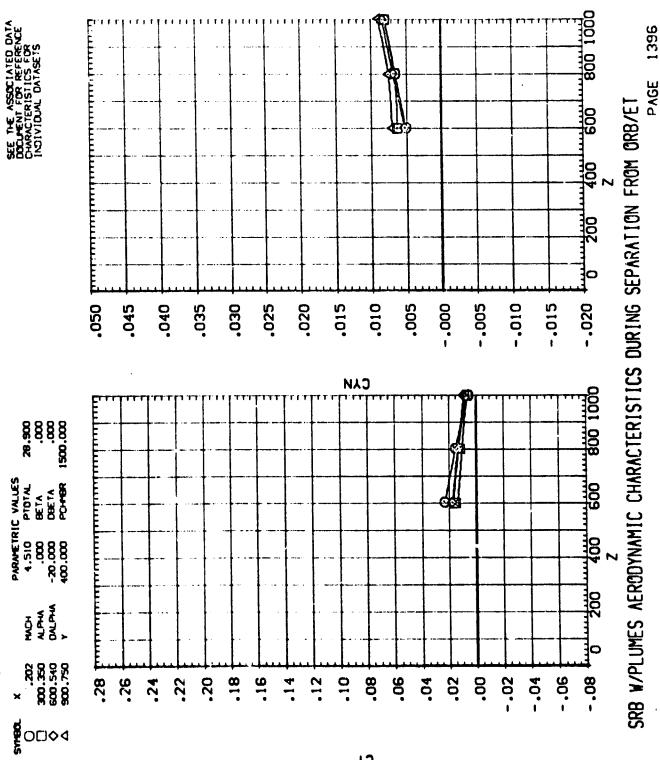
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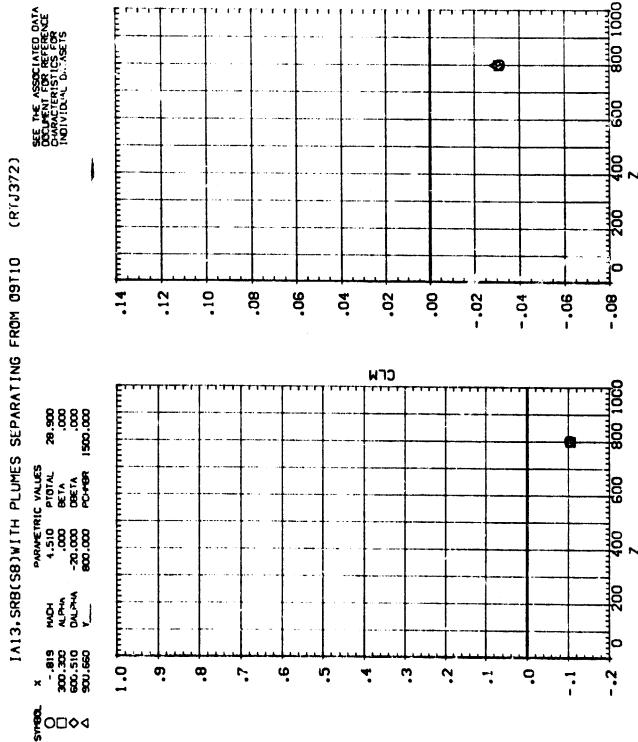
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4.510 PTOTAL
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400.000 PCHYBR 1 9 200 MACH ALPHA DALPHA 10 300.350 500.540 900.750 0. ທຸ ? œ 9 ດຸ . .. CM

(RTJ371) IA13, SRB(SB) WITH PLUMES SEPARATING FROM 09110



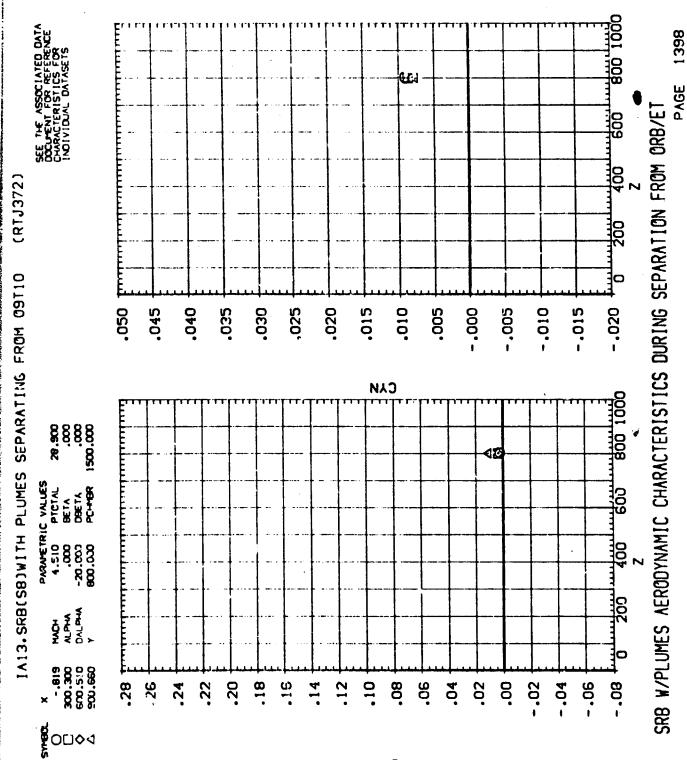
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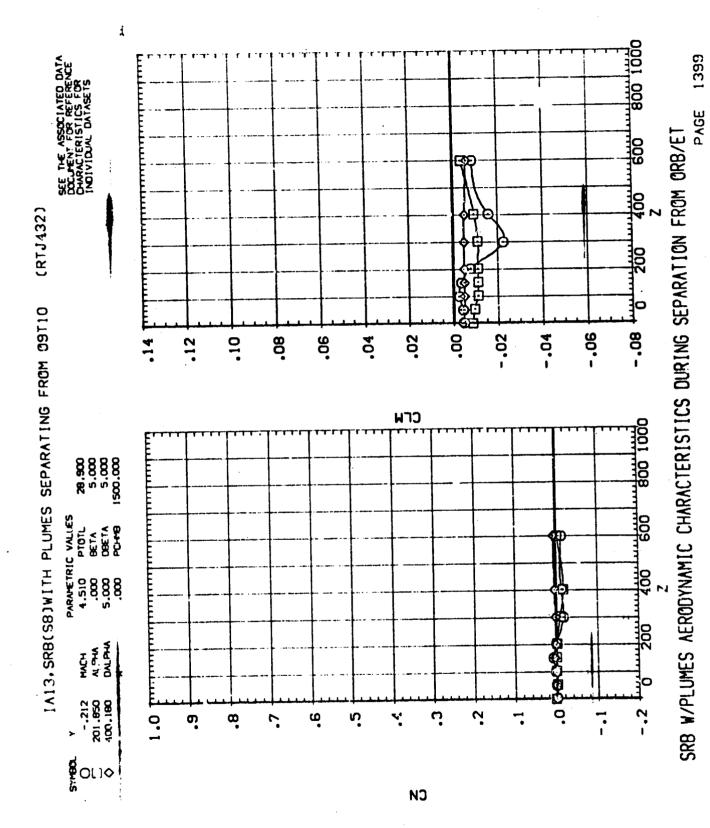
PAGE SRB W/PLUMES AERODYNAMIC CHARACIERISTICS DURING SEPARATION FROM ORB/ET

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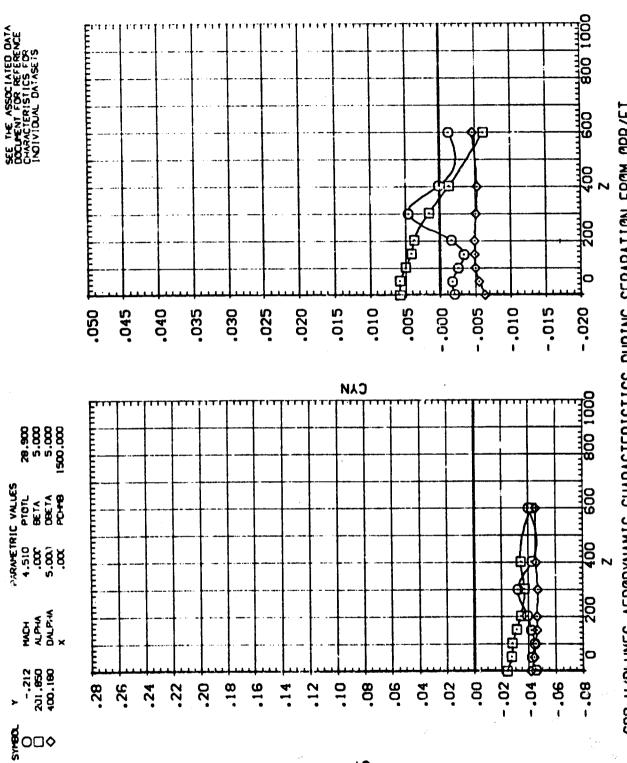


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IA13, SRB(S8) WITH PLUMES SEPARATING FROM 09110 (RTJ432)



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL, DATASETS (RTJ433) 200 ្នាំ IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710 **9** .12 .10 80. 90. **.** .02 3 -.02 -.08 20. CLM 800 100 28.900 5.000 1500.000 A.SIO PTOT.
.000 BETA
5.000 DBETA
.000 PCP-8 8 8 200 MACH PALPAN DALPAN .415 199.680 402.890 • • Q o, 8 Ģ ທຸ 4 'n ? . -:1 СИ

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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM GRB/ET

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(RTJ433)

SEPARATING FROM 09110

IA13. SRB(SB)WITH PLUMES

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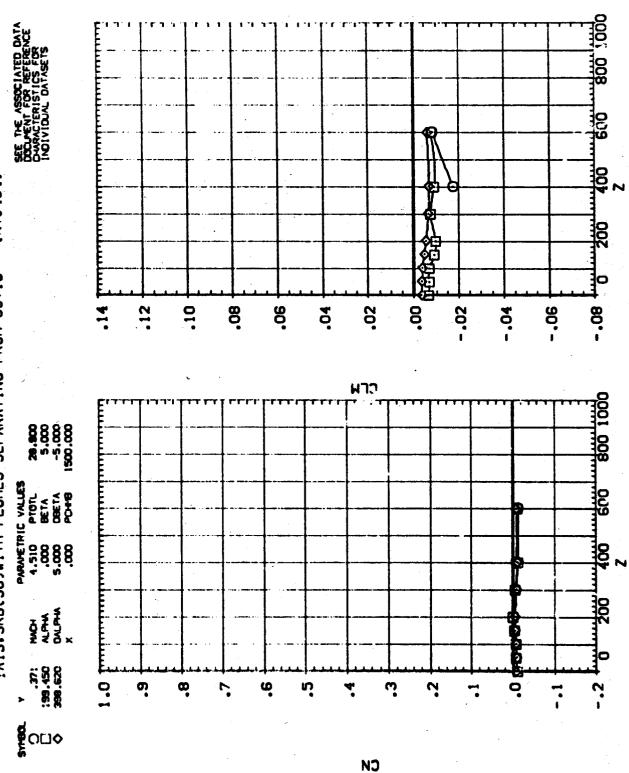
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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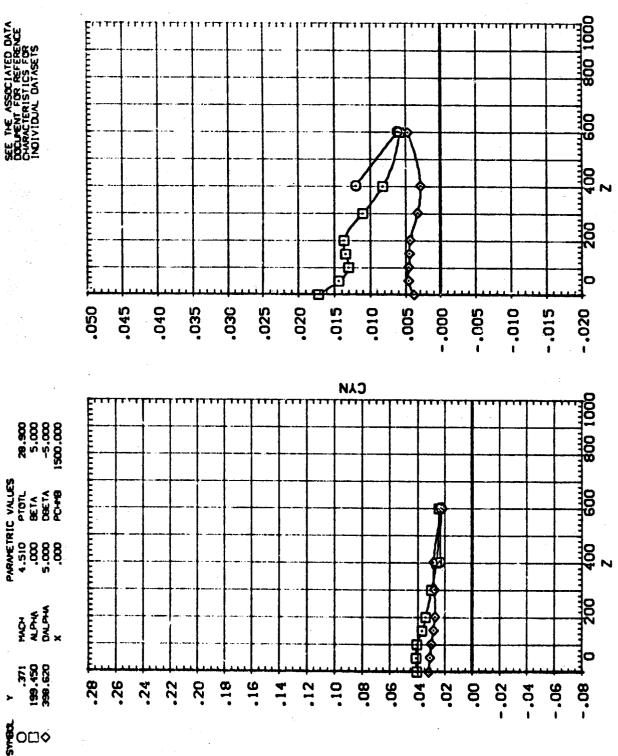
:A13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ434)



PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ434)



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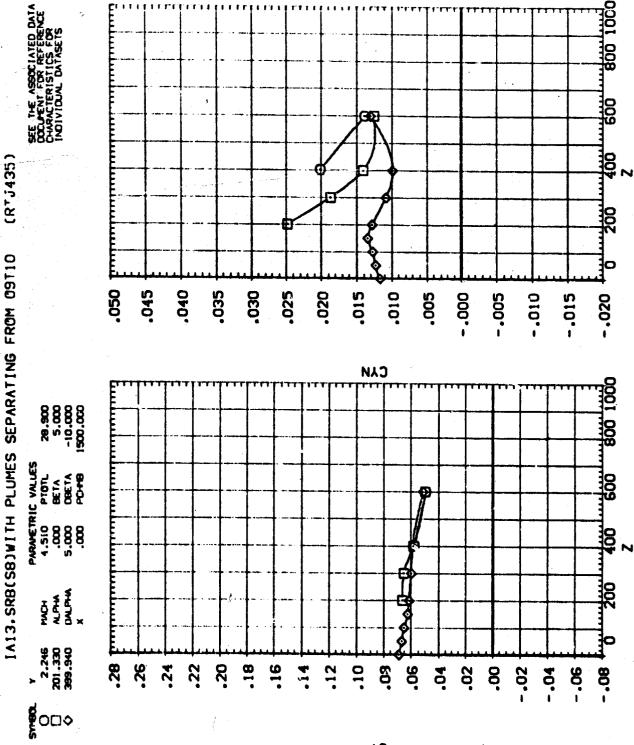
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4.510 PTOT.
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.000 POPB 8 18 18 A PAR PACPAR PACPAR TO PAC 0 2.246 201.330 399.940 ó ٥ æ Ģ 'n . ņ o ? - 2 ---

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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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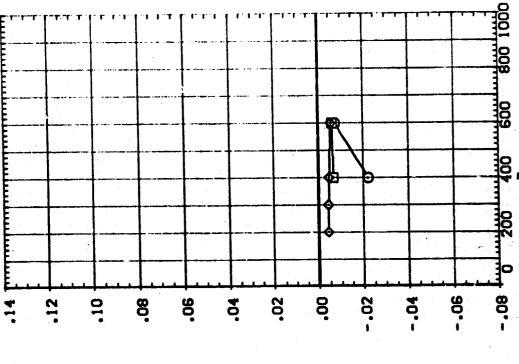
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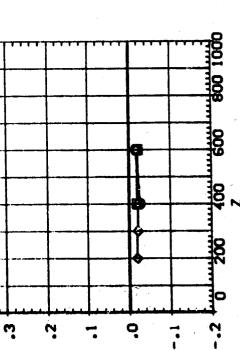
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ436) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710 .14 .12 28.900 5.000 -20.000 1500.000 PARAYETRIC VALLES
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 8 2

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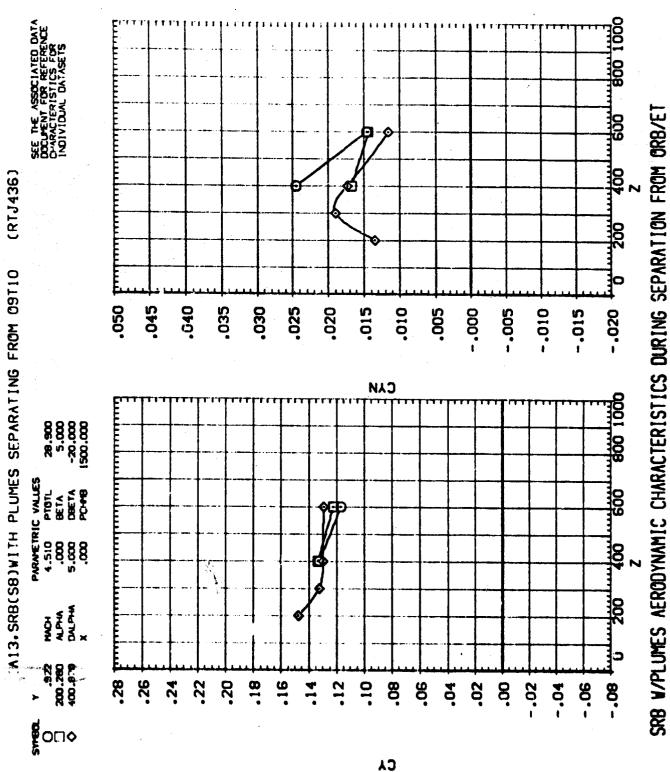
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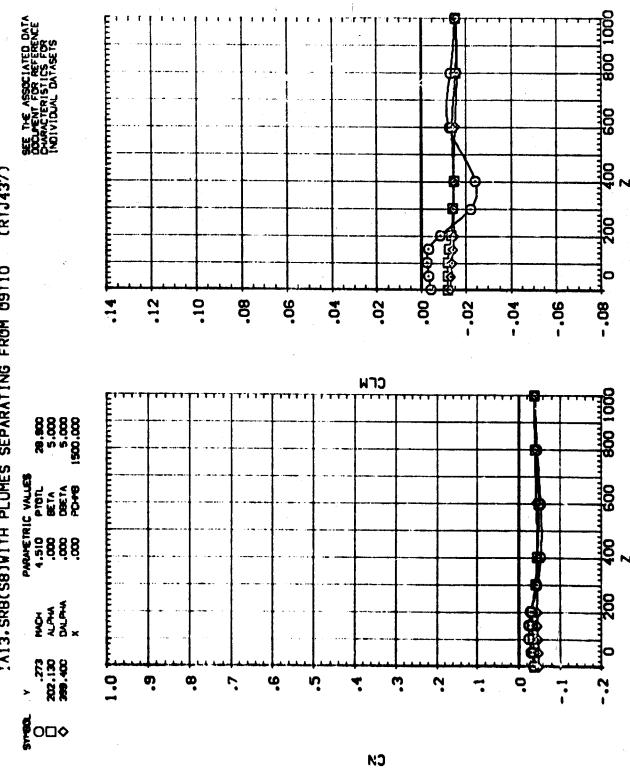
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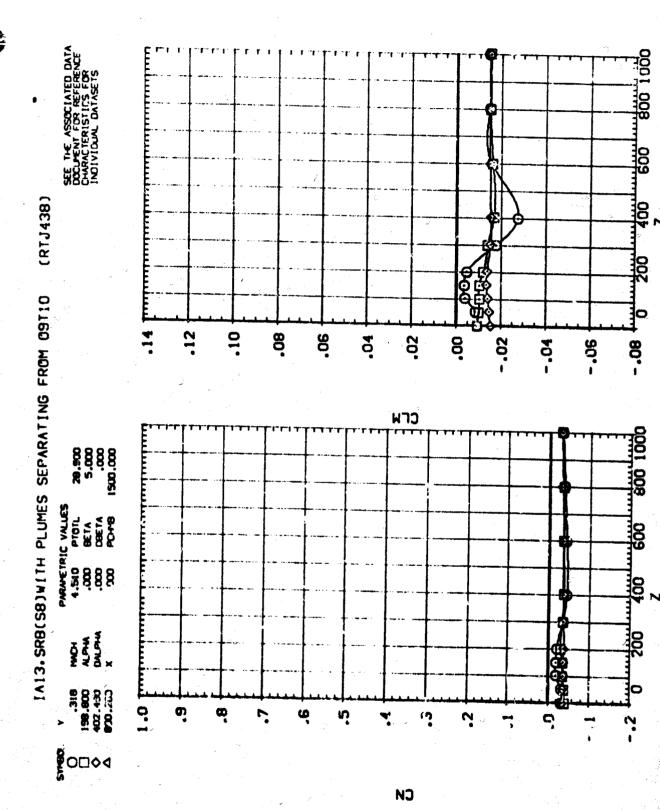




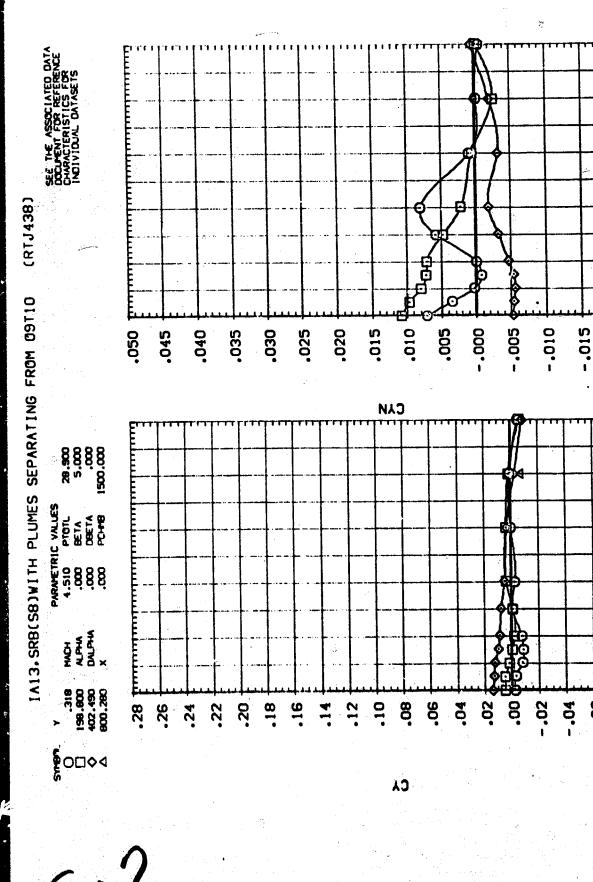
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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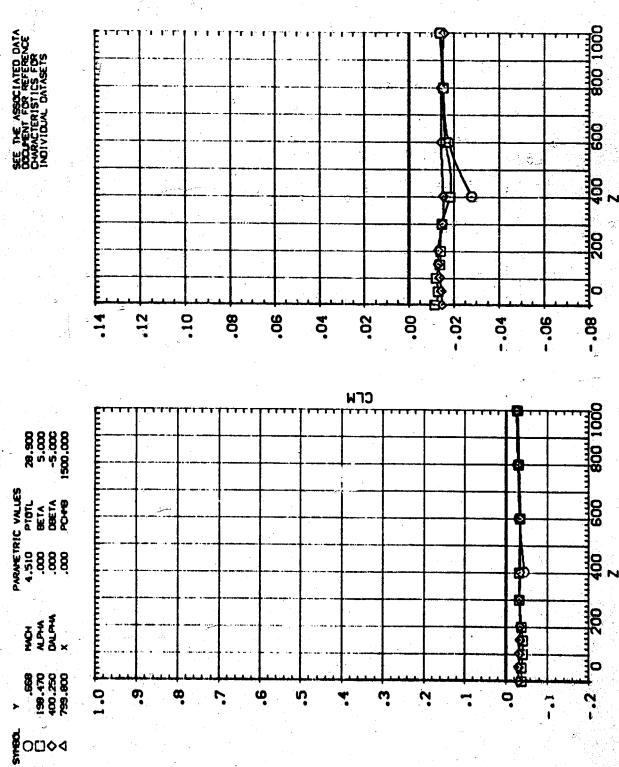
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(RTJ439) SEPARATING FROM 09110 IA13. SRB(SB)WITH PLUMES

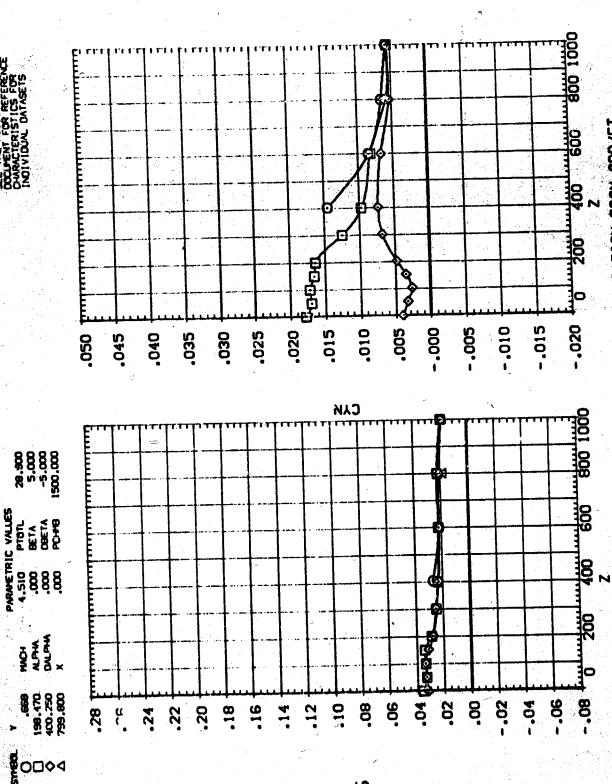


SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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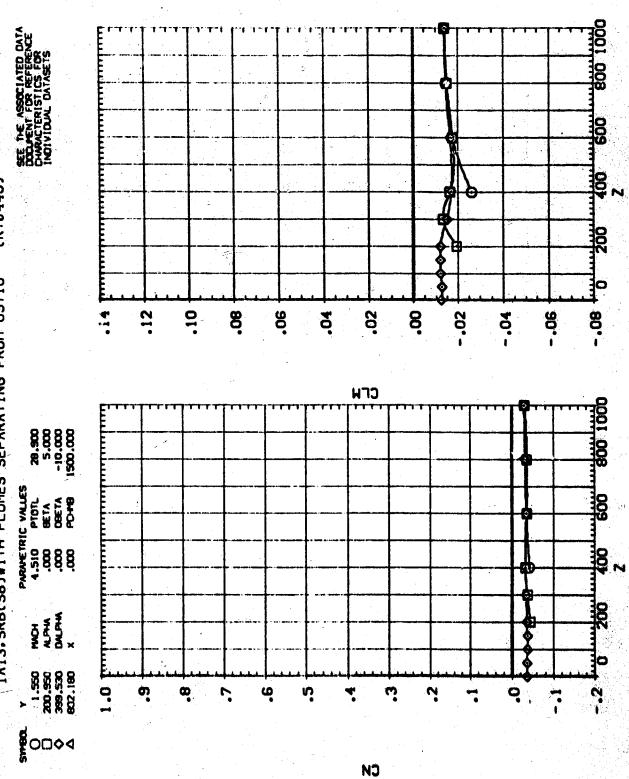
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PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

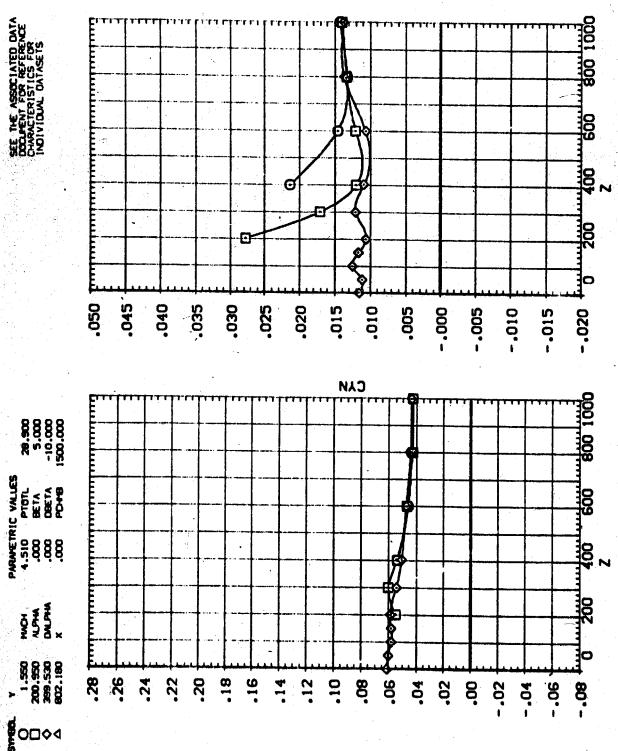
(RTJ440) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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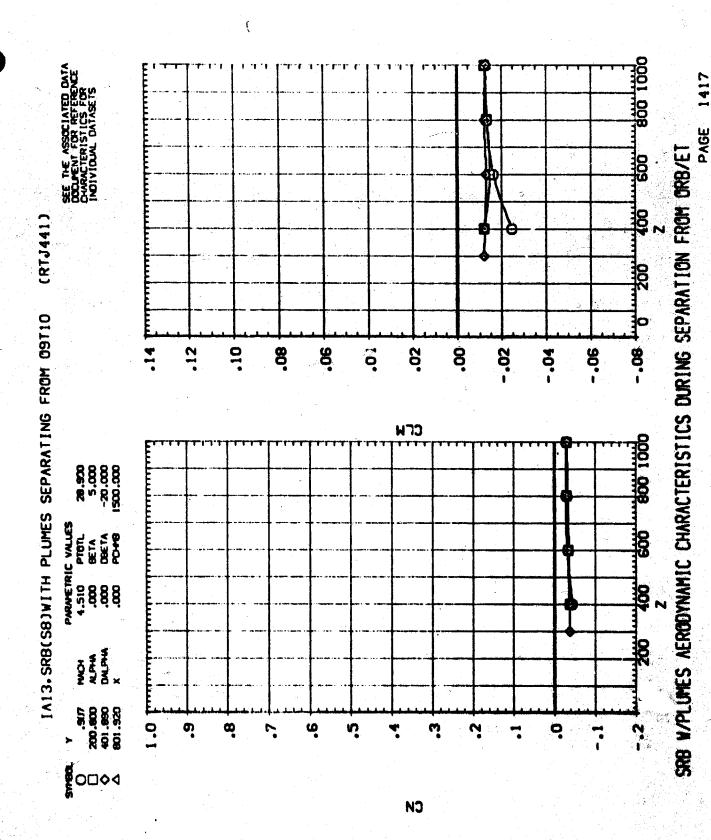
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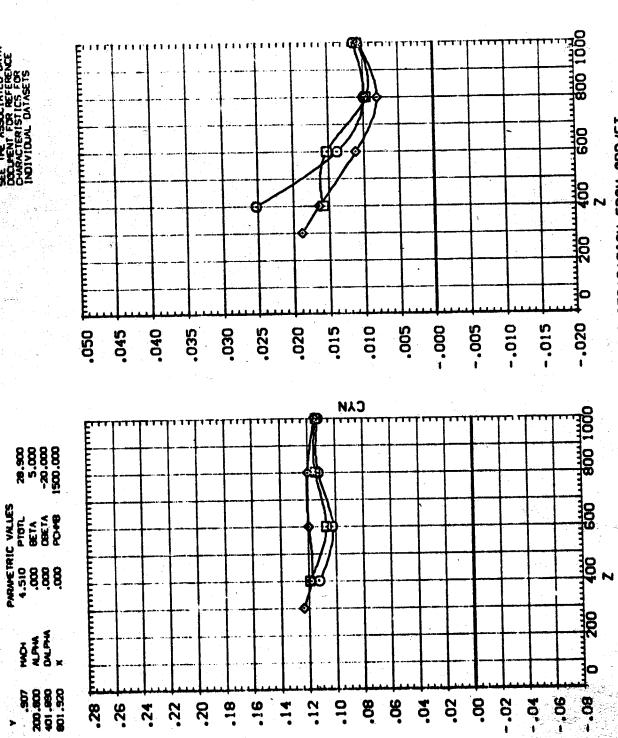
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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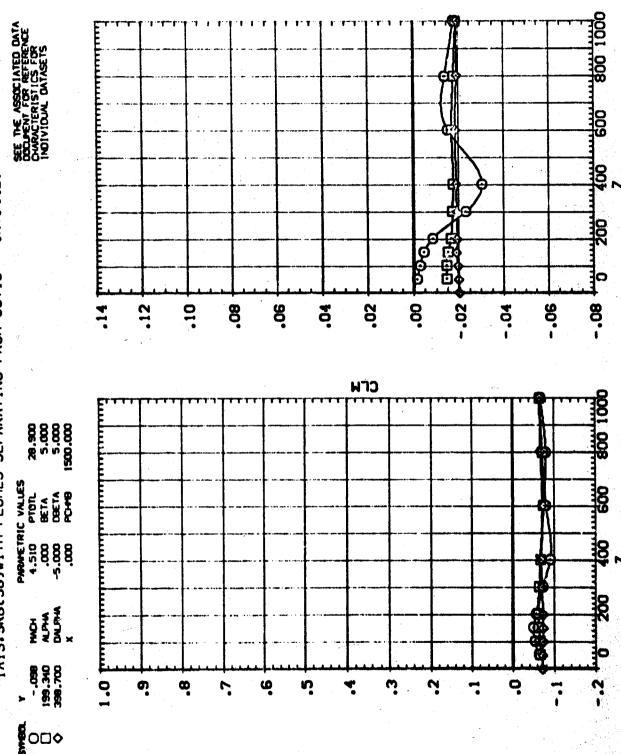


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SRB WYPLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

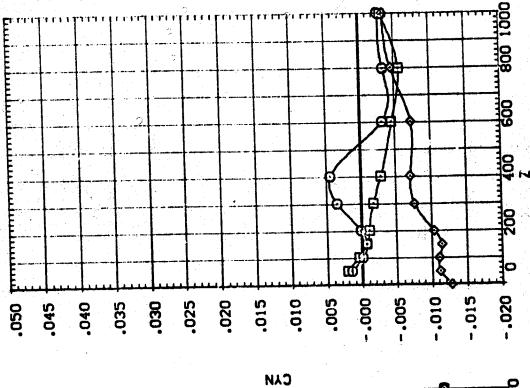
(RTJ442) IA13. SRBISBINITH PLUMES SEPARATING FROM 09110

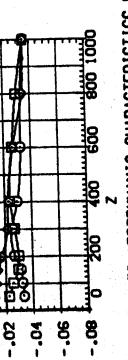


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PAGE SRB W/PLUMES AFRODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

SEE THE ASSOCIATED DATA
DOCUMENT FOR REFERENCE
CHARACTERISTICS FOR
INDIVIDUAL DATASETS (RTJ442) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 .040 .045 .050 28.900 5.000 5.000 1500.000 PARACETRIC VALLES
4.510 PTGTL
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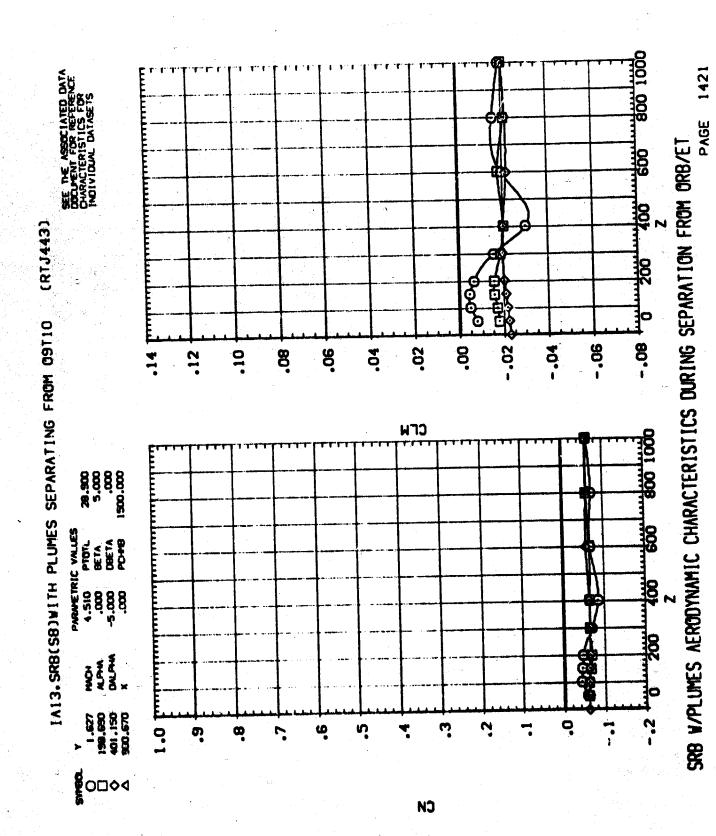
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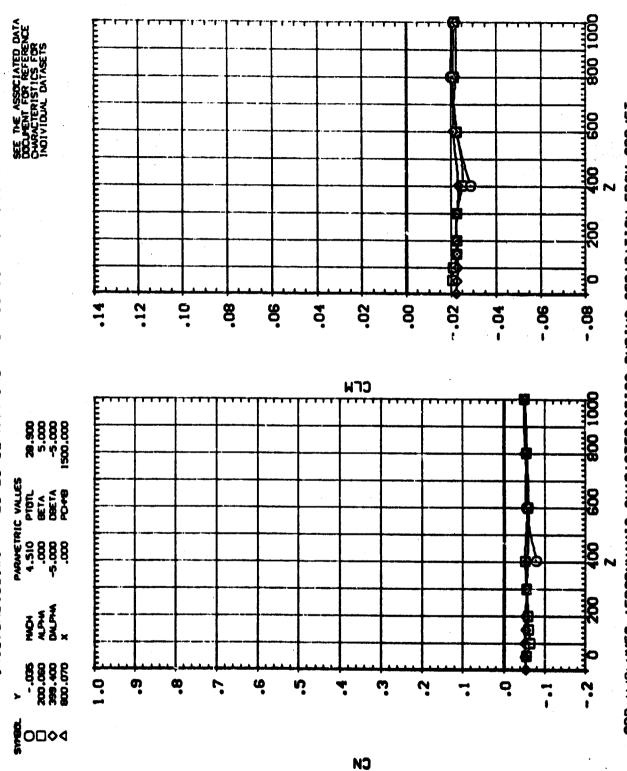
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(RTJ444) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110

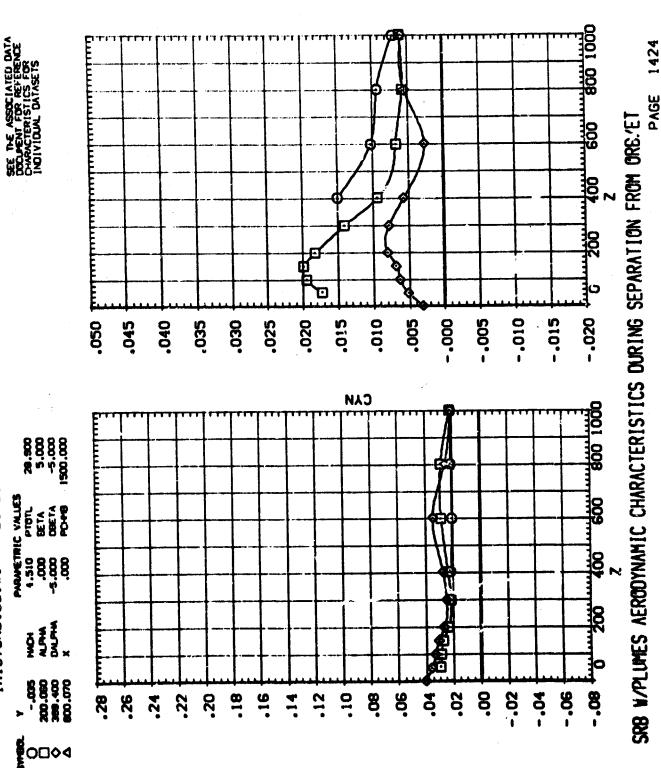


SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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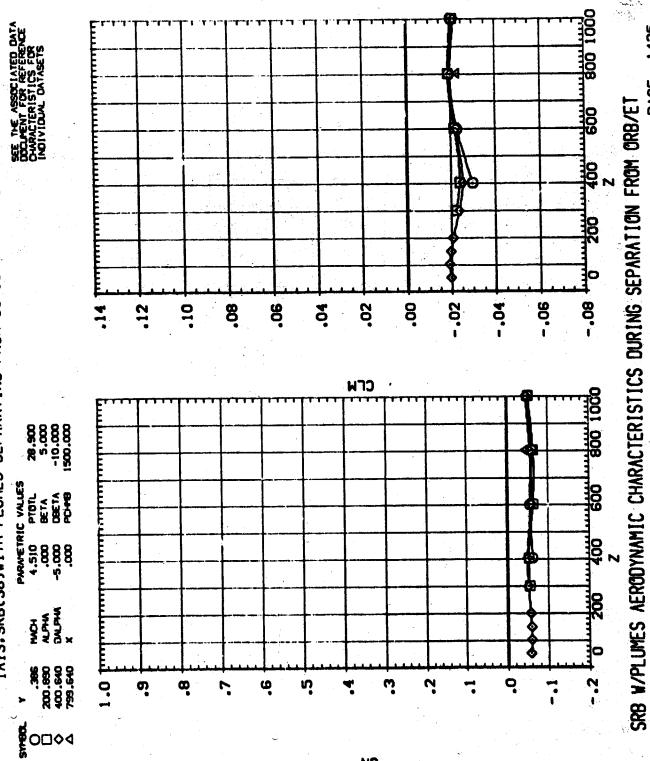
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ444)



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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ445)

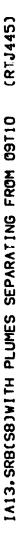
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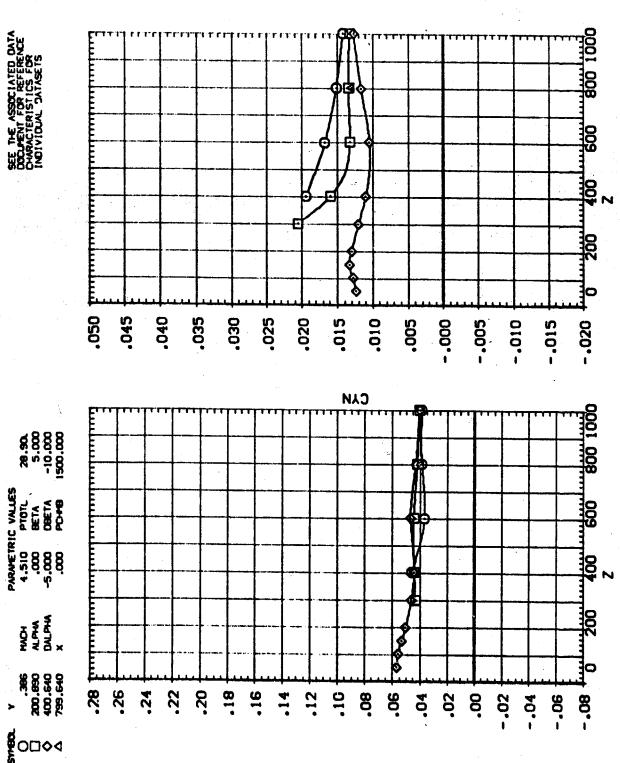


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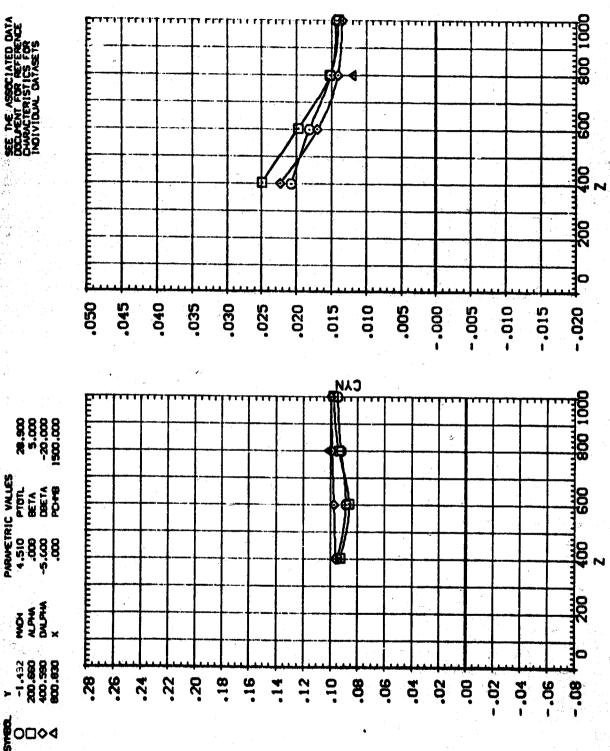
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

800 1000 100 **400** 2 (RTJ446) 200 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 -.08 -.02 90. .14 01. 8 .02 8 -.04 .12 8 5 כרא 800 1000 28.900 5.000 -20.000 1500.000 PARAFERIC VALLES
4.510 PTOT.
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.000 PO+8 009 MOT WENT 1.0 1 7 -1.492 200.660 400.590 800.830 ທຸ n o ດຸ œ 4 **2**0□ ◊ 4 CM

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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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PAGE SOB WAPLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORBAET

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(RTJ447) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 .050 .040 -.000 -.005 .045 .010 -.010 .035 030 .025 .020 .015 .00° CAN 28.500 28.200 28.200 200.000 BETA DBETA POPE 4.510 P1 4.510 P1 .000 91 -10.000 0 58 .22 .16 .12 .10 8 8 8 -.8 .24 .20 9 20. **∮**O□◊ CA

PAGE 1430 SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM DRB/ET

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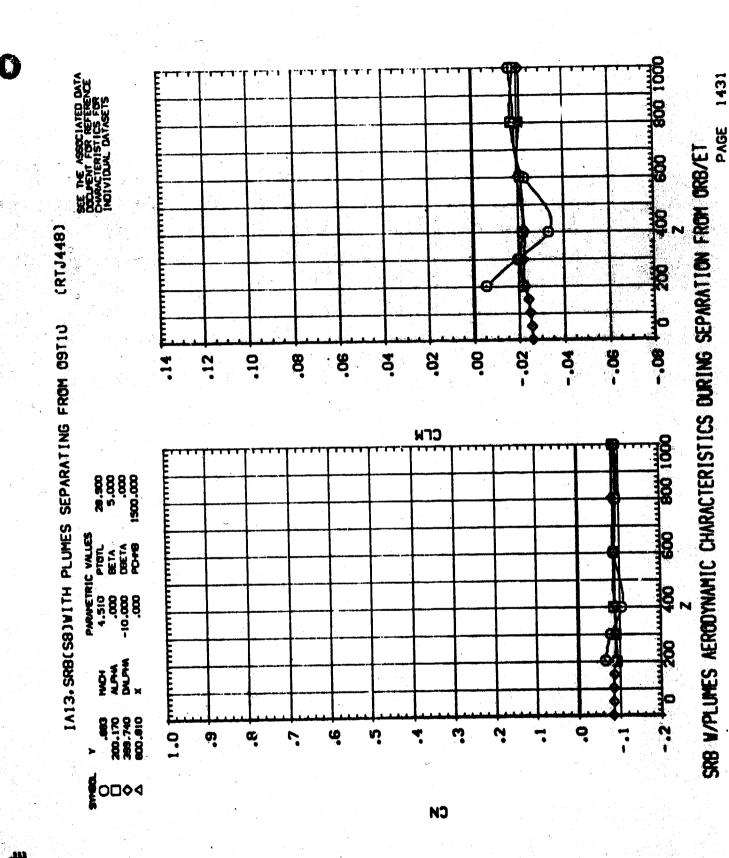
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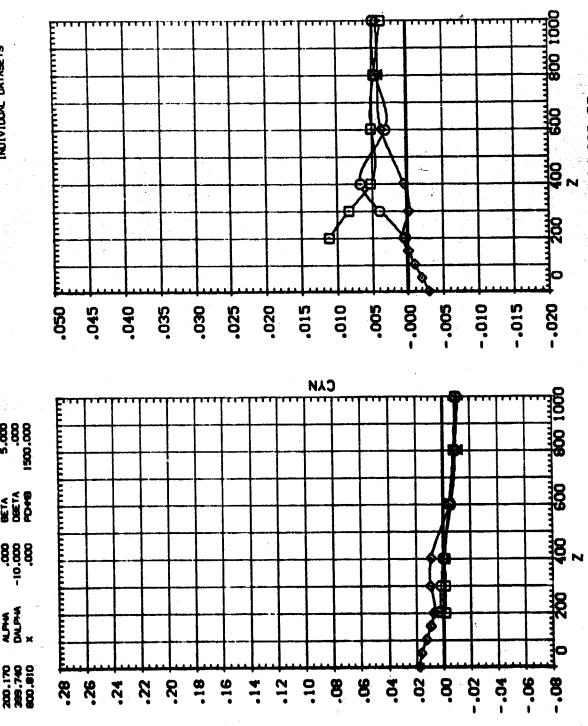
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SAB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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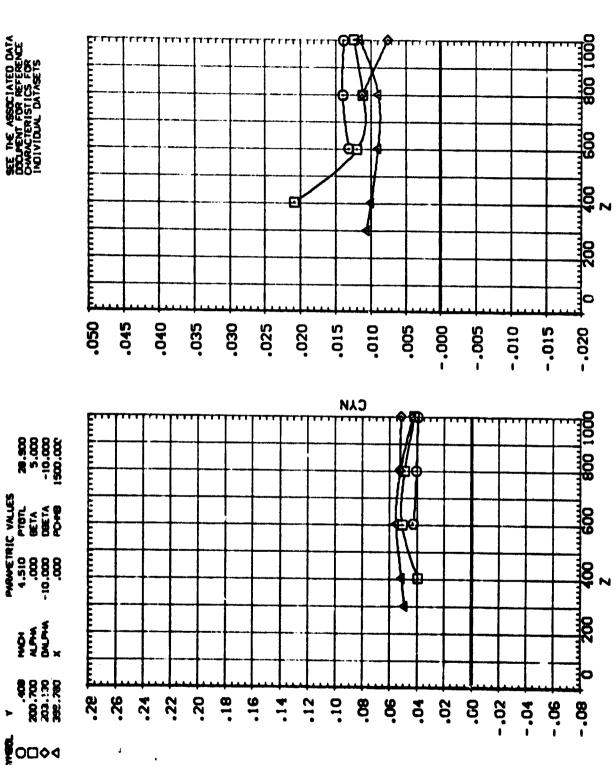
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800 1000 DECEMBER OF REFERENCE DARACTERISTICS FOR INDIVIDUAL DATASETS 1434 PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET 9009 超~ (RTJ449) 別 þ IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710 -.020 -.015 -.000 -.010 . 00. -.005 .020 .015 010 .050 .045 .040 .035 030 .025 CAN 800 1000 26.900 26.900 26.900 1500.000 PARMETRIC VALLES 4.510 PT01. .000 BETA -10.000 DBETA .000 PO48 100 18 × PER × 90-8 8 8 -.08 .16 .12 8 9 .02 -.02 97. .22 2 .18 .14 .10 40.-.24 000 CA

800 1000 DECLENT FOR REFERENCE CHRACTERISTICS FOR INDIVIDUAL DATASETS SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET **18** ~ (RTJ450) SEPARATING FROM 09110 .14 .12 9: 8 8 .02 8 9. -.02 -.06 -.08 -.04 CLM 800 1000 26.600 25.800 10.000 10.000 10.000 IA13. SRB(SB) WITH PLUMES 18 4.510 4.510 8.000 10.000 9.000 8 18 200.00 200.00 200.00 200.00 0. ų. ຫຸ ú 1 Ģ ທ່ 4 ņ o 7 **E**O□◊4 CM

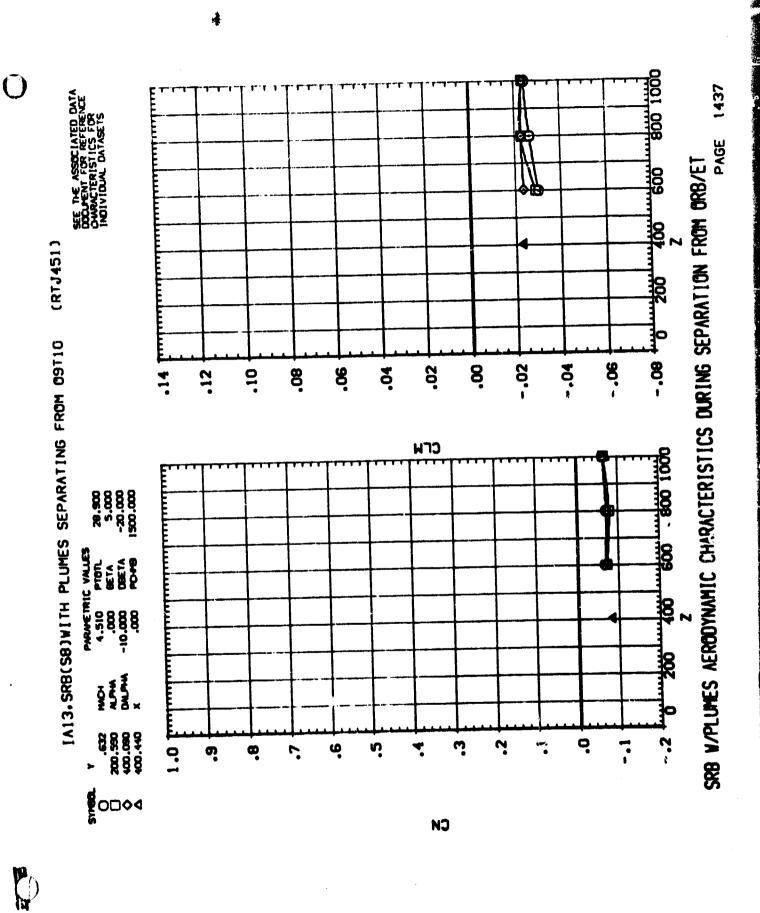
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PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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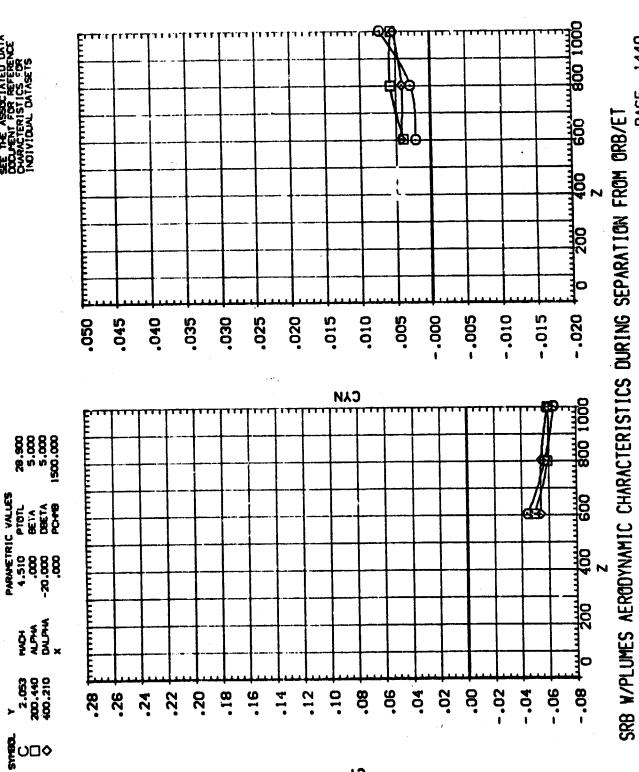
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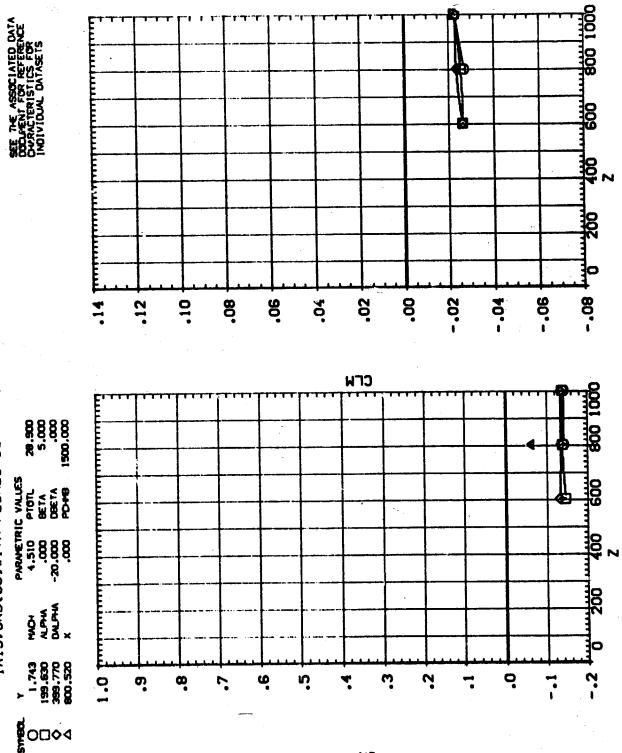




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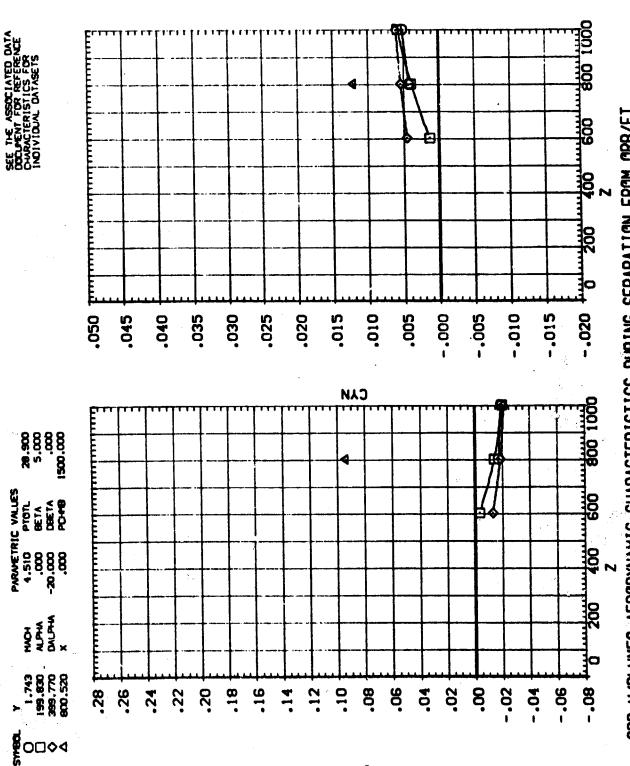
(RTJ453) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110



PAGE SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

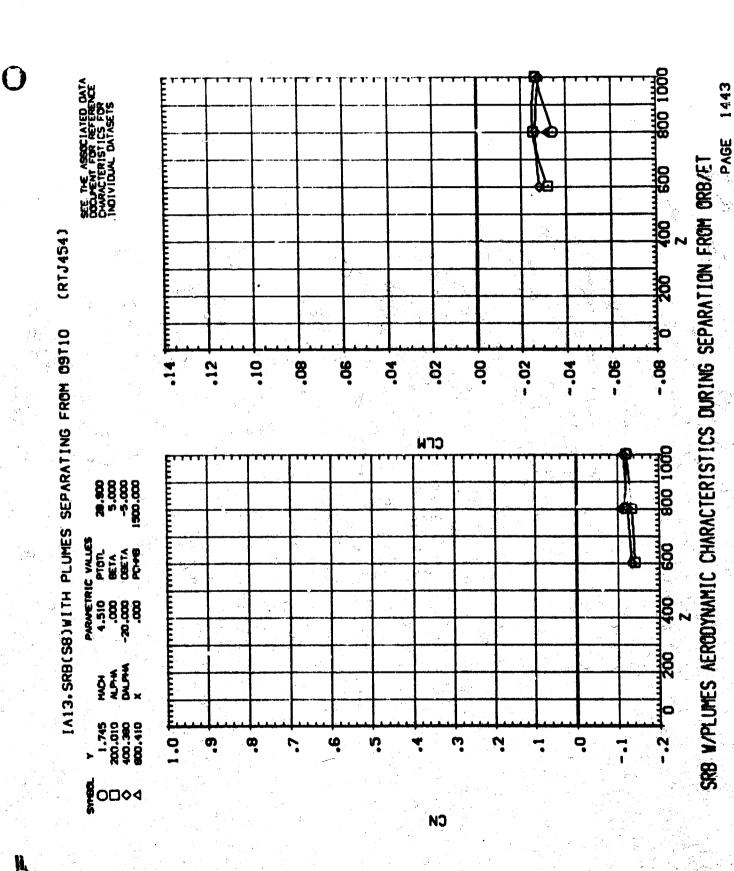
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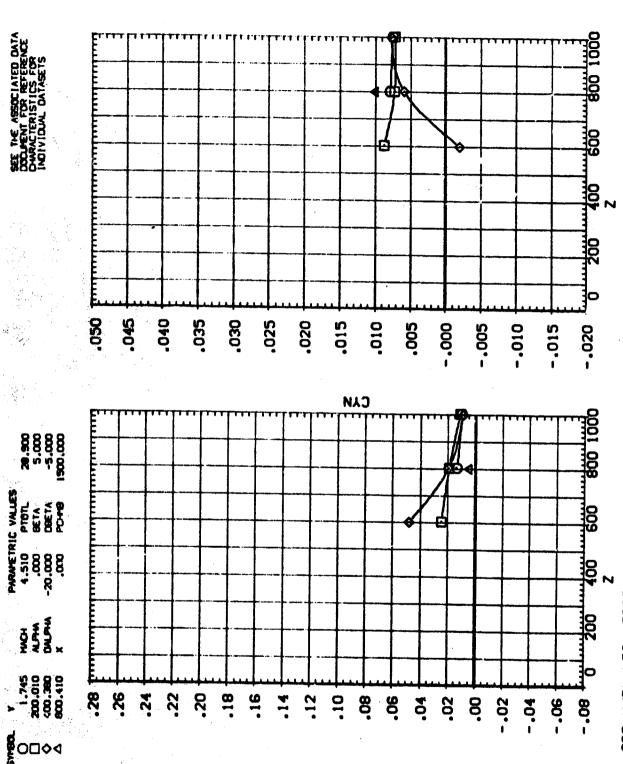




SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

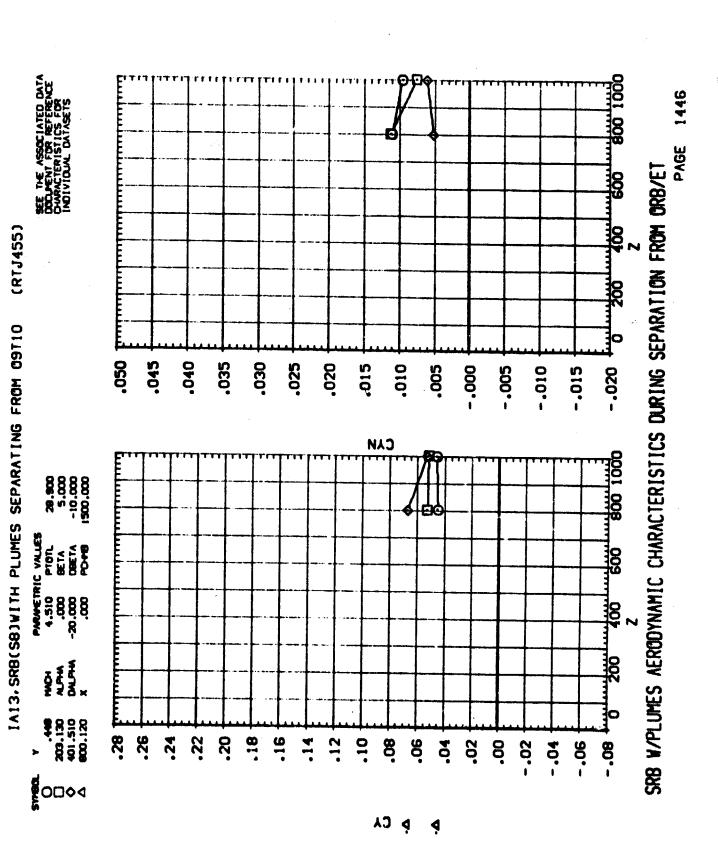
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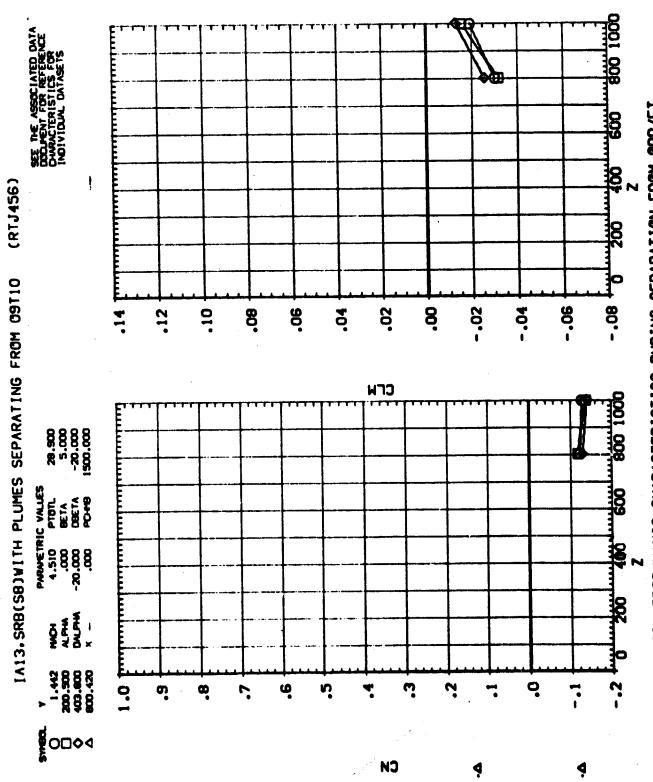




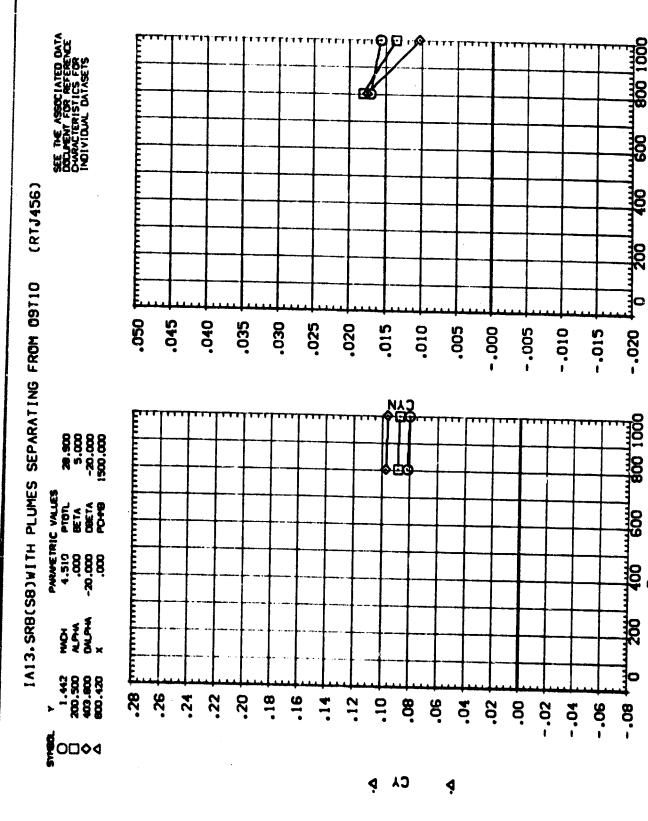
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE DAMANCTERISTICS FOR INDIVIDUAL DATASETS SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET (RTJ455) 祭 IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 -.06 8 .12 9: 8 90. 9 .02 -.02 -.04 כרש 800 1000 26.90 26.90 10.000 190.000 PARMETRIC VALLES
4.510 PTOT.
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.000 PO-48 18 18 203.130 401.510 800.120 0.1 9 n 7 o œ ທຸ O, **€**0□◊4 CM Þ Ģ





SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

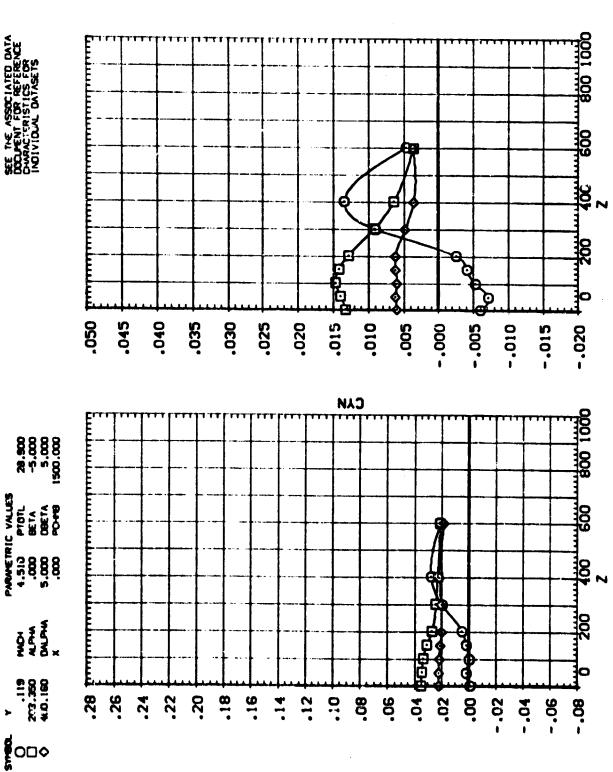
PAGE 1448

800 1000 SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS (RTJ459) 8 .Оф IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09710 80. -.06 2: -.02 8 8 .02 8 -.04 .14 .12 9. CLM 800 1000 26.50 26.50 26.00 PARAFERIC VALLES 4.510 PTOT. .000 BETA 5.000 DBETA .000 PO+9 89 8 18 MON X .1:9 203.360 400.160 ó 0. œ ห่ -: Ģ 7 n ? ດ. . **ਊ**0□◊ CM

SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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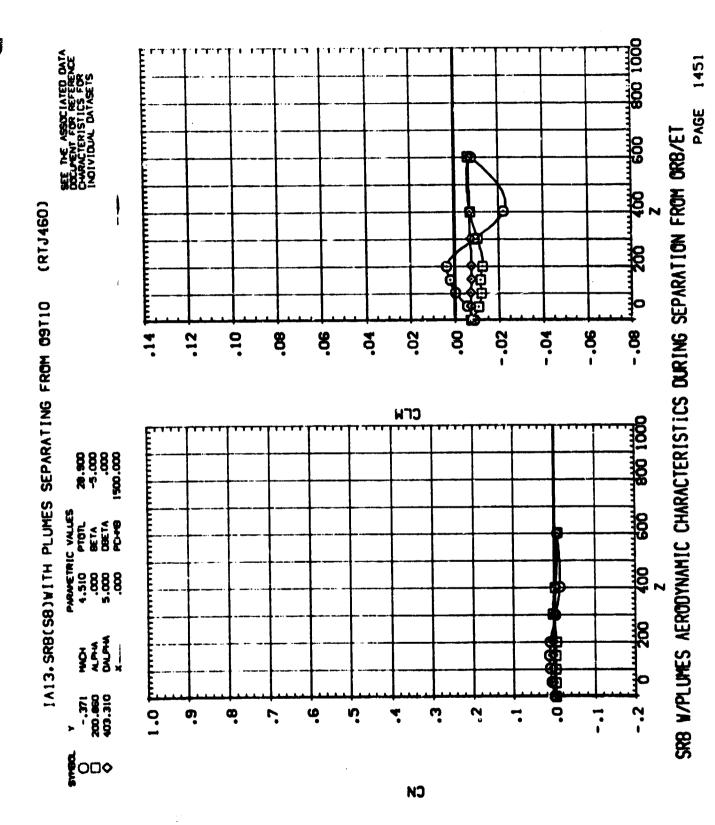
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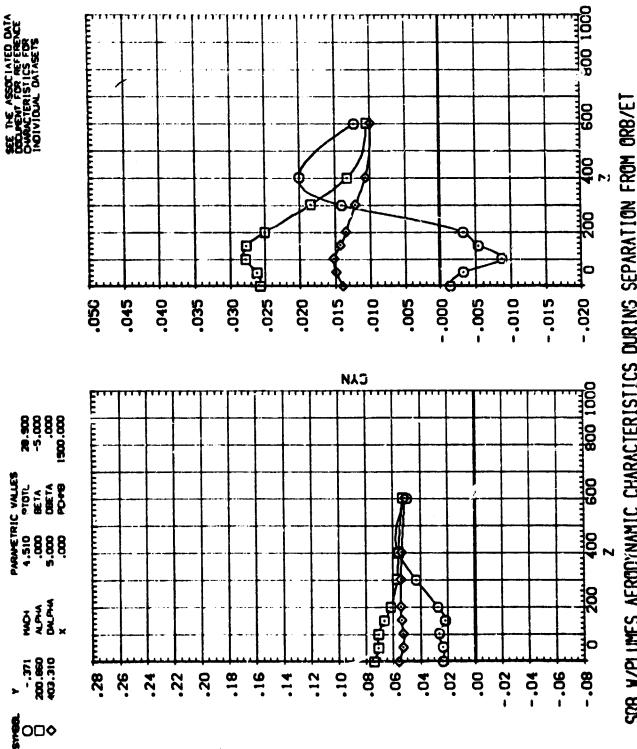
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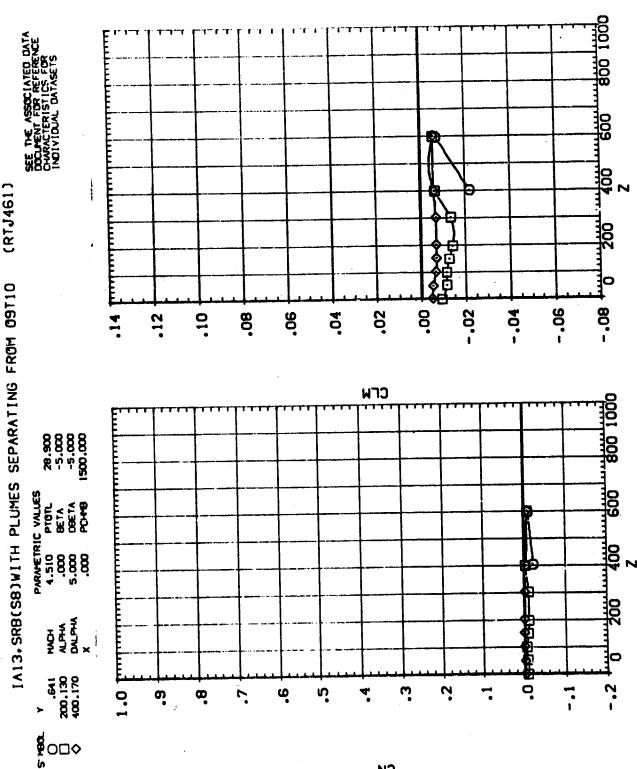


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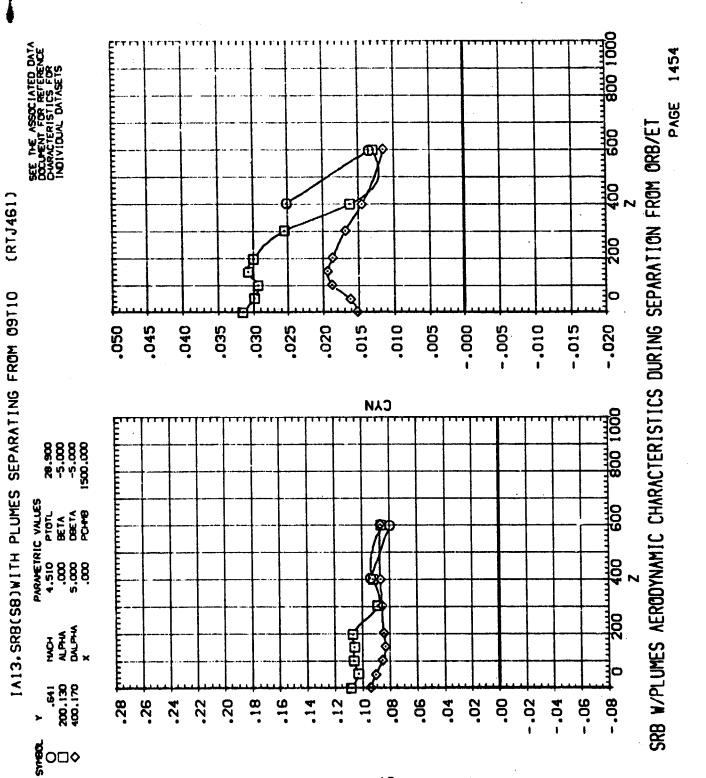
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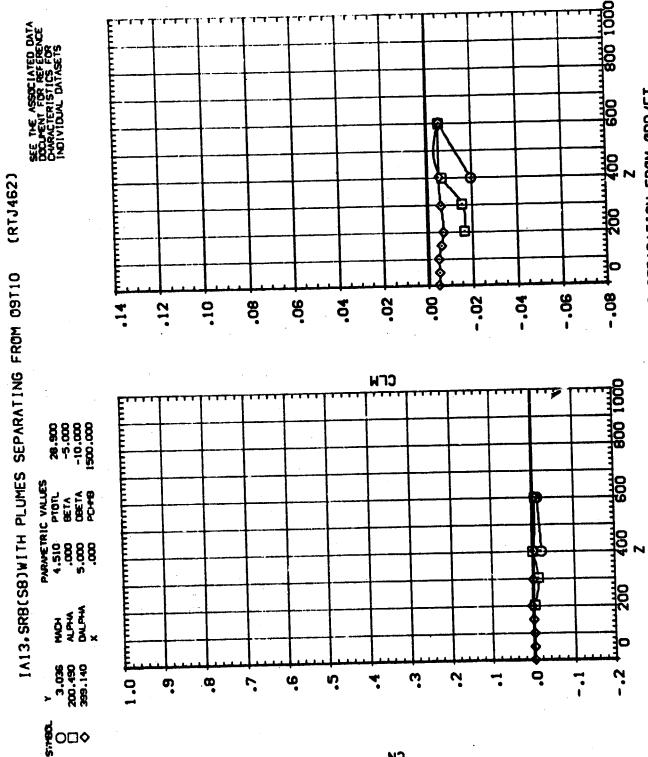
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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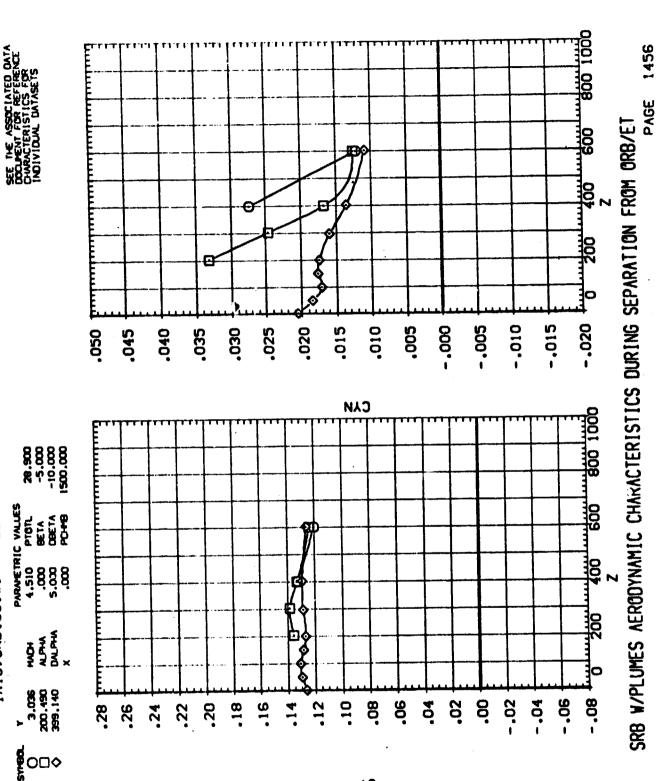
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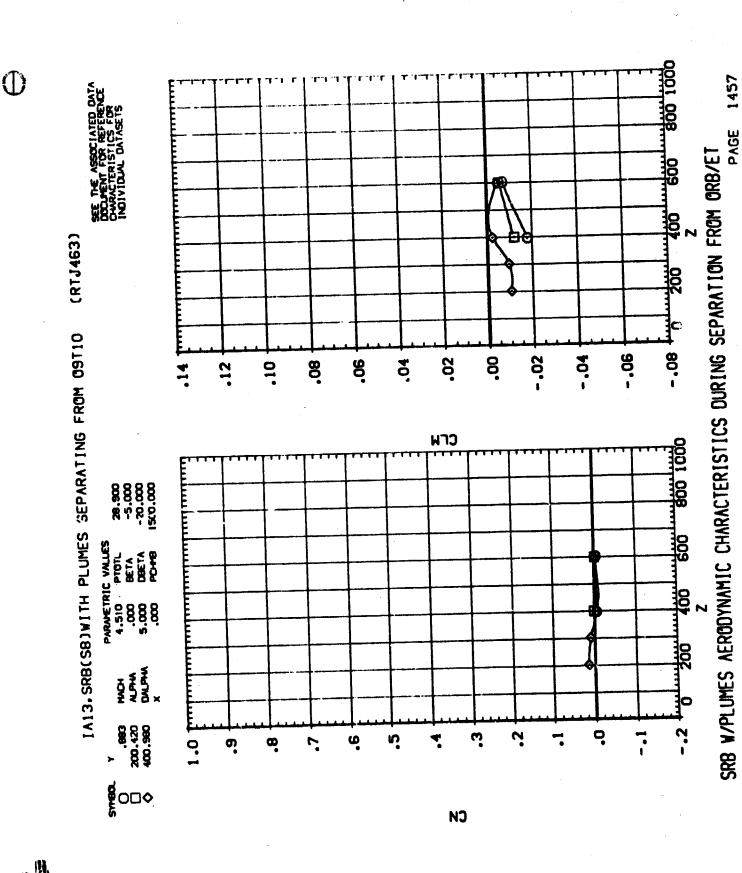


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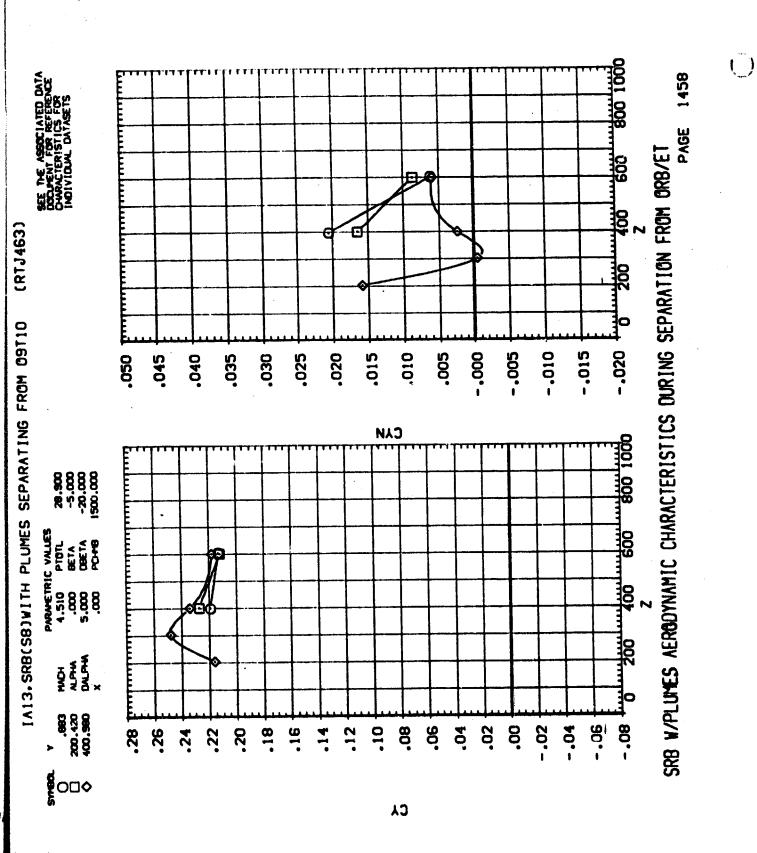
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IA13.SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ462)

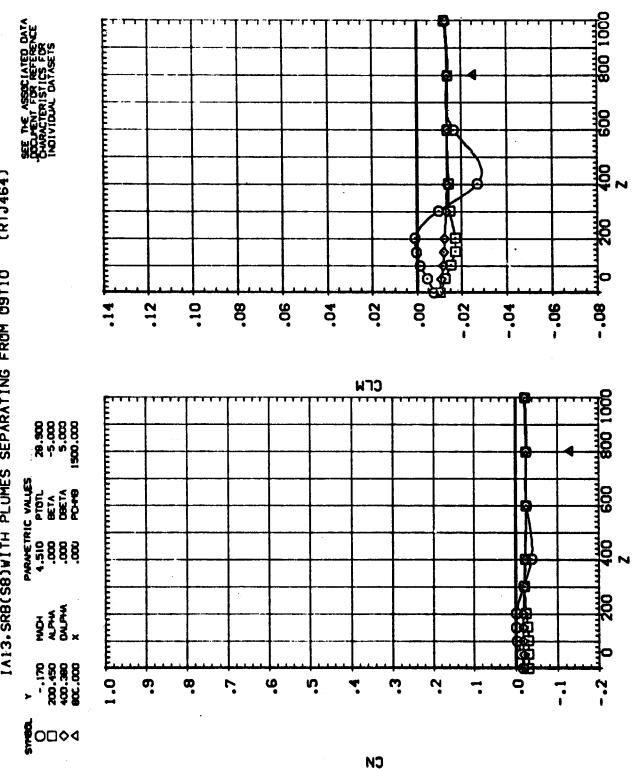




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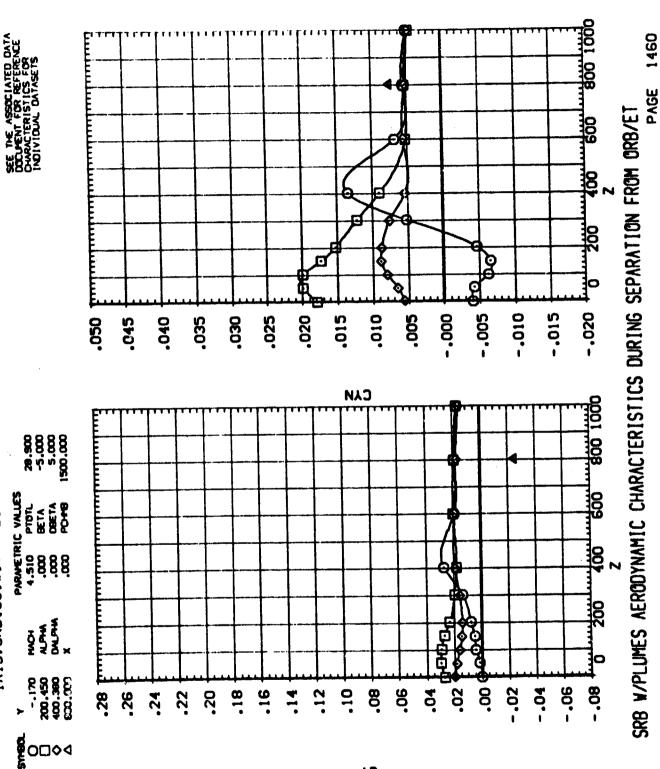
(RTJ464) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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(RTJ464) IA13, SRB(SB) WITH PLUMES SEPARATING FROM 09110



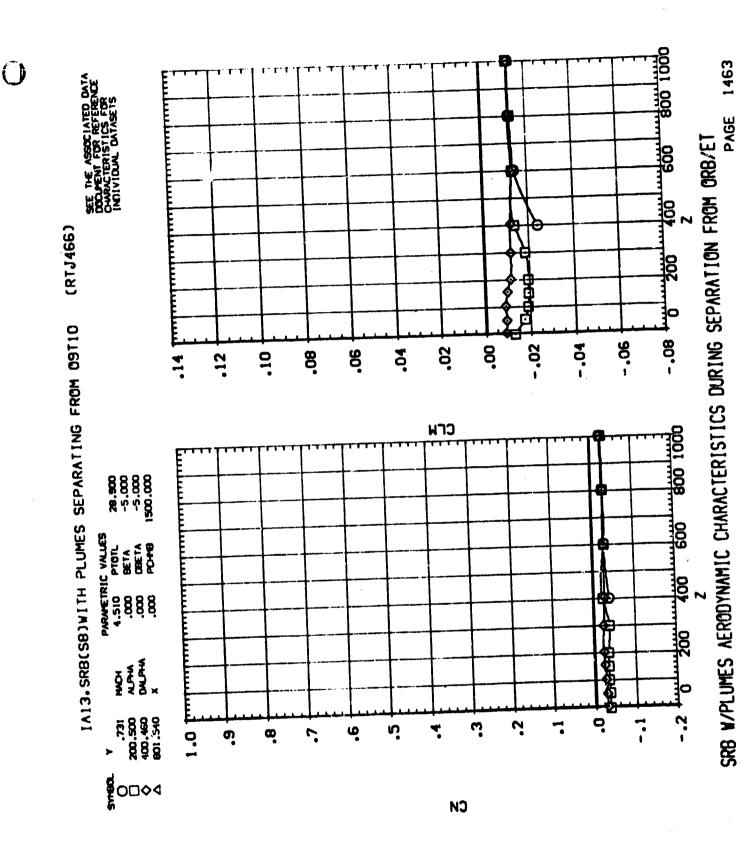
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SEE THE ASSOCIATED DATA DOCUMENT FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL, DATASETS 800 1000 (RTJ465) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 09110 1. .12 .10 8 8 .02 9 8 -.02 90.--.0**.** CLM 800 1000 28.90 25.900 150.000 1 PARAVETRIC VALLES
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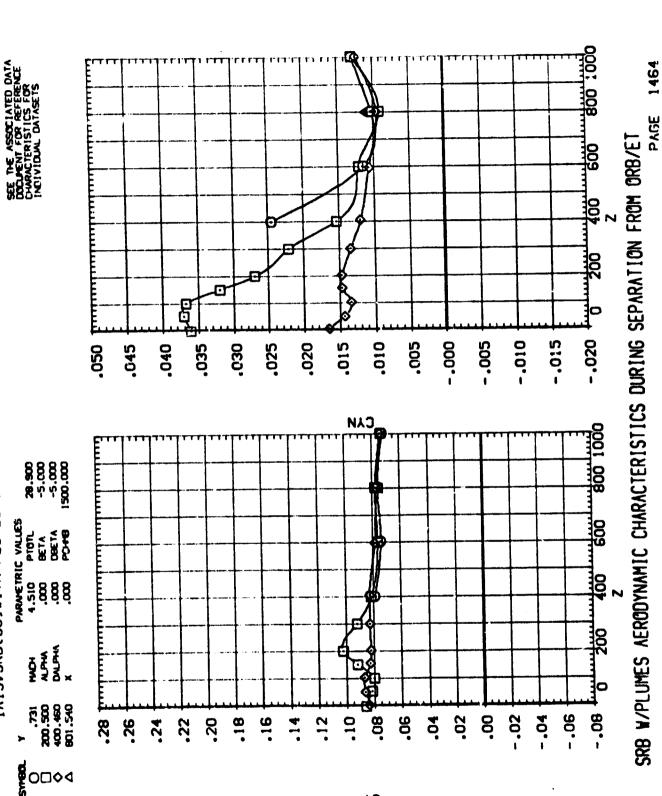
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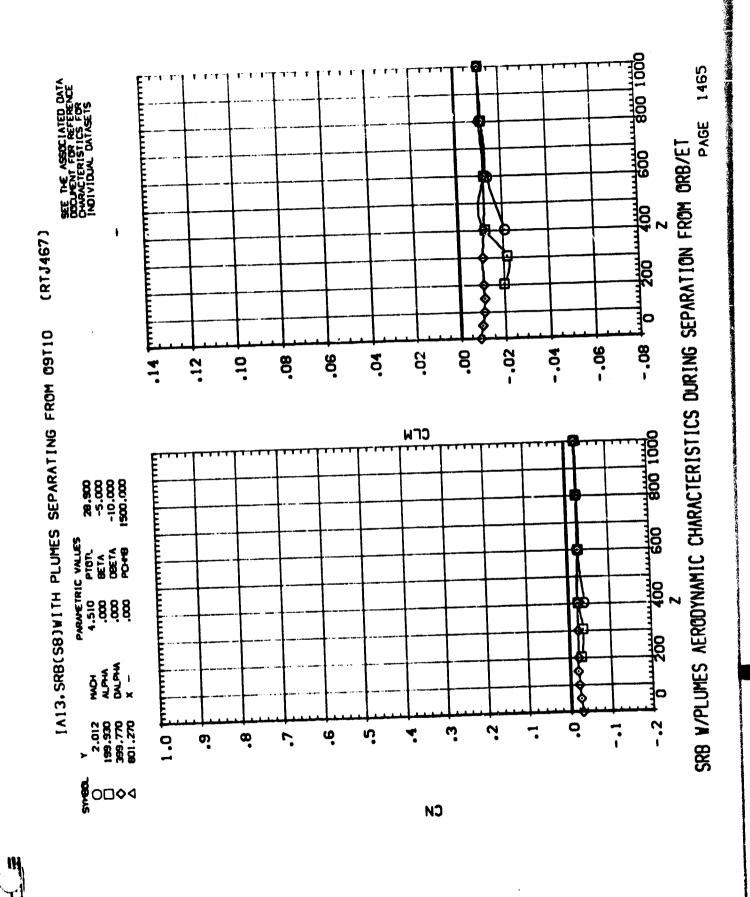


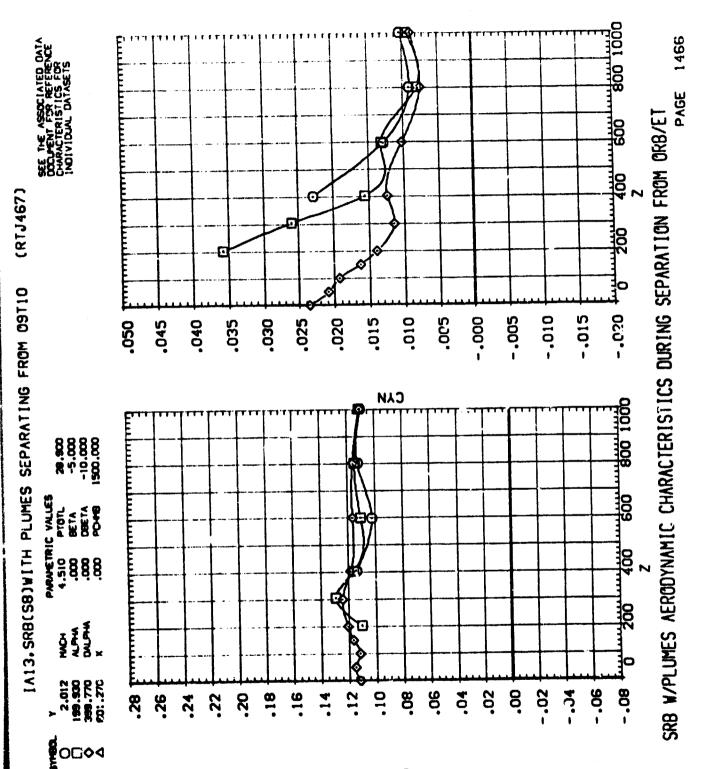
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ466)







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CHARACTERISTICS FOR
INDIVIDUAL DATASETS (RTJ468) SEPARATING FROM 09110 1. .12 .10 80. 90. .02 8 • -.02 90.--.04 כרא 28.30 -5.000 -20.000 -20.000 -20.000 IA13. SRB(SB)WITH PLUMES PARAFERIC VALUES
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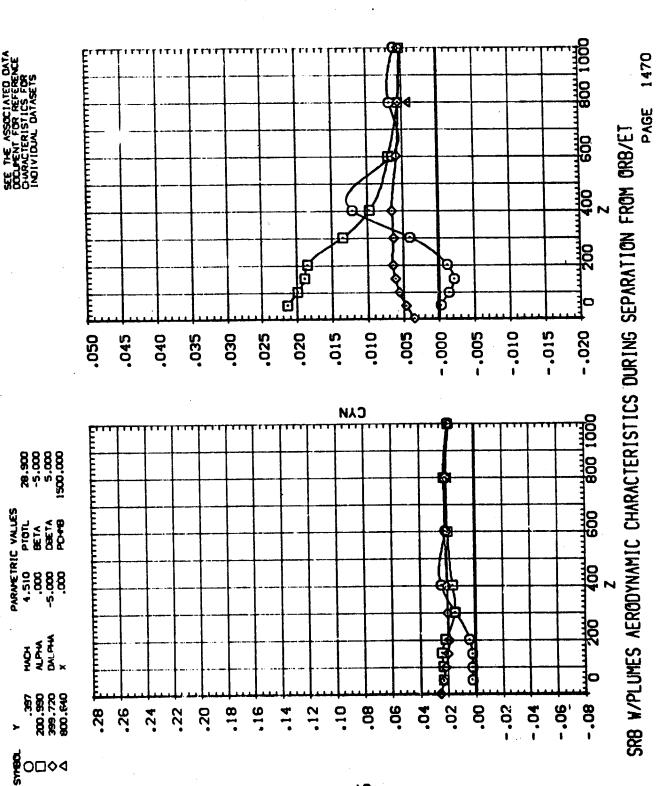
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SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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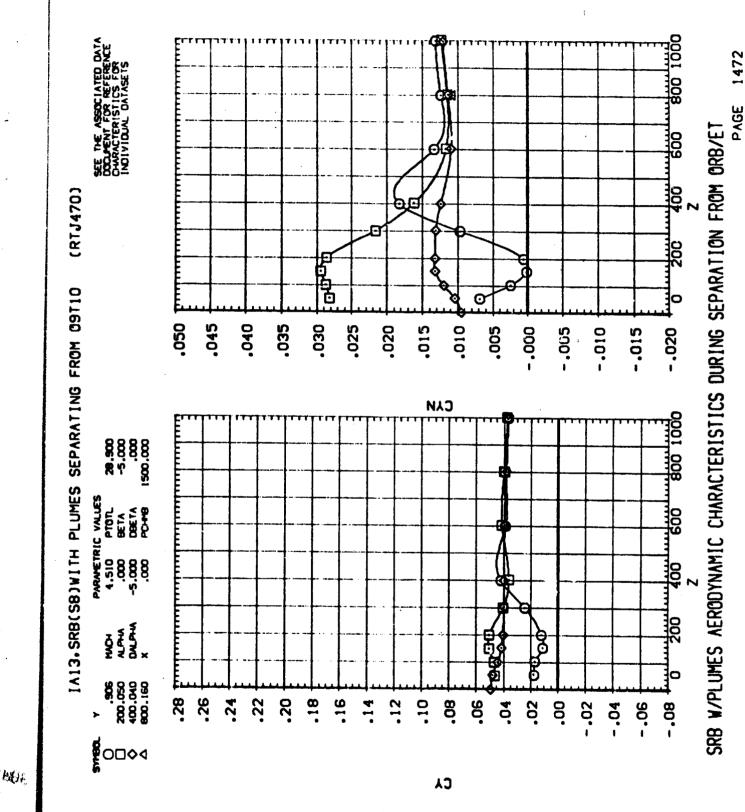
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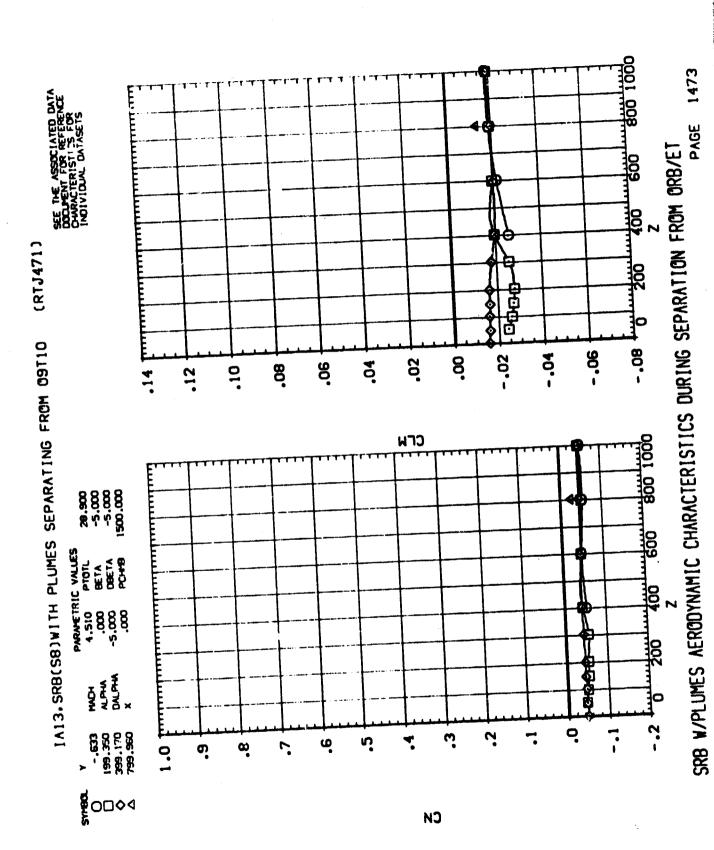
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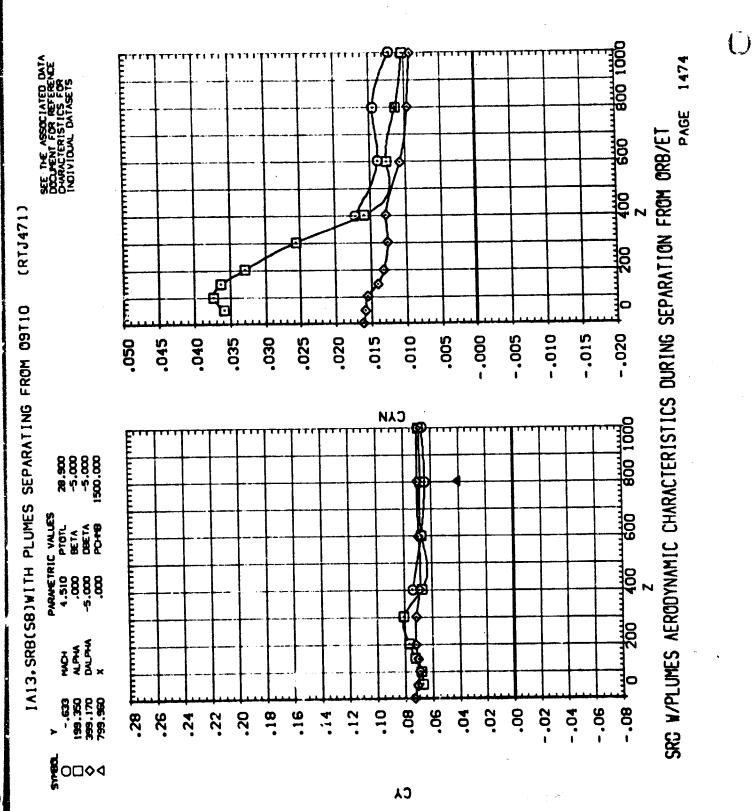
SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

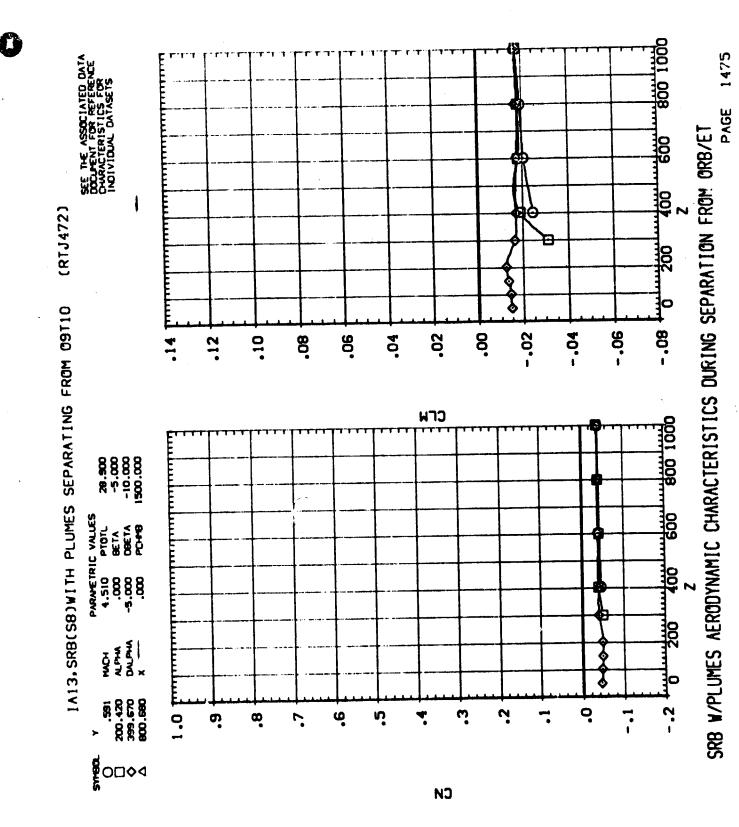




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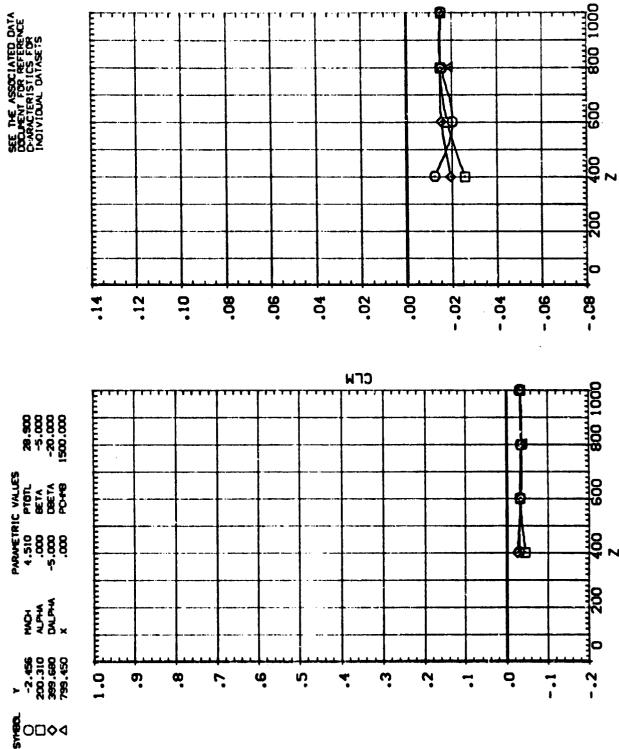




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(RTJ473) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110

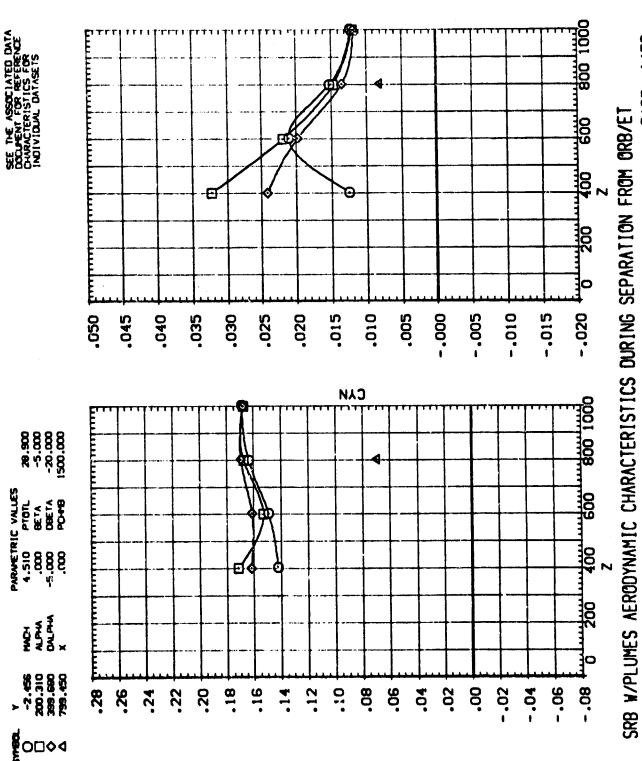


SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM GRB/ET

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(RTJ473) IA13. SRB(SB) WITH PLUMES SEPARATING FROM 0971C

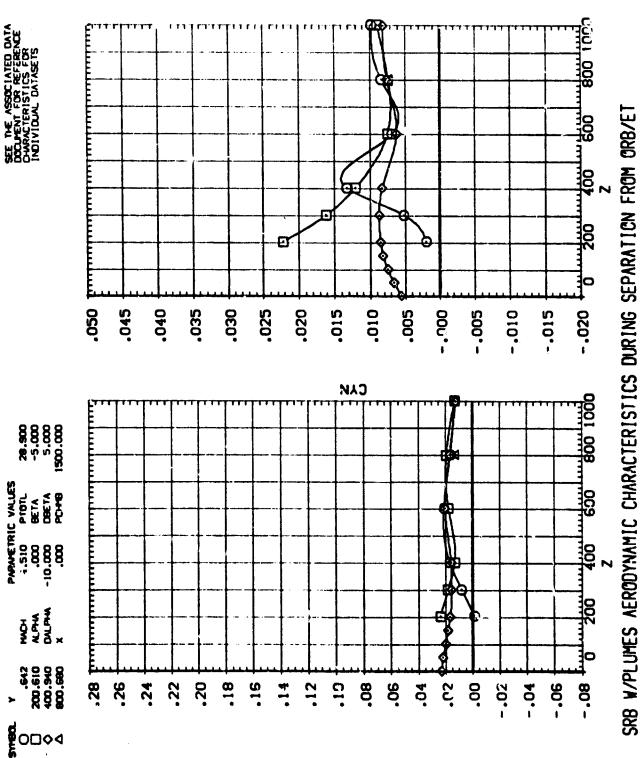


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SEE IN ASSOCIATED DATA DOCUMEN. FOR REFERENCE CHARACTERISTICS FOR INDIVIDUAL DATASETS SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET (RTJ474) 0 IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09710 -.08 -.02 90.-.02 8 -.04 90. • 14 .12 .10 90. CLM 800 1000 28.900 -5.000 5.000 1500.000 PTOTL BETA DBETA POMB A.Si0 P1 4.Si0 P1 .000 B2 -10.000 D2 7 00 7 A PARA 200.510 200.510 400.940 ó 0. ø 'n ų. ~ 4 ٥ œ . **€**0□◊4 CM

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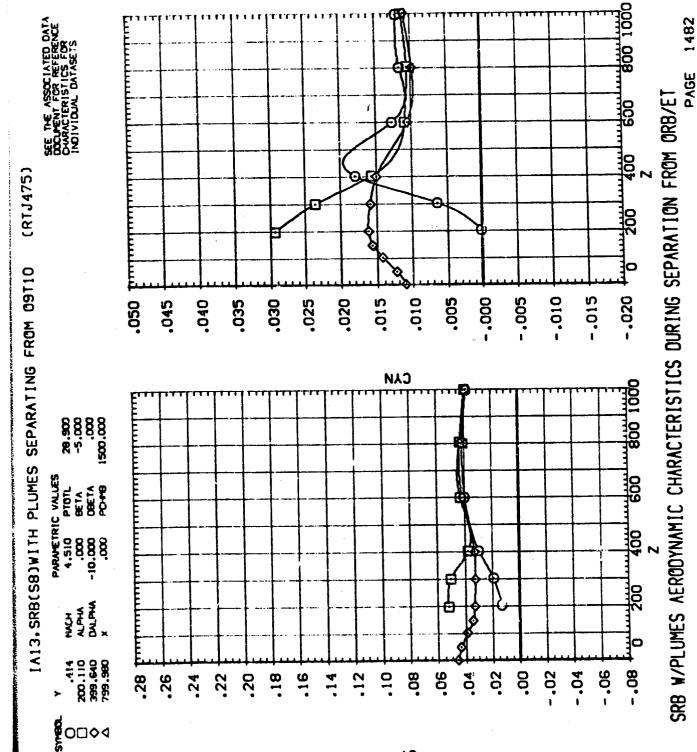
IA13.SRB(SB)WITH PLUMES SEPARATING FROM 09T10 (RTJ474)



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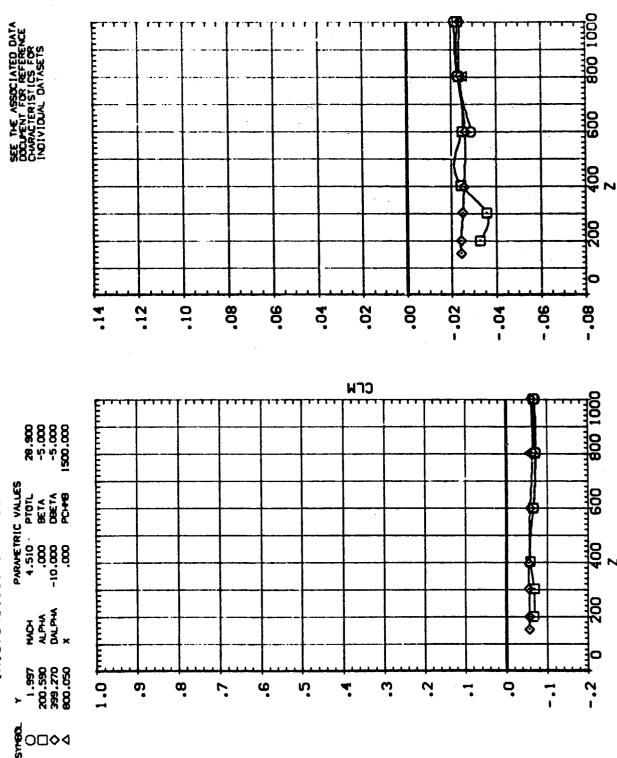
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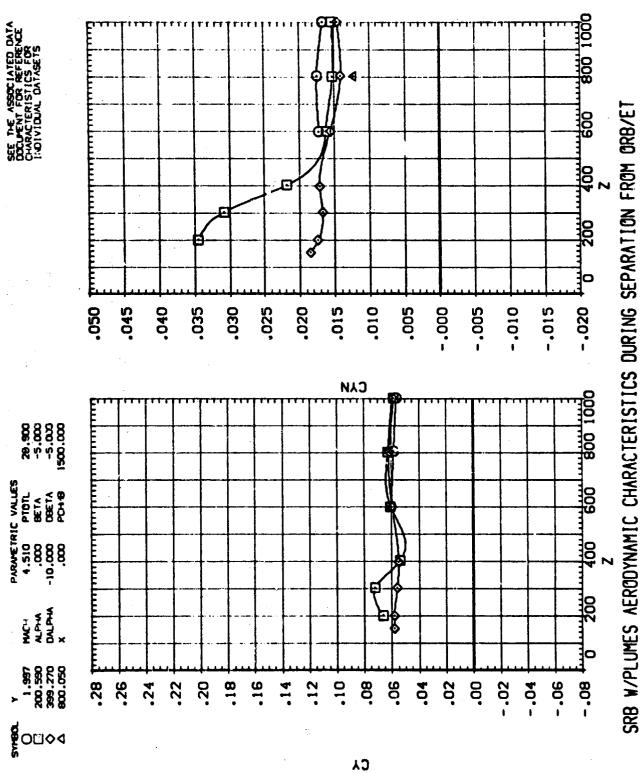
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ476)



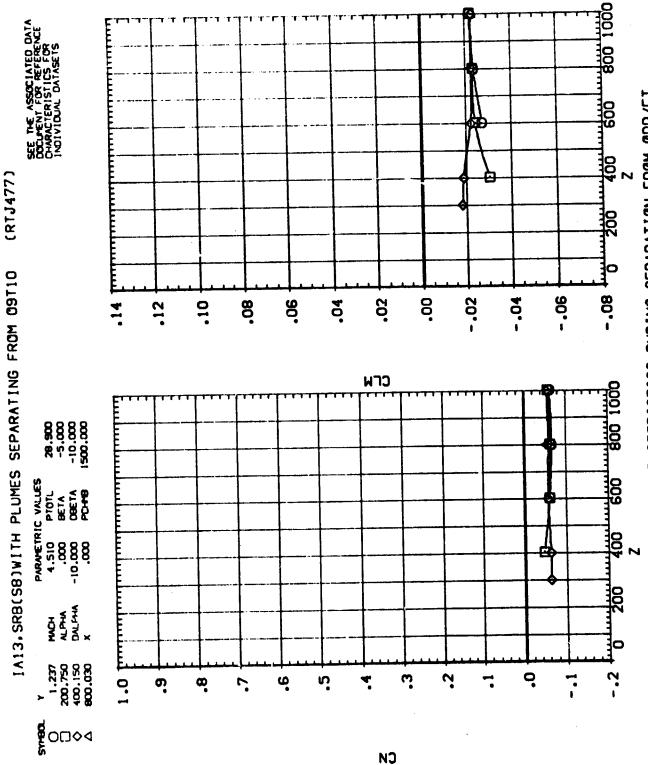
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ476)



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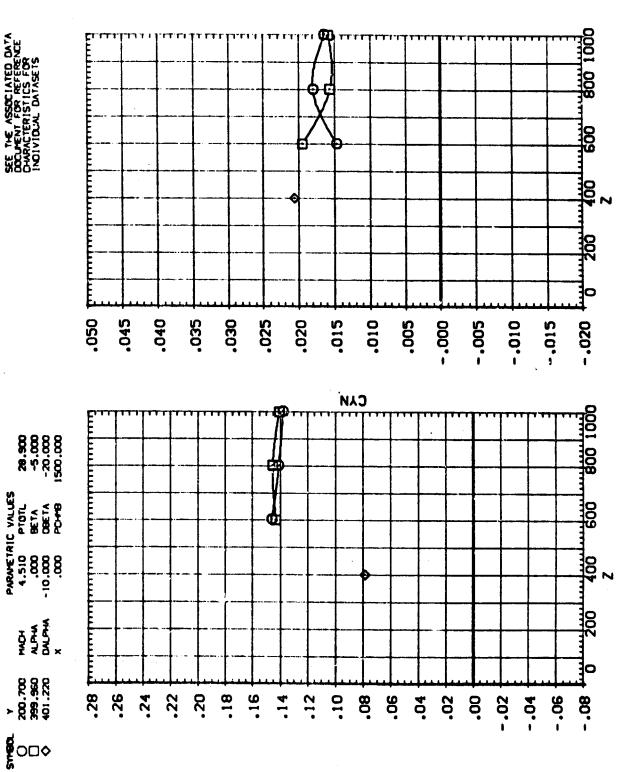
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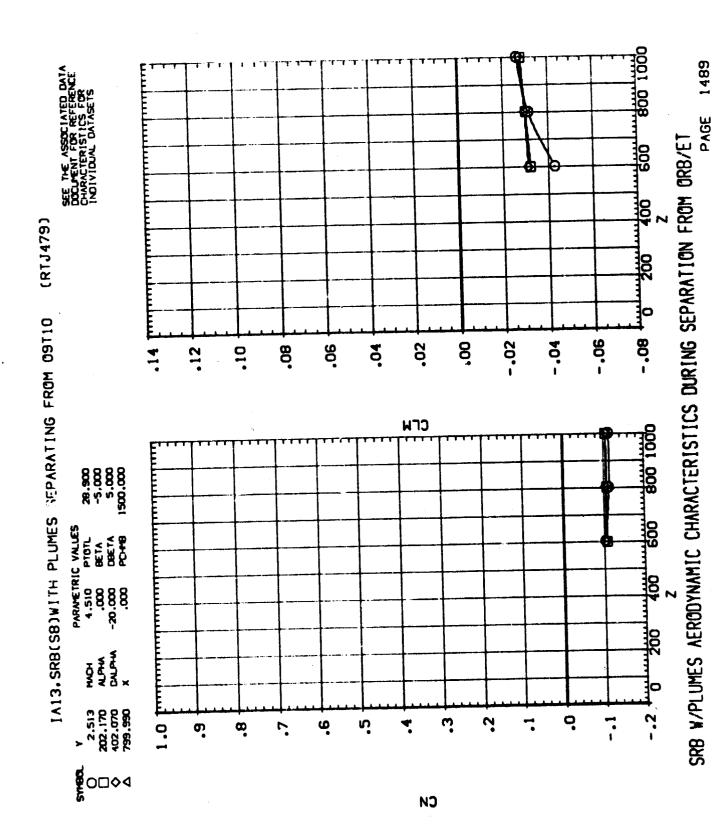
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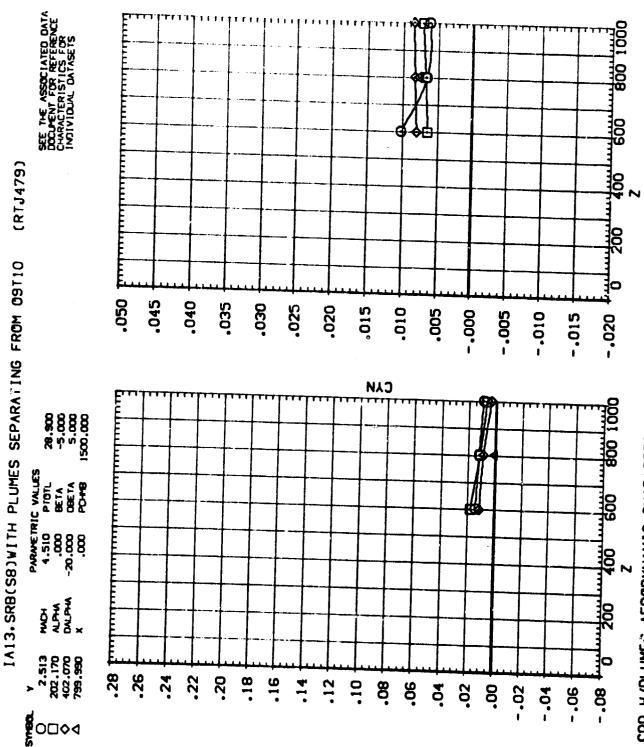
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ478)



SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET



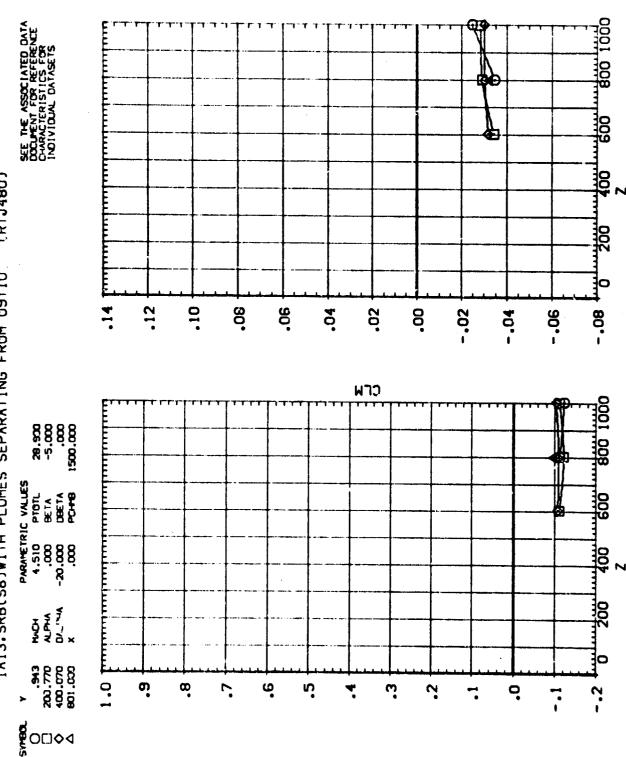




SRB W/PLUMES AERODYNAMIC CHARACTERISTICS DURING SEPARATION FROM ORB/ET

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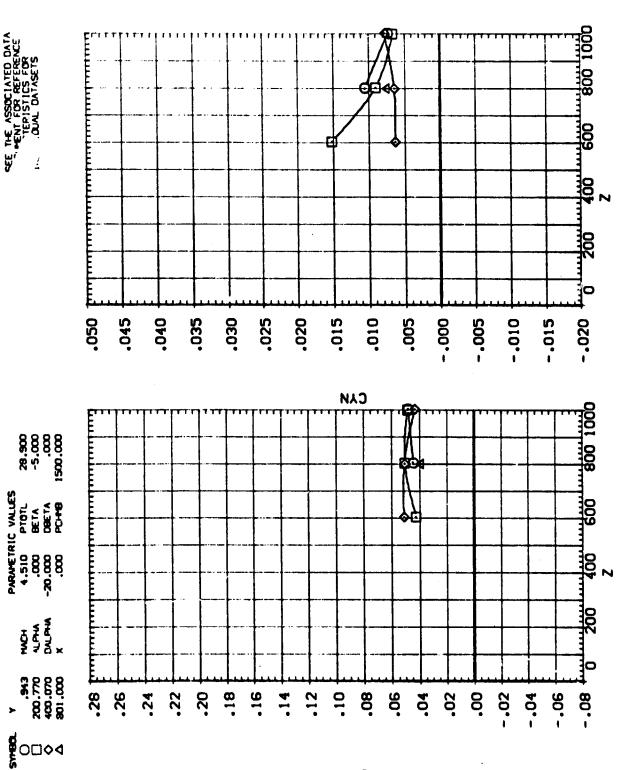
(RTJ480) IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110



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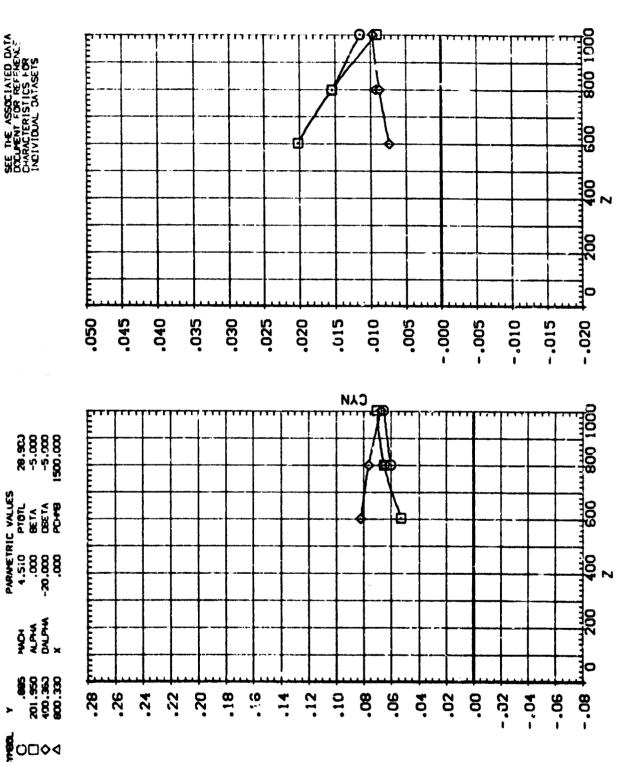
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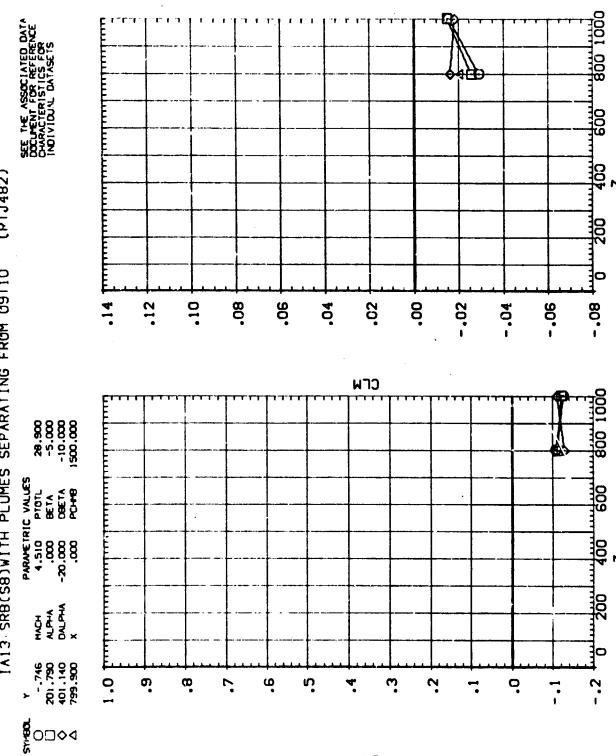
IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ481)



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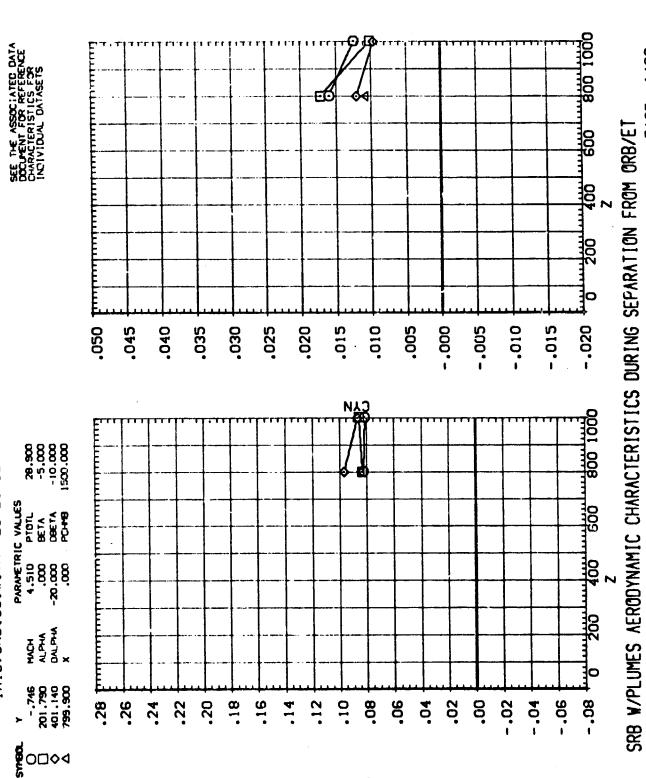
(PTJ482) IA13 SRB(SB)WITH PLUMES SEPARATING FROM 09710

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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ482)

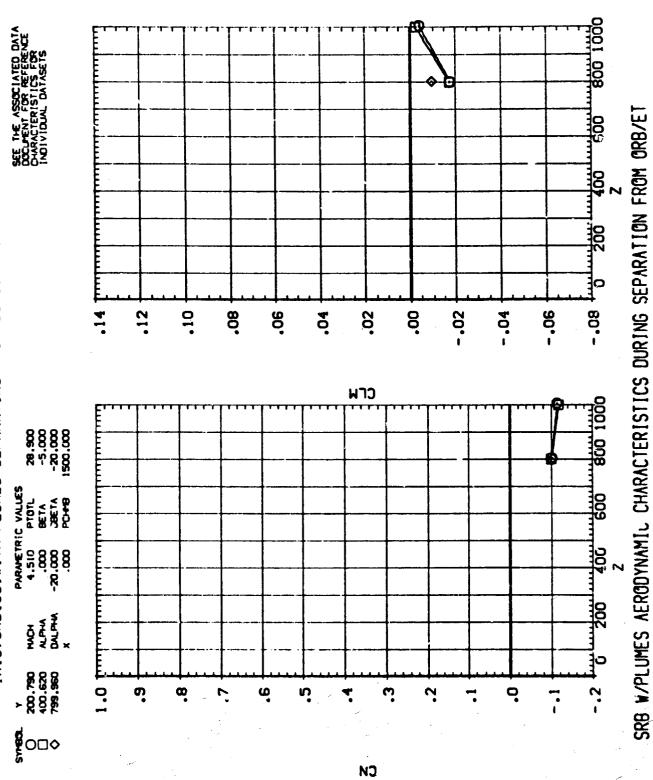


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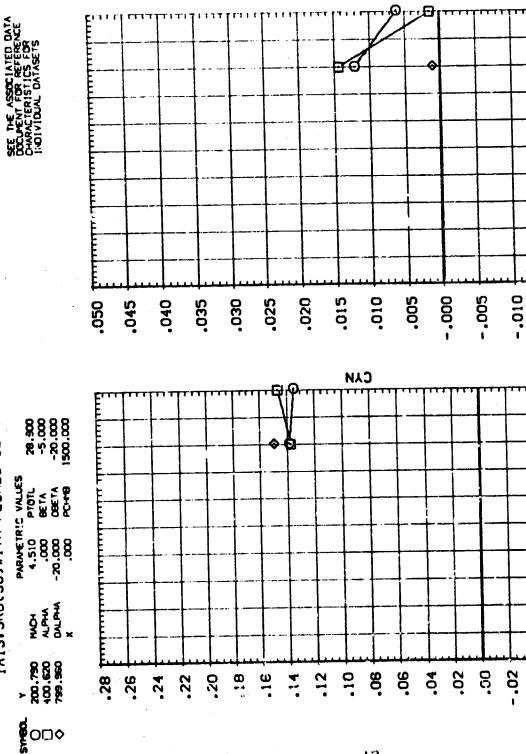
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IA13. SRB(SB)WITH PLUMES SEPARATING FROM 09110 (RTJ483)



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